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COMPARATIVE VERTICAL IMPACT TESTING OF THE F/FB-111 CREW RESTRAINT SYSTEM AND A PROPOSED MODIFICATION

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MARCH 1982



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FOR THE COMMANDER

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An impact test program was conducted to evaluate the operational F/FB-111 crew seat and restraint system and a proposed modification. A primary objective of the program was to compare human response to vertical impacts in the two restraint harnesses. A total of 67 human impact tests were performed on the Vertical Deceleration Tower up to 10 G peak, $26 \, \text{ft/sec}$. Subjects were exposed to comparable impacts at different seat elevations in both harnesses to allow parametric analysis of the test results. Measured data included seat

Block 20 (continued).

acceleration and velocity, head and chest translational acceleration components, triaxial forces acting on the seat and footrest, forces acting at the restraint harness attachments, and displacements of various body segments. The resultant head and chest accelerations were significantly greater in the modified harness than in the operational harness, regardless of seat elevation. On the basis of this comparison, the proposed modification to the F/FB-111 crew seat and restraint is not recommended for implementation. Future restraint harness modification proposals should be based on careful evaluation of all unconventional design features of the operational harness and should address all mechanisms by which adverse loads may be imposed on the seat occupant. In addition, future redesign efforts of the F/FB-111 escape system should provide improved landing impact attenuation.

PREFACE

This report was prepared by the Biomechanical Protection Branch, Biodynamics and Bioengineering Division of the Air Force Aerospace Medical Research Laboratory. On the basis of the test results described herein and other data presented elsewhere (Brinkley et al., 1981; Hearon et al., 1981), the proposed, modified F/FB-111 restraint system described in this report was not incorporated into the F/FB-111. The Engineering Change Proposal for this modification was cancelled at a Configuration Control Board in April 1981.

The impact facilities and data collection equipment were operated by the Scientific Services Division of the Dynalectron Corporation under Air Force Contract F33615-79-C-0523. Mr. Harold F. Boedeker was the Engineering Supervisor for the Dynalectron Corporation.

The test fixtures used during the experimental phase of the effort were designed and built by General Dynamics, Fort Worth Division. Mr. Andrew Shafer was the on-site engineering representative of General Dynamics during the test program.

Photographic support was provided by the 4950th Test Wing, Technical Photographic Division. Special acknowledgement is given to Mr. Paul Creiger for operation of the high speed motion picture cameras and to the many personnel who provided still photography coverage.

Anthropometric measurements of the test subjects were collected by Mr. Charles E. Clauser, Dr. Kenneth W. Kennedy, and Lt Col Maureen Lofberg of the Workload and Ergonomics Branch, Human Engineering Division of the Air Force Aerospace Medical Research Laboratory.

The authors wish to express their gratitude to the personnel of the Biomechanical Protection Branch who participated in the planning, preparation, and performance of the research program and in the preparation of this report. Special commendation is also given to the Air Force officers and airmen who volunteered to participate in the impact tests. The devotion, skill, and professionalism of the entire team of government and contractor personnel were vital to the successful and safe accomplishment of this evaluation.

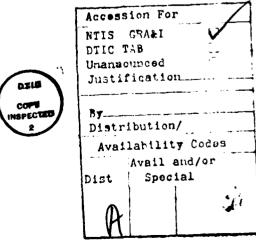


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Section 1

INTRODUCTION

A. BACKGROUND

The F/FB-111 escape system was developed between 1963 and 1967. It was the first multi-place escape system to provide emergency egress by severing the entire crew station from the aircraft fuselage. Earlier efforts to provide an encapsulated escape capability for multi-place aircraft, such as the B-58 and B-70 bombers, utilized individual systems for each crew station.

Numerous aeromedical problems were encountered during the development of the F/FB-111 escape system. Perhaps the most significant of these was the severity of the acceleration environment experienced by the crewmembers during the escape sequence. The accelerations encountered during the ejection and the landing impact phases of the escape sequence were, under certain conditions, higher than levels considered to be acceptable. These acceleration environments have recently been reviewed in considerable detail (Brinkley et al., 1981).

Ejection tests of the crew module, for example, accomplished from a rocket propelled sled, revealed that the amplitude and duration of the acceleration measured at the crew seat increases as the ejection airspeed increases (Carney & Melvin, 1966; Hatcher, 1966; Hefti, 1967; McCauley, 1966; and McCauley & Melvin, 1966). Utilizing only the vertical axis acceleration data obtained in these tests, the probability of vertebral compression fracture was estimated as a function of ejection airspeed. The probability of vertebral fracture was estimated using the Dynamic Response Index (Payne, 1965). The Dynamic Response Index (DRI) is based on a simple lumped parameter mathematical model that has been correlated with a probability of vertebral fracture in operational USAF ejection seats (Brinkley et al., 1971). Since only the vertical component of the acceleration data was used in these estimates and there were large accelerations in the X axis at higher airspeeds, it was assumed that the calculated DRI values would tend to underestimate the actual probability of vertebral fracture in these cases.

A DRI value of 18 corresponds to an estimated probability of vertebral compression fracture of 5%. Therefore, the calculated DRI values plotted in Figure 1 show that below 450 KEAS (knots equivalent airspeed), the probability of such injury is estimated to be less than 5%. However, from 450 KEAS to 550 KEAS, the estimated probability of spinal injury increases from 5% to 50%.

The potential for vertebral fracture during landing impact of the crew module was also recognized during the development of this escape system. The calculated DRI as a function of crew module weight is shown in Figure 2 (Gossick & Mohr, 1968). These 23 values were obtained from the vertical acceleration data measured at the seat of the crew module in various landing impact tests (for example, Fricker, 1966). In 7 tests the probability of vertebral fracture was estimated to be lower than 5%. The probability of such injury was estimated to be between 20% and 50% in 12 of these tests and greater than 50% in 4 tests.

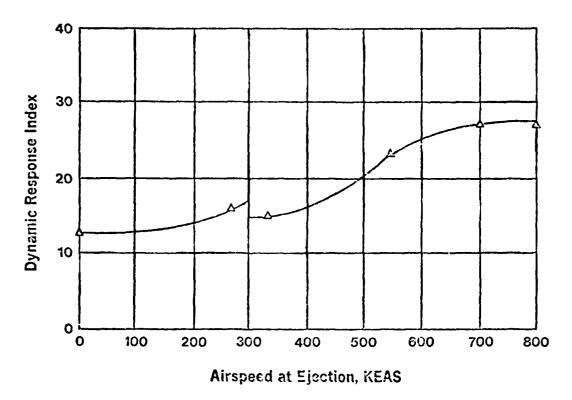


Figure 1. Dynamic Response Index Values Calculated from Qualification Ejection Test Data. (Discontinuity in curve occurs at the airspeed where a secondary rocket nozzle is opened to reduce +G_Z accelerations. Adapted from Gossick & Mohr, 1968.)

During module descent, the combination of surface winds and parachute oscillations may cause the horizontal velocity of the module at the time of impact to be as high as 43 ft/sec. Consideration of the sideward and/or fore-aft components of landing impact acceleration which occur as a result of this horizontal velocity, in addition to the vertical acceleration component, indicates that the resultant acceleration measured at the crew seat during this phase of the escape is beyond the human exposure limits given in USAF Military Specification MIL-C-25969B (Johnson, 1968; Peterson & Roberts, 1972). In order to decrease the module impact velocity and, in turn, the severity of the acceleration during this phase of the escape, incorporation of a retro-rocket system, which would act just prior to landing impact, was proposed. Subsequent landing impact tests of the module revealed that the impact tolerance criterion in MIL-C-25969B with such a modification could be met without a requirement for supplemental impact attenuation air bags (Peterson & Roberts, 1972). Owing to cost considerations, however, this modification was not pursued. The attenuation air bags were retained, leaving the spinal injury predictions noted above.

During the development of the F/FB-111 escape system, the restraint harness also came under close scrutiny. In fact, the current operational harness was not the restraint system which was originally utilized in the F/FB-111. The current harness was developed by the Royal Air Force Institute of Aviation Medicine in Great Britain. It was installed in the F/FB-11 crew module in 1970 after

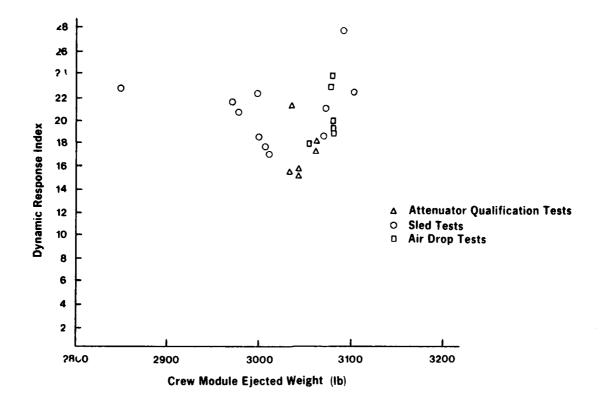


Figure 2. Dynamic Response Index Values Calculated from Ground Landing Impact Test Data. (Adapted from Gossick & Mohr, 1968.)

fore-aft and sideward demonstration impact tests indicated that the harness generally provided adequate restraint and subjectively was more comfortable than the original harness (Reader, 1967; Zaborowski, 1965). The vertebral fractures which occurred in the early F/FB-111 ejection experience with the original harness were apparently in the same general area of the spine (mid-thoracic) as the injuries experienced with the current operational harness. The original harness, it should be noted, did not have a shoulder harness yoke and reflection straps. The two harnesses are described in detail by Brinkley et al. (1981).

In 1977, a review of the operational ejection data by Kazarian indicated that the vertebral fracture rate among F/FB-111 ejectee survivors was even higher than anticipated. An injury rate of 40.3% (25 of 62) was reported for the period October 1967 to June 1977. Despite the previous concerns regarding the severity of the acceleration environment during ejection and landing impact described above, the majority of these injuries was attributed to the retraction phase of the escape sequence. Twenty-nine percent (18 of 62) of the survivors were believed to have incurred hyperextension or combined hyperextension-hyperflexion vertebral fractures (Kazarian et al., 1979). Furthermore, this injury mechanism assessment implicated negative (with respect to the reference waterline) inertia reel strap angles as the cause of these hyperextension injuries. It was suggested that elevation of the inertia reel and reflection strap anchor points would alleviate the hyperextension injuries presumed to

occur during retraction and the smaller number of hyperflexion injuries presumed to occur during landing impact.

The Life Support System Program Office of Aeronautical Systems Division (ASD/AES), Air Force Systems Command, Wright-Patterson Air Force Base, was assigned the responsibility for development of appropriate modifications to reduce or eliminate the high spinal injury rate associated with operational use of the F/FB-111 escape system. This task was defined in the Air Force Systems Command Program Management Document R-P 6097 (3) P.E. 64212F/2229 entitled "F-111 Crew Restraint System Redesign". The objectives of the restraint system redesign effort were to (1) eliminate the downward component of vertebral loading caused by the shoulder straps during retraction, (2) reduce forward and downward rotation of shoulders and back on landing impact, and (3) extend the seat back to provide upper back support during powered inertia reel retraction.

After a feasibility study exploring possible approaches to improve the crew seat and restraint sytem, General Dynamics Corporation, under Contract No. F33657-78-C-0651, proposed hardware modifications to achieve the aforementioned redesign objectives. The details of this proposed modification have been described elsewhere (Brinkley $\underline{et\ al.}$, 1981).

In June 1979, the Air Force Aerospace Medical Research Laboratory (AFAMRL) initiated a human impact test program to evaluate the proposed, modified F/FB-111 crew seat and restraint system. The research objectives of this program were to (1) assess the adequacy of the restraint as an impact protection device, (2) quantify the shoulder harness geometry for a range of subject anthropometry, and (3) uncover any areas of performance degradation. This demonstration test series revealed that negative inertia reel strap angles were still possible for some subjects in some seat configurations in the modified system and that lateral and vertical impact protection performance had apparently been degraded by the modification (Brinkley et al., 1981). As a result of these findings, additional impact tests were conducted to clarify the test results (Brinkley et al., 1982) and to directly compare the current operational harness to the proposed, modified harness. The results of the latter study are reported herein.

B. PROGRAM OBJECTIVES

The purpose of this research effort was to compare the operational F/FB-111 crew seat and restraint system with the proposed modification of the F/FB-111 crew seat and restraint system under identical impact test conditions. The experimental effort was designed to measure differences in the impact responses of volunteer test subjects as a function of restraint configuration, vertical position of the seat, and subject anthropometry. The influence of the angle of the inertia reel and reflection straps on the impact responses of the subjects was of particular interest.

The investigative effort also was intended to collect data which can be used as baseline data for future protective equipment evaluations and to provide data for ongoing research efforts to develop mathematical models to predict human impact responses.

C. PURPOSE AND SCOPE OF REPORT

This report (1) describes the impact tests accomplished to meet the program objectives outline above, (2) presents analysis and interpretation of the collected data, (3) summarizes the results of the evaluation, and (4) presents a final recommendation regarding the proposed modification.

Section 2

TECHNICAL APPROACH

A. EXPERIMENTAL DESIGN

The null hypothesis evaluated during this test program was that human inertial response to impact acceleration in the operational F/FB-111 harness is not significantly different from such response in the proposed, modified F/FB-111 harness. Vertical acceleration tests were considered to be crucial in this evaluation, since a significant number of the vertebral compression fractures incurred by ejectees were attributed to the inadequacy of the operational restraint on landing impact (Kazarian, 1977) and since the largest component of the resultant module acceleration on landing impact is in the Z axis (Brinkley et al., 1981). Also, the vertical impact tests which had recently been performed with the modified harness revealed increased vertical head acceleration with increasing inertia reel strap angles. This was interpreted as being indicative of degraded vertical impact protection performance in the modified harness. These facts, in addition to the absence of a vertical impact data base for the operational harness, made a direct comparison between the two harnesses mandatory. The vertical $(+G_7)$ impact profile selected for use during this program has been used extensively at AFAMRL in the evaluation of other restraint systems with human subjects.

In order to minimize the potential for injury to human subjects, these tests were conducted at what is considered to be subinjury impact acceleration levels. The nominal 8 G peak carriage acceleration level was chosen as the orientation level for inexperienced subjects who had never before participated in tests on the Vertical Deceleration Tower (VDT). The nominal 10 G peak carriage acceleration level was selected as the experimental level, since some comparable test data at this level were already available and since, on the basis of prior experience, the risk of subject injury at this level was still acceptably low. At the same time, the forces acting on the subject at this exposure level are generally sufficient to overcome the forces created by voluntary muscle contraction, thereby producing a response suitable for comparative parametric analysis.

The sample of subjects selected to participate in this test program is comparable to the USAF flying population in terms of age, sex, and anthropometry. As the number of female flyers has increased, efforts have been made to introduce qualified female subjects into AFAMRL impact test programs. One female subject was among the eighteen volunteers who participated in this test program. The medical screening of all subjects prior to participation continues to be more highly selective than a routine USAF Flying Class I evaluation, resulting in a panel of supranormal volunteers (Hearon & Raddin, 1981). This difference in the populations of interest has a negligible influence on the significance of results of tests such as these, since all tests were conducted below anticipated injury threshold, even for a normal population. Such a conservative approach to subject screening is necessary to assure subject safety.

The experimental matrix for this test program is shown in Table 1. The factors influencing human inertial response which were investigated in this study were (1) the type of F/FB-111 harness (modified or operational) and (2) the seat vertical adjustment (which, in turn, directly determines the inertia reel strap

angles with respect to a reference horizontal for a given subject). All tests in the matrix were conducted with the plane of the backrest perpendicular to the plane of the seat and parallel to the impact vector, in the 90° seat back angle condition. This seat configuration was achieved by adjusting the headrest full forward and the seat pan full aft. (See Figure 3.) In this position, the contact point of the subject's flight helmet was 2½ inches forward of the plane of the seat back. Upper extremity bracing was not a factor in this test program, since all subjects folded their hands in their laps during all exposures in the matrix. It should also be noted that tests in the C cell of the experimental matrix were completed as part of the F/FB-111 test program which preceded this study (Brinkley et al., 1982).

TABLE 1. EXPERIMENTAL MATRIX

VERTICAL	F/FB-111 HARNESS		
SEAT ADJUSTMENT	MODIFIED	OPERATIONAL	
71	С	G	
22	Н	J	

In Table 1, z_1 is defined as the seat vertical adjustment (relative to full down or z=0) at which a subject had the largest positive inertia reel strap angle in the modified harness. This condition, in fact, corresponded to the full down seat position for all subjects. However, those subjects with relatively small sitting heights did not receive adequate helmet support in the full down seat position. Adequate helmet support was assumed if the approximate Frankfort horizontal plane (defined by the lowest points in the inferior orbital rims and the midpoint of the line connecting the highest points in the margin of the auditory meati) of the subject was in contact with the headrest. If necessary, the seat was elevated until minimum helmet support, according to the above definition, was provided. This seat elevation was designated as z_1 for that subject. It should be noted that the vertical position of the test seat was adjustable in one inch increments over a range of five inches. (See Figure 3.) The operational seat is continuously adjustable. The vertical position of the headrest, however, was fixed, as it is operationally.

In Table 1, z_2 is defined as the seat vertical adjustment at which a given subject had the largest negative inertia reel strap angle in the operational harness. Upward seat vertical adjustment was limited for all subjects by impingement of the inertia reel straps on the lower aspect of the headrest. Contact of no more than one-half the width of the inertia reel straps (inboard portions) was considered acceptable in the determination of z_2 and in the conduct of tests during the program.

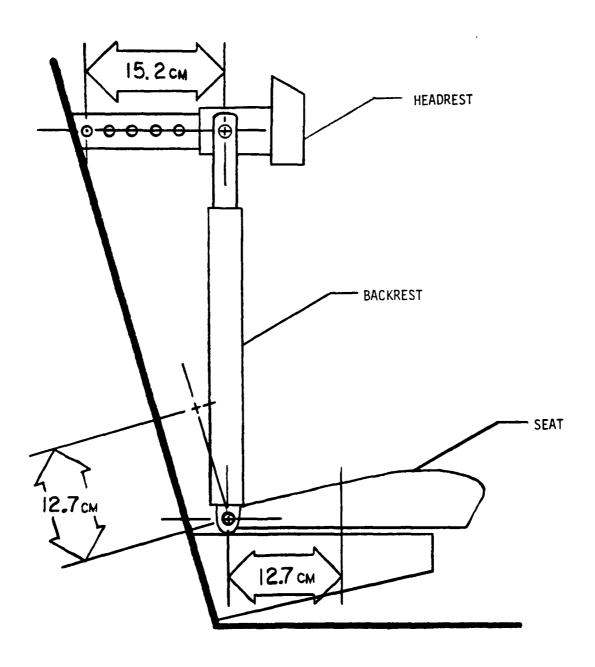


Figure 3. F/FB-111 Crew Seat Geometry.

The test conditions described in the preceding paragraphs were chosen in order to best achieve the objectives of this test program. For example, the 90° seat back angle position was selected for investigation because previous vertical impact tests of the modified harness had demonstrated that measured human inertial response could be expected to be more severe in this condition than in similar exposures with the seat back reclined. In particular, forward head accelerations and vertical seat pan reaction loads were significantly higher with the seat back angle at 90° than with the seat back angle at 103° or 110° (Brinkley et al., 1981). This allowed greater observability of the effects of harness variations.

Similarly, the seat vertical adjustments selected in the experimental design were considered crucial to achieve the program objectives. In the J cell of the matrix, for example, the largest negative inertia reel strap angle for each subject in the operational harness was explored. This condition is of considerable interest because, based on a review of the operational ejection data, Kazarian (1977) theorized that negative inertia reel strap angles were causally related to vertebral compression fractures incurred by ejectees not only during landing impact, but also during inertia reel retraction. This presumed correlation, in fact, resulted in a highly conservative approach which avoided negative inertia reel strap angles in the previous tests of the modified harness (Brinkley et al., 1981). However, a more recent review of the operational ejection data (Hearon et al., 1982) revealed that the likelihood of spinal injury is more probably related to the severity of the acceleration environment on landing impact and did not reveal a correlation with restraint harness geometry, as had been previously reported. (See also Section 5B.) The restriction of not exposing volunteers to conditions with negative inertia reel strap angles was, therefore, removed in the present study.

Also of considerable interest is the C cell condition of the experimental matrix, because the relatively large positive inertia reel strap angles in the modified harness in this test condition have been demonstrated to result in larger vertical head accelerations than similar exposures at higher seat elevations (Brinkley et al., 1981). In summary, the seat vertical adjustments were chosen to investigate the conditions in which the operational harness was originally presumed to provide less than adequate restraint and in which the modified harness was shown to degrade vertical impact protection. In addition, these conditions are operationally relevant, since the aircraft commander often adjusts the seat upward to assure adequate over-the-nose visibility and since the weapons systems operator often adjusts the seat downward to facilitate radar work. However, since crewmembers are not constrained by incremental vertical adjustment or by inadequate helmet support on the low side or strap impingement on the high side, the entire range of seat vertical adjustment that may used operationally was not evaluated in this test program.

Impact tests were conducted in all cells of the experimental matrix using an anthropomorphic dummy prior to initiating tests with volunteer subjects. As an additional safety precaution, a dummy test was performed each day prior to testing with human subjects.

The controlled variables during these experiments were the carriage drop height (and, in turn, the carriage acceleration and impact velocity), the type of F/FB-111 restraint harness (operational or modified) and the seat vertical adjustment. For the 8 G orientation exposures, the carriage drop height was

 $8.5~{\rm ft}$ and for the $10~{\rm G}$ experimental exposures, the carriage drop height was $11.0~{\rm ft}$.

The observable variables which were measured during these experiments included the restraint harness geometry (eg., inertia reel strap angles), the restraint harness static preloads, the restraint harness loads during impact (eg., reflection strap and lap belt loads), the forces (horizontal, lateral, and vertical) reacted at the seat pan and footrest during the impact, the triaxial translational acceleration components measured at the seat pan and at the subject's head and chest, and the displacements of photometric targets on the subject's body segments. The potential measurement error of the accelerometers, load cells, strain gages, and other devices utilized to make these measurements is detailed in Appendix A.

Significant unobservable variables during these experiments included the motion of each vertebral body and the force distribution along the vertebral column during the impact event.

B. EVALUATION CRITERIA

The electronic measurements obtained during these experiments included the tension-time histories of the various restraint harness straps measured at their attachment points, the force-time histories of the loads reacted into the seat pan and footrest, and the acceleration-time histories of the subject's head and chest and of the seat pan and drop carriage as well. It should be noted that the accelerometer arrays, attached to the subjects were, in general, rotating measurement coordinate frames which measured translational acceleration components summed with translational components resulting from angular motions. One implication of this observation is that, as the head rotated down and forward, vertical acceleration of the head with respect to the laboratory reference frame transitioned from a Z axis (vertical) to an X axis (fore-aft) measured acceleration with respect to the head. This situation is illustrated in Fig. 4. Another implication is that separation of translational accelerations of the effective center of gravity and translational acceleration components resulting from rotational motions cannot be achieved with the three orthogonal linear accelerometers utilized in these experiments. The relevant equations have been summarized elsewhere (Simons et al., 1979). For the purposes of this test program, it was adequate to measure the mixed translational data and assess rotational motion photometrically.

Evaluation of the entire measured acceleration-time histories of chest and head was accomplished by calculating Severity Indices (Gadd, 1966). These single parameters, which were derived by a weighted integral of the acceleration-time function taken over the interval of the impact (SI = $\int a^n(t)dt$, where n = 2.5), were used to compare the severities of impact responses. No exposure limit values were assigned to the chest or head acceleration Severity Indices. Instead, they were used only in a relative sense for purposes of comparison.

The Wilcoxon paired-replicate rank test (Wilcoxon & Wilcox, 1964) was the statistical technique selected to compare the peak values of specific measured parameters and to establish the statistical significance of observed trends in the data. Experimentally-measured parameters for each subject were arithmetically compared with the same parameters measured for the same subject in a

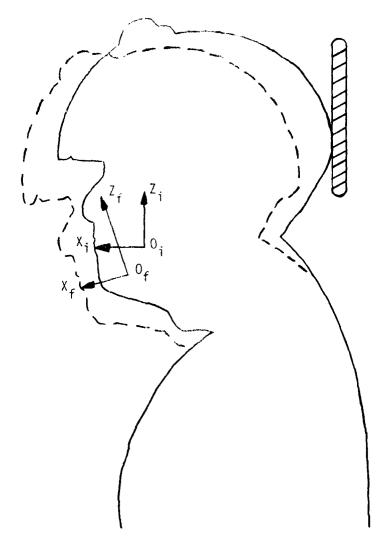


Figure 4. Rotation of the Head Accelerometer Array (Located at the Origin of the Coordinate System) During Vertical Impact.

different (but comparable) test condition, thereby establishing pair differences. When a sufficient number of pair differences for a specific parameter changed in the same direction for a variety of subjects, a trend was established as statistically significant by the Wilcoxon technique. The 90% confidence level was defined as the level of statistical significance for rejection of the null hypothesis, assuming a two-tailed test.

The advantages of employing this statistical technique are particularly noteworthy in these experiments. The technique is comparative and therefore is readily applied to the comparison of two restraint harnesses. Also, the method establishes each subject as his own control, thereby reducing the effects of biological variability on the data. In addition, a relatively small number of paired-replicates (6) is the minimum number required to permit a valid conclusion at the chosen significance level. This limits the total number of tests required to detect statistically significant trends in the test results.

The disadvantages of the Wilcoxon technique, however, must also be considered. Although the trend (direction) of a statistically significant difference in a given parameter is indicated, the magnitude of that difference is not quantified by the technique. (The difference between the means of the two sets of parameters being compared may be easily computed, however.) The method is also less powerful than, for example, the analysis of variance. As in any statistical technique, statistical significance can be computed, but practical significance must be judged.

Statistically significant trends in test parameters between two test conditions were critical in this comparative restraint harness evaluation. Generally, trends in specific parameters differ in practical importance. In this test program, for example, there was limited interest in the loads reacted into the lap belt, since all tests were conducted in the Z axis with an unreclined seat back, thereby minimizing variation in lap belt reaction loads. However, in view of the relatively high incidence of vertebral fractures experienced by F/FB-111 ejectees operationally, the more crucial considerations were the trends in the experimentally-measured seat pan reaction loads and the head and chest accelerations. At this time, the loads reacted at the seat pan are the best indirect measurement of the magnitude of vertebral column loading during impact.

In the final analysis, the overall distribution of statistically significant trends in all test parameters being compared generally assumes greater importance than the trend of any single parameter. At times, a "beneficial" trend in one parameter may be accompanied by a "detrimental" trend in another. In this circumstance, a careful evaluation of the "trade-offs" among parameters is necessary, in order to accurately assess which test configuration is "best" or perhaps which is the lesser of two evils. It is conceivable that, in some circumstances, such a determination may be impossible.

For ethical and moral reasons, it is not possible to design and conduct impact experiments in the laboratory with human subjects at operational exposure levels where there is a significant probability of serious injury. These tests, therefore, were performed at subinjury impact levels which have been demonstrated to be well within human tolerance and where the risk of injury is acceptably low. However, the levels were sufficiently high to overcome voluntary muscle resistance and approach the operational range. Extension of the impact accelerations could be expected to lead to increases in response until a nonlinearity occurs in the form of injury. The statistically significant trends reported herein for this experimental level cannot be extrapolated to operational levels for the purpose of predicting injury rates. However, the trends discovered at this experimental level should be valid with increasing levels of impact until the non-linearities associated with injury are encountered.

Section 3

TEST EQUIPMENT, METHODS, AND FACILITIES

A. VERTICAL DECELERATION TOWER

The AFAMRL Vertical Deceleration Tower (VDT), shown in Figure 5, was used for the entire impact test series. This facility consists of a 60 ft vertical steel tower, which supports a guide rail system, an impact carriage, a hydraulic deceleration device, and a test control and safety system. The impact carriage which is used to carry the test specimen can be elevated to a maximum height of 42 ft prior to release. After release, the carriage falls until a plunger attached to the carriage enters a water-filled cylinder located at the base of the tower. The deceleration profile produced as the plunger displaces the water in the cylinder is a function of the free fall distance, the carriage and test specimen mass, the shape and size of the plunger, and the diameter of the cylinder orifice.

A typical acceleration-time history recorded on the impact carriage during this test program is shown in Figure 6. The 10 G test level mean carriage acceleration for the entire vertical test series was 10.5 G with an estimated standard deviation of 0.19.

B. CREW SEAT AND RESTRAINT SYSTEM

The crew seat used in the test program is a unit that was salvaged from an F-111 crew escape module. The seat was attached to the impact facility using a special structure that was designed and fabricated by the General Dynamics Corporation under Contract No. F33657-78-C-0651. This fixture, shown in Fig. 7, supported the seat and aircraft rudder control pedal footrest. The seat pan and the rudder pedals were instrumented to measure the loads reacted by the subject into these structures during the experiment. The design configuration of the seat and its components is described more completely elsewhere (Brinkley et al., 1981).

A modified headrest, shown in Figure 8, was used in all tests of the proposed, modified crew seat and restraint system. This headrest differs from the operational headrest in that 2.66 inches of the helmet support structure has been removed from the lower portion of the headrest. The aeromedical implications of this modification are described elsewhere (Brinkley et al., 1981).

The inertia reel was not used in the test program. It was replaced by a simple webbing clamp bar located at the centerline of the reel.

The same restraint harness was used for all tests in this series. However, the attachment geometry of the straps connecting the shoulder harness to the inertia reel and seat structure anchor points was varied. The modified restraint configuration is shown in Figure 8 and the operational configuration is shown in Figure 9. The reflection straps of the operational harness are attached to the upper portion of the seat back 24.09 inches above the seat reference axis (the intersection of the planes of the compressed backrest cushion and the seat cushion). The operational reflection strap attachment points move vertically as

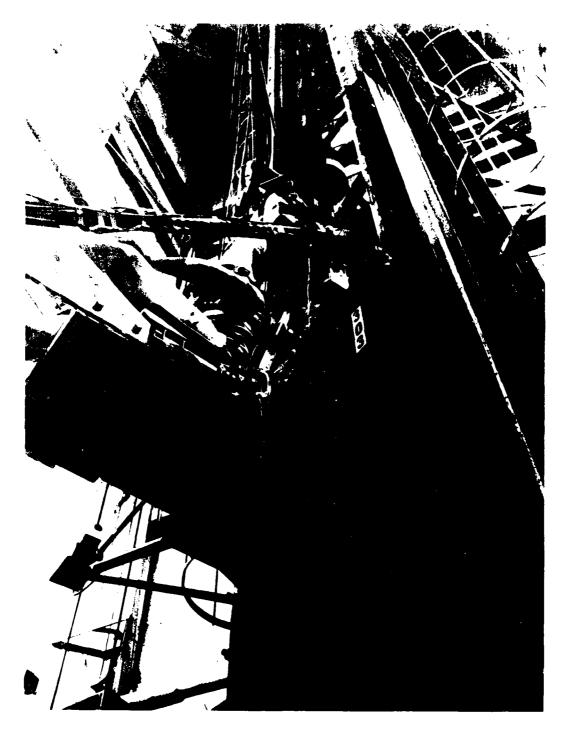


Figure 5. AFAMRL Vertical Deceleration Tower and F/FB-111 Test Fixture Viewed from Below.

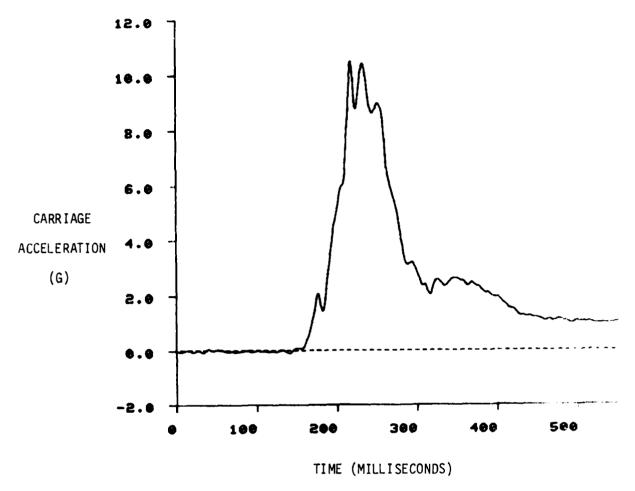


Figure 6. Typical Carriage Acceleration Profile.

the seat is moved up or down. In the proposed modification of the crew seat and restraint system, the reflection strap attachment points are located in the headrest structure (which does not move with seat vertical adjustment). These proposed attachment points (located at waterline 203.2) are 2.94 inches above the highest elevation which the attachment points can reach in the operational configuration and 7.94 inches above the lowest elevation. In addition, they have been moved 2.81 inches toward the center of the headrest. The inertia reel attachment points of the operational restraint system are located on a waterline (200.75) which is 27.5 inches above the lowest position of the seat reference axis. The proposed modification raises the load reaction point of the inertia reel straps by 1.9 inches by the use of two sets of rollers. The lower portion of the headrest has been removed in order to provide clearance for the higher strap locations.

Each restraint system was pretensioned prior to the impact experiment. The lap belt pretension was 20 ± 5 lb measured by strain gages mounted on the lap belt end attachment fittings. The total load acting on each shoulder strap was set at 14 ± 5 lb by measuring the loads at the end fittings of each reflection strap and using Lebow gages attached to the inertia reel straps. This preload procedure imposes a load on the subject which is lower than the maximum load (50 lb



Figure 7. Test Fixture with Rudder Pedal Support Structure.

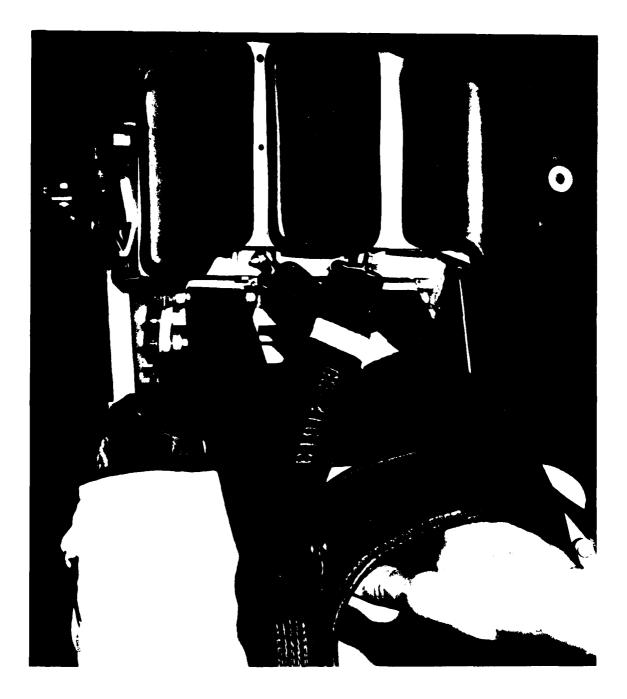


Figure 8. Proposed Crew Seat and Restraint System Modifications.

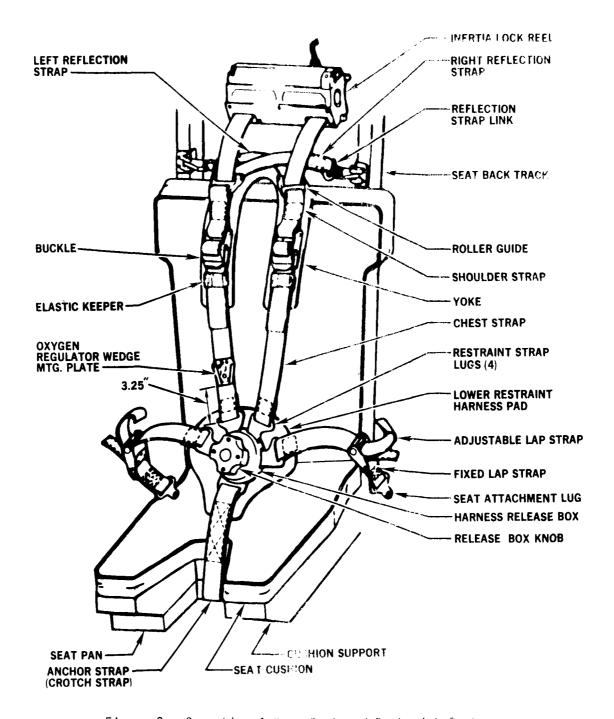


Figure 9. Operational Crew Seat and Restraint System.

in each shoulder strap) expected during operational use of the inertia reel. However, previous tests of similar restraints in England (Reader, 1967) and at Holloman AFB (Zaborowski, 1965) resulted in subject complaints when preloads of 50 lb or greater per strap were imposed. In view of these reports, and since imposition of static preloads on the subject was required for relatively long periods of time (approximately 15 minutes) prior to the impact event, imposition of preloads of such magnitude was neither practical nor desirable. In addition, previous experience at AFAMRL has shown that significant variations in restraint performance do not occur unless the pretension is well below 20 lb. Therefore, the aforementioned pretensions were deemed adequate.

C. DATA ACQUISITION SYSTEM

Electronic data collected during the test program included impact carriage acceleration and velocity, test fixture loads and acceleration, subject head and chest acceleration, harness loads, and single-lead electrocardiograms. Detailed descriptions of the instrumentation, electronic data processing equipment, mounting procedures, and calibration techniques are provided in Appendix A. The following information summarizes the electronic instrumentation that was used to acquire the test data.

Carriage acceleration was measured using three miniature, piezoresistive accelerometers mounted to the structure of the VDT carriage. Vertical velocity was determined at the point of impact, i.e., the point where the carriage plunger contacted the water in the deceleration cylinder.

The test fixture was instrumented to measure the forces reacted into the seat, restraint, and footrest by the subject. Triaxial acceleration was measured on the seat pan structure to quantify the impact exposure. The seat pan structure included three load cells and three load links to measure the vertical and horizontal forces reacted through the structure. Forces were measured in the restraint system using strain gages bonded to the seat attachment hardware or Lebow belt load cells. Leg forces were measured by three triaxial load cells which were incorporated within the rudder pedal support structure.

Triaxial accelerometer arrays were used to measure acceleration on the head and chest of each subject. The chest accelerometer package was held tightly against the subject's sternum by a Velcro chest strap. The subject's head accelerometers were mounted on a dental bite block, which was held in the subject's mouth during the test. This technique has proven to be not only a safe means of providing intraoral/dental protection during impact, but also an effective way of minimizing movement of the accelerometer package relative to the subject's head during impact.

The electronic data obtained from the transducers described above were encoded into pulse code modulation digital format and then transmitted by telemetry to a word formatter. The word formatter reformatted the serial data into parallel data which was transmitted to a PDP 11/34 computer for recording and processing.

Photometric data were collected using two high speed (500 frames per second), 16 mm Milliken cameras mounted on the impact carriage. One camera was mounted to the right of the subject perpendicular to the sagittal plane. The second camera was mounted above the footrest to provide a frontal view of the subject.

The movements of the subject's helmet, head, shoulders, arms, and the chest accelerometer package were quantified by tracking the motion of fiducials attached to these sites. The fiducials which were attached to the subject and to the test fixture consisted of a one-half inch diameter black circle printed on a one inch diameter white target. The locations of the fiducials generally followed the guidelines provided in "Film Analysis Guides for Dynamic Studies of Test Subjects, Recommended Practice" (SAE J138, March 1980). More complete descriptions of the fiducial locations as well as the photometric instrumentation system are provided in Appendices A and D. Timing reference marks were recorded on the 16 mm film once every 0.01 sec. These reference marks were synchronized with the electronic instrumentation recordings.

A video camera was also used to document the tests. This camera and accompanying recorder operate at 120 frames per second with an effective shutter speed of 10 microseconds or less. Use of this system allowed the investigators to evaluate the kinematic response of each subject immediately after each test. This system is described in Appendix A.

Photographs of the test subject and equipment configuration were taken prior to each test. Items of special interest were photographed as required.

D. TEST SUBJECTS

The test dummy used for this program was an Alderson Research Laboratories, Inc., model VIP-95 dummy (serial number 124), which was designed to represent a 95th percentile adult male. The dummy was originally built for $-G_X$ automotive crash testing, based on specifications furnished to Alderson by the National Highway Traffic Safety Administration. It was designed to reproduce the headneck response of human cadavers in forward facing impacts, but was not designed to produce meaningful response dynamics in vertical impacts. This limitation was not a critical factor in the current study, since the dummy was used only to verify the structural integrity of the test apparatus prior to human testing. The dummy's joints were adjusted to a nominal one G value, in accordance with the U.S. Department of Transportation Federal Motor Vehicle Safety Standard No. 208.

All human volunteer subjects who participated in this test program were members of the AFAMRL Impact Acceleration Stress Panel. This panel is composed of volunteer active duty Air Force members whose primary duties do not involve participation as subjects. A total of 18 subjects were utilized during this test program. There were no special technical qualifications or training requirements for subjects. However, all subjects were qualified to participate only after successfully completing an intensive medical screening evaluation (Hearon & Raddin, 1981). This evaluation was directed by the panel physician and consisted of medical history screening, physical examination, visual acuity testing, audiometry, blood pressure measurement, routine laboratory examination (blood work and urinalysis), standard 12-lead electrocardiogram, pulmonary function tests, electroencephalogram, treadmill exercise stress test, and x-rays, including chest, skull, and complete spine films. The x-rays were reviewed by the panel physician in consultation with a radiologist (and orthopedic surgeon, as necessary) to assure elimination of individuals with disqualifying radiographic findings. Female subjects had a negative pregnancy test documented and underwent a pelvic exam by a gynecologist, to assure there were no

gynecologic contraindications to their participation. Relevant abnormalities in any part of the medical evaluation led to elimination of the candidate or specialty consultation and further examination, as required. Annual requalification of panel members was accomplished with a limited medical evaluation, including a physical examination and other relevant medical tests.

The generic human use protocol under which these impact tests were conducted was AFAMRL Protocol No. 80-01, "Generic Impact Acceleration Protocol, 1980". This document presented a survey of available human biodynamic test data, established broad generic exposure limits for human impact testing, and described the generic medical risks associated with such tests. Following review by the AFAMRL Human Use Review Committee (HURC) on 10 January 1980, this protocol was recommended for approval by higher authority. Subsequently, the protocol was approved by AFAMRL/CC and, as SGO R-80-001, it was approved by USAF/SG on 7 March 1980.

The specific human use protocol under which these impact tests were conducted was AFAMRL Protocol No. 80-37, "Evaluation of the Influence of Negative Shoulder Harness Angles in the Operational F/FB-111 Crew Seat and Restraint System During $+G_Z$ Impact Acceleration". This document summarized the operational F/FB-111 ejection data, the previous experience with human impact testing in the F/FB-111 harness, and the specific medical risks associated with the proposed human tests. The overall risk of injury to human subjects was judged to be acceptable when compared to minimizing F/FB-111 crewmember morbidity during emergency escape. This protocol also specified that subjects who had not previously participated in vertical impact tests in the F/FB-111 restraint harness would first be exposed at the orientation 8 G level prior to 10 G exposures. In addition, it was noted that the proposed orientation and test levels were well below accelerations experienced operationally. This human use protocol was reviewed and recommended for approval by the AFAMRL/HURC on 9 October 1980 and was subsequently approved by AFAMRL/CC.

The tests which comprise the C cell of the experimental matrix were conducted from 1 August to 3 September 1980 as part of another F/FB-111 test program (Brinkley et al., 1982). The specific human use protocol under which these tests were conducted was AFAMRL Protocol No. 80-23, "Evaluation of the F/FB-111 Crew Seat and Restraint System Headrest Position", which was reviewed and recommended for approval by AFAMRL/HURC on 26 June 1980 and which was subsequently approved by AFAMRL/CC. Protocols 80-23 and 80-37 were specific protocols submitted under the "Generic Impact Acceleration Protocol, 1980", Protocol 80-01, and as such required local consideration and approval only, in accordance with AFR 169-3, "Use of Human Subjects in Research Development, Test, and Evaluation" (February 1979).

Ongoing informed consent was provided by all subjects during the test program. Prior to testing, subjects received a thorough briefing on the experimental procedures and potential medical risks of participation. The subjects signed a witnessed consent form attesting to the fact that a detailed briefing was received and summarizing its content. Throughout the test program, the medical investigator continued to stress that any subject was free to withdraw at any time for any reason.

Anthropometric measurements of the subjects participating in this test program were made by Dr. Kenneth Kennedy, Mr. Charles Clauser, and Lt Col Maureen Lofberg of AFAMRL/HEG. Table 2 is a summary of selected anthropometric values for each subject. The mean and standard deviation computed from each set of dimensions compare favorably with the mean and standard deviation of the dimensions obtained from an anthropometric survey of USAF personnel conducted in 1967 and published in AFSC Design Handbook 2-2. Forty-nine anthropometric measurements were obtained from each subject. The mean, standard deviation, and range of selected group measurements are listed in Table 3. Weight is expressed in pounds and all other parameters are expressed in inches.

E. EXPERIMENT SEQUENCE

The controlled parameters during this program were the carriage drop height and the seat vertical adjustment. The relation between carriage drop height and resulting acceleration exposure had been well established during preceding test programs in which the same test fixture was utilized (Brinkley et al., 1981, 1982). The seat vertical adjustments for each subject were determined prior to the test program, as described in Section 2A of this report.

The specific parameters for each test were provided to the test conductor and other personnel at the beginning of each day of testing. The conduct of all human exposures was the responsibility of a qualified and experienced test conductor. The test conductor directed the activities of all other personnel in the test area in accordance with a detailed checklist.

The first test of each day was done with an anthropomorphic dummy using the equipment configuration and test level planned for the first human test of the day. If no abnormalities were detected, the test personnel proceeded with preparations for tests with volunteer subjects. High speed motion picture cameras were loaded and mounted on the test fixture. Seat vertical and footrest adjustments were made to obtain the appropriate seat configuration based upon the test plan and the anthropometry of the individual test subject. Video recording equipment was readied to permit immediate review of the test by the investigators. The accelerometer packages were then oriented in their respective reference planes and reference zero values were sampled using the data acquisition system.

Subject preparation was concurrent with preparation of the test fixture and instrumentation. Prior to every impact exposure, each subject provided a brief interval medical history and was physically examined. Emphasis was placed on neck or back symptoms, medications, abnormalities of recent sleep patterns, or recent overindulgence in food or alcoholic beverages. No subject was exposed with symptoms which may have obscured detection of test-related injury or which may have indicated predisposition to such injury.

All subjects wore orange, cut-off, long underwear to allow mounting of camera targets and instrumentation. Male subjects wore athletic supporters. The female subject wore a bathing suit. Each subject was instructed to void prior to entering the test area.

TABLE 2. INDIVIDUAL SUBJECT ANTHROPOMETRY SUMMARY

SUBJECT NUMBER	WEIGHT (1b)	STATURE (in)	SITTING HEIGHT (in)	MID-SHOULDER SITTING HEIGHT (in)
NUMBER D-1 F-3 F-2 F-4 G-3 G-2 H-3 H-5 H-4 K-1 M-2 M-10 M-11 M-13 P-3 R-2 R-3 S-3	(1b) 203 167 159 142 164 117 186 139 192 169 162 140 145 169 198 148 146 167	(in) 73.6 68.6 67.1 67.0 67.1 62.9 73.9 68.5 67.7 67.1 66.1 65.7 69.5 73.0 72.8 68.1 66.2 69.6	(in) 39.7 36.4 37.5 36.4 34.8 33.3 38.0 35.6 37.0 35.7 35.2 36.1 35.7 37.3 39.1 35.9 35.2 36.6	(in) 28.0 25.5 26.3 24.7 25.0 23.2 26.1 24.1 25.7 24.8 24.0 24.8 25.4 26.3 27.7 24.3 23.9 25.6
MEAN	162	68.6	36.4	25.3
STD DEV	22.7	3.02	1.52	1.26

TABLE 3. COLLECTIVE SUBJECT ANTHROPOMETRY SUMMARY

ANTHROPOMETRIC MEASUREMENT	MEAN	STD DEV	RANGE
Weight Stature Cervicale Height Trochanteric Height Tibiale Height Chest Circumference Waist Circumference Buttock Circumference Acromion-Radiale Length Radiale-Stylion Length Sitting Height Mid-Shoulder Sitting Height Buttock-Knee Length Knee Height, Sitting Head Length Head Breadth Head Circumference Hip Breadth, Sitting	161	22.7	117 - 203
	68.6	3.02	62.9 - 73.9
	58.7	2.84	54.4 - 63.8
	35.9	2.29	32.3 - 39.9
	17.6	1.15	15.8 - 20.3
	38.2	2.34	33.7 - 42.7
	33.3	2.64	29.6 - 39.1
	36.0	8.27	35.0 - 42.4
	12.2	2.87	11.7 - 14.4
	9.4	2.89	8.5 - 11.3
	36.4	1.52	33.3 - 39.7
	25.3	1.26	23.2 - 28.0
	23.8	1.24	21.9 - 26.3
	21.4	1.30	18.7 - 23.7
	7.3	1.78	7.2 - 8.2
	6.0	0.21	5.6 - 6.4
	22.4	0.64	21.5 - 23.6
	14.4	0.77	13.2 - 15.9

A disposable dental bite block (made of Optosil placed over a stainless steel frame) was molded for each subject prior to each impact test. An electrically-isolated accelerometer array was then mounted on the metal frame of the bite block. This metal frame was recently modified with a metal arm which extends from the mouth of the subject to permit the mounting of a photometric target (fiducial). This modification was implemented to permit more precise quantification of subject head displacement and to directly couple the electronic and photometric data obtained from a given impact event.

The medical instrumentation of each subject was standardized as follows. Three stick-on EKG electrodes were placed on the subject, one on the upper posterior aspect of each arm and a third on the right lateral chest, sixth intercostal space, mid-axillary line. The snap-on lead from each of these electrodes was plugged into a telemetry transmitter, which, in turn, was strapped to the left upper extremity of the subject. Continuous remote transmission of a single-lead EKG to a portable EKG machine located near the VDT was assured prior to each impact. Sitting and standing tracings were obtained immediately pre-impact (and post-impact) and a continuous tracing was obtained during test countdown and impact. Coincident with EKG recording, pretest (and post-test) sitting and standing blood pressure determinations were made for each subject by the medical technician using a sphygmomanometer. These pressures were recorded on the appropriate EKG tracing.

The subject was then fitted with the appropriate size (medium or large) USAF HGU-26P flight helmet. After mounting the test fixture platform, the subject was asked to exhale and the chest accelerometer array was secured against the chest with a Velcro strap. The subject was then seated in the proper, upright position and, the restraint harness fitted according to the procedure described in the F/FB-111 Technical Order. The inertia reel straps were pretensioned to 14 ± 5 lb and the lap belt straps were pretensioned to 20 ± 5 lb. Stick-on photometric targets were placed on the subject at pre-determined locations and the positions of these targets relative to one another and to targets mounted on the test fixture were measured. Finally, the inertia reel strap angles (relative to a reference horizontal) were measured.

The subject was instructed not to brace with his upper extremities prior to impact. He was instructed to simply rest his folded hands in his lap. However, tensing of the leg muscles by bracing the feet against the rudder pedals was permitted. The subject was also instructed to brace his helmeted head against the headrest, in an attempt to minimize the forward and downward displacement of his head during the impact.

The final pretest activity consisted of documentation of the test configuration by still photographs, measurement of subject blood pressure, evaluation of the electrocardiographic tracing by the medical monitor, and final safety checks of the test equipment and facility by the designated safety monitor. The test carriage was then elevated to an intermediate height while the water brake was filled with water. Finally, the carriage was raised to the specified drop height, the test area was cleared, a countdown was initiated, and the carriage was allowed to fall onto the water brake to produce the desired impact.

The subject was provided with a foot switch which was connected to the control system of the VDT in such a way that the carriage could not be released unless the switch was depressed. In this manner, the subject was required to

consciously provide his ongoing informed consent throughout the immediate preimpact period (including the countdown) until carriage release, in order for the test to proceed. After carriage release, of course, it was no longer possible to abort the test.

A physician monitor, who was responsible for assuring subject safety during testing, was present for each test and reserved the right to cancel any test at any time for any reason. Such reasons may have included a recent history of neck or back strain, pretest pre-syncope, pretest arrhythmia, or any other condition of the subject, equipment, or procedure which was deemed by the monitor to place the subject at undue risk. The medical monitor was provided a finger-operated switch similar in function to the subject's switch. It had to be depressed prior to carriage release in order for the test to proceed. Agreement of both the subject and the medical monitor that the test should proceed was thus assured.

During testing, an ambulance crew was alerted and standing by within one-half mile of the test facility. In addition, emergency medical equipment was arranged in the test area for use by the physician monitor in the event of an emergency. This equipment included a defibrillator, oxygen equipment, intubation equipment, IV solutions and equipment, appropriate emergency drugs, backboard, harness cutters, and bandages.

Following the impact exposure, the subject was released from the harness. The physician monitor assured that the subject was uninjured. Post-test blood pressures and EKG (single-lead) were obtained and a brief post-test physical examination was accomplished. The subject was then provided with contacts to obtain later medical care as required or to ask questions relating to his participation. Impact exposures for each subject occurred no more frequently than once in any five-day period to allow time for detection of any occult injury.

Regarding the chronology of this test program, as previously indicated, all but one of the tests in cell C of the experimental matrix were conducted from 1 August to 3 September 1980, as part of another F/FB-111 test program (Brinkley et al., 1982). The remaining human tests were accomplished between 29 October and $\overline{25}$ November 1980. It was impractical to randomize the presentation of these tests due to the length of time required to change the harness configuration from operational to modified. Therefore, in general, the tests with the operational harness (cells G and J) were conducted first, followed by tests with the modified harness (cell H and one additional C cell test).

Two deviations from the original test plan are noteworthy. In test #471, an 8 G orientation exposure, the inertia reel straps were not rigged as planned, being directed to the operational anchor points rather than to the anchor points designated by the modification. This did not affect subject safety or data analysis, since the data obtained in this test was not utilized in any statistical comparisons. In test #457, the seat vertical adjustment was incorrect (one inch lower than planned). This deviation from the test plan also did not affect subject safety or data analysis. The exposure for that subject was repeated at the proper seat height later in the test program and the data from the latter test was utilized in data analysis.

Section 4

TEST RESULTS AND ANALYSIS

A. PRETEST MEASUREMENTS

The vertical seat adjustment and resulting inertia reel strap angles (measured by inclinometer) for each subject in the various test conditions are indicated in Table 4. The vertical seat height of the test fixture was adjustable in one inch increments only. Therefore, for some subjects in the J cell test condition (operational harness), there was minimal contact or impingement of the inertia reel straps on the lower aspect of the headrest, as shown in Figure 10. (Refer to Table 1 for experimental matrix cell designations.) Contact with the headrest of as much as one-half these straps was permitted. Greater impingement of the inertia reel straps on the headrest (such that the inertia reel strap angle was distorted) resulted in lowering the seat by one increment. When performed, this adjustment generally eliminated the contact between the inertia reel straps and headrest. Measurements made with ten of the subjects (D1, F2, H3, H4, K1, M2, M10, P3, R3, S3) in an F/FB-111 simulator indicated that the seat adjustments investigated experimentally would not have resulted in helmet-canopy contact operationally.

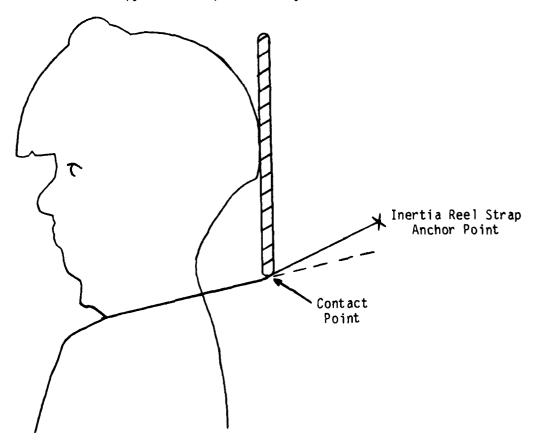


Figure 10. Contact of Inertia Reel Straps with Lower Aspect of Headrest.

Downward adjustment of the seat in the C cell test condition (modified harness) was limited by the requirement for adequate head support. Subjects with relatively small sitting heights were not tested in the full down seat configuration due to inadequate head support. For a given subject, adequate support was assumed to be present when the Frankfort plane intersected the headrest.

These limitations in the vertical seat adjustment (upward in the operational system and downward in the proposed, modified sytem) resulted in a relatively small change in seat height for some subjects from the lowest seat position (z_1) to the highest seat position (z_2). The average change in seat vertical adjustment was 1.82 inches. The largest change was three inches. Three subjects could be elevated only one inch from the "down" to the "up" test condition. Two subjects (Dl and P3), with relatively large sitting heights, could be exposed only in the full down seat position, since a seat elevation of as little as one inch resulted in significant impingement of inertia reel straps on the lower aspect of the headrest. These two subjects, therefore, could not be exposed to test conditions in the H and J cells of the experimental matrix. Also, due to the short duration of this test program and the occasional non-availability of subjects, it was not possible to expose all remaining subjects to all four test conditions. (See Table 4.)

Measured inertia reel strap angles as a function of harness type (operational or modified), vertical seat adjustment (in one inch increments), and seat back angle (90°, 103° , 110°) are given for each subject in Appendix E. (Due to scheduling conflicts, subjects D1 and M2 were not measured in the modified harness.) The inertia reel strap angles of a subject have been found to be a function not only of harness type, seat configuration, and subject mid-shoulder sitting height, but also of subject body habitus and harness adjustment. It should also be noted that measuring inertia reel strap angles by means of an inclinometer is, at best, probably accurate to only $+1^{\circ}$.

B. ELECTRONIC DATA

The electronically measured and computed data obtained during this test program are summarized in Table B-1 in Appendix B. Typical analog data sets from each cell of the experimental matrix and data summaries of each test at the experimental level are also presented in Appendix B.

A statistical analysis of the test results by the Wilcoxon paired-replicate rank test was performed. The means and standard deviations of each parameter in each comparison are summarized in Tables C-1 through C-6 in Appendix C. Statistically significant trends in the measured and computed response parameters for each comparison are summarized in Table 5. The arrow designates a statistically significant change in a parameter at the 90% confidence level for a two-tailed test. The arrow also indicates the direction of the trend from the cell smaller in magnitude. The number indicates the percentage increase in the parameter mean. Typical Wilcoxon computations from each comparison are also presented in Appendix C.

The potential operational significance of these experimentally determined trends should be noted. Among all non-fatal F/FB-111 ejections in a normally functioning crew module, 29.5% of the ejectees incurred vertebral compression fractures (Hearon et al., 1981). In addition to these crewmembers, others were

TABLE 4

MEASURED SEAT ELEVATIONS AND CORRESPONDING

INERTIA REEL STRAP ANGLES FOR EACH SUBJECT

ļ		CELL OF MATRI	Χ
•	C and G	C_	G
SUBJECT	Z 1	ΘC	θ _G
D 1 F 3 F 2 F 4 G 2 H 5 H 4 K 1 M 2 M 10 M 11 M 13 P 3 R 2 R 3 S	0 0 0 2 2 0 1 0 0 2 1	8 17 19 * 12 18 * * * 15.5 15 21.5 14 13 3.5 18.5 14	-4.5 6.5 9.5 0.7.5 4.75 8.5 5.5 7.5 3.0.5 -4.25 8.5 2.25
MEANS STD DEV N =	0.61 0.85 18	14.4 4.67 14	4.24 4.28 18
RANGES	0 to 2	3.5 to 21.5	-4.25 to 9.5

z₁ indicates seat elevation (in inches from full down).

 $[\]theta$ indicates the average inertia reel strap angle (in degrees) obtained by averaging measured left and right inertia reel strap angles.

^{*} Subject was not tested in this cell of the matrix.

TABLE 4 (continued)

MEASURED SEAT ELEVATIONS AND CORRESPONDING

INERTIA REEL STRAP ANGLES FOR EACH SUBJECT

		CELL OF MATRIX	
CELL OF MATRIX	H and J	H	J
SUBJECT	Z 2	θн	θ̈́J
D 1 F 3 F 4 3 G 3 2 H 5 4 H 4 1 M 10 M 11 M 13 P 3 R 3 S	+ 323342*314331+243	† 3.5 5 6 8 7.5 6 * 1.25 10.5 3 3.75 5 8.5 † 9.5 2.75 1	+ -4.25 -1.75 * -2.25 -0.5 -2.25 * -4.5 1 -3.5 3.75 -2.25 0.5 + 3
MEANS STD DEV N = RANGES	2.73 0.96 15 1 to 4	5.42 2.93 15 1 to 10.5	-1.52 2.71 14 -4.5 to 3.75

z₂ indicates seat elevation (in inches from full down).

 $[\]theta$ indicates the average inertia reel strap angle (in degrees) obtained by averaging measured left and right inertia reel strap angles.

^{*} Subject was not tested in this cell of the matrix.

[†] Subject could not be tested in this cell of the matrix due to his large sitting height.

TABLE 5

SUMMARY OF STATISTICALLY SIGNIFICANT TRENDS FROM THE WILCOXON COMPARISONS

AND PERCENT INCREASE IN PARAMETER MEANS

MATRIX CELL F-111 HARNESS SEAT POSITION	C G Mod Oper Down Down (n = 14)	H J Mod Oper Up Up (n = 14)	C J Mod Oper Down Up (n = 12)
CARRIAGE ACCELERATION CARRIAGE VELOCITY SEAT ACCELERATION CHEST ACCELERATION -X axis		> 45	> 2
+X axis +Z axis Resultant CHEST SEVERITY INDEX HFAD ACCELERATION	23 < 17 < 17 <	39 < 8 < 8 <	50 < 19 < 20 < 12 <
-X axis +X axis -Z axis Resultant HEAD SEVERITY INDEX	46 < 7 < 6 <	25 < > 29 4 < 3 <	47 < 4 < 4 <
STRAP LOADS Reflection Straps Inertia Reel Straps Total Shoulder Straps Total Lap Belt Crotch Strap SEAT PAN LOADS	56 < 45 < 54 <-	29 <	75 < 28 < 48 < > 13
-X axis +Z axis Resultant FOOTREST LOADS -X axis +Z axis Resultant		5 <	> 4 > 4 14 <
SEE APPENDIX C	Table C-1	Table C-2	Table C-3

TABLE 5 (continued)

SUMMARY OF STATISTICALLY SIGNIFICANT TRENDS FROM THE WILCOXON COMPARISONS

AND PERCENT INCREASE IN PARAMETER MEANS

MATRIX CELL F-111 HARNESS SEAT POSITION CARRIAGE ACCELERATION CARRIAGE VELOCITY SEAT ACCELERATION CHEST ACCELERATION -X axis +X axis +Z axis Resultant CHEST SEVERITY INDEX HEAD ACCELERATION -X axis +X axis -Z axis Resultant HEAD SEVERITY INDEX STRAP LOADS Reflection Straps Inertia Reel Straps Total Shoulder Straps Total Lap Belt Crotch Strap SEAT PAN LOADS	H G Oper Up Down (n = 15) 5 < 5 < 6 < 23 < 16 < 21 < 15 < 35 <	C	G Oper Oper Down Up (n = 14) > 28 32 < 34> 6 > 25> 15> 20
Total Shoulder Straps Total Lap Belt Crotch Strap	21 < 15 <	31 <	> 15.
Resultant SEE APPENDIX C	Table C-4	Table C-5	Table C-6

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symptomatic, complaining of back pain following ejection. In view of these facts, it may be assumed that other ejectees narrowly averted vertebral fractures during emergency escape. Therefore, altering the current operational restraint in any manner which degrades impact protection may also increase the vertebral injury rate.

The impact test conditions were controlled by utilizing the same carriage plunger for all tests and by maintaining a constant drop height for all tests done at the experimental level. The means and standard deviations of carriage acceleration, seat acceleration, and carriage velocity change are indicated in Table B-1. During this test program, the peak carriage acceleration ranged from 10.2 G to 11.1 G and the peak seat acceleration ranged from 10.0 G to 11.5 G. The Wilcoxon analysis, as expected, revealed no statistically significant differences in these parameters from one test condition to another. (See Table 5.)

However, a similar analysis of the velocity change of the carriage at impact, which ranged from 24.3 ft/sec to 26.3 ft/sec, revealed a statistically significant difference between test conditions on two occasions (comparisons C-J and C-H), as shown in Table 5. Fortunately, these differences represented at most a 2% increase in velocity change and, in all cases, the direction of the increase was opposite to the directions of the trends of the other measured parameters. Therefore, the statistically significant increases in these parameters were observed in spite of the small but statistically significant decreases in carriage velocity. The changes in carriage velocity may be attributed to variations in rail friction on the VDT.

A direct comparison of the operational and the proposed, modified F/FB-111 harness was achieved by comparing the test results from cells C and G and cells H and J of the experimental matrix. In the C-G comparison, the harnesses are compared in the "down" seat adjustment. Resultant accelerations measured at the chest and the head were increased by 17% and 6%, respectively, in the modified harness. Both increases were due to statistically significant increases in both +X and +Z components of acceleration. Note that forward and downward rotation of the head during the impact event results in a portion of the +Z head acceleration being reflected as -X head acceleration. (See also Section 2B.) A statistically significant increase (54%) in total shoulder harness loads (as a result of increases in both inertia reel strap and reflection strap loads) was also demonstrated in the modified harness. The acceleration findings appear to indicate degraded impact protection performance in the modified harness.

In comparison H-J, the harnesses are compared in the "up" seat adjustment. There was an 8% increase in resultant chest acceleration and a 4% increase in resultant head acceleration in the modified harness. The increases were the result of statistically significant increases in the +X and +Z components of these accelerations, respectively. In addition, the chest Severity Index (SI) was increased in the modified harness and the component of chest acceleration in the -X direction (toward the seat back) was increased in the operational harness. These findings are all consistent with the interpretation of degraded impact protection in the modified harness. Although the reflection strap loads are again significantly higher (by 29%) in the modified harness, there is no significant change in inertia reel strap loads or in total shoulder harness loads. This may be related to artificial inflation of the inertia reel strap loads as a result of impingement of these straps on the lower aspect of the headrest for some subjects (6 of 14) in the J cell test condition.

The operational and modified harnesses were also compared along the diagonals of the experimental matrix, ie. comparisons C-J and H-G. (See Tables 1 and 5.) Since the "down" position compared to the "up" position for the modified harness has been previously shown to result in degraded vertical impact protection (Brinkley et al., 1981), and since the "up" position for the operational harness is associated with negative inertia reel strap angles which have been theorized to act in the genesis of vertebral fractures during emergency escape (Kazarian, 1977), the C and J cells of the matrix may provide relatively poor protection in the modified and operational harnesses. It follows that the test conditions in cells H and G may provide relatively better protection in the modified and operational harnesses. Thus, it is reasonable to pursue these comparisons.

In comparison C-J, the +X component, +Z component and resultant accelerations measured at the chest and the head are significantly increased in the modified harness. The computed chest SI is also significantly increased in the modified harness, as are all measured shoulder strap loads (inertia reel straps, reflection straps, and total shoulder strap loads). The acceleration findings are again consistent with the interpretation of degraded impact protection performance of the modified harness.

In addition, the resultant load reacted into the seat pan is significantly greater (4%) in the "up" position (operational harness) than in the "down" position (modified harness). This difference is due to the Z component of the load reacted into the seat pan. It is accompanied by a statistically significant increase in the resultant load reacted at the footrest (due primarily to the X component) in the modified harness. Since these findings were not observed in the previous comparisons of the modified and operational harnesses in which the seat vertical adjustment was held constant, these results may be attributed to the difference in vertical seat adjustment for the cells being compared. As the seat pan is lowered with respect to the fixed footrest, a smaller portion of the thigh is supported by the seat pan. Therefore, the inertial loading of the lower extremities was transferred from the seat pan (in cell J) to the footrest (in cell C). Similar findings related to vertical seat adjustment have been reported in the previous vertical test program of the modified harness (3rinkley et al., 1981).

Essentially the same findings in the C-J comparison are present in the H-G comparison. The +Z axis and resultant chest accelerations, the chest SI, and the -X axis, +Z axis, and resultant head accelerations are all significantly increased in the modified harness. Inertia reel strap, reflection strap, and total shoulder strap loads are all significantly increased in the modified harness. The acceleration findings again indicate degraded performance in the modified harness. Also, the vertical component and the resultant load reacted into the seat pan are significantly greater (3%) in the "up" position (modified harness) than in the "down" position (operational harness). These findings are accompanied by an increase in the X component of the load reacted at the footrest in the operational harness. The findings at the seat pan and footrest may be attributed to the difference in vertical seat adjustment, in the test conditions being compared, as described in the previous paragraph.

In all four comparisons of the operational and modified F/FB-111 harnesses, the resultant accelerations measured at the chest and head and the total loads carried in the reflection straps were significantly increased in the modified harness. The acceleration findings are indicative of degraded impact protection

performance in the modified harness and, furthermore, are independent of vertical seat adjustment. Similar trends would be expected in comparisons of test results obtained at intermediate vertical seat adjustments. Finally, these findings are not entirely unexpected, considering the large change in the restraint geometry in the proposed modification, as a result of relocating the reflection strap anchor points well aft and above the present locations of these anchor points on the seat back.

In comparison C-H, the modified harness in the "down" position is compared to the modified harness in the "up" position. The +Z component and the resultant chest acceleration, the -X component head acceleration, and the inertia reel strap, reflection strap, and total shoulder strap loads were all significantly increased in the "down" position. The acceleration findings are consistent with degraded impact protection performance of the proposed, modified harness as the vertical seat adjustment is lowered, thereby increasing the inertia reel strap angle.

These findings of the C-H comparison are at variance with the previously reported findings of similar vertical impact tests of the modified harness (Brinkley et al., 1981). In particular, the finding of increased Z component and resultant head acceleration at the lower seat adjustment in the previous study of the modified harness (the primary test result cited as being indicative of degraded vertical impact protection performance) was not demonstrated in the present study. On the other hand, an adverse change in chest acceleration was observed in this study, but was not demonstrated previously.

The variance in these two sets of experimental findings are apparently related to two important differences in the test conditions of the two studies. In the original evaluation, the range of vertical seat adjustment was, on the average, broader than the range in the present study. This is true because the "up" condition in the first test program was defined as the vertical seat adjustment at which a specific subject had a 00 inertia reel strap angle. The actual mean inertia reel strap angle for all subjects tested in that condition was +2.590. However, in this test program, the "up" condition for the modified harness was determined by the vertical seat adjustment in the operational harness (for the J cell). The actual mean inertia reel strap angle for all subjects tested in the comparable test condition in this study (H cell) was +5.42°. The "down" test conditions were specified in the same manner in both studies. It is conceivable, therefore, that the broader range of vertical seat adjustment between the test conditions investigated in the previous test program may have contributed to the statistically significant increase in resultant head acceleration reported in that study.

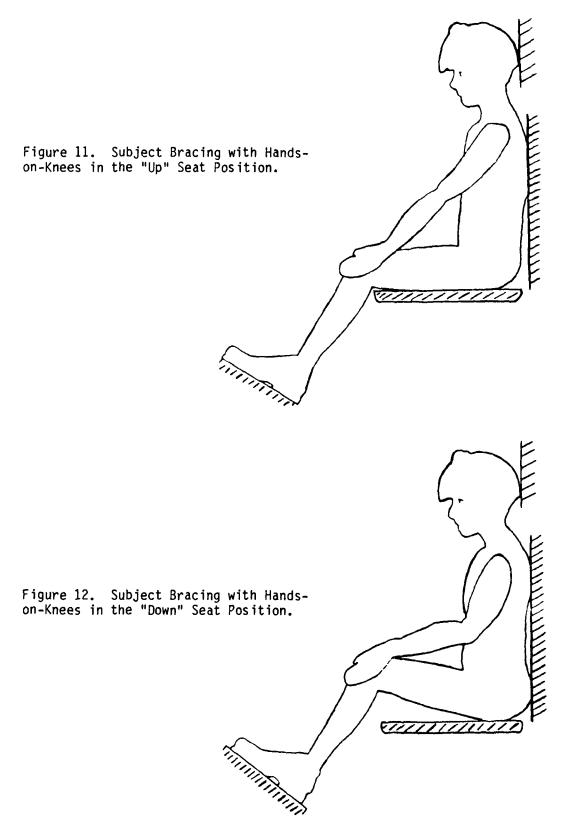
It is also likely that the observed differences in test results may be attributed to the different upper extremity bracing techniques utilized by the subjects in the two test programs. During the original evaluation of the modified harness, in addition to bracing their helmeted heads against the headrest and their feet against the rudder pedals, the subjects were instructed to extend their upper extremities and to brace their hands against their anterior thighs or knees, as shown in Figure 11. The influence of this "hands-on-knees" bracing technique on human response during 16, impacts in the modified F/FB-111 harness and a standard USAF double show decreaped to be left configuration has been measured experimentally and has the extended by Brinkley et al. (1932). In general, this bracing technique increases had accepted for the passes chest

acceleration, increases loads reacted at the footrest, and decreases loads reacted at the seat pan, compared to a posture which precludes upper extremity bracing (eg., hands-in-lap). Findings similar to these were reported during the original vertical tests of the modified harness as the seat vertical adjustment was lowered from the "up" to the "down" position. In retrospect, these findings may be at least partially attributed to less effective upper extremity bracing in the lower seat position, and not solely based on the change in vertical seat height. This is true because lowering the seat pan may raise the knees of the subject relative to the seat pan, thereby making it more difficult for the subject to fully extend his arms and brace effectively. (See and compare Figures 11 and 12.)

In the present study, the influence of upper extremity bracing was eliminated by instructing the subjects to simply fold their hands in their laps, as shown in Figure 13. The findings in the C-H comparison reflect the absence of hands-on-knees bracing and are consistent with results expected for the hands-in-lap posture and observed in earlier comparisons. Finally, it is not clear why expected changes in seat pan and footrest loads with variation in vertical seat adjustment in the two test conditions are not observed. It may be that the relatively small vertical changes were not sufficient to demonstrate the trends.

The final comparison, G-J, concerns only the operational harness. Statistically significant increases are seen in resultant head acceleration and head Severity Index as the seat vertical adjustment is increased. (See Section 5A for analysis of this finding.) Again the inertia reel strap and total shoulder harness loads are increased in the J cell. However, this finding may be artifactual, since there was inertia reel strap impingement on the lower aspect of the headrest for some subjects in J. The Z component and resultant seat pan load were increased in the "up" condition, while the X component footrest load was increased in the "down" condition. These tandem findings, like those reported in the C-J and H-G comparisons, appear to be related to the amount of thigh support provided by the seat pan in each condition. There was no significant change in the reflection strap loads, which suggests that, as expected, the function of this strap is relatively unaffected by vertical seat adjustment.

The observed correlations are summarized in Table 5. Further discussion of test results and the implications of these results is presented in Section 5A.



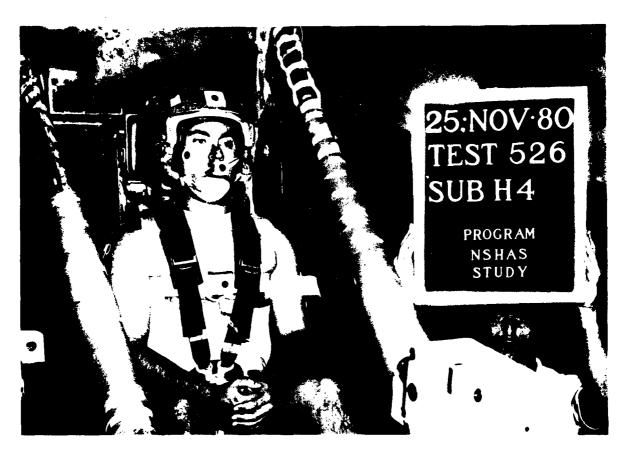


Figure 13. Hands-in-Lap Position.

C. PHOTOMETRIC DATA

The lateral photometric data obtained in this test series were processed to obtain the displacement-time, velocity-time, and acceleration-time histories of various photometric targets. In addition, the X-Z trajectories of these targets were analyzed. These data were utilized to (1) confirm the presence of subject head rotation during the impact, (2) demonstrate subject-specificity in the X-Z trajectories of the photometric targets, (3) verify subject headstrikes suggested by the electronic data, and (4) analyze the subject head motion associated with various head acceleration waveforms. (See also Section 4D.) Typical photometric data obtained in this test program are presented in Appendix D.

D. MEDICAL FINDINGS

The findings of 67 human impact tests are presented in this report. Thirteen of these exposures were at 10 G in the C cell of the experimental matrix and were originally conducted as part of another test program (Brinkley et al., 1982). Five of the tests were 8 G orientation exposures and the remaining tests were at the 10 G experimental level. Noteworthy medical findings were annotated immediately following each experiment by one of the two physicians who participated as medical monitors during this test program.

Three subjects reported transient paracervical pain during the impact, but no residual discomfort following the event. Two subjects incurred mild paracervical muscle strains, one requiring approximately 10 days to resolve. Another subject incurred a mild trapezius muscle strain. No contusions or abrasions were documented. The medical findings noted above were considered to be of no clinical consequence. Thus, the test conditions investigated in this program were considered to be well within human tolerance. No subject attrition was experienced during the testing.

Noteworthy is the observation that the X component of head acceleration was, in general, a characteristic waveform for each subject. This electronically measured parameter, of course, reflects both translational and rotational acceleration, as previously described (Section 2B). Nevertheless, analysis of the waveform patterns is instructive. Three basic patterns are apparent, as shown in Figure 14. The majority of subjects typically exhibited a biphasic waveform (positive acceleration followed by negative acceleration). Relatively few subjects exhibited either positive (primarily +X acceleration) or negative (primarily -X acceleration) waveform patterns.

The head motions associated with these waveforms may be appreciated by analysis of correlated photometric data. In Figures 15 and 16, the X and Z head displacement (measured at the subject's cheek) is plotted as a function of time for three subjects with typical waveforms. Initially, the X displacement increases (as the head moves forward) and the Z displacement decreases (as the head moves downward) at approximately the same rates in all three cases. However, each subject regains control of his head motion during the inertial response differently.

In the negative waveform case, the X and Z displacements are essentially held constant after the inflection points N and N', respectively. The subject, therefore, simply maintains a fixed head position after regaining control of his head motion. This pattern was observed in subjects who did not effectively brace their head against the headrest, those who simply forgot to brace, or those who "lost" their brace during descent of the test carriage.

In the biphasic case, the X displacement decreases and the Z displacement increases after the inflection points B and B', respectively. More effective contraction of the posterior paracervical muscles by this subject, as he regains control of his head motion, tends to return his head to its initial position. This pattern was observed in the majority of subjects, ie., those who were able to effectively brace their heads against the headrest.

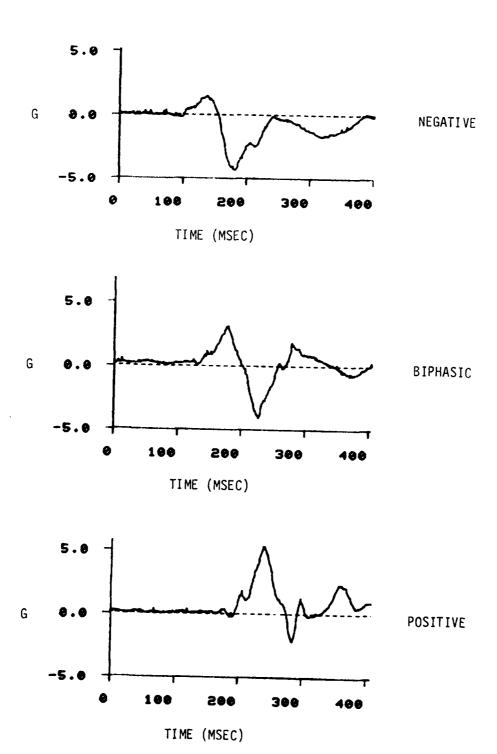
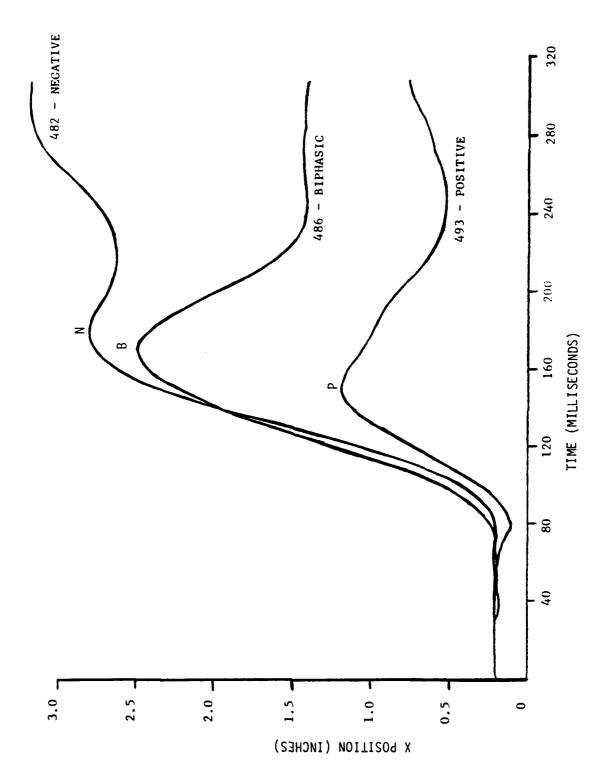


Figure 14. Head X Acceleration Waveforms.



The second second

Head X Position as a Function of Time for Three Subjects with Different Head Acceleration Waveforms. Fig. 15.

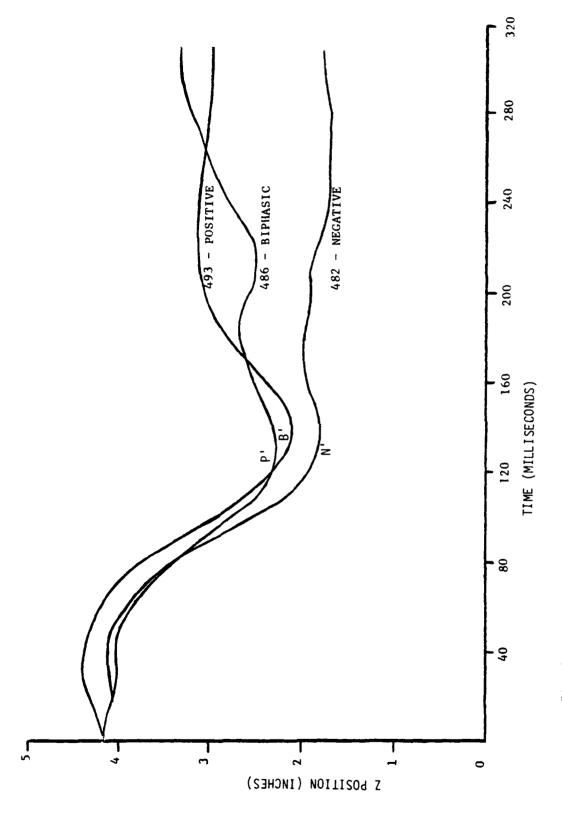


Fig. 16. Head Z Position as a Function of Time for Three Subjects with Different Head Acceleration Waveforms.

Finally, in the positive waveform case, the X displacement decreases and the Z displacement increases after the inflection points P and P', respectively. Very effective neck muscle contraction by this subject tends to return his head to its initial position earlier and more effectively than in the biphasic case. In fact, the photometric data from this particular test (#493) indicate that the subject's helmeted head was in contact with the headrest during the latter portion of the event. (Note the X-Z trajectory of the upper helmet fiducial shown in Figure 17.) The relatively small horizontal displacement of the head during this inertial response resulted in a hyperextension of the cervical spine and paracervical musculature. At this experimental impact level, the outcome was a muscle strain which resolved in approximately ten days. At operational impact levels, a more severe muscle strain under these circumstances would be likely. However, more serious cervical spine injury is considered to be unlikely, in view of the absence of such injury among survivors in the operational ejection experience (Hearon et al., 1981, 1982).

The frequency of occurrence of the three waveforms described above is indicated for each subject in Table 6. In most cases, a single pattern was characteristic for each subject. This may be a reflection of the biological variability among subjects and underscores the desirability of utilizing each subject as his own control when comparing inertial responses. Such biological variability is also demonstrated by characteristic or subject-specific X-Z trajectories of photometric targets mounted at various anatomic landmarks.

Headstrikes of the helmet on the headrest were observed during this test program. These were indicated electronically by positive spikes in the X component of head acceleration and were verified by correlation with the photometric data. The amplitudes of these spikes were relatively small, the largest being approximately 6 G. Subjects did not report being aware of significant helmet-headrest contact during impact. The frequency of these headstrikes is also indicated for each subject in Table 6. The tendency of some subjects to experience headstrikes is likely another evidence of biological variability or voluntary procedural variation among subjects.

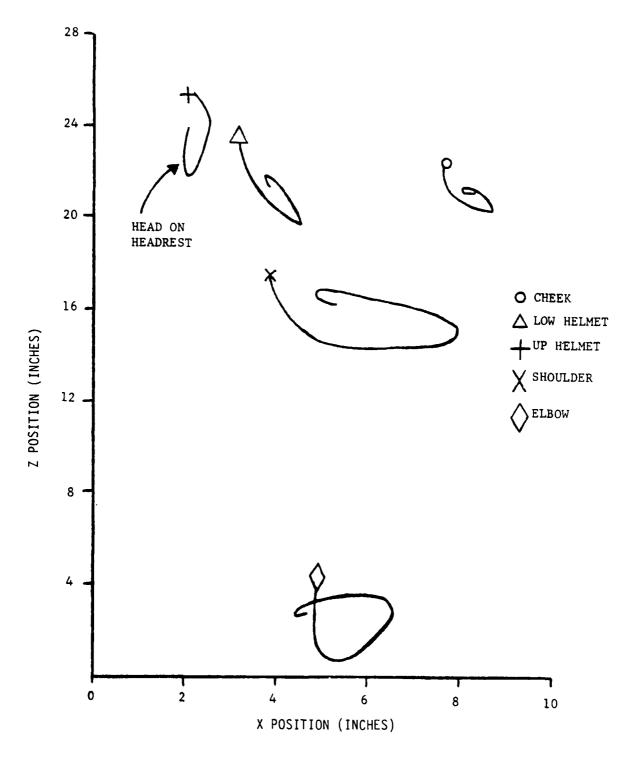


Figure 17. X-Z Trajectories of Photometric Targets in Test #493.

TABLE 6

HEAD X ACCELERATION WAVEFORMS AND

NUMBER OF HEADSTRIKES FOR EACH SUBJECT

		WAVEFORMS		TOTAL	TOTAL
SUBJECT	NEGATIVE	BIPHASIC	POSITIVE	EXPOSURES	HEADSTRIKES
D-1 F-3 F-2 F-4 G-3 G-2 H-5 H-4 K-1 M-10 M-13 P-3 R-2 R-3 S-3	1 1 0 2 1 4 0 0 0 4 4 0 1 2 2 4 3 1	1 3 4 0 3 0 3 1 0 0 0 4 3 2 0 0 1 3	00000003000000000	2 4 4 2 4 4 3 1 3 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	2 1 4 0 0 0 0 0 0 0 0 4 3 1 2 4 1 0

Section 5

DISCUSSION

A. IMPLICATIONS OF TEST RESULTS

The proposed modification to the F/FB-111 restraint system was based on an injury mechanism assessment which theorized that vertebral fractures among ejectees were the result of negative inertia reel strap angles in the operational harness (Kazarian, 1977). During powered inertia reel retraction in the presence of negative inertia reel strap angles, the direction of force application at the crewmember's shoulders is downward and medial. This unconventional shoulder strap geometry was theorized to be the primary etiologic factor in the vertebral fractures occurring during retraction (hyperextension injuries) and also was implicated as a contributing factor in the genesis of the landing impact fractures (hyperflexion injuries) as well. In fact, concern that negative inertia reel strap angles were causative in the operational back injuries was so great that no human volunteer subjects were exposed with negative angles in the initial impact tests of the modified harness (Brinkley et al., 1981). However, that concern was mitigated when the original contention was analyzed more carefully and when several plausible alternate injury mechanisms, unrelated to inertia reel strap angles, were identified during the initial testing. Finally, that concern was further mitigated by the findings of a more recent F/FB-111 vertebral injury analysis (Hearon et al., 1981, 1982). This human test program was justified from a medical risk standpoint on the basis of the results of this later analysis. (See Section 5B.)

Previous evaluation of the F/FB-111 harness by Dr. Kennedy of AFAMRL/HEG has revealed that the critical anthropometric dimensions which determine the inertia reel strap angle geometry for a given subject are sitting height and midshoulder sitting height. As a result of the pretest measurement of strap angles accomplished during this test program, this geometry also appears to be a function of a subject's individual body habitus and the manner in which the harness is adjusted. For example, in Table 7, two subjects matched for sitting height and mid-shoulder sitting height were found to have different inertia reel strap angles at the same seat elevation. The shoulder harness geometry is, of course, also a function of seat adjustment. The inertia reel strap angle increases as the seat is lowered and as the seat back is reclined. In Appendix E, this variation is documented for each subject in both the operational and modified harnesses.

TABLE 7

MEASURED INERTIA REEL STRAP ANGLES IN SUBJECTS WITH SIMILAR ANTHROPOMETRY

Subject ID	K-I	G-3
Sitting Height (cm)	90.6	88.5
Mid-Shoulder Height (cm)	63.0	63.5
Seat Vertical Position (in)	2	2
Right Inertia Reel Strap Angle (Degrees)	-3.5	+2.5
Left Inertia Reel Strap Angle (Degrees)	-3	+2

In the operational horness, apward vertical seat significant in a decleration the largest negative inertia reel strap angle for a given subject was limited by contact of these straps with the lower aspect of the headrest, as shown in Figure 4. Strap "contact" with the headrest was defined as minimal headrest contact with, at most, one-half of the inboard aspect of either or both inertia reel straps. Strap "impingement" was defined as significant headrest contact, such that the path of either or both inertia reel straps was grossly distorted. Strap contact in test condition J was allowed. Strap impingement, however, was not accepted and was eliminated by lowering the vertical seat adjustment one inch. Utilizing these acceptance criteria, the largest negative inertia reel strap angle (average of measured right and left strap angles) for any subject in test condition J, was found to be -4.5°. (See Table 4.) The largest negative angle (with strap contact) documented in the pretest measurements was -8° in the 90° seat back angle configuration and -12° in the 110° seat back angle configuration. (See Appendix E.)

The relatively small inertia reel strap loads (mean = 258 lb at -65° F and 13 inches retraction distance) which occur during inertia reel retraction (Whitney & Haas, 1970) appear to be insufficient to cause vertebral fracture, regardless of the inertia reel strap angle. This impression is supported by the AFAMRL experience with live human retractions using the Body Positioning and Restraint Device (BPRD). Subjects retracted from forward-leaning positions in 150 msec by forces up to 358 lb have indicated no untoward effects and, in fact, have generally reported the experience as benign (Raddin et al., 1979). Operationally, in the F/FB-111, for seat adjustments beyond the impingement point, significant frictional losses would be expected, resulting in slower, more benign retractions than those experienced without such impingement. Furthermore, higher seat positions would result in contact of the helmet with the canopy during flight, and would therefore not be used. It is problematic, therefore, that the presence of negative inertia reel strap angles causes vertebral injury during inertia reel retraction.

The human impact test series documented in this report was designed to assess the influence of negative inertia reel strap angles in the operational harness on inertial responses during ${}^+\!G_Z$ acceleration, such as that experienced during landing impact. The initial impact tests of the modified harness revealed changes in the performance of the harness with changes in vertical seat adjustment. Specifically, in the 90° seat back angle condition, the resultant head acceleration and the head Severity Index were increased with increasing inertia reel strap angles (ie., as the seat was lowered). These results naturally led to speculation regarding the outcome of a similar comparison of the operational harness. Statistically significant trends established by analysis of the electronic test results are revealing in this regard. (See Table 5.)

The pertinent comparison is G-J, the results of which have been described in Section 4B. The resultant head acceleration and head Severity Index were found to be significantly greater in the J condition (that associated with negative inertia reel strap angles in the operational harness). The chest acceleration data, however, indicated improved performance in the J condition, since the chest acceleration in the -X direction (into the seat) was significantly higher and the chest acceleration in the +X direction (away from the seat) was significantly lower in that cell than in the G condition. In this comparison, the evidence that negative inertia reel strap angles adversely influence inertial response during vertical impacts is conflicting. However, if a problem exists

here, it apparently cannot be rectified by the proposed modification since, in comparisons with the operational harness, the modification allowed both higher resultant head accelerations and higher +X (forward) chest accelerations for similar seat positions.

A final observation regarding shoulder harness geometry is noteworthy. The importance of the influence of inertia reel strap angles (as determined by vertical seat adjustment) relative to the type of F/FB-111 harness (operational or modified) utilized may be assessed by examination of the statistically significant trends in the six comparisons in Table 5. Comparisons C-G and H-J, in which the modified harness was directly compared to the operational harness at the same seat vertical adjustment, revealed statistically significant increases in resultant chest and head accelerations in the modified harness. On the other hand, comparisons C-H and G-J revealed that lowering the seat in either harness adversely affects measured chest acceleration. In general, then, these four comparisons indicated some evidence of degraded performance as the seat was lowered for either harness alone and for the modified harness compared to the operational harness at the same seat vertical adjustment.

Examination of the remaining two comparisons in Table 5 revealed, as expected, degraded performance when the modified harness in the "down" condition (C cell) was compared to the operational harness in the "up" condition (J cell). This was evidenced by statistically significant increases in both chest and head accelerations in the C cell. However, in the H-G comparison, when the modified harness in the "up" condition was compared to the operational harness in the "down" condition, the adverse trends in chest and head acceleration were correlated with the modified harness rather than the "down" vertical seat adjustment. Therefore, it appears that the adverse influence on performance due to the modified harness itself is more significant than the influence of vertical seat adjustment.

The statistically significant trends summarized in Table 5 consistently indicate degraded performance in the modified harness, as evidenced by increases in resultant chest and head accelerations and increases in the chest Severity Index. The relative magnitudes of these increases in the various comparisons are also shown. These indications that human inertial response in the modified harness is more severe than in the operational harness are disturbing, particularly in the face of the relatively high vertebral injury rate (29.5%) experienced operationally. When the injury rate associated with a mechanical force environment is high, a reasonably safe presumption is that subclinical vertebral injuries or near-injuries, in addition to those diagnosed vertebral fractures, are also occurring. The operational ejection data confirm that not all ejectees who experienced back pain as the result of emergency escape were found to have vertebral fractures (Hearon, 1981). Those crewmembers, who were diagnosed as having paravertebral muscle strains, in fact, may have narrowly averted a vertebral fracture. Any factor shown to degrade impact protection performance at the experimental level, therefore, may adversely influence the injury rate operationally by increasing the severity of the inertial response.

Despite the consistent trends observed in the resultant chest and head accelerations, the vertical and resultant loads reacted at the seat pan were not increased when the modified harness was compared to the operational at the same vertical seat adjustment. (See comparisons C-G and H-J.) This observation does

not mitigate the aforementioned acceleration findings, which indicate some level of degraded performance in the modified harness.

A final observation concerns the shoulder strap loading. In the modified harness, the anchor points of both inertia reel straps and reflection straps are relocated upward and the reflection strap anchor points are moved well medial and aft of their locations on the seat back in the operational harness. This results in a statistically significant increase in inertia reel strap and reflection strap loading in the modified harness. (See comparisons C-G, C-J, and H-G.) The operational reflection straps carry significantly less load because of the change in strap arrangements and load application points. However, they apparently provide more effective upper torso restraint in vertical impacts, as evidenced by the changes observed in chest and head accelerations. The importance of the operational reflection straps in providing lateral restraint during sideward impact has been previously documented (Brinkley et al., 1981).

B. FUTURE CONSIDERATIONS

This comparison of the operational and modified F/FB-111 harnesses was limited in that the assessment was confined to the vertical axis and four test conditions. The conditions selected for evaluation, however, as previously noted in Section 2A, were believed to be appropriate for several reasons. For example, the vertical axis was chosen for investigation because the Z component of resultant module acceleration on landing impact exceeds both X and Y components of acceleration (Brinkley et al., 1981) and because excessive vertical loading is now believed to be of considerable importance in the etiology of the vertebral fractures observed operationally (Hearon et al., 1982). Also, the 90° seat back angle condition was selected for evaluation because human response in the modified harness was shown to be more severe in that seat configuration than in reclined seat positions (Brinkley et al., 1981). The results of this limited test program, in addition to previous evidence of degraded lateral impact protection in the modified harness (Brinkley et al., 1981), provide the basis for our present recommendation not to pursue further comparative impact testing with the currently proposed modification and the operational harness.

On the basis of the foregoing test results and discussion, the currently proposed modification to the F/FB-111 crew seat and restraint system is not recommended for implementation. In the face of a restraint system which departs from standard design practice in several areas, as the F/FB-111 restraint does, it would be unlikely that the operational vertebral injury rate could be significantly improved by the simple expedient of partially eliminating negative inertia reel strap angles. Future proposals for modification of this restraint should thoroughly address mechanisms by which undesirable forces may be imposed on the crewmember. Numerous unconventional design features of the operational harness which may contribute to such adverse loading have been identified and discussed elsewhere (Brinkley et al., 1981). These include the shoulder harness yoke, the crossing reflection straps, the location of the headrest forward of the plane of the seat back, and the wide latitude for seat adjustment, including the independent motion of the seat with respect to the headrest.

For the present, it is recommended that a "buddy check" procedural change be adopted. Following harness and seat adjustment prior to take-off, each F/FB-111 crewmember should be instructed to verify that the inertia reel straps of the other crewmember are not in contact with the headrest. Downward seat adjustment is recommended if such contact is present, so long as adequate over-the-nose vision is maintained. In the event of ejection initiation in the presence of such contact, the resulting increased drag on the inertia reel straps could be sufficient to slow retraction, or in extreme cases, to stall the inertia reel and preclude retraction prior to ejection. This situation could thereby predispose an ejectee to a back injury during module separation or landing. Several crewmembers have, in fact, testified that they were not adequately retracted following initiation of the ejection sequence and, consequently, experienced forward flexion during module separation. Two of these crewmembers incurred vertebral fractures during retraction-ejection and a third may have incurred a vertebral fracture during this phase of the escape (Hearon et al., 1981; Hearon, 1981). Reasons for these apparent inertia reel failures were not delineated in the accident investigation reports. Further restriction of the range of seat adjustment for a specific anthropometrically-determined subset of flyers does not appear to be necessary, since there is no compelling evidence of performance degradation in the operational harness with negative inertia reel strap angles, as shown by the results of comparison G-J.

Any future modification to the F/FB-111 escape system for the purpose of reducing the vertebral injury rate should be based on a comprehensive aeromedical analysis of these injuries. Furthermore, design changes should be based on well established design practice or, if based on hypotheses, the changes should first be tested to verify the hypotheses. The importance of this approach cannot be overstated.

The findings of the most recent vertebral injury analysis (Hearon et al., 1981) are at variance with the earlier analysis (Kazarian, 1977; Kazarian et al., 1979). The salient differences between the vertebral injury analyses are shown in Table 8. The reasons for the marked disparity between the two assessments are described in more detail elsewhere (Hearon et al., 1982). It is apparent that the original review overestimated the magnitude of the vertebral injury problem and attributed a disproportionate number of vertebral fractures to the retraction phase of the escape sequence. Furthermore, the analysis attributed these retraction injuries to a presumed hyperextension injury mechanism and stated that negative inertia reel strap angles were the specific design feature of the operational harness responsible for this injury mechanism. Reassessment of the operational data, experience measuring inertia reel strap angles experimentally during this test program, and inertia reel retraction force data which showed relatively low force levels indicated that this contention was unlikely.

In addition, the re-examination of the operational ejection data revealed only two cases in which the F/FB-111 crew seat and restraint system itself was implicated as a cause or possible cause of a vertebral injury. One injury was incurred by a crewmember with a relatively large sitting height (37.75 inches) and was attributed to contact of his shoulders with the lower aspect of the headrest during retraction. The potential location of the plane of the headrest up to $2\frac{1}{4}$ inches forward of the plane of the seat back may have been a factor in the etiology of this particular injury. Another crewmember may have been injured by the so-called "horsecollar" mechanism, which involves an initial maladjustment of the restraint harness by the crewmember and which has been

described elsewhere (Brinkley et al., 1981). The unconventional design features of the restraint system to which these injuries have been attributed are not addressed by the proposed modification.

TABLE 8

COMPARISON OF F/FB-111 VERTEBRAL INJURY ANALYSES

	INITIAL ASSESSMENT (1977)	REASSESSMENT (1981)
OVERALL VERTEBRAL FRACTURE RATE AMONG SURVIVORS	40.3% (25 of 62)	31.3% (25 of 80)
RETRACTION-EJECTION VERTEBRAL FRACTURE RATE	29.0% (18 of 62)	11.3% (9 of 80)
MECHANISM RESPONSIBLE FOR MAJORITY OF FRACTURES	Hyperextension or combined hyperextension-hyperflexion	Axial compression- flexion
PHASE OF ESCAPE DURING WHICH MAJORITY OF FRACTURES OCCURRED	Retraction	Landing Impact
FRACTURES CORRELATED WITH NEGA- TIVE INERTIA REEL STRAP ANGLES	Yes	No

At least 11 of 23 injured crewmembers sustained vertebral fractures during landing impact of the crew module and several other crewmembers also may have been injured during this phase of the escape sequence. Crewmember testimony indicates that many of these ejectees were asymptomatic during the retraction-ejection and module descent phases of the escape and experienced the onset of back pain with landing impact. In addition, the spinal radiographic findings strongly implicate axial compression and hyperflexion (not hyperextension) as the primary causative mechanisms in these fractures (Hearon et al., 1981). The available evidence, therefore, indicates that the observed vertebral fractures in this population are frequently the result of excessive vertical loading during landing impact.

Over a decade ago, the severity of landing impact in the F/FB-111 crew module was recognized as a potential cause of crewmember vertebral injury. In fact, several concepts for modifying the crew module to achieve greater impact attenuation on module landing were considered (Johnson, 1968). However, none of these proposals were pursued.

To date, the operational ejection data indicate that the anticipated potential for injury on landing impact has been realized. Futhermore, it is likely that a significant reduction in the F/FB-111 spinal injury rate may be achieved only by decreasing the acceleration stresses imposed on the crewmember during landing impact (Brinkley et al., 1981; Hearon et al., 1981, 1982). A re-evaluation of the options available for improving impact attenuation on module landing should be pursued in future redesign considerations.

Finally, as an encapsulated escape system, the F/FB-111 aircraft provides the crewmember excellent protection against flail injuries due to windblast, even in the event of canopy loss from birdstrike. Such protection is becoming more important in view of increasing airspeeds at ejection. Windblast protection must continue to hold a high priority in the design of future escape systems and in the performance evaluation of current systems. In this regard, the absence of extremity flail injuries and the relative clinical benignity of the majority of F/FB-111 vertebral fractures should be carefully considered when comparing F/FB-111 performance to that of open ejection seat escape systems.

Section 6

SUMMARY

A. PROGRAM OBJECTIVES

This test program was designed to achieve the following objectives (Section 1B).

- 1. Establish the range of inertia reel strap angles for a population of volunteer subjects in the proposed modified and the operational F/FB-111 restraint harnesses as a function of seat adjustment.
- 2. Comparatively evaluate human inertial response in the modified and operational harnesses during vertical impact accelerations.
- 3. Obtain human impact data for use in the development of current and future mathematical models intended to predict human inertial response to impact.

B. TEST PROGRAM

- 1. A factorial experimental design was utilized to elucidate the effects of F/FB-111 harness type (modified or operational) and vertical seat adjustment (and, in turn, inertia reel strap angle) variation. The chosen test conditions investigated the 90° seat back angle position at the extremes of vertical seat adjustment (Section 2A).
- 2. The Vertical Deceleration Tower (VDT) was utilized to provide nominal experimental $+G_Z$ impacts of 10 G (26 ft/sec). (Section 3A).
- 3. The operational crew seat and restraint harness was salvaged from a F/FB-111 crew module. The proposed modification test item was provided by General Dynamics. Instrumentation was provided by AFAMRL (Sections 3B and 3C).
- 4. Human volunteer subjects were medically qualified and utilized in accordance with applicable human use regulations (Section 3D).
- 5. Relevant accelerations, forces, and loads were measured electronically. Appropriate physiological data were obtained. Subject motion was documented by high speed cameras (Sections 3C, 3D, and 3E).
- 6. A total of 54 human tests was conducted between 29 October and 25 November 1980. The results of 13 additional comparable human impact tests conducted during a previous test program (Brinkley et al., 1982) are also reported. (Sections 3D and 3E).
- 7. The Wilcoxon paired-replicate rank test was utilized in data analysis to establish the statistical significance of test results (Section 2B).

C. TEST RESULTS

- 1. All accelerations, forces, and loads measured at these impact levels in these test conditions were considered to be well within human tolerance (Section 4D).
- 2. Upward vertical seat adjustment in the operational harness was limited by contact of the inertia reel straps with the lower aspect of the headrest (Section 4A).
- 3. Measured inertia reel strap angles were found to be subject-specific. Measured inertia reel strap angles were not less than -12° in the static measurements taken over the entire range of seat adjustment and were not less than -4.5° (average) for the 90° seat back adjustment condition during any impact test (Section 4A).
- 4. Resultant head and chest accelerations were significantly greater in the modified harness than in the operational harness, regardless of seat vertical adjustment (and, in turn, inertia reel strap angles). (Section 4B).
- 5. There was no significant difference in seat pan loads when the modified harness was compared to the operational harness at the same seat elevation (Section 4B).
- 6. Resultant head acceleration in the operational harness increased with increasing seat vertical adjustment. However, resultant chest acceleration in the operational harness was independent of vertical seat adjustment (and, in turn, inertia reel strap angles). (Section 4B).
- 7. Evaluation of the modified harness at different seat elevations revealed statistically significant trends at variance with those reported in the initial test program (Brinkley et al., 1981). However, performance was still found to be degraded at the lower seat elevation. The differences in test results are attributable to variations in test conditions (including subject bracing) between the two studies (Section 48).
- 8. Biological variabilty among subjects was evidenced by characteristic (subject-specific) X component head acceleration waveforms (Section 4.D).

D. RECOMMENDATIONS

- 1. The current proposed modification to the F/FB-111 crew seat and restraint is not recommended for implementation. This recommendation is based on the degraded vertical impact protection performance of the modified harness compared to the operational harness documented in this study, the previous test program (Brinkley et al., 1981), and the findings of the most recent (Hearon et al., 1981) vertebral injury analysis (Sections 5A and 5B). (This recommendation has been accepted. The Engineering Change Proposal for the modified harness was cancelled at a Configuration Control Board in April 1981.)
- 2. Further human impact tests of the currently proposed modification are not recommended (Section 5B).

- 3. F/FB-111 crewmembers should be instructed to avoid vertical seat adjustments which produce contact of the inertia reel straps with the lower aspect of the headrest.
- 4. Future restraint harness modification proposals should consider all unconventional design features of the operational harness and should address all the mechanisms by which adverse loads may be imposed on the ejecter (Section 5B).
- 5. Future redesign efforts of the F/FB-111 escape system should be based on the findings of the most recent vertebral injury analysis (Hearon et al., 1981) and, therefore, should address landing impact attenuation as well as design variances in the crew seat (Section 5B).

APPENDIX A DATA ACQUISITION EQUIPMENT AND METHODS

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Wesley M. Waldron

Dynalectron Corporation
Scientific Services Division

INTRODUCTION

Under Contract F33615-79-C-0523, Dynalectron was requested by the Air Force Aerospace Medical Research Laboratory/Biomechanical Protection Branch to instrument a test fixture fabricated by General Dynamics Corporation and to collect impact test data in the comparative evaluation of the operational and proposed, modified F/FB-111 crew seat and restraint systems. The testing was conducted in one axis of acceleration on the Vertical Deceleration Tower Test Facility located at the Air Force Aerospace Medical Research Laboratory, Building 824, Area B, Wright-Patterson Air Force Base. The following is a discussion of the equipment and techniques used in acquiring and processing data that describes the kinematic and inertial responses of the human body. Installation and sensor specifications are also included in the discussion.

DATA MEASUREMENT DEVICES

This evaluation program was instrumented using thirty-seven transducers. The Digital Instrumentation Requirements sheets of Figures A-1 through A-3 contain the pertinent data for each channel.

SUBJECT INSTRUMENTATION

Each subject was instrumented with six accelerometers. These accelerometers were configured in groups of three to create two triaxial measuring packages. Each package was mounted to indicate accelerations in the X, Y and Z axes. Figure A-4 shows the coordinate system utilized and the corresponding output polarity for an applied acceleration.

The accelerometer package used to measure head accelerations was designed to be inserted into the subject's mouth. It consisted of three Endevco accelerometers, Model 2264-200, mounted to a plastic block with dimensions of 7/16 x 7/16 x 7/16 inches. This assembly was covered with a medical grade silicone rubber sealant to provide electrical isolation. The three accelerometer cables were routed to one end of the block. A dental bracket that had been custom fitted to the subject's mouth was mounted to the block. The approximate weight of the completed package was 50 grams. When the dummy subject was used the dental bracket was removed and the package was mounted to a bracket at the approximate center of the dummy's head. Specifications for the accelerometers used in this package are shown in Figure A-5.

The accelerometer package used to measure chest accelerations was designed to be attached externally to the subject's chest. It consisted of three Endevco accelerometers, Model 2264-150, mounted to an aluminum block that measured approximately $5/8 \times 5/8 \times 3/4$ inches. This

assembly was inserted into an aluminum protection shield that was attached to a length of Velcro fastener strap. In use, the completed package was placed over the subject's sternum while the Velcro strap was wrapped around the subject and fastened. Specifications for the accelerometers used in this package are shown in Figure A-6.

HARNESS INSTRUMENTATION

Figure A-7 shows the test fixture, seat and restraint harness used during this evaluation program. This harness was used in two different configurations. Figure A-8 shows the modified configuration and Figure A-9 shows the operational configuration. Also, two different headrests were used, one for the modified and one for the operational harness.

A total of seven load cells were used to instrument the F-III harness. Two of the transducers used were Lebow automotive belt load cells, Model 3419. These load cells monitored the load applied to the left and right inertia reel straps as shown in Figure A-10. Specifications for these load cells are shown in Figure A-11.

The five remaining load cells utilized the restraint harness hardware. Four 350 ohm resistive strain gages were bonded to each piece of harness hardware and wired in a bridge configuration. Figure A-12 shows the strain gage placement and wiring diagram. Figure A-10 shows the two reflection straps and Figure A-13 shows the lap and crotch strap units. The output polarity of each load cell corresponds to an applied load in accordance with the coordinate system shown in Figure A-14.

SEAT PAN INSTRUMENTATION

The seat pan instrumentation measured both acceleration and load. The acceleration measurements were performed using three Endevco accelerometers, Model 2264-200. The accelerometers were mounted to a plastic block, $3/4 \times 1 \times 1$ inch, to form a triaxial package. This package

was secured to the seat pan assembly to indicate accelerations in the X, Y and Z axes as shown in Figure A-15. Figure A-4 shows the coordinate system utilized and the corresponding output polarity for an applied acceleration. Figure A-5 shows the specifications for the accelerometers used in this package.

The load measurements were made utilizing two types of load cells to fit the physical size limitations of the seat pan. Z-axis load measurements were taken using three Strainsert Flat Load Cells, Model FL2.5U-2SKPT. These cells were used in a three point mounting configuration as shown in Figure A-15. Specifications for these load cells are shown in Figure A-16. The X-axis and Y-axis loads were measured using load links specifically designed for this application by General Dynamics. These load links were instrumented with resistive strain gages as shown in Figure A-17. Each load link had four resistive arms with 2 arms active. Each end of the load links housed a swivel ball to eliminate cross-axis load effects on the measurements. The output polarity of each load cell corresponds to an applied load in accordance with the coordinate system shown in Figure A-14.

FOOT REST INSTRUMENTATION

The foot rest assembly, as shown in Figure A-18, was instrumented using three GSE load cells, Model T-10952C. These triaxial load cells were capable of measuring 2500 lbs. in the Z-axis and 500 lbs. in both the X and Y-axis. Figure A-19 illustrates the location and orientation of these load cells. The output polarity of each load cell corresponds to an applied load in accordance with the coordinate system shown in Figure A-14.

CARRIAGE INSTRUMENTATION

For acceleration measurements the carriage was instrumented with a triaxial accelerometer package. This package consisted of three accelerometers mounted to a $3/4 \times 1 \times 1$ inch block. The accelerometers used were all Endevco transducers with the following Model numbers and axis measurements; 2262A-200 for Z-axis, 2264-200 for Z-axis, and 2264-150 for Z-axis.

Specifications for these accelerometers are shown in Figures A-20, A-5 and A-6 respectively. This package was securely mounted to the underside of the carriage. Figure A-4 shows the coordinate system utilized and the corresponding output polarity for an applied acceleration.

Carriage velocity measurements were obtained by means of a velocity wheel running against the rail. This unit consisted of a Globe Industries tachometer, Model 22A672, and a wheel mounted on its shaft. The wheel was aluminum with a rubber "O"-ring around the circumference. To insure continous rail contact the wheel assembly was spring loaded against the rail. The wheel was calibrated to output voltage as a function of velocity.

CALIBRATION

Strainsert Load Cells were calibrated on a periodic basis at the Precision Measurement Equipment Laboratories (PMEL), Wright-Patterson Air Force Base. The PMEL returns each device with a certificate providing current sensitivity and linearity data. Factory calibration data for the GSE Triaxial Load Cells were used for this evaluation program.

All accelerometers, load links, Lebow belt load cells and harness hardware load cells were calibrated at the AFAMRL/BBP Laboratory, Wright-Patterson Air Force Base. These calibrations were performed prior to (pre) and upon completion of (post) the evaluation program. This calibration data is shown in Figures A-21 and A-22.

Accelerometers were calibrated by using the reciprocity method to determine accelerometer frequency and phase characteristics as well as sensitivity. This method utilized a shaker table to which a "standard" accelerometer and the accelerometer to be calibrated were mounted. This "standard" accelerometer is calibrated yearly to standards traceable to the National Bureau of Standards. The sensitivity was determined by comparing the outputs of the standard and test accelerometer at 100Hz and 40G. The frequency and phase response was determined by driving the shaker table with a random noise generator and analyzing the output data by

Fourier Analysis via the PDP 11/15 and Time Data unit. The natural frequency and the dampening factor of the test accelerometer were both determined from this information.

The load cells mentioned previously in this section were all calibrated on a special test fixture. The sensitivity and linearity of each load cell was obtained by comparing its output with the output of a "standard" load cell output placed under an identical tension load. This "standard" load cell is calibrated on a yearly basis by standards traceable to the National Bureau of Standards.

SEAT GEOMETRY

The seat geometry drawings in Figure A-23 and A-24 show the polarity of the various output signals. Included in the drawings are the location dimensions for each fixed load cell and the variables introduced by the seat height and seat pan adjustment.

		DIG	DIGITAL INS	INSTRUMENT	ATION	REQUIREMENTS	ENTS							•
PROGRAM	A Negative	Shoulder	PROGRAM Regative Shoulder Harness Angle Study	igle Study		DATE 28 Oct 80	- }	THRU 25 NOV 80	lov 80	Д	YNAI	ECTF	NON	DYNALECTRON (CORPORATION
FACILITY	Vertical	Vertical Deceleration Tower	tion Tower			NOW	454	THRU 5	528	1				
DATA CHANNEL	DATA POINT	X DUCEN MFG A TVPE	#/s	XDUCER SENS	EXCITE V CHAN	SERIES	OAIN	SAMPLE	F.S. SENS	FILTER	ZENO	BRIDGE BALANCE RESISTORS	BRIDGE COMPLETION RESISTONS	SPECIAL NOTATIONS
-	Carriage Z	Endevco 2262A-	FR42	4.161 mV/9	10.00	9	27/25		24.03 9	120	2,50			
~	Head	Endevco 2264- 200	01d 8	2.496 #¥/9	20.00	8/2	8/2	/-	20.03 9	120		680K -into Gnd	1,65K	
	Head Y	•	BQ42	2.713 mV/9	0.00) 09	8 5	¥-	9.21 9	120		114K -into 10V	3	
-	Head 2	I	1508	6/Nm	B:52	03	200	¥ \	39.17 g	120		250K -into Gnd	•	
s	Chest X	Endevco 2264- 150	9228	2.786 mV/9	10.00	5 09	92	¥ \	17.95 9	120		1.2 M -into Gnd		
ى	Ches t		8813	2.430 mV/9	10.00	3	200	¥	10.29 g	120		220K -into 10V	z	
,	Chest 2	•	2A20	2.619 mV/g	10.00	2 09	\$2	¥ \	38.18 g	120		155K -into Gnd		
æ	left Lap	Micro-Med EA06-125 BZ 350	13	15.10 uV/1b	10.00	8 09	201	IK 1	824 1b	120		39.5K -into Gnd	•	
6	Right Lap	2	3 1	13.66 ul//lb	10.00	609	6 102	¥ \	91 116	120	· -	40 K -into Gnd	•	
6	N-G Strap	2	143377	1.80 uY/1h	10.00	01	900	1 1	1736 16	120		470 K -Into Gnd	•	
=	Left Seat Zan	Strainser FL2.5u 25PKT	t 3294-3	8.040 uV/15	10.00	11	201	¥ \	1547 lb	120		•	•	
12	Right Seat Pan	8	3294-4	7.988 uV/1b	10.00	21	01	¥ \	1557 16	120	-	-	•	Use Tension Calibration Sensitivity
13	Center Seat Pan	•	3294-6	8.011 uV/1b	10.00	00 13	3/3	<u>*</u> -	1553 lb	120		-	•	
7	Left Micro-Mea ReflectionEA-06-125 Strap BZ 350	Micro-Mea EA-06-125 EZ 350	02-10	26.32 uV/1b	10.00	3/4	3/2/	¥ \	950 lb	120		-	•	
Foor 1604	outer Compa ified Harne	rator Stal	Computer Comparator Start 0 -3; Off 0 +1 Podified Harness Tests: 470-478; 495-515; 52	f e +1 95-515; 52	51; 528		•							

Figure A-1 - Digital Instrumentation Requirements

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		Dig	DIGITAL INS	INSTRUMENTATION		REQUIREMENTS	l		00					-
PROGRA	PROGRAM Regative Shoulder Harness Angle Study	Megative Shoulder Harness A	ton Tower	igle study		9	1	- 1	328 528	A	XNX	ECTE	Z	CORPORATION
	100	200	1011			NO.								
DATA CAANNEL	Point	ABUCER MFG & TYPE	\$	BENS	EXCITE CHAN	FILTER SERVES	OVIN S	a Tudante	F.S. SENS 2 5 Volt	FILTER	1	BALDGE BALANCE RESISTONS	BAIDGE COMMETION RESISTONS	SPECIAL MOTATIONS
15	Right Micro-Mea RefrectionEA-06-125 Stran	Hicro-Mea EA-06-125 E1-350	01-3	34.04 uV/1b	10.00	51	20 21		734 1b	120	2.50	+5.0 20K	'	
16	Left Inertia Peel Strap	Letow 3419-3.5K	363	7.86 uV/16	91 00.01	91 09	405	<u>-</u> =	d1 167	120			,	
11	Right Inertia Feel Strap		364	7.54 uV/1b	10.00	(1 09	405		825 1b	120		•		
18	Left Load Link X	Micro-Mea EA-06-062 TJ-350	100	10.79 uV/1b	81 00.01	81 09	11 205	<u>-</u>	ol 978	120		106K Into Gnd	•	
19	Right Load Link X	•	200	10.11 dV/lb	61 00.01	61	13	¥ \	91 519	120		55 K Into Gnd	,	
8	Left Foot	GSE T-10952C	100	27.64 uV/1b	10.00	80	22	¥ \	904 Jb	120		,	,	
12	Left Foot Load Y		100	28.61 uV/1b	10.00	12	22		874 lb	120				
22	Left Foot Load Z	•	100	16.93 uV/1b	10.00	22 09	50	¥ \	d1 E2953	120		•	,	
23	Right Foot Load X		200	28.36 uV/1b	10.00	60 23	80	¥ \	882 1b	120			•	
92	Right Foo Load Y	•	500	28.16 uV/1b	10.00	12 DS	BE C	<u>-</u>	q. 888	120				
82	Right Foot Load Z		200	16.61 uV/1b	25 25		02 e1	¥ \	3010 16	120			,	
56	Cen Foot Load X	•	603	27.94 uV/1b	10.00		100	1,	91 S68	120			•	
27	Cen Foot Load Y	•	003	28.08 uV/1b	10.00	12 03	100	, ×	q1 068	120			•	
28	Cen Foot Load Z	•	003	16.50 uV/1b	10.00	82 28	30 20	<u>-</u> <u>*</u>	3030 lb	120		٠	·	
														PAGE 2 OF 3
1/4/1				ï	•	•		7 6 7 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7				•		

Figure A-2 - Digital Instrumentation Requirements

		DIG	DIGITAL IN	INSTRUMENT	ATION	REQUIREMENTS	IENTS							
PROGRA	M Hegativ	PROGRAM Megative Shoulder Harness Angle Study	Harness A	ngle Study		DATE 28 Oct 80		THRU 25 NOV	Nov 80	A	IXNXI	ECTE	ONO	DYNALECTRON (CORPORATION
FACILITY	Vertica	Vertical Ceceleration Tower	tion Tower			NOM		THEC	528	I)) 		
DATA	POINT	XDUCER MFG & TYPE	\$	XDUCER	EXCITE CHAN	ENIES SAN	N/N S/N	NATE NATE NATE	F.S. SENS	FILTER	ZENO ZENO ZENO ZENO	BRIDGE BALANCE RESISTONS	BRIBGE COMPLETION RESISTONS	SPECIAL NOTATIONS
83	Velocity	Globe 22A672	3	.5019 vlts/fps	. 29	3/	-	<u>-</u>	62.2 FPS	120	5.00			Signal Attenuated by 6.242. Prior to Signal Conditioner Ampl. Neg. Outbut Sens Solly 6.2420804V/FPS
33	Carriage Y	Endevco 2264-150	1188	2.354 mV/g	10.00	3	50	_ `	21.249	120	2.50	1M -into GND	1.65K	
*	Seat	Endevco 2264-200	8V63	2.564 mV/g	10.00	<u>د</u> ا		_ <u>×</u>	19.509	120		397K -into Gnd	1.47k	
33	Seat	•	BV41	3.298 mV/g	33		58 2	\ <u>*</u>	16.169	120		125K -into Gnd	1.47K	
*	Seat 2	•	BN63	2.825 mV/9	B.:0	98 X	250	Ž	17.709	120		512K -into Gnd	1.63к	
35	Cen Load Link Y	Mfcro-Pes EA-06-C62- T.1-350	100	10.23 uV/1b	10.00	35	200	_ <u>×</u>	df 809	120		82K - into Gnd		
*	Carriage X	Endevcc 2264-200	BX49	2.581 mV/9	36.00	8 8	28	_ <u>×</u>	19.379	120		127K -tnto Gnd	1.47K	
4.7	2.5 Volt Bias			·			-	-\ <u></u>	2,5 Volt	360				
a	10 Volt Exc.			٠		BE .	-\	<u>*</u>	5 Volt	360	0.0			
														PAGE 3 OF 3
1/8/78					Etauro A-3	1	4,0,0	1,00	Oist 1 Tar to month the Document		1000			

Figure A-3 - Digital Instrumentation Requirements

ACCELEROMETER COORDINATE SYSTEM

ACCELERATION

Accelerometers will be oriented and wired to provide an output corresponding to the applied acceleration. Use this table as a reference:

BARE SLED AND MACHINE TESTS

Accelerometers will be oriented to provide outputs to agree with track coordinate system with polarities as noted in test log.

<u>Acceleration</u>	Output			
+Gx -Gx +Gy -Gy +Gz -Gz	Positive Negative Positive Negative Positive Negative	+Z		RACK
Accelerometers for and test fixture wi in this manner.				
Sled and carriage a will be wired and i the same manner as and test fixture (i profiles).	dentified in the subject			
	+Y		-X	
•		*	-Y	
+X		-Z		

AMRL BBP COORDINATE SYSTEM (Left Hand Rule)

Figure A-4

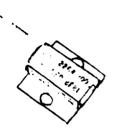
MODEL 2264-200

± 200 a One gram MINIATURE **PIEZORESISTIVE** ACCELEROMETER

The Model 2264-200 is a very low mass, piezoresistive accelerometer designed for modal studies, flutter testing and similar applications requiring good low frequency response and minimum mass loading.

With only a small amount of damping, the Model 2264-200 has no phase shift over its useful frequency range of steady state to 1200 Hz. Protection against overranging results from the high environmental rating of ±1000 g peak. The accelerometer can be operated over a temperature range of 0°F to 150°F (-18°C to 66°C).

The 2264-200 utilizes Piezite* Element Type P-11 gages in a half bridge circuit providing a low imperance nominal output of 500 mV full scale at 10 Volts de excitation.



SPECIFICATIONS FOR MODEL 2264-209 ACCELEROMETER

DYNAMIC

RANGE -200 g to +200 g

SENSITIVITY (at rated excitation)1...2.5 mV · g, nominal; 2,0 mV/ g, minimum

MOUNTED RESONANCE FREQUENCY

..... 1700 Hz, nominal

AMPLIFICATION FACTOR, Q. 10. maximum, at resonance and 75°F

FREQUENCY RESPONSE:

TRANSVERSE SENSITIVITY3°, maximum
LINEARITY AND HYSTERESIS! 2°° of reading maximum 0.16.200.0 = 2.5°+ of reading maximum, 0 to 200 g.

THERMAL SENSITIVITY SHIFT.... ±40 mV max.. at 0 F and 150 F (=48°C and 66°C), rof. 75°F (24°C)

WARMUP TIME 1 minute

ELECTRICAL

10 0 V dc

RESISTANCE FOR ARM ...1700f2 20%, at 75°F (24°C) ZERO MEASURAND OUTPUT 50 mV dc max , at 75 F

INSULATION RESISTANCES 10M 12 minimum at 100 V dc

NOTES

Measured with stoomy state accounts....

Measured with stancy state acconstant in this special measurements, minimum puller dustation for half a new or thank or howes and usual exceed 1.5 millisemonds. In land exercting that the quenty from the sundered Prezonny strip Ap-ceter meter Manual.

Centry netter Manka ()

Unit is cathbraned at 10.0 V no. (Carring rists ovoltance may no unit but should be known on it time of order (Lipp risting CONTYMER 4.13 M KAR Supply, Or Mindel 4470 Signel Gundleon is us esculation source.

*Due to self neuring of the programmative comments the measured resistance is serviced to the applied voltage.

ENVIRONMENTAL

ACCELERATION LIMIT!

(in any direction)

Static: * 1000 g.

Sinusoidal: :::1000 g.pk.
Shock: :::1000 g.pk. 1.5 millisecond duration or longer
CAUTION: Keep protective steeve on ancelerometer until rindy to use.

TEMPERATURE

Operating: 0°F to 150°F (= 18°C to 56°C')
-Operating: 65°F to 200°F (= 54°C to 33°C)

Non-Operating:

HUMIDITY **ALTITUDE**

Epoxy Sealed Not Affected

Figure A-5 - Accelerometer Specifications

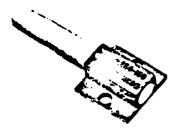
MODEL 2264-150

±150 g One gram MINIATURE **PIEZORESISTIVE ACCELEROMETER**

The Model 2264-150 is a very low mass, piezoresistive accelerometer designed for modal studies, flutter testing and similar applications requiring good low frequency response and minimum mass loading.

With only a small amount of damping, the Model 2264-150 has no phase shift over its useful frequency range of steady state to 1200 Hz. Protection against overranging results from the high environmental risting of + 1000 q sk. The accelerometer can be operated over a temperature range of

The 2264-150 utilizes Plezite® Element Type P-11 gages in a half bridge circuit providing a low impedance nominal output of 375 mV full scale at 10 Volts de excitation.



TWO TIMES ACTUAL SIZE

SPECIFICATIONS FOR MODEL 2264-150 ACCELEROMETER (According to ANSI and ISA Standards)

RANGE-200 g to +200 g

SENSITIVITY (at rated excitation)1...2.5 mV/g, nominal; 2.0 mV/g, minimum

AMPLIFICATION FACTOR, Q.....10, maximum, at resonance and 75°F

FREQUENCY RESPONSE

TRANSVERSE SENSITIVITY3% maximum

THERMAL SENSITIVITY SHIFT.... ±10% max, at 0°F and = 150°F, ref. +75°F WARMUP TIME1 minute

ELECTRICAL.

EXCITATION 10.0 V de
RESISTANCE PER ARM 17000 ± 20%, at +75°F (24°C)
ZERO MEASURAND OUTPUT ±50 mV de max., at +75°F

THERMAL ZERO SHIFT = 50 mV max., over rated temperature range

INSULATION RESISTANCE 10M Ω minimum at 100 V dc

MOTER

I Management with standy state accularation

Tin shock measurements, minimum culse duration for half asse or triangular pulses smould exceed 1.8 milliseasms. To several service high frequently ringing, (See Endevice Pretriessance Accelerations Measure).

Flinkt is calibrated at 10.0 V cc. Lawer excitation vultages may be used but should be specified at time of erise. Use ENDEVCOP Model 403 Power Buesty, 820 B ridgesec, or Model 4470 Eignal Contiltioning as excitation source

Oue to self healing of the piezoretistive ele-ments, the measured resistance is sensitive to the applied voltage.

Massured between all leads tied together and should ar case.

ENVIRONMENTAL

ACCELERATION LIMIT

 \pm 1000 g pk shock pulse, one millisecond duration or longer. CAUTION: Keep protective sleeve on accelerometer until ready to use.

TEMPERATURE

Operating: 0°F to 150°F (-18°C to 66°C)
Non-Operating: -65°F to 200°F (-54°C to 93°C)

HUMIDITY

Epoxy Sealed

Figure A-6 - Accelerometer Specifications

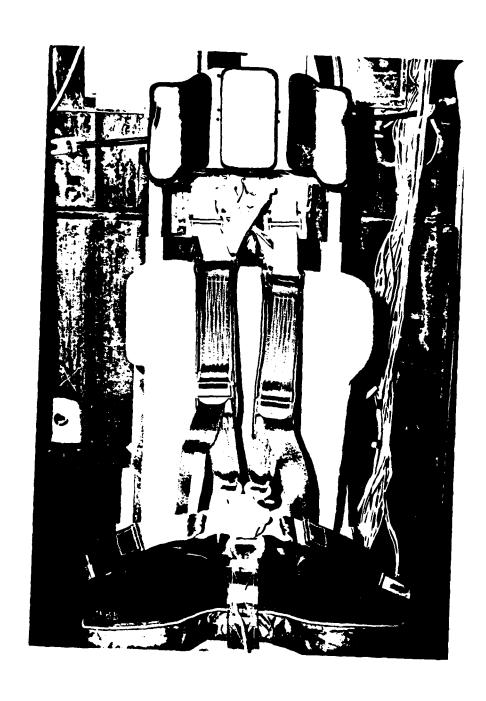


Figure A-7 - HARNESS ASSEMBLY

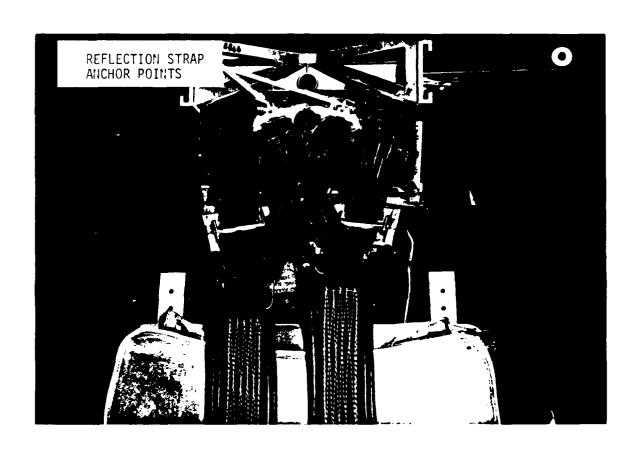


Figure A-8 - HARNESS CONFIGURATION (MODIFIED)



Figure A-9 - HARMESS CONFIGURATION (OPERATIONAL)



Figure A-10 - HARNESS INSTRUMENTATION

AUTOMOTIVE LOAD CELLS



Model 3419 Capacity Available 3500 lbs.

SPECIFICATIONS

Output at rated councily: millivolts per volt, nominal	<u>+</u> 2
Nonlinearity: of rated output	± 2%
Hysterisis: of raced output	± 4%
Repeatability: of rated output	<u>+</u> 1.0%
Zero balance: of rated output	± 2%
Bridge resistance: ohms noromal	350
Temperature range, compensated: OF	+ 30 to + 150
Temperature range, useable: ^O F	- 65 to + 260
Temperature effect on output: of reading per OF	± 0.003%
Temperature effect on zero: of rated output per OF	± 0.603%
Overload rating, sale: of rated capacity	150%
Excitation voltage, maximum: volts DC or AC rms	20
Insulation resistance, bridge/case: megohms at 50 VDC	1000
Belt thickness: (maximum) inches	0.10
Belt width: (maximum) inches	2.00
Weight: in ounces	8
Available canacities: pounds	3500

Figure A-11 - LOAD CELL SPECIFICATIONS

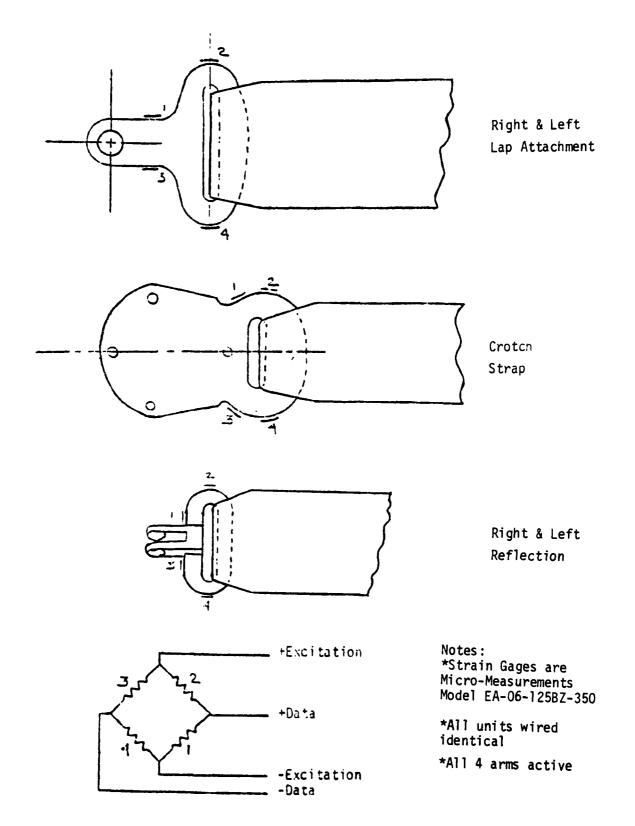


Figure A-12 - LOAD CELL SPECIFICATIONS (HARNESS HARDWARE)

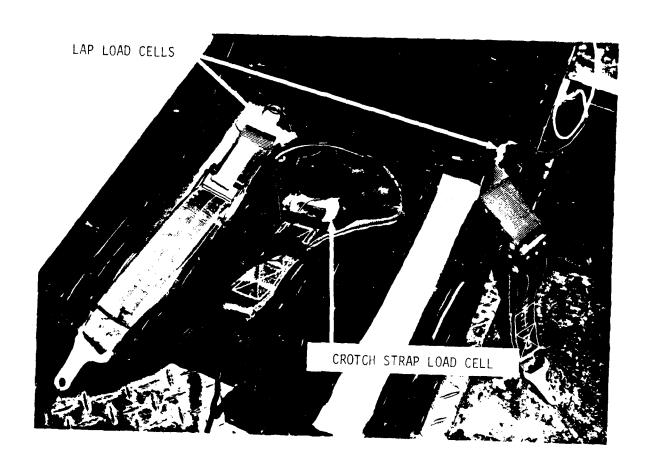
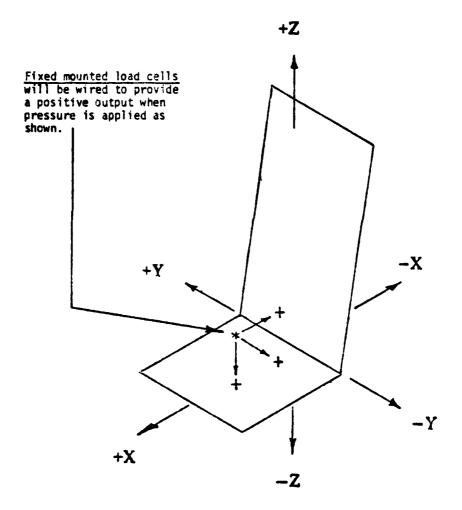


Figure A-13 - HARNESS INSTRUMENTATION

Swivel mount and Lebow belt load cells will be wired to provide a positive output when the belt is pulled.



AMRL BBP COORDINATE SYSTEM (Left Hand Rule)

Figure A-14

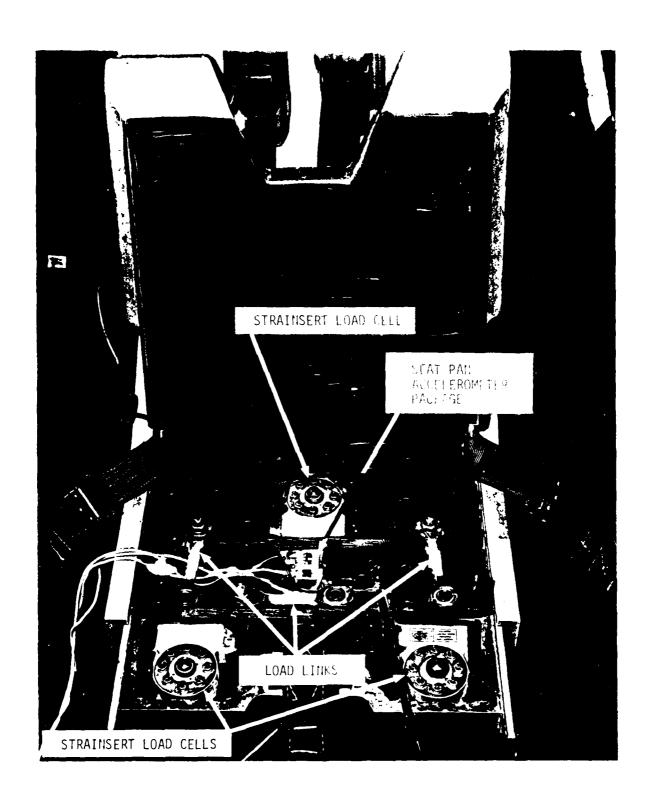


Figure A-15 - SEAT PAR INSTRUMENTATION

STRAIRSERT CALIBRATION DATA

U. S. Air Force Wright-Patterson AFB Dayton, Ohio Q-3294 Strainsert Job No.

Date: 10/16/76

Customer P.O. No. F33-601-76-86950

Sign: CGH

Transducer: Universal Flat Load Cell, Model FL2.5U-2SPKT 2,500 lb. Capacity, 2 mv/v, 350 Ohms

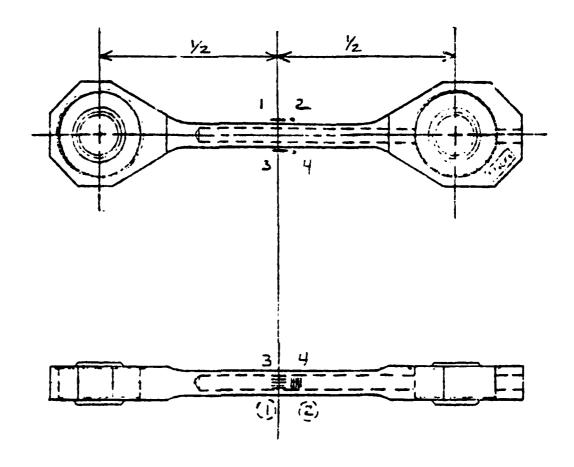
Gages: EA-06-104ZA-175 Service Temp.: 150°F Max. Calib. Temp.: 73°F Type: C (Bendix PTO2H-10-6P)
Ins. Res.: Over 10,000 megohms.

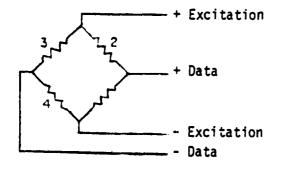
lib. Temp.: 73°F S/N: Q3294-6

Load	Straight	Straight Deviation, µv/v								
LBS.	Signal Mv/v	Run 1	Run 2	Run 3	Rep. µV/V					
0	, 0	0	0	0	0					
500	0.400	- 15	0	0	انغ					
1,000	0.800	+ 1/2	+ 1/2	+15	0					
1.500	1.200	0	0	0	0					
2,000	1.600	-1	-1	-1	0					
2,500	2.000	- kg	- l ₂	-15	0					
2,000	1.500	-1	- lş	- l ₂	15					
1,500	1.200	+1	+1	+1	0					
1,000	0.800	+11/2	+] ½	+] ⅓	0					
500	0.400	+15	+1	+1	ĮŽ.					
0	0	0	0	0	0					
	Hysteresis	1	1	1						

Calibration Analysis:						
Non-Linearity:	1	parts	fn	2,000	*	.05%
Repetition						
Loading :	15	parts	in	2.000	•	. 03%
Unloading:	پز	parts	ín	2,000		.03%
Zero Load:	0 ئۆ	parts	1 n		=	
Max. Load:	Ŏ	parts		2,000	•	
End Point :	, ig	parts	in	2,000		.03%
Hysteresis :	1	parts	in	2,000		.05%

Hold Down Bolts: 10-32NF; Torque = 6 ft. lb. lubricated





NOTES: *Strain Gages are Micro-Measurements

Model EA-06-062TJ-350 *Z arms active

Figure A-17 - LOAD CELL SPECIFICATIONS (LOAD LINK)

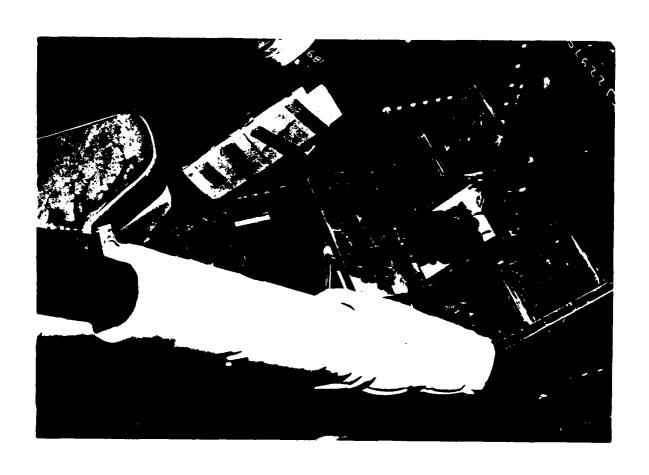
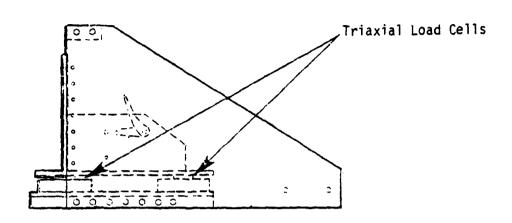


Figure A-18 - FOOT REST ASSEMBLY



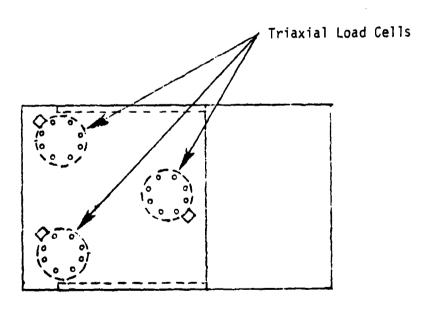


Figure A-19 - FOOT REST LOAD CELL LOCATIONS

22G2A-200 2262CA-200

Damped, Overload Stops **PIEZORESISTIVE**

ACCELEROMETERS



SPECIFICATIONS FOR MODEL

2262A-200 and 2262CA-200 ACCELEROMETERS

RANGE

OVERRANGE LIMITING

SENSITIVITY

MOUNTED NATURAL FREQUENCY (AT 75°F)

FREQUENCY PESPONSE

DAMPING RATIO

TRANSVERSE SENSITIVITY

THERMAL SENSITIVITY SHIFT

LINEARITY AND HYSTERESIS

ELECTRICAL

EXCITATION

INPUT RESISTANCE (AT 75°F)

OUTPUT RESISTANCE (AT 75°F)

INSULATION RESISTANCE ZERO MEASURAND OUTPUT Models 2262A-200 (2202CA-200)*

-200 g to 200 g

±300 to ±1 200 g

25 mV/g typical (1.2 mV/g typical)

2 mV-g minimum (1 niV-g minimum)

7 000 Hz typical

±5% maximum-0 to 3 000 Hz at 75°E, -05°c-10% typical at 0.200 °F and 3 000 Hz

0.7 typicai

3% maximum

•2% of reading, maximum,

to 200 g

10 00 Vdc

1 500 t typical (1 000 t tylical)

1 200 1 typical (1 000 1 typical)

100 Git minimum

±25 m/V maximum

FHVIRONMENTAL

ACCELERATION LIMITS (In any direction)

0°F to -200°F

Static 2 000 g Sinusoidal 1 000 g pk Shock 2 000 g half sine pulse

TEMPERATURE

Compensated 0°F to -200°F (-18°C to -93°C) Nonoperating -20°F to -220°F (-29°C to +104°C)

YIJOILTY

Sealed by glass to metal fusion and webling.

Figure A-20 - ACCELEROMETER SPECIFICATIONS

PROCRAM Negative Shoulder Harness Angle Study DATE 28 Oct 80

VOT FACILITY RUN NO'S 454-528

_						 	
SENAFACO							
Z CHANGE			. +	+ 1 + w.4.0	+ 1 1 4.8°E		
CAL	SENS	2.569 2.353 4.124	2.492 2.715 2.559	2.795 2.420 2.634	2.574 3.280 2.822		
PUST CAL	DATE	22 Dec 80 22 Dec 80 22 Dec 80	22 Dec 80 22 Dec 80 22 Dec 80	22 Dec 80 22 Dec 80 22 Dec 80	23 Dec 80 23 Dec 80 23 Dec 80		
CAL	SKAR.	2.581 2.354 4.161	2.496 2.713 2.553	2.786 2.430 2.619	2.564 3.298 2.825		
PRE	DATE	30 Sep80 30 Sep80 09 Oct80	30 Sep80 30 Sep80 30 Sep80	29 Sep80 29 Sep80 29 Sep80	30 Oct80 30 Oct80 30 Oct80		
2	s ò	8X49 8811 FR42	BP10 BA42 BQ51	BC26 BB13 2A20	BV63 BV41 BN63		
TRA::SDUCER	Mec & Model	Endevco 2264-200 Endevco 2264-150 Endevco 2262A-200	Endevco 2264-200 Endevco 2264-200 Endevco 2264-200	Endevco 2264-150 Endevco 2264-150 Endevco 2264-150	Endevco 2264-200 Endevco 2264-200 Endevco 2264-200		
T THI DE VIEW		Carriage X Ende Carriage Y Ende Carriage Z Ende		Chest X Ende Chest Y Ende Chest Z Ende	Seat X Endo Seat Y Endo Seat Z Endo		

Figure A-21 - PROGRAM CALIBRATIONS (PRE AND PUST)

:

PROGRAM Negative Shoulder Harness Angle Study

VDT FACTLITY RUN NO'S 454-528

DATE 28 Oct 80

SENTENCO															
a UN VIII A	A CIMINGE	9	9	٠.3	4.1-	2	-5.0	6	2:-	5	4				
CAL	SENS	10.73	10.05	10.20	14.89	13.63	F	25.47	33.97	7.52	7.51				
MIST CAL	DATE	29 Dec 80	29 Dec 80	29 Dec 80	29 Dec 80	29 Dec 80	29 Dec 80	29 Dec 80	29 Dec 30	29 Dec 80	29 Dec 80		 -		
CAI.	3.E).}p	10.79	19.11	10.23	15.10	13.66	1.30	26.32	34 04	7.86	7.54				
PRE	DATE	C3)ct 80	D3 Jct 80	C3 Oct 80	C2 Oct 80	02 Oct 80	03 Oct 30	02 3ct 30	02 Set 30	U2 Oct 80	02 Oct 80				
2	2	100	205		<i>,</i> ~1		143377	32-10	-	363		-	 		
TRANSDUCER	NEG & MODEL	-5 coad cink \[M.M.EA-06-062-	350		4, 4, EA-26-3 2582-	7	÷			,	Tap				
17100	Terror criss	IF Load Link N	ST Load Cirk (Sen Load Linky	GE 3D		Y-6 Strap	ils Ref. Strap :	ign Ref. Strip	LF inert RESI	RT Inert Rt Strap				

Figure A-22 - PROGRAM CALIBRATIONS (PRE AND POST)

AND WE AND COMMERCIALS OR SEVE AND WELLINGS MODIFIED

				-T-	ARS PRSITTO			T MIS
l			1	1	•	5	•	BINE
					-19.53			
					-19.33			
23					-7.07			
25					-2: 35			
ן ד					22.51			
سا	•	-17.92	46.84	-14.92	-3.12	-21.72	-22.72	-6.45

TO ALL MANAGES OF FORTIST ASSESSED WITHOUT PILE.

"I" ARLS PROFINEST POSITION											
7	7	1	1	1 1	•	7					
4.6	47.0	410	-44 0	م دی.ا	-91.6	42.0					

SET OF STATE STATE

Figure A-23 - SEAT GEOMETRY

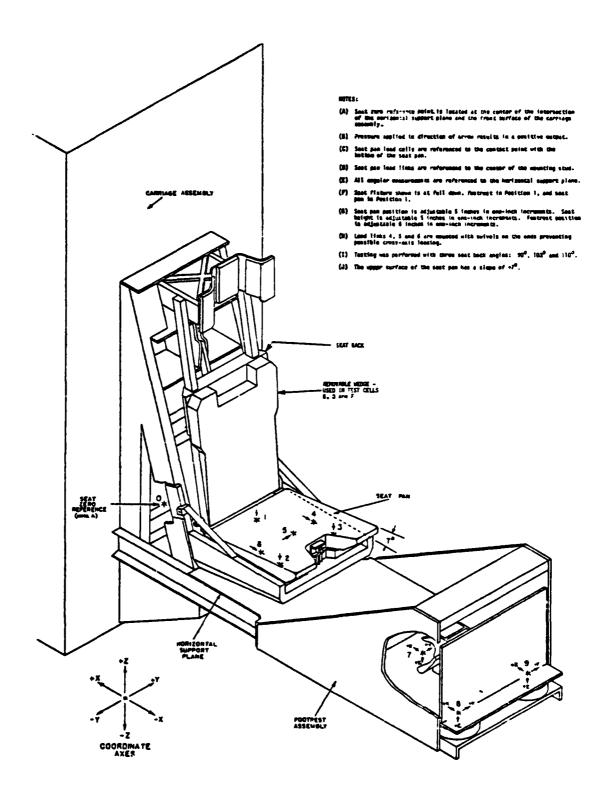


Figure A-24 - SEAT GEOMETRY

AUTOMATIC DATA ACQUISITION AND CONTROL SYSTEM

CARRIAGE DIGITAL DATA ACQUISITION SYSTEM EQUIPMENT

Figure A-25 is a photograph of the Carriage Digital Data Acquisition System. Figure A-26 shows the block diagram of the Carriage Digital Data Acquisition System. This system consists of four parts: the power conditioner, the signal conditioner and sensors, the encoder and the junction box. The power conditioner requires a 28 vdc, 4A power source and provides several regulated supplies. They are the +15 and -12 vdc (0.8A) supply for the signal conditioners, the 5 vdc and the 10 vdc bridge excitation voltages (1.2A total), and the 2.5 vdc signal output bias voltage (0.1A). The 28 vdc source also powers the pulse code modulator (PCM) encoder (0.24A).

The signal conditioner consists of 48 signal modules. Each module is capable of processing a sensor (transducer) signal which can be a voltage generating source or a bridge-type sensor. If a bridge-type sensor is used, the bridge excitation voltage is selectable from the 5V or the 10V source. By connecting the proper external resistors to the module input connector a half bridge is completed. A full or half bridge is balanced by connecting external resistors to its module input connector.

The signal conditioning module consists of a amplifier section and a filter section. The amplifier gain can be selected by inserting one of seven external gain plugs. These gains provide the capability of covering an input dynamic range from 50 mV up to 5 V. The filter section can be programmed by inserting one of four external filter plugs. These filter plugs are in accordance with the SAE recommended classes 60, 180, 600 and 1000.

The 48 channel data signals are time multiplexed and digitized via an encoder into 48 11-bit digital words. Two additional 11-bit synchronization (sync) words are added to the data frame. The 50-word frame is then sampled at a rate of 1000 samples/second. These serial digital data along with three additional synchronization pulse trains (bit sync, word sync, and frame sync) are connected to the computer room by four twisted pairs incorporated into a drag cable. They pass through a junction box to the digital computer interface to allow recording and processing.

PDP 11-34 DATA COLLECTION AND STORAGE

The PDP 11-34 minicomputer is the main control for all electronic data collection and storage functions. The block diagram of Figure A-27 shows the processor and its related equipment. All data transfer in the data collection system are under software control by the central processor unit. Serial data are constantly being received by the data formatter unit from the carriage data encoder. These data are converted by the data formatter from serial to parallel for input via a buffered data channel to computer memory for storage on disk. Finally, the data are transferred from disk to magnetic tape for permanement storage following the test event.

QUICK LOOK INERTIAL DATA

After each test, the data were sampled and checked. This check was made using the <u>Single Channel ANalysis</u> (SCAN) routine for the PDP 11-34 processor. This routine allows the operator to access and plot up to 2000 points of data for any of the 48 data channels. The operator selects the channel to be processed and enters its location description as well as the start and stop points to be processed. A maximum of 2000 milliseconds or 2000 data points may be accessed for each plot. The program converts the raw data into the appropriate units of measure and calculates the minimum and maximum values during the sample interval. If the sample is acceleration data, the velocity will also be calculated using an integration process.

An added optional reature is a digital smoothing routine which can smooth the data to remove any excess high frequency component that may be present.

FOOT REST AND SEAT PAN CORRECTION

Dynamic foot and seat pan loads were corrected by removing the effects of the foot support fixture and seat pan loading on supporting load cells.

A series of tests was conducted to determine the percentage of the total force resting on each cell. The weight assessed each load cell and multiplied by the carriage acceleration was subtracted from the acquired test data in the processing.

In practice, the load cell outputs are zeroed with the foot support fixture and seat pan weight resting on the load cells. During a drop, with no payload, the sum output of the load cells would reflect the weight of the fixture as a negative load (fixture weight removed from the load cell). The data were processed to remove this effect and thus reflect a zero output during a drop with no payload.

The final foot and seat pan loads were processed to provide corrected values which represent actual loads encountered by the human or dummy subjects.

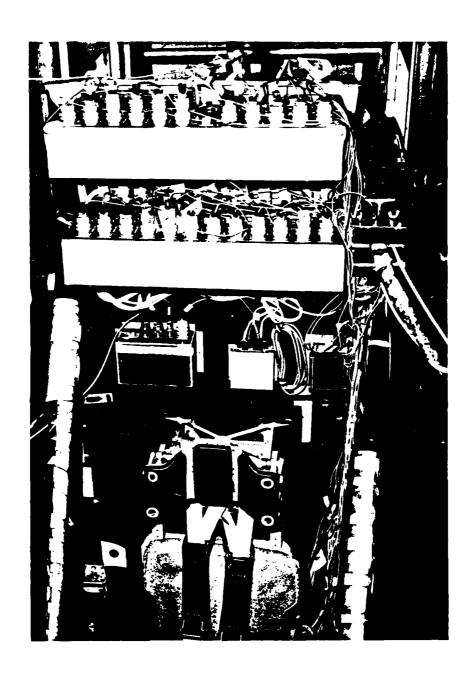


Figure A-25 - CARRIAGE DIGITAL DATA ACQUISITION SYSTEM

AIR FORCE AEROSPACE MEDICAL RESEARCH LAB WRIGHT-PATT--ETC F/G 1/3 COMPARATIVE VERTICAL IMPACT TESTING OF THE F/FB-111 CREW RESTRA--ETC(U) MAR 82 B F HEARON, J W BRINKLEY, J H RADDIN AFAMRL-TR-82-13 NL AD-A113 957 UNCLASSIFIED 2 of 4

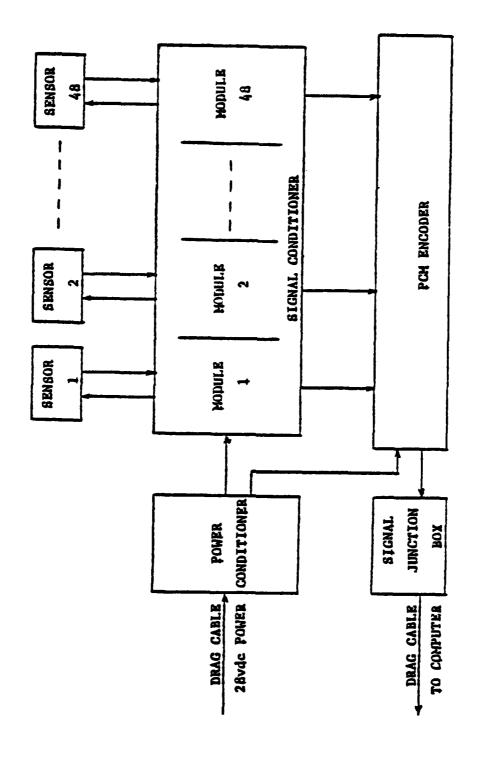


Figure A-26 - CARRIAGE DIGITAL DATA ACQUISITION SYSTEM BLOCK DIAGRAM

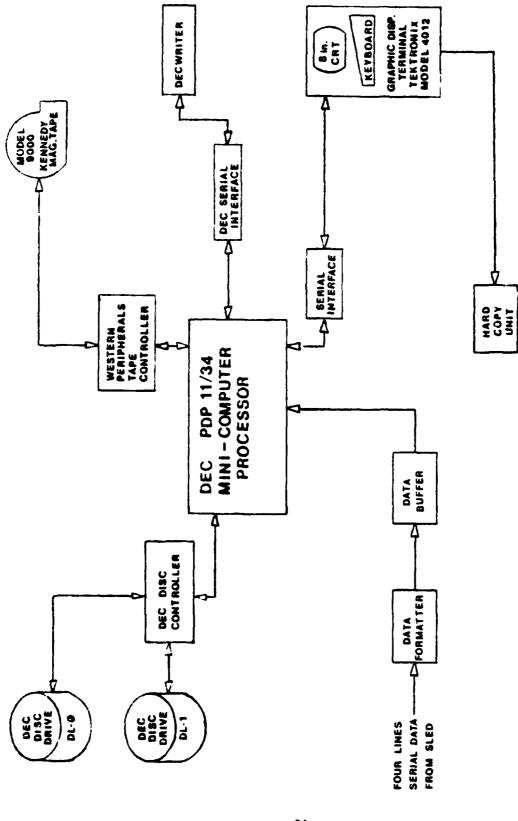


Figure A-27- CENTRAL DATA ACQUISITION AND STORAGE SYSTEM

KINEMATIC DATA ACQUISITION SYSTEM

HIGH SPEED CAMERAS AND CONTROL

Kinematic data were acquired through the use of high speed 16mm cameras operating at a rate of 500 frames per second. The cameras were Teledyne Milliken Model DBM45 pin registered units which were capable of withstanding 25 G. Two cameras were mounted to the carriage, one to provide a frontal view and one to provide a right lateral view of the subject. During a test, the cameras were started and stopped automatically by the Camera and Lighting Control Station which is part of the impact facility safety and control system. The cameras were started at a preset time in the test sequence and run for a period of 8 seconds.

AUTOMATIC FILM READER

The AFR subsystem was developed by Photo Digitizing Systems, Inc. It automatically extracts photo data, digitizes it and records it on magnetic tape. The subsystem consists of:

Film motion analyzer with 16mm projection head Electronic scanning camera
Control Unit
Alphanumeric Cathode Ray Tube (CRT)
Line printer
Magnetic tape transport

The film reader recognizes quandrant or circular fiducial targets. It automatically tracks targets and extracts data for up to twelve targets per film frame at a minimum rate of one-half frame per second. Film may be processed through the reader manually or automatically.

Figure A-28 is a block diagram of the Automatic Film Reader System (AFR). The X-Y coordinate position of each target on each film frame is input to the computer and recorded on magnetic tape.

A NOVA 3/12 computer controls the AFR which contains 16K, 16-bit, words of core memory, a CRT terminal, and a magnetic tape transport with suitable interface. In addition, a parallel data link is provided between the NOVA 3/12 and the PDP 11/34.

An alphanumeric CRT (DGC 6052) automatically displays the AFR control information. The CRT display and its keyboard function are used as separate devices. The keyboard is a transmit-only device and the display is a receive-only device but has the additional capability of transmitting cursor position information on program request.

A hard copy device, LA36 Decwriter II, provides hard copies of the information presented on the 6052 CRT. The LA36 is medium-sized interaction terminal with a low-speed impact printer and a standard ASCII keyboard consisting of alphanumeric characters and non-printing system control codes.

Either the Decwriter or the 6052 CRT output may be assigned to the PDP 11/34A. Programs can also be established which can "down load" from the disc on the PDP 11/34A to the NOVA, or digital film data can be loaded on the PDP 11/34A for processing or disc storage.

QUICK LOOK KINEMATIC DATA

The Instar (Instant Analytical Replay) System is a highperformance video recorder and display device designed for the analysis
of high speed motion. It is a compact, portable, fully transistorized
instrument that combines the long recording capacity and instant replay
features of video tape. The system records 120 frames/second with an
effective shutter speed of 10us or less and will playback all recordings
in real time, stop action, reverse slow motion, and variable slow motion
(2%-15% of real time). Each of the frames is sequential and non-interlaced.

Instar incorporates two cameras and a special effects generator for the added flexibility of split screen. The simultaneous display of two events offers the precise evaluation of three dimensional problems

or the referencing of one physical event to an instrument (i.e., digital clock or oscilloscope). Other features include:

End of tape sensing
Foolproof logic control sequences
Dynamic braking
Interscene blanking
Video logic signal processing modules

The Instar System was utilized to record each impact event. This video tape was available for review by the test conductor and/or medical monitor immediately after the impact event.

TIMING REFERENCE

A 100 PPS timing signal was an integral part of the Kinematic Data Acquisition System. The Camera and Lighting Control Station started the timing signal at T = 0. An event signal was generated less than one second after T = 0. This event signal performed two functions. It triggered a photo flash unit which marked the film frame at the beginning of the impact event. Second, it started the 100 PPS signal to the LED drivers, LM Dearing Model 2/3/3R. The LEDs, located in the high-speed cameras, were pulsed every 10 ms which produced a .75 ms timing bar on the edge of the film. The diagram of Figure A-29 shows the 100 PPS signal, the event signal and the LED driver signal. Figure A-30 illustrates the event and timing bar in relationship to the film.

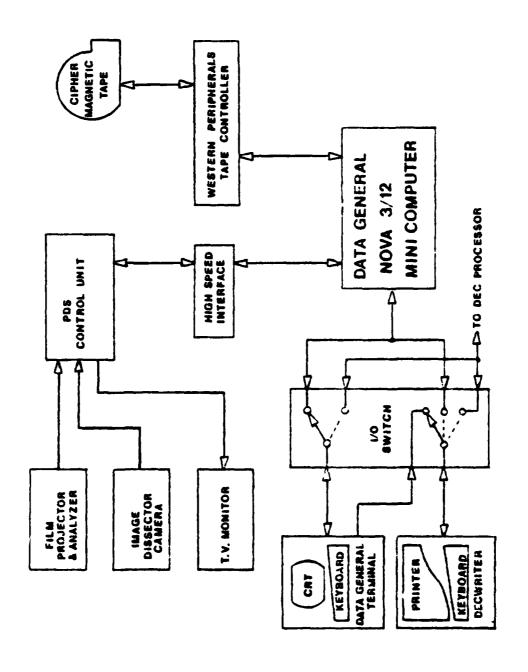


Figure A-28 - AUTOMATIC FILM READER SYSTEM BLOCK DIAGRAM

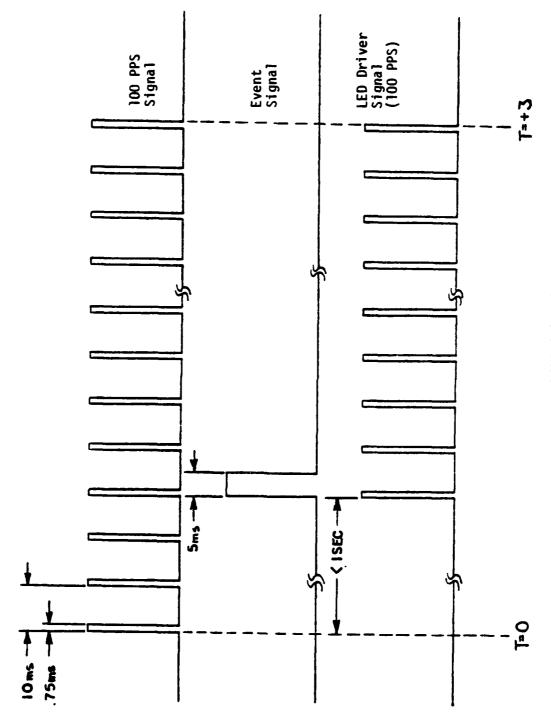


Figure A-29 - TIMING REFERENCE

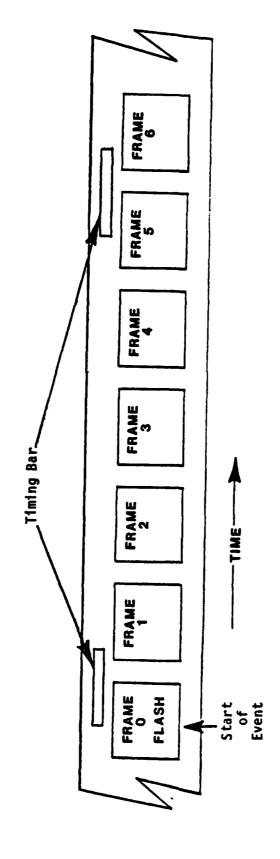


Figure A-30 - TIMING REFERENCE

APPENDIX B

SUMMARY OF ELECTRONIC DATA

The means and condard deviations of all peak measured and computed parameters from each cell of experimental matrix are shown in Table B-1. (The cell designations are explained in the experimental design matrix, Table 1, is the body of the report.) In addition, the maximum and minimum values of each parameter are tabulated for each test conducted at the experimental level. The times at which these values were achieved during the impact are also tabulated. These data are grouped according to test condition. Finally, a set of analog data from each test condition is presented. To permit comparability among these data, the test results of the same subject, S3, are shown in each test condition. This subject was selected because his tabulated maxima and minima in each of the four tests were not beyond 2.5 standard deviations of the mean and because the subject's body weight, standing height, and sitting height were close to the means of those anthropometric measurements for the sample under investigation. (See Table 2.)

All electronic data derived from this test program will be maintained by the Biomechanical Protection Branch of AFAMPL until this work unit is retired. These experimental results will eventually be recorded in a permanent data bank within the Laboratory.

TABLE B-1 $\mbox{SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA}$ (Peak values are tabulated for velocity, accelerations and loads.)

MATRIX CELL F/FB-111 HARNESS SEAT POSITION	MOD I	FIED		G OPERATONAL DOWN			
	(n =	15)		(n =	18)		
	MEAN	s		MEAN	s		
CARRIAGE ACCELERATION (G)	10.5	0.14		10.6	0.20		
CARRIAGE VELOCITY (ft/sec)	-25.8		j	-25.9			
SEAT ACCELERATION (G)	10.6	0.14		10.8	0.33		
CHEST ACCELERATION (G)	İ				}		
-X axis	-1.98	1.02		-2.11	0.62		
+X axis	4.51	1.26		3.42	1.02		
+Z axis	20.0	2.92		17.1	2.20		
Resultant	20.3			17.3			
CHEST SEVERITY INDEX	35.2	5.72		32.5	4.13		
HEAD ACCELERATION (G)	ŧ				ì		
-X axis	-4.79			-3.18			
+X axis	1.22			1.72			
+Z axis	13.2	0.91		12.5			
Resultant	13.3	0.91		12.6	0.80		
HEAD SEVERITY INDEX	19.6	2.14		19.4	2.05		
STRAP LOADS (1b)	1	į					
Reflection Straps	104	24		67	20		
Inertia Reel Straps	111	27		78	25		
Total Shoulder Straps	206	52	·	137	35		
Total Lap Belt	95	29	·	89	27		
SEAT PAN LOADS (1b)	1		ŀ		1		
-X axis	-290	70		-276	63		
+Z axis	1760	237		1740	257		
Resultant	1780	240		1760	258		
FOOTREST LOADS (1b)	1		j		1		
-X axis	-390	117	l	-400	96		
+Z axis	467	72	ſ	471	75		
Resultant	558	120	l	573	107		

TABLE B-1 (continued)

SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA

(Peak values are tabulated for velocity, accelerations and loads.)

MATRIX CELL F/FB-111 HARNESS SEAT POSITION	H MOD IF I ED UP		J OPERATONAL UP		
SEAT FUSITION	(n =			(n =	
	MEÀN	s		MEÀN	s
CARRIAGE ACCELERATION (G)	10.5			10.5	
CARRIAGE VELOCITY (ft/sec)	-26.1			-26.2	
SEAT ACCELERATION (G)	10.7	0.38		10.6	0.30
CHEST ACCELERATION (G)	1 77	0.00		2 50	0.07
-X axis	-1.77 3.66	0.86 1.37		-2.59 2.68	0.97 1.34
+X axis +Z axis	18.3	1.96		17.2	2.04
Resultant	18.6	1.88		17.4	2.02
CHEST SEVERITY INDEX	34.4	4.25		32.1	2.92
HEAD ACCELERATION (G)		.,	1	02,1	
-X axis	-4.07	1.03		-3.24	1.07
+X axis	1.72			2.31	
+Z axis	13.0	0.94		12.6	
Resultant	13.2	0.97		12.9	0.96
HEAD SEVERITY INDEX	19.8	2.20		20.4	1.91
STRAP LOADS (1b)	00	10		63	12
Reflection Straps	80	13		63 05	13
Inertia Reel Straps	86 156	22 35		95 150	34 45
Total Shoulder Straps Total Lap Belt	97	20	1	102	27
SEAT PAN LOADS (1b)	37	20		102	- '
-X axis	-295	68		- 281	57
+Z axis	1760	204		1770	215
Resultant	1790	207		1790	217
FOOTREST LOADS (1b)					j
-X axis	-338	94		-324	79
+Z axis	461	83		443	71
Resultant	530	107		508	99

NEG 9-10 -443 -453	120 : 30	J J.	∧1+ 210.0	5: 10	SP: 1 CELL:	3
DATA ID		¥ax 	5 1 tv	•	š	2.5
LOV EXT PWR CARRESE Y CARRESE Y CARRESE Z (SM) CARRESE VEL SEAT Z SEAT Z SEAT Z SEAT Z CHEST X CHEST Y CHEST SI HEAD X HEAD Z		10.35 10.746 10.51 -1.981 10.579 10.579 -0.777 27.593 -0.777 27.593 -0.8777 27.593 -0.8777 27.593 -1.593	9.00 48 -25.882 -25.899 -0.160 -1.602 -1.2277 -4.28	0.000000000000000000000000000000000000	48000000000000000000000000000000000000	100m 200000 0000 2000
HEAD RES HEAD SI HEAD HIC SHD REFL LF		14.02 21.47 16.33 59.61	~1.32 0.67 23.19	3876.00 3876.00 3831.00 3850.00 3884.00	3698.00 3988.00 3984.00 3921.00 3973.00	14
SHO REEL LF LF SHOULDER SHO REFL RT SHO REEL RT RT SHOULDER TOTAL SHLO REFL TOTAL SHLO REEL TOTAL SHOULDER		56.03 97.16 54.51 41.06 92.78 195.09 189.95	29.135 55.239 16.10 2.93 37.11 46.37 13.03	3931.00 3907.00 3902.00 3909.00 3907.00 3909.00 3909.00	3872.00 3983.00 41000.00 3860.00 3974.00 3869.00	16 15 17
TOTAL SHD / WT LF LAP BELT AT LAP BELT TOTAL LAP TOTAL LAP / WT		0.90 56.81 43.91 95.53 0.45	0.45 32.18 27.65 59.92 0.29	3907.00 3970.00 4C52.00 3970.00 3970.00	3860.00 3858.00 3859.00 3859.00 3858.00	8
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X		283.31 52.29 30.01 69.06	-13.43 -172.19 -94.32 -265.36	3960.00 4095.00 3937.00 3937.00	3880.00 3878.00 3673. 00 3873.00	1 C 1 8 1 9
SEAT LNK Y LF SEAT PAN Z AT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z TOTAL SEAT Z WT RES SEAT FORCE RES SEAT FORCE / WT		78.12 587.40 457.49 1166.34 2210.53 2226.09 10.60	-42.33 38.59 35.90 91.54 173.85 0.83 175.45 0.84	3924.00 3876.00 3875.00 3876.00 3876.00 3876.00 3876.00	3876.00 3695.00 3615.00 3615.00 3602.00 3602.00	12
LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		28.44 11.81 30.50 67.21	-164.13 -199.67 -207.42 -571.22	3824.00 3823.00 3825.00 3824.00	3876.00 3876.00 3876.00 3876.00	26 23 26
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y		156.30 26.09 19.83 46.38	-23.64 -169.14 -37.13 -48.84	3869.00 3824.00 3896.00 3896.00	3832.00 3868.00 3835.00 3833.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		196.13 251.58 171.56 599.02 765.67	-34.90 0.89 -77.91 -37.04 40.29	3869.00 3861.00 3884.00 3869.00 3869.00	3964.00 3972.20 3835.00 3810.00 3973.00	22 25 28

NEG SHLD HAR ANG	TEST: 406	SUBJ: F-3	WT: 163.0	G: 10	GP: 1 CELL:	С
DATA ID		MAX 	MIN	T 1	12	9 4
10V EXT PHR CARRIAGE X CARRIAGE Y CARRIAGE Z		10.06 1.60 0.30 12.20	9.96 -0.90 -1.04 -0.29	773.00 3862.00 3904.00 3893.00	2861.00 3872.00 3851.00 3792.00	48 36 31 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT T SEAT Z		10.45 -1.00 1.82 0.77 11.46	-0.12 -25.96 -1.40 -0.97 -0.24	3908.00 4181.00 3865.00 3900.00 3900.00	3794.00 3863.00 3870.00 3868.00 3677.00	29 32 33 34
SEAT Z (SM) CHEST X CHEST Z CHEST RES		10.42 3.03 -0.15 19.12 19.21 27.58	-0.15 -1.46 -1.66 -1.36 0.76	3901.00 3917.00 3944.00 3926.00 3926.00	3678.00 3945.00 3929.00 3823.00 3859.00	5 5 7
CHEST SI HEAD X HEAD T HEAD Z HEAD RES HEAD SI		.56 1.43 13.56 13.58 19.71	-3.95 -0.32 -0.98 0.80	3861.00 3999.00 4029.00 3915.00 3915.00	3990.00 3941.00 3911.00 3689.00 3859.00	2 3 4
HEAD HIC SHO REFL LF SHO REEL LF LF SHOULDER SHO REFL AT SHO REEL AT		16.52 70.99 49.35 116.10 64.35 84.46	27.43 11.38 48.83 26.83 17.36	3889.00 3935.00 3942.00 3940.00 3932.00 3942.00	3957.00 4007.00 3915.00 4032.00 4003.00 3914.00	14 16 15 17
RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER		142.94 134.82 133.82 259.04 1.56 56.89	61.93 55.25 28.83 117.64 0.71 16.45 33.04	3941.00 3934.00 3942.00 3941.00 4005.00 4004.00	3908.00 4004.00 3914.00 4033.00 4033.00 3907.00	B 9
TOTAL LAP TOTAL LAP / HT CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		131.86 0.79 89.57 39.45 37.04	50.00 0.30 -68.46 -233.96 -68.73 -300.22	4004.00 4004.00 3999.00 4003.00 3866.00 3860.00	3906.00 3906.00 3914.00 3910.00 3910.00	10
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / HT RES SEAT FORCE		69.22 473.50 341.49 857.01 1664.52 10.03 1690.93	-106.31 28.70 25.98 55.89 119.72 0.72 128.09	3979.00 3918.00 3909.00 3918.00 3918.00 3918.00	3915.00 3614.00 3652.00 3610.00 3601.00 3601.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		10.19 8.55 21.55 52.52 71.17	120.07 -137.01 -78.90 -147.83 -363.74 -20.07	3918.00 3864.00 3861.00 3864.00 3864.00	3601.00 3908.00 3908.00 3908.00 3908.00 4001.00	20 23 26
LF F00T T RT F00T T CT F00T T T0TAL F00T T LF F00T Z RT F00T Z		139.77 24.74 22.39 37.98 225.56 162.12	-121.68 -54.14 -72.71 -7.37 -6.15	4192.00 3866.00 3885.00 3697.00 3922.00	3912.00 3912.00 3915.00 3925.00 3994.00	21 24 27 22 25
CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		178.41 425.90 515.42	-107.45 -109.47 55.47	3862.00 3897.00 3921.00	3854.00 3854.00 4123.00	28

NEG SHLD HAR ANG	TEST: 422	SUBJ: F-2	WT: 161.0	G: 10	GP: 1 CELL:	С
DATA ID		MAX	MIN	T!	T2	CH
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z		10.05 1.20 0.56 12.34	9.37 -0.85 -0.87 -0.15	128.00 3840.00 3839.00 3833.00	167.00 3809.00 3818.00 3641.00	48 36 31
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Y SEAT Z		10.52 -0.90 1.15 0.84 12.52	-0.06 -26.04 -1.34 -1.72 -0.22	3834.00 4111.00 3799.00 3797.00 3840.00	3691.00 3799.00 3848.00 3811.00 3670.00	29 32 33 34
SERT Z (SM) CHEST X CHEST Z CHEST RES		10.66 3.68 -0.03 22.88 23.17	-0.13 -2.56 -2.04 -0.78 0.71	3840.00 3860.00 3820.00 3856.00 3856.00	3669.00 3886.00 3914.00 3673.00 3791.00	5 6 7
CHEST SI HEAD X HEAD T HEAD RES HEAD SI		38.10 1.95 0.78 14.82 14.82 23.25	-4.47 -1.77 -0.78 0.56	3797.00 3854.00 3899.00 3854.00 3854.00	3932.00 3887.00 3850.00 3674.00 4125.00 3954.00	2 3 4
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER		18.36 44.59 62.49 98.11	14.82 6.41 26.04	3833.00 3869.00 3888.00 3887.00	3894.00 3945.00 3847.00 3975.00	14 16
SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHCOLDER		64.46 95.08 156.48 107.94 141.31 240.70	24.93 9.86 42.43 40.94 16.84 72.54	3873.00 3880.00 3880.00 3871.00 3883.00 3881.00	4097.00 3852.00 3846.00 4100.00 3846.00 3846.00	15 17
TOTAL SHD / WT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT		1.50 30.96 34.80 62.81 0.39	0.45 7.11 14.43 22.38 0.14	3881.00 3956,00 3942.00 3947.00 3947.00	3846.00 3840.00 3843.00 3834.00 3834.00	8
CHOTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		156.17 20.27 8.62 11.90	-23.13 -268.10 -130.55 -395.60	3892.00 4177.00 3660.00 3626.00	3851.00 3852.00 3855.00 3853.00	10 18 19
SEAT LNKTY LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z TOTAL SEAT Z RES SEAT FORCE		54.61 504.76 407.75 795.21 1706.14 10.60 1754.69	-112.45 17.53 8.48 26.82 65.48 0.41 68.86	3936.00 3836.00 3852.00 3856.00 3855.00 3855.00	3852.00 3608.00 3611.00 3636.00 3611.00 3611.00	35 11 12 13
RĒS SĒRT FORCĒ / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		10.90 1.14 -4.71 -22.86 -30.83	0.43 -100.12 -150.10 -182.07 -425.15	3855.00 3801.00 3800.00 3801.00 3801.00	3611.00 3849.00 3851.00 3851.00 3850.00	20 23 26
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y		120.43 23.55 66.02 84.42	-35.92 -176.12 -14.08 -80.63	3836.00 4039.00 3875.00 3815.00	3808.00 3844.00 3993.00 3845.00	21 24 27
LF FOOT Z AT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		209.04 201.53 189.14 537.13 616.31	6.58 18.96 -81.82 -2.50 104.25	3837.00 3844.00 3843.00 3844.00	3794.00 3952.00 3664.00 3793.00 4188.00	22 25 28

NEG SHLO HAR AND	TEST: 387	5UBJ: G-3	WT: 159.0	G: 10	GP: 2 CELL:	С
DATA ID		MAX 	MIN	T1	<u></u>	CH
IOV EXT PHR CARRIAGE X CARRIAGE Y CARRIAGE Z		10.04 1.52 0.33 12.32	9.97 -0.88 -0.94 -0.22	29.00 3841.00 3880.00 3871.00	409.00 3854.00 3979.00 3686.00	48 36 31 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z		10.39 -1.04 1.76 1.27 11.59	-0.08 -25.85 -1.29 -1.41 -0.19	3886.00 4191.00 3842.00 3838.00 3877.00	3796.00 3828.00 3885.00 3844.00 3709.00	29 32 33 34
SEAT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES		10.44 5.09 -0.40 21.86 22.29 38.79	-0.10 -1.93 -3.30 -0.91 0.69	3878.00 3895.00 3701.00 3905.00 3904.00	3707.00 3925.00 3903.00 3710.00 3600.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD RES		38.79 0.65 1.48 13.02 13.03 19.41	-6.27 -0.14 -1.29 0.83	3835.00 3698.00 3951.00 3892.00 3892.00	4083.00 3935.00 3920.00 3705.00 4191.00 4013.00	2 3 4
HEAD SI HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT		14.07 66.46 65.09	15.39 2.57 27.89 22.98	3871.00 3918.00 3920.00 3920.00 3920.00	3912.00 4008.00 3879.00 4007.00 3987.00	14 16
SHD REEL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER		52.91 72.27 125.08 119.16 137.25 256.32	3.97 34.13 42.45 6.69 65.36	3919.00 3920.00 3919.00 3920.00 3920.00	3874.00 3874.00 3987.00 3873.00 3874.00	15 17
TOTAL SHO / HT LF LAP BELT AT LAP BELT TOTAL LAP TOTAL LAP / HT		1.61 51.56 54.51 103.27 0.65	0.41 9.40 13.17 23.07 0.15	3920.00 3990.00 3986.00 3988.00	3874.00 3881.00 3879.00 3880.00 3880.00	8 9
CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		45.95 33.06 15.01 42.27	-54.45 -193.72 -87.67 -279.54	3933.00 3785.00 3835.00 3778.00	3894.00 3887.00 3893.00 3867.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / NT		61.96 684.22 678.57 478.97 1841.30 11.58	-70.60 56.95 34.21 29.18 129.65 0.82	4069.00 3896.00 3896.00 3896.00 3896.00	3898.00 3600.00 3606.00 3604.00 3601.00	35 11 12 13
RES SERT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		1861.29 11.71 11.63 31.70 54.08 74.32	133.23 0.84 -121.28 -93.42 -123.02 -337.72	3896.00 3896.00 3844.00 3841.00 3843.00 3842.00	3601.00 3601.00 3900.00 3900.00 3900.00	56 53 50
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y		127.82 20.29 18.65 64.09	-20.67 -104.84 -75.68 -77.65	3882.00 3842.00 3842.00 3861.00	3851.00 3890.00 3893.00 3893.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		139.63 170.15 154.20 370.12 464.89	-57.57 -16.50 -111.23 -153.26 24.99	3875.00 3891.00 3896.00 3896.00 3891.00	3833.00 3859.00 3852.00 3852.00 3965.00	22 25 28

NES SHID HAR AND	TEST: 380	SUBJ: G-2	WT: 120.3	3: 10	GP: 2 CELL:	: C
DATA ID		MAX 	MIN	T1	T 2	CH
10Y EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z		10.05 1.20 0.77 12.41	9.96 -1.02 -1.00 -0.!8	2845.00 3856.00 3853.00 3847.00	905.00 3846.00 3955.00 3763.00	48 36 31 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT T SEAT Z		10.66 -0.91 1.36 0.71 11.91	-0.10 -25.62 -1.37 -0.80 -0.27	3847.00 4151.00 3857.00 3967.00 3853.00	3654.00 3815.00 3862.00 3961.00 3663.00	29 32 33 34
SEAT Z (SM) CHEST X CHEST Y CHEST Z CHEST RES		10.81 3.85 0.93 22.74	-0.17 -2.13 -2.42 -0.68 0.63	3854.00 3864.00 3886.00 3879.00 3879.00	3663.00 3910.00 3879.00 3786.00 3726.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD RES HEAD SI		22.87 35.13 0.83 12.39 12.42	-5.75 -0.87 -0.90 0.40	3813.00 3973.00 3920.00 3868.00 3868.00 3819.00	3910.00 3910.00 3871.00 3621.00 4129.00 3943.00	2 3 4
HEAD HIC SHO REFL LF SHO REEL LF LF SHOULDER		15.33 37.27 45.28 78.74	13.18 6.43 24.44	3842.00 3888.00 3903.00 3904.00	3921.00 3956.00 3848.00 3978.00	14 16
SHO REFL AT SHO REEL AT RT SHOULDER TOTAL SHLO REFL TOTAL SHLO REEL TOTAL SHOULDER		37.95 53.32 88.38 72.40 98.47 167.12	15.70 2.81 19.94 31.40 9.92 46.43	3911.00 3904.00 3904.00 3897.00 3904.00	3983.00 3852.00 3986.00 4089.00 3858.00 3879.00	15 17
TOTAL SHD / WT LF LAP BELT RT LAP BELT TOTAL LAP		1.39 25.95 28.92 54.69	0.39 2.28 6.88 10.05	3904.00 3926.00 3938.00 3938.00	3979.00 3847.00 3850.00 3847.00	8 9
TOTAL LAP / HT CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		0.46 41.98 41.51 32.97	0.08 -27.87 -123.11 -42.49 -163.13	3938.00 3948.00 3969.00 3813.00 3790.00	3847.00 3868.00 3862.00 3865.00 3862.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT		49.04 56.70 355.95 360.11 623.02 1329.03	-25.62 17.04 15.26 36.73 82.45 0.69	3912.00 3865.00 3865.00 3870.00 3865.00	3864.00 3667.00 3681.00 3652.00 3667.00 3667.00	35 11 12 13
RES SEAT FORCE RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		1338.79 11.16 15.91 7.91 19.32 24.23	90.65 0.76 -39.24 -118.98 -103.22 -246.99	3865.00 3865.00 3943.00 3810.00 3811.00	3667.00 3667.00 3863.00 3865.00 3865.00	20 23 26
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y		65.39 17.77 50.07 34.70	-15.84 -97.60 -22.91 -41.51	3850.00 3629.00 3862.00 4013.00	3931.00 3858.00 3960.00 3915.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		119.82 151.38 181.70 389.29 395.06	-23.65 -20.92 -53.52 -68.76 7.68	3874.00 3859.00 3855.00 3857.00 3857.00	3993.00 3959.00 3909.00 3959.00 4141.00	22 25 28

NEG SHED HAR AND	TEST: 371	ეიცე: ∿ 1	w1: 175.0	3: 13	GP: 2 CELL:	:
DATA 10		MAX 	HIN	<u>T1</u>	15	۲-
10V EX PWA CARRIAGE X CARRIAGE T CARRIAGE Z		10.06 1.25 0.42 12.74	9.97 +0.81 -0.94 -0.71	13.00 3983.00 3888.00 3882.00	389.00 3832.00 3837.00 3802.00	46 36 31
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Y SEAT Z		10.59 -1.10 1.01 0.86 11.77	-0.20 -25.63 -1.21 -1.15 -0.31	3882.00 4156.00 3847.00 3859.00 3888.00	3803.00 3842.00 3897.00 3855.00 3711.00	29 32 33 34
SEAT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES		10.55 6.35 0.12 18.62 19.23	-0.17 -1.03 -1.75 -0.81 0.58	3889.00 3906.00 3881.00 3915.00 3915.00	3710.00 3945.00 3914.00 3712.00 4006.00	5 6 7
CHEST SI HEAD X HEAD T HEAD RES HEAD SI		37.60 0.53 1.65 12.31 12.43 16.85	-6.03 -1.19 -1.45 1.07	3847.00 3699.00 3973.00 3904.00 3904.00 3861.00	3997.00 3937.00 3903.00 3651.00 4200.00 3987.00	2 2 2
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT		13.07 35.36 33.20 62.35 44.85	17.67 4.58 29.80 12.08	3880.00 3913.00 3980.00 3954.00 3913.00	3948.00 4014.00 3891.00 3884.00 3992.00	14 16 15
SHO REEL RT RT SHOULDER TOTAL SHLD REEL TOTAL SHLD REEL TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER		49.95 82.29 80.21 74.81 137.30 0.78 46.98 56.09	4.64 29.68 39.47 66.638 29.55	3931.00 3928.00 3913.00 3968.00 3931.00 3985.00 3984.00	3888.00 4002.00 3994.00 3890.00 3884.00 4075.00	17 8 9
TOTAL LAP TOTAL LAP TOTAL LAP / HT CROTCH STRAP LF SERT LNK X RT SEAT LNK X TOTAL SEAT X		102.82 0.59 117.28 16.07 4.55 10.62	55.46 0.32 -51.40 -215.32 -177.92 -389.61	3985.00 3985.00 3978.00 3734.00 3671.00 3702.00	4075.00 4075.00 3898.00 3904.00 3902.00	10 18 19
SEAT LAKTY LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z		45.60 670.04 669.31 739.19 2064.17 11.80 2099.55	-74.85 42.43 34.19 26.23 115.55 0.66	3967.00 3967.00 3903.00 3906.00 3906.00 3906.00	3906.00 3607.00 3610.00 3620.00 3603.00 3603.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		12.00 10.63 13.02 34.01 55.86	0.66 -91.53 -90.07 -117.15 -296.95	3906.00 3845.00 3846.00 3847.00 3846.00	3603.00 3899.00 3900.00 3900.00 3900.00	20 23 26
LF FOOT Y AT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z AT FOOT Z		115.42 21.12 22.37 38.79 130.62 165.72	-19.09 -134.18 -55.05 -57.46 -18.69 -8.79	3884.00 3722.00 4081.00 3924.00 3885.00 3885.00	4038.00 3884.00 3894.00 3894.00 4146.00	21 24 27 22 25 28
TOTAL FOOT Z TOTAL FOOT Z RES FOOT FORCE		164.51 431.87 468.29	-44.85 -34.03 21.83	3905.00 3885.00 3885.00	4146.00 4004.00 3840.00 4116.00	28 23

NEG SHLD HAR ANG	TEST:	424	SUBJ: M-2	WT: 163.	0 G; 10	GP: 1 CELL	.: C
DATA 10			MAX	MIN	T1	72	СН
10V EXT PHR CARRIAGE X CARRIAGE T CARRIAGE Z CARRIAGE Z (SM)			10.05 0.97 0.77 12.25 10.43	9.96 -0.95 -0.68 -0.22 -0.11	1821.00 3867.00 3831.00 3860.00	1474.00 3837.00 3836.00 3722.00	48 36 31 1
CARRIAGE VEL SEAT X SEAT Y SEAT Z SEAT Z (SM) CHEST X			-1.20 1.32 0.77 11.97 10.63 1.93	-25.85 -1.17 -0.98 -0.14 -0.09 -4.10	3861.00 4184.00 3830.00 3827.00 3867.00	3757.00 3818.00 3874.00 3834.00 3615.00 3717.00	34 33 53 53
CHEST Y CHEST Z CHEST RES CHEST SI HERD X			0.24 20.06 20.15 33.38	-1.28 -0.95 0.94	3867.00 3917.00 3698.00 3898.00 3827.00	3917.00 3848.00 3682.00 3811.00 3963.00	5 6 7
HEAD Y HEAD Z HEAD SI HEAD HIC			2.16 1.16 11.31 11.63 15.06 9.48	-5.99 -1.89 -1.23 0.55	3881.00 3951.00 3878.00 3878.00	3925.00 3883.00 3646.00 4148.00 4020.00	2 3 4
SHD REFL LF SHO REEL LF LF SHOULDER SHO REFL RT			86.72 71.06 149.23	24.03 10.64 42.20	3857.00 3920.00 3909.00 3911.00	3901.00 3990.00 3872.00 3863.00	14 16
SHD REEL AT AT SHOULDER TOTAL SHLD REEL TOTAL SHLD REEL TOTAL SHDU PEEL TOTAL SHDU / WT LF LAP BELT			59.78 84.29 143.54 142.32 152.73 292.51 1.79	34.12 12.99 48.52 62.21 24.30 91.82 0.56	3910.00 3913.00 3912.00 3917.00 3911.00 3911.00	3860.00 3868.00 3860.00 3870.00 3863.00	15 17
RT LAP BELT TOTAL LAP TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X			36.54 56.07 91.44 0.56 126.53	11.47 19.84 31.33 0.19 -45.20	3974.00 3941.00 3967.00 3967.00 3950.00	3863.00 3867.00 3868.00 3868.00	8 9
TOTAL SEAT X SEATLINK X			44.61 26.21 35.57	-177.00 -91.31 -267.07	4124.00 3831.00 4124.00	3885.00 3876.00 3882.00 3876.00	10 18 19
LF SEAT PAN Z AT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE LF FOOT X			78.45 363.10 403.62 904.02 1665.17 10.22 1686.06 10.34	-56.53 34.14 33.58 72.29 150.83 0.83 153.18 0.94	3949.00 3877.00 3877.00 3887.00 3881.00 3881.00 3881.00	3882.00 3774.00 3623.00 3627.00 3623.00 3623.00 3623.00	35 11 12 13
RT FOOT X CT FOOT X TOTAL FOOT X LF FOOT Y			10.79 -6.78 0.30 -14.55 101.27	-57.01 -130.15 -135.65 -312.88 -26.26	3827.00 4181.00 3830.00 3828.00	3623.00 3675.00 3878.00 3877.00 3878.00	58 53 50
AT FOOT T CT FOOT Y TOTAL FOOT Y LF FOOT Z			101.27 16.75 47.49 56.55	-135.88 -30.83 -90.77	3862.00 3740.00 3899.00 3843.00	3835.00 3871.00 3927.00 3835.00	21 24 27
RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE			192.38 166.41 172.21 431.83 504.36	-25.40 -5.58 -78.52 -24.68	3864.00 3878.00 3828.00 3872.00 3872.00	3834.00 3647.00 3844.00 3821.00	25 25 28

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DATA ID		MAX	¥ . %	ī: 	÷	35
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z (SM)		10.08 1.95 0.46 12.46	9.97 -1.76 -1.17 -0.30	535.30 38.30 3839.00 38.00	3580.30 3802.30 3803.00 3708.	46 36 31
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z SEAT Z (SM)		10.56 -1.08 2.27 0.58 11 25	-1035.58 -25.58 -25.58 -25.58 -25.58 -25.77	38 4 3 3 3 3 4 3 3 6 6 6 6 6 6 6 6 6 6 6 6	3802.00 3722.00 3775.00 3802.00 3802.00	2923 H
SEAT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES		10.36 5.16 0.23 17.67 17.74	-0.17 -2.17 -1.75 -1.02 0.43	3629.00 3845.00 3879.00 3858.00 3658.00	3643.00 3879.00 3863.00 3715.00 3698.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD RES		31.71 2.09 1.73 14.37 14.49	-3.43 -1.31 -1.20 0.54	3789.00 3849.00 3891.00 3841.00	4015.00 3882.00 3855.00 3981.00 3794.00	2 3 4
HEAD SI HEAD HIC SHO REFL LF SHO REEL LF LF SHOULDER		22.92 17.94 39.03 38.36 69.99	6.1± 1.29 11.41	3797.00 3820.00 3862.00 3875.00 3875.00	3936.00 3862.00 4093.00 3826.00 4094.00	14
SHD REFL AT SHD REEL AT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER		43.35 42.37 75.98 73.21 71.71 133.52	19.73 -3.30 18.09 26.39 -1.15	3844.00 3863.00 3662.00 3861.00 3872.00 3865.00	4:00.00 3977.00 3976.00 4092.00 3977.00 3976.00	15 17
TOTAL SHO / WT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT		0.95 37.62 32.43 69.12 0.49	33.24 0.40 12.40 14.27 27.19	3865.00 3879.00 3891.00 3879.00	3976.00 3821.00 3831.00 3831.00 3830.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X		118.65 5 0.24 44.66	-26.46 -148.92 -120.49	4055.00 4052.00 3793.00	3844.00 3835.00 3837.00	10 18 19
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z		44.16 57.26 340.32 889.95 1612.33	-266.53 -33.53 10.37 6.22 41.95 65.68	4060.00 3894.00 3839.00 3838.00 3841.00	3837.00 3834.00 3629.00 3601.00 3636.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X RT FOOT X		11.52 1634.33 11.67 6.36 29.03	70.37 70.50 -136.49 -82.61	3839.00 3839.00 3839.00 3793.00 3793.00	3601.00 3601.00 3601.00 3850.00 3841.00	20 23
CT F00T X TOTAL F00T X LF F00T Y BT F00T Y CT F00T Y		31.81 63.60 133.33 17.22 36.76	-142.61 -353.10 -19.53 -139.86 -42.45	3792.00 3792.00 3833.00 3653.00 3812.00	3840.00 3840.00 3781.00 3841.00 3838.00	26 21 24 27
TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		45.77 154.93 162.95 204.06 445.71 515.64	-65.28 -65.36 -43.46 -117.63 -200.67 28.82	3869.00 3849.00 3841.00 3790.00 3790.00 3841.00	3905.00 3801.00 3783.00 3805.00 3783.00 3616.00	22 25 28

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01 87 80		MAX	MIN	Ţ 1	T 2	0H
IOV EXT PHR CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE VEL SEAT X SEAT Z SEAT Z CHEST X CHEST X CHEST X CHEST X CHEST SI HEAD X HEAD X HEAD SI HEAD RES HEAD RES HEAD REF HEAD REF SHD REFL LF SHD REFL LF			9.1.413 -0.1.413 -0.1.413 -0.1.509 -1.1.413 -0.1.509 -0.1.416 -0.1	11800000000000000000000000000000000000	3436.00 3891.00 3893.00 3716.00 3716.00 3716.00 3923.00 3934.00 3739.00 3739.00 3739.00 3739.00 3974.00 3974.00 3964.00 3964.00 3964.00 3964.00 3964.00 3964.00 3964.00	
LF SHOULDER SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULD WT LF LAP BELT TOTAL LAP TOTAL LAP CROTCH STRAP LF SEAT LNK X TOTAL SEAT LNK X TOTAL SEAT LNK X TOTAL SEAT PAN Z CT SEAT PAN Z TOTAL SEAT PAN Z TOTAL SEAT PAN Z TOTAL SEAT PAN Z TOTAL SEAT FORCE / WT RES SEAT FORCE / WT RES SEAT FORCE LF FOOT X RT FOOT X RT FOOT X		120.196 804.196 16350.5.706 16350.5.706 16350.5.706 16350.5.709 16350.5.709 1782.05.709 17	21.78 28.6561 46.259 46.259 46.259 50.078 50.039 50.628 50.628 50.628 60.475 60.4475 7706.63 740.666 71076 71060 71060 71060	10000000000000000000000000000000000000	3913.00 3910.00 3912.00 3924.00 3912.00 3913.00 3913.00 3916.00 3916.00 3917.00 3916.00 3917.00 3916.00 3917.00 3917.00 3917.00 3917.00 3917.00 3917.00 3917.00	157 17 89 100189 11223 2036
TOTAL FOOT X LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		~4.64 144.84 25.10 33.57 59.66 184.02 192.24 172.54 484.01 570.14	-414.06 -25.49 -144.40 -47.42 -65.82 9.37 13.61 -106.41 -13.86 88.70	3879.00 3912.00 3806.00 3892.00 3951.00 3912.00 3913.00 3921.00 3921.00	3928.00 3766.00 3911.00 3922.00 3922.00 3989.00 4029.00 3669.00 3869.00	21 27 22 25 28

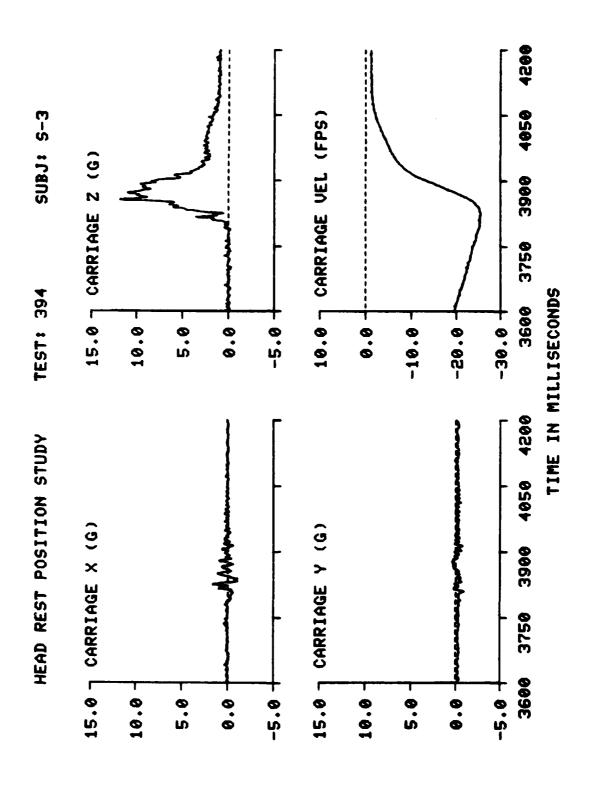
NEG SHLD HAR ANG	TEST: 5:0	3080: 3-8	WT: 200.0	G: 10	GP: 2 CELL:	į
DATA ID		MAX	MIN	T 1	T2 	C m
IOV EXT PWA CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE Z (SM)		10.05 1.01 0.88 12.31 10.25	9,96 -1.03 -0.56 -0.30 -0.11	270.00 3868.00 3819.00 3862.00 3862.00	490.00 3862.00 3815.00 3683.00 3631.00	48 36 31
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z		-0.93 1.45 0.73	-25.92 -1.20 -1.28 -0.16	4161.00 3829.00 3841.00 3869.00	3818.00 3876.00 3832.00 3675.00	29 32 33 34
SEAT Z (SM) CMEST X CMEST Y		11.72 10.52 5.14 -0.38 120.04	-0.08 -0.55 2.51 -0.94 0.62	3869.00 3885.00 3834.00 3693.00 3893.00	3677.00 3919.00 3890.00 3791.00 3822.00	5 6 7
CHEST AES CHEST SI TEGE X HEAD Z HEAD BES HEAD SI		34.76 .76 1.43 12.43 12.47 18.70	-4.30 0.26 -1.19 0.74	3825.00 3845.00 4033.00 3879.00 3879.00	4076.00 3907.00 3948.00 3806.00 3830.30 3961.00	2 3 4
HEAD HIC SAD REEL LE SAD REEL LE LE SAOULDER		15.70 57.18 56.30 108.85	24.54 12.68 48.43	3856.00 3900.00 3910.00 3909.00	3922.00 4100.00 3680.00 3974.00	14 16
SHO REFL RT SHO REEL RT 51 SHOULDER TOTAL SHLO REFL TOTAL SHLO REEL TOTAL SHOULDER		56.04 53.67 108.59 109.74 109.72 216.66	20.85 5.88 47.14 51.23 99.35	3908.00 3917.00 3911.00 3900.00 3911.00 3910.00	4002.00 3878.00 4007.00 3991.00 3879.00	15 17
TÖTGL SHO / WT LF LGP BELT BT LGP BELT TOTAL LGP TOTAL LGP / WT		1.08 70.96 72.01 139.43 0.70	20.78 24.54 46.52 0.23	3966.00 3960.00 3962.00 3962.00	3998.00 3886.00 3882.00 3884.00 3884.00	8 9
CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		211.79 53.84 25.16 51.34	34.52 -230.76 -127.94 -358.13	4065.00 4140.00 3935.00 4115.00	3681.00 3680.00 3877.00 3877.00	10 :8 :9
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE / WT		61.42 494.21 569.64 1097.27 2146.80 10.73 2175.70 10.88	-89.28 40.59 44.68 94.49 191.78 0.96 192.97	4053.00 3886.00 3885.00 3886.00 3886.00 3880.00 3880.00	3876.00 3608.00 3603.00 3603.00 3603.00 3603.00	35 11 12 13
LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		2.57 21.31 0.34 18.05	-144.80 -80.91 -218.79 -433.66	3828.00 3827.00 3828.00 3828.00	3879.00 3889.00 3889.00 3889.00	20 23 26
LF F00T Y RT F00T Y CT F00T Y T0TRL F00T Y		126.48 18.57 16.04 47.05	-14.15 -104.79 -59.61 -52.35	3864.00 3914.00 3849.00 3847.00	3954.00 3672.00 3883.00 3883.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		196.02 228.66 151.90 505.08 664.38	-3.23 -3.37 -73.53 -55.78 59.01	3889.00 3888.00 3885.00 3888.00 3889.00	3819.00 3847.00 3820.00 3820.00 4149.00	22 25 28

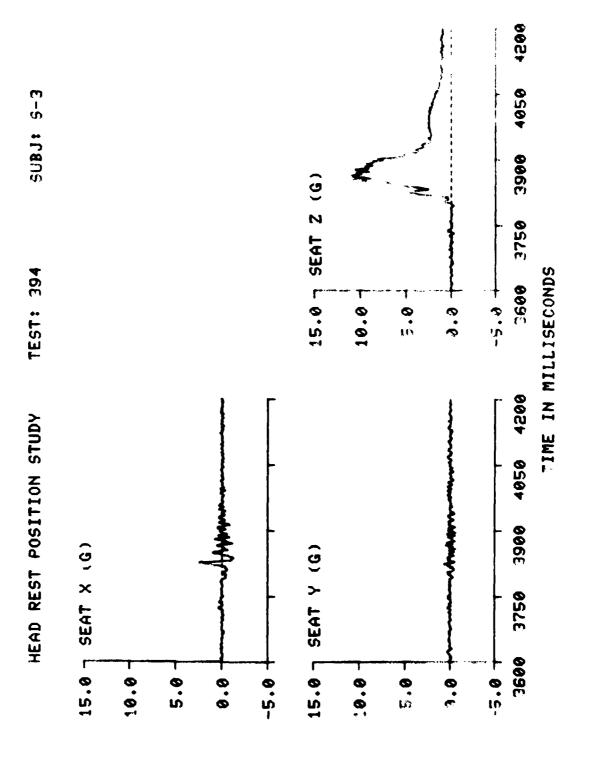
NEG SHLD HAR ANG	TEST: 370	SUBJ: R-1	WT: 194.0	G: 10	GP: 2 CELL:	С
DATA ID		MAX	MIN	T1	T2	CH
10V EXT PWR CHRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z CARRIAGE X		10.06 1.56 0.86 13.06	9.97 -1.19 -1.02 -0.34 -0.10	11.00 3830.00 3830.00 3825.00 3825.00	28:3.00 3824.00 3935.00 3608.00 3632.00	48 36 31
CARRIAGE VEL SEAT X SEAT X SEAT Z SEAT Z (SM)		-0.98 1.36 0.93 11.69 10.69	-25.75 -1.72 -1.60 -0.31 -0.19	4102.00 3829.00 3785.00 3832.00 3832.00	3790.00 3824.00 3794.00 3657.00 3643.00	29 33 34
CHEST X CHEST Y CHEST Z CHEST RES		6.35 -0.46 16.50 17.39	-0.94 -3.26 -1.00 1.05	3851.00 3955.00 3863.00 3861.00	3901.00 3868.00 3645.00 3745.00	5 6 7
CHEST SI HEAD X HEAD Y MEAD Z HEAD RES HEAD SI		33.06 2.68 0.66 12.31 12.39 17.09	-3.84 -1.16 -0.79 0.35	3795.00 3839.00 3687.00 3851.00 3848.00 3795.00	3911.00 3880.00 3847.00 3655.00 4074.00	3 3 5
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHO REFL RY		71.44 75.12 125.86 43.43	14.54 13.10 46.42 13.44	3823.00 3861.00 3903.00 3864.00 3865.00	3888.00 4092.00 3837.00 4092.00 3955.00	14 16
SHO REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOLDER TOTAL SHOULDER TOTAL SHOULDER		64.36 97.73 114.14 133.12 212.68 1.10	13.82 22.10 32.95 15.50 72.58 0.37	3913.00 3912.00 3862.00 3910.00 3910.00	3834.00 3835.00 4100.00 3835.00 3827.00	15 17
LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT		65.20 75.76 135.70 0.70	44.68 46.80 93.58 0.48	3867.00 3892.00 3892.00 3892.00	3835.00 4099.00 3834.00 3834.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X		140.71 44.85 11.78 35.35	-111.74 -135.89 -163.69 -296.68	3918.00 3939.00 3791.00 4070.00	3857.00 3845.00 3847.00 3845.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / HT RES SEAT FORCE		34.36 426.12 797.90 1896.12 877 1918.20	-30.41 28.98 57.02 59.88 159.50 0.82 159.64	4115.00 3849.00 3846.00 3848.00 3848.00 3848.00	3800.00 3656.00 3600.00 3611.00 3600.00 3600.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		9.89 -32.19 -26.81 -58.76 -159.61	0.82 -220.24 -193.35 -273.42	3848.00 3609.00 3678.00 3787.00	3600.00 3842.00 3843.00 3842.00	20 23 26
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y		189.61 33.65 25.04 48.47	-677.09 -25.26 -191.76 -43.46 -62.09	3787.00 3827.00 3638.00 3943.00 3869.00	4085.00 3835.00 3837.00 3846.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		314.87 262.70 148.60 622.76 838.06	33.27 47.59 -117.42 34.37 194.89	3828.00 3835.00 3834.00 3835.00 3844.00	3965.00 4163.00 3799.00 3797.00 4190.00	22 25 28

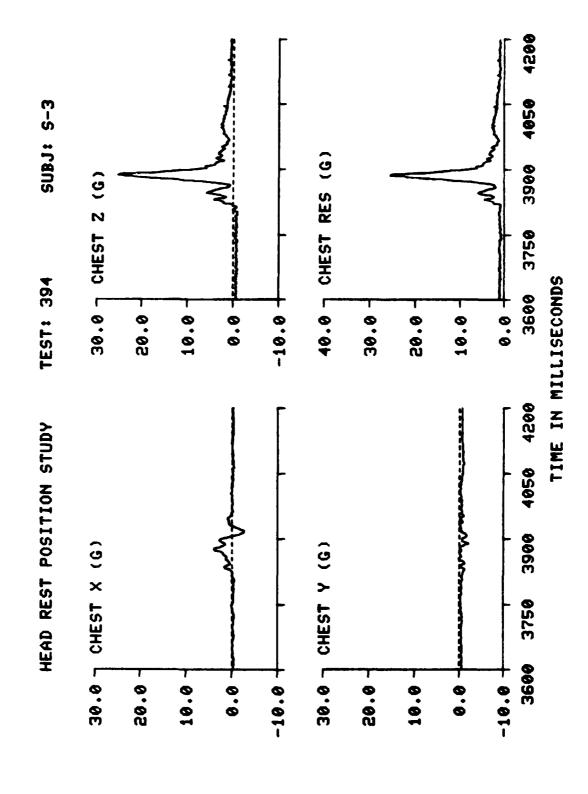
NEG SHLD HAR AND	TEST: 39	? SUBJ: 3-2	₩T: 145.0	3: 10	6P: 1 CELL:	ε
DATA ID		MAX	MIN	T 1	T2	SH
10V EXT PWR CARRIAGE X CARRIAGE T CARRIAGE Z CARRIAGE Z (SM)		10.04 1.38 0.79 12.73	9.96 -1.10 -0.72 -0.17	1457.00 3829.00 3826.00 3821.00	1428.00 3820.00 3771.00 3605.00	48 36 31
CARRIAGE VEL SEAT X SEAT Z SEAT Z		10.71 -0.96 1.13 0.59 11.89	-0.10 -25.67 -1.60 -0.74 -0.24	3821.00 4126.00 3782.00 3782.00 3828.00	3629.00 3786.00 3820.00 3845.00 3659.00	29 32 33 34
SERT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES CHEST SI		10.77 5.69 1.06 18.25 18.75 32.99	-0.11 -0.79 -2.38 -1.26 0.28	3829.00 3840.00 3834.00 3856.00 3857.00	3646.00 3882.00 3856.00 3690.00 3780.00	5 6 7
HEAD X HEAD Y HEAD Z HEAD RES HEAD SI		1.14 14.21 14.41 20.90	~4.49 -1.16 -0.76 0.37	3785.00 3637.00 3909.00 3839.00 3839.00 3793.00	3924.00 3881.00 3841.00 3603.00 4182.00 4052.00	đ 3
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT		14.43 33.01 27.90 51.99	11.48 5.96 26.29 15.15	3818.00 3869.00 3923.00 3869.00 3860.00	3861.00 4094.00 3828.00 4100.00 4062.00	14
SHD REEL BY BY SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER TOTAL SHOULDER		512.05 399 400.106 652.10 1200.70	1.59 25.32 28.19 7.66 53.70	3875.00 3873.00 3849.00 3876.00 3869.00	3829.00 3825.00 4080.00 3828.00 3823.00	15 17
LF LAP BELT AT LAP BELT TOTAL LAP TOTAL LAP / WT		0.83 34.00 39.74 73. 47 0. 50	0.37 0.00 4.75 7.57 0.05	3669.00 3933.00 3934.00 3933.00 3933.00	3623.00 3634.00 3629.00 3632.00 3832.00	8
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X		28.20 54.82 52.35 77.04	-34.16 -119.01 -53.42 -171.85	3915.00 3936.00 3792.00 3894.00	3846.00 3838.00 3837.00 3837.00	10 18 19
SEAT LNK Y LF SEAT PAN Z AT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z AT SEAT SEAT Z AT SEAT FORCE		71.21 274.53 340.17 979.76 1579.04 10.82 1587.72	-8.09 11.26 6.00 76.80 106.76 0.73 115.40	3888.00 3837.00 3847.00 3839.00 3839.00 3839.00	3795.00 3602.00 3651.00 3653.00 3602.00 3602.00 3649.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X AT FOOT X CT FOOT X TOTAL FOOT X LF FOOT Y		10.87 -5.17 9.52 1.31 1.18	0.79 -175.15 -124.42 -195.47 -495.03	3839.00 3675.00 3787.00 4200.00 3787.00	3649.00 3839.00 3838.00 3839.00 3839.00	26 23 20
RT FOOT Y CT FOOT Y IOTAL FOOT Y LF FOOT Z		147.70 22.75 16.66 59.84 188.53 196.30	-16.51 -129.00 -53.65 -62.77 -6.72	3833.00 3891.00 3789.00 3814.00 3849.00	4026.00 3822.00 3842.00 3842.00 3916.00	21 27 27
AT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		196.30 130.90 430.27 585.77	-7.42 -91.00 -67.71 22.88	3850,00 3831.00 3849.00 3841.00	3798.00 3778.00 3778.00 3925.00	22 25 28

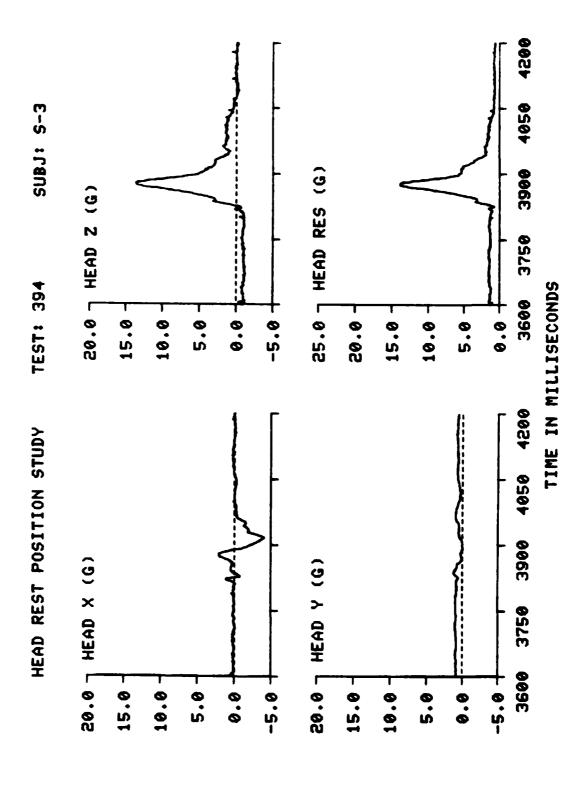
NEG SHED HAR ANG	TEST: 400	SUBJ: R-3	WT: 146.0	G: 10	GP: 2 CELL:	Ċ
CATA ID		м е, х	→ IN	T 1	T 2	G M
10V EXT TWR CORRIAGE T CARRIAGE T CARRIAGE Z CARRIAGE Z CHRRIAGE Z SECT X SECT T		10.05 1.52 0.49 12.11 10.41 -1.15 1.30 0.80	9.95 1.17 1.29 1.29 1.35 25.55 1.36	1308,000 2863.00 3893.00 3893.00 3893.00 4119.000 3899.00	2158.00 3870.00 3870.00 3655.00 3654.00 3859.00 3871.00 3914.00	865
SERI Z SERT Z (RM) CHEST X		11 94 10:65 5:12	0.21 -0.15 -1.98	3900.00 3900.00 3918.00	3659.00 3696.00 3954.00	34 5
CHEST Y CHEST Z CHEST RES CHEST SI		0.86 16.92 17.30 27.34	2 94 1.31	3896.00 2932.00 1932.00 3857.00	3908.00 3698.00 3711.00 4020.00	5 6 7
HEAD X HEAD Z HEAD RES HEAD SI		0.70 1.36 13.32 13.33 1 9. 22	5.39 1.17 9.31 0.33	3859.00 3988.00 3917.00 3917.00 3867.00	3957.30 3927.30 3527.00 4198.00 4061.00	2 3 4
HEAD HIC SHO REFL LF SHO REEL LF LF SHOULDER		12.69 64.04 49.27	20.65	3889.00 3948.00 3951.00	3935.00 4027.00 3895.00	14 16
SHD REFL RT SHD REEL RT RY SHOULDER TOTAL SHLO REFL TOTAL SHLO REFL		112.33 48.32 67.03 114.30 110.82 114.37	33.49 22.66 11.65 46.31 43.31 45.34 83.94	3950.00 3944.00 3948.00 3946.00 3947.00	4026.00 4027.00 3909.00 4047.00 3909.00	15 17
TÖTAL SHO ZUTT LE LAP BELT BT LAP BELT TOTAL LAP		224.50 1.54 56.90 68.34 125.24	0.57 16 65 25 64 43.92	3948.00 3948.00 3993.00 3993.00	3903.00 3903.00 3903.00	89
TOTAL LAP / WI CROTCH STRAP LF SEAT LNK X AT SEAT LNK X		0.86 34.06 50.51 19.93	0.30 50.85 140.58 -73.47	3993.00 3996.00 4145.00 3981.00	3903.00 3893.00 3911.00 3910.00	10 18 19
TOTAL SEAT Y SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z		56.90 66.79 420.16 417.09 732.72 1554.18 10.65	-213.48 -82.11 32.04 46.04 67.24 162.18 1.11	3995.00 3977.00 3913.00 3910.00 3914.00 3912.00	3910.00 3915.00 3608.00 3624.00 3619.00 3624.00	35 11 12 13
RES SERT FORCE RES SERT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X LF FOOT Y		1570.47 10.76 9.38 14.82 62.66 74.44 115.05	104.00 1.13 -98.21 -72.41 -130.53 -300.25 -21.21	3912.00 3912.00 3865.00 3866.00 3866.00 3896.00	3624.00 3624.00 3911.00 3912.00 3912.00 4004.00	20 23 26
RT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		26.56 25.72 85.74 194.45 162.35 135.98 423.95 481.05	-110.77 -63.28 -80.70 -57.03 300.57 -50.48 7.92	3961.00 3966.00 3883.00 3920.00 3912.00 3898.00 3920.00	3904.70 3906.00 3906.00 3967.00 3867.00 3865.00 3874.00	27 22 25 28

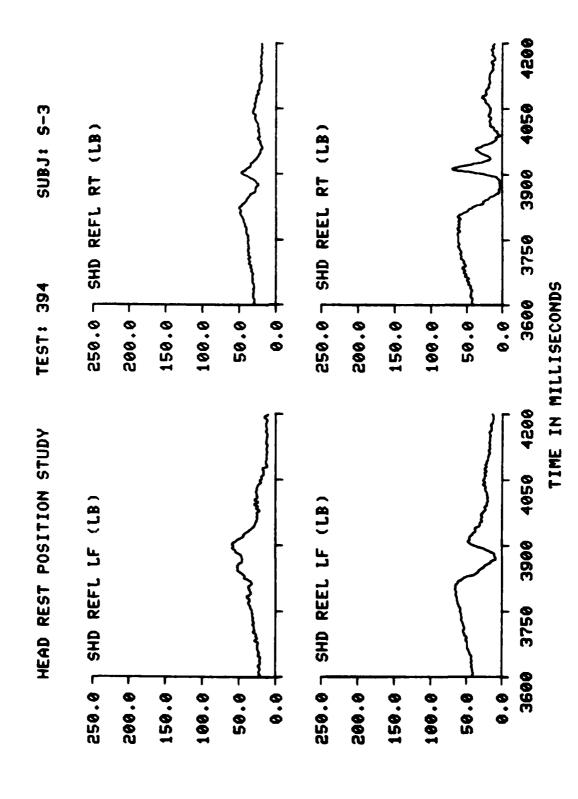
The composition The compos	NEG SHLD HAR ANG	TEST: 394	SU8.: \$-3	AT: 165	.C G: 10	3 0. 3. 65.	
100 EXI PAR CRAPA ROSE X CRARA ROSE X CRACA ROSE X CRARA ROSE X CRACA	ФАТА 10 27		чях				
9ES COOT FORCE 446.70 196.21 3849.00 3837.00 28	R (S) P X Y Z X E R (S) P X Y Z X E R (S) P X Y Z X Y Z X E F R R R R R R R R R R R R R R R R R R		537 0.637 10.201.3996714286369971428699933730844217773884.0947.1049699.694128699933750844217773884.0474.0486997.078.594388421774.048699.6941299411299429999999999999999999999999	9:15195251826881 20047 9:15195251826881 20047 9:100:5:363166881 20047 9:100:5:363166881 20047 9:100:5:363166881 2004682000000000000000000000000000000000	9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9957.000 9959.0000 9959.000 9959.000 9959.0000 99590.0000 99590.00	2-3000000000000000000000000000000000000	H 8611 9234 567 234 46 57 89 089 5123 232 222 2
· · · · · · · · · · · · · · · · · · ·	BES ก็ต่ตรั้งตลับย		4 46.70 -82.06	-196.21	3849.00 3875.00	3837.00	28

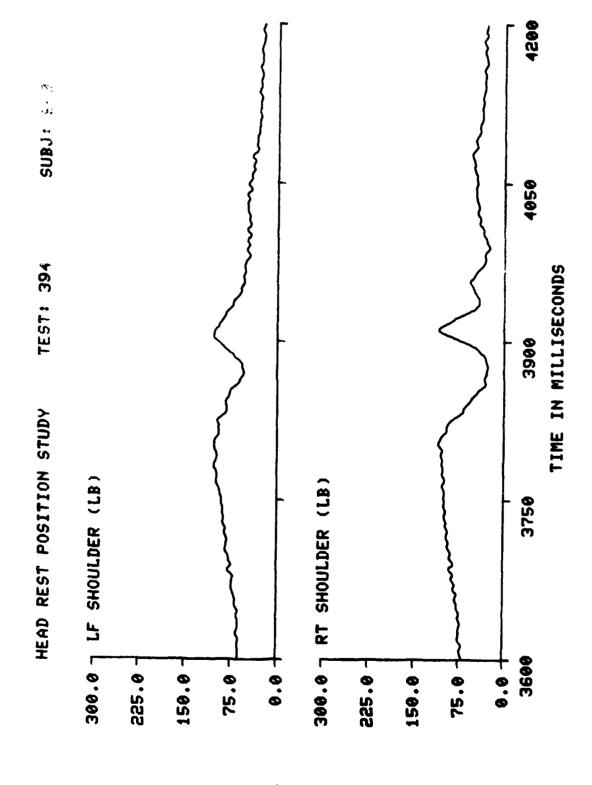


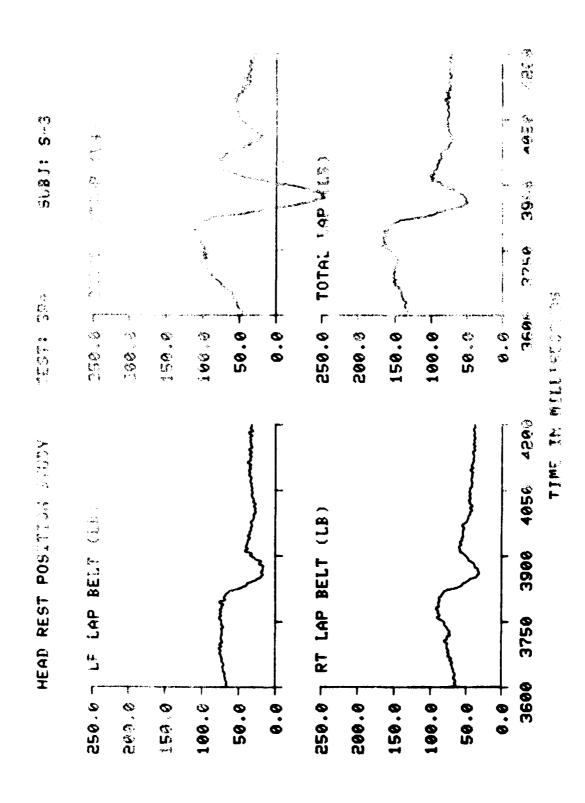


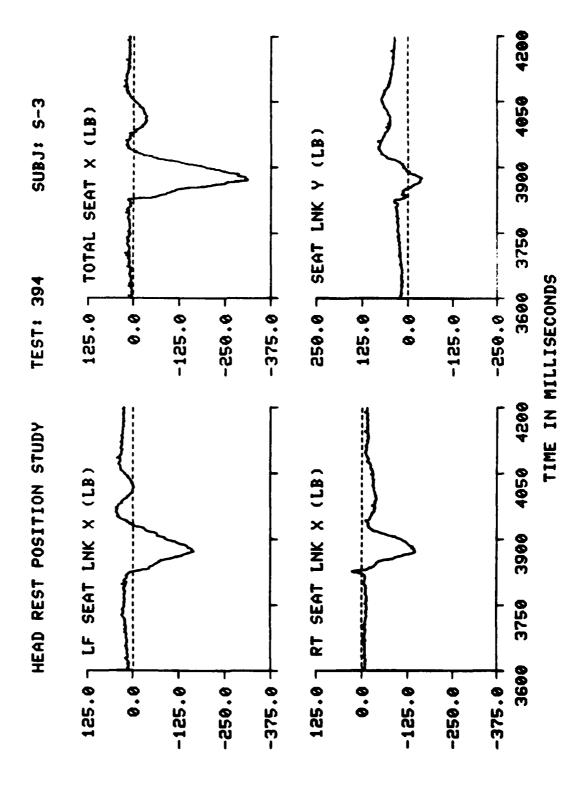


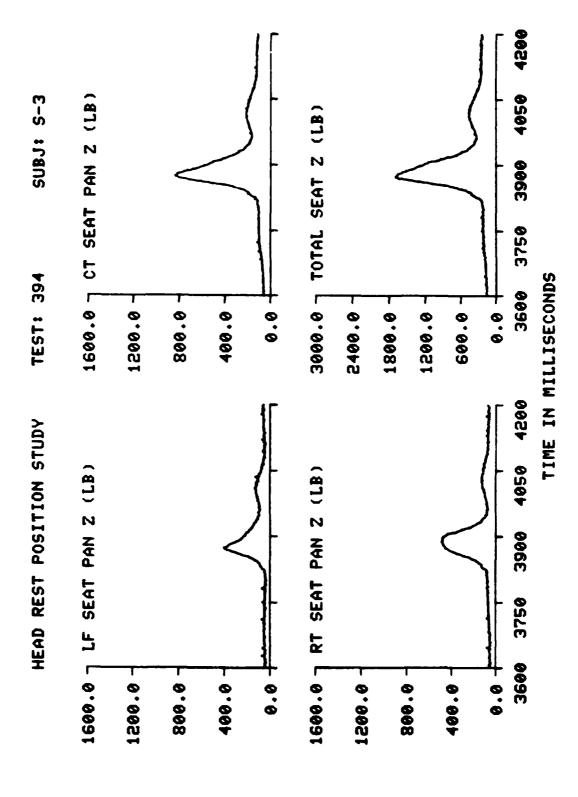


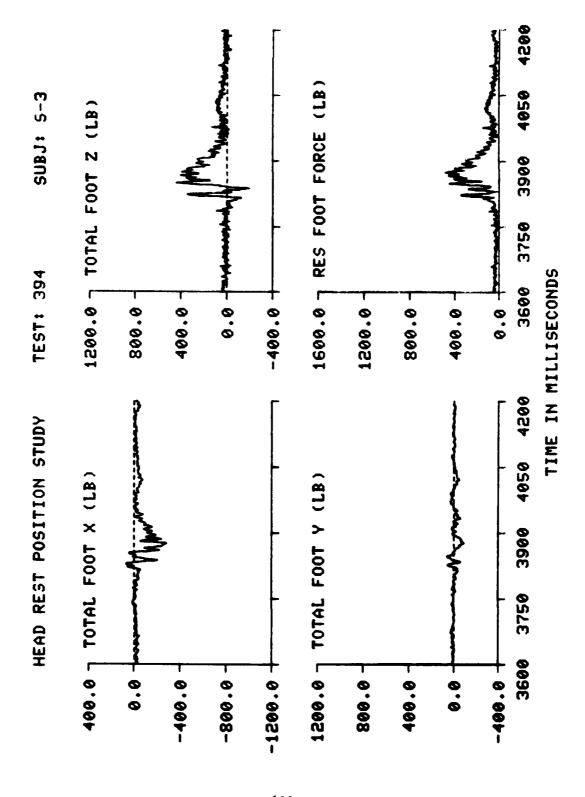












NEG SHLO HAR AND TEST: 401	SUBJ: D-1 ₩1	: 216.6	11 3* 1		
D678 10	MAX	MIN	":	* 2 	2.5
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE VEL SEAT X SEAT X SEAT Z SEAT Z CHEST X CHEST Z CH	10.02 1.168 1.168 1.256 1.256 1.279 1.1.679 1.1.679 1.1.697 2.383 1.4.997 2.385 1.1.157 1.1.157	9.93677624413 9.0041624413 -004162413 -0041641	00000000000000000000000000000000000000	100 400 100 100 100 100 100 100 100 100	The second control of
HERC HIC SHO REFL LF SHO REFL LF SHO REFL RT SHO REFL RT SHO REFL RT RI SHOULDER TOTAL SHLO REFL TOTAL SHLO REFL TOTAL SHLO REFL TOTAL SHOULDER TOTAL LAP / WT LF LAP BELT TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X RT SEAT LNK X RT SEAT LNK X SEAT LNK Y LF SEAT PAN Z TOTAL SEAT Z / HT RES SEAT FORCE	14.15 613.197 89.901 827.30 83.14 122.10 85.11 172.00 953.70 97.049 22.20 71.06 541.79 1194.79 1194.79 2194.16	2591-996563746391021 151-8563746391021 161-86-8746391021 161-86-1229-9811398 161-995 17787-1786 1786-1786	00000000000000000000000000000000000000	1.000 1.	100 cm 1 0 cm 1
RES SEAT FORCE RES SEAT FORCE / WT LF FCOT X RT FOOT X TOTAL FOOT X LF FCOT Y RT FCOT T CT FOOT T CT FOOT Z RT FCOT Z RES FCOT FORCE	213.44 10.25 16.41 -34.47 160.88 29.34 -1.30 40.94 219.56 2642.31 556.06 733.02	250.25 1.16 1.16 1.159.70 -297.57 -1.59 -124.19 -68.064 -20.43 24.29 -65.92 74.99	38 6 9 0 0 0 38 7 0 0 0 38 7 0 0 0 38 7 0 0 0 38 7 0 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 38 7 7 0 0 0 38 7 7 0 0 0 38 7 7 0 0 0 38 7 7 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	16 16 16 16 16 16 16 16 16 16 16 16 16 1	1.00 (

NEG SHLD HAR ANG	TEST: 456	SUBJ:	F-3 WT:	161.0	G: 10 GP:	1 CELL: 3	
DATA ID			MAX	MIN	T 1	T2	CH
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z CARRIAGE Z (SM) CARRIAGE VEL			10.02 1.32 0.80 12.07 10.39 -0.86	9.98 -1.06 -0.77 -0.23 -0.09 -24.54	1491.00 3797.00 3789.00 3831.00 3846.00 4192.00 3796.00	118.00 3809.00 3810.00 3751.00 3753.00 3773.00	\$8 31 1 29
SERT X SERT Y SERT Z SERT Z (SM) CHEST X CHEST X CHEST Z CHEST Z CHEST Z CHEST SI			11.29 10.56 2.07	-1.02 -1.15 -0.23 -0.13 -2.67 -2.27 -0.72	3794.00 3794.00 3837.00 3838.00 3848.00 3833.00 3861.00	3808.00 3803.00 3655.00 3655.00 3845.00 3655.00	92334 567
CHEST SI HERD X HERD Y HERD Z HERD RES HERD SI HERD HIC		٠	16.11 16.19 28.36 2.81 11.86 11.86 18.56	-1.43 -0.07 -1.07 1.68	3801.00 3838.00 3923.00 3854.00 3854.00 3807.00 3825.00	381500 385900 385900 403400 391400	2 3 4
SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT SHD REEL RT AT SHOULDER			11.86 118.895 118.895 116.863 116.863 117.057 117.057 1185.15	3.10 10.49 16.46 19.97 8.53 39.26	3914.00 3889.00 3890.00 3929.00 3890.00	3834.00 3844.00 3844.00 3833.00 3959.00 3853.00	14 16 15 17
TOTAL SHLD REFL TOTAL SHLD REFL TOTAL SHDULDER TOTAL SHD / WT LF LAP BELT TOTAL LAP TOTAL LAP TOTAL LAP TOTAL LAP			41.36 52.34 93.54	23.09 23.30 57.95 0.36 4.47 7.02	3927.00 3889.00 3890.00 3994.00 3927.00 3926.00	3834.00 4086.00 3845.00 3845.00 3840.00 3840.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X SEAT LNK Y			0.58 144.89 41.75 34.64 47.69 66.81 439.69	0.07 13.65 -170.84 -62.93 -233.16 -46.22 26.68	3922.00 3925.00 3800.00 3925.00 3901.00 3854.00	3841.00 3851.00 3845.00 3851.00 3854.00 3613.00	10 18 19 35 11
AT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE	N T		322.55 814.30 1572.26	28.19 69.90 134.78 0.84	3853.00 3855.00 3853.00 3853.00 3853.00 3853.00	3655.00 3603.00 3613.00 3613.00 3613.00	12
LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X LF FOOT Y	W I		-1114 14	0.90 -197.57 -70.23 -297.66 -558.37	3837.00 4180.00 4158.00	3613.00 3647.00 3849.00 3847.00 3846.00 4031.00	21 23 26 20 20
AT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z AES FOOT FORCE			224.57 225.60	-6.95 -129.67 -77.78 -60.61 12.15 27.47 -121.92 -26.75 118.84	3693.00 3800.00 3891.00 3834.00 3858.00 3835.00 3835.00	3650.00 3855.00 3653.00 3790.00 4178.00 3606.00 3769.00 4178.00	24 27 22 25 28
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NEG SHLD FOR AND TEST:	467 SUBJ: F-2 MT:	158.0 G: 1	O GP: 1 CELL:	3
DATR 10	MAX	MIN	T1 T2	CH
10V EXT PWR CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE Z (SM)	10.02 1.68 1.05 12.29	-1.15 38 -1.07 38 -0.64 39	21.00 1121.00 78.00 3900.00 74.00 3870.00 13.00 3833.00	48 35 31
CARPIAGE VEL SEAT X SEAT Z	10.42 -1.33 1.67 0.91 11.51	-26.23 41 -1.45 39 -1.19 38 -0.20 39	14.00 3833.00 93.00 3873.00 07.00 3899.00 84.00 3809.00 19.00 3743.00	29 32 33 34
CHEST X CHEST Y CHEST Z CHEST RES	10.42 3.53 -0.12 16.31 16.57	-2.19 39 -1.98 39 -0.26 39	20.00 3744.00 34.00 3981.00 84.00 3923.00 41.00 3742.00 41.00 3763.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD BES HEAD SI	36.38 2.74 2.11 13.10 13.36 21.32 17.85	-2.85 39 0.33 40 -2.37 39 1.63 39	81.00	2 3 4
HEAD HIC SHO REEL LF SHO REEL LF LF SHOULDER	22.07 28.74 50.46	1.44 39 1.53 39 5.13 39	11.00 3972.00 75.00 3925.00 77.00 3929.00 77.00 3926.00	14 16
SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REFL	31.81 37.18 68.89 53.74 62.66 116.39	0.67 39 11.12 39 10.73 39 3.32 39 16.27 39	73.00 3918.00 71.00 3942.00 71.00 3925.00 75.00 3925.00 75.00 3930.00 75.00 3926.00	15 17
TOTAL S-COLDER TOTAL SHO / WT LE LAP SELT RT LAP SELT TOTAL LAP TOTAL LAP / WT	0.74 40.95 59.22 99.35 0.63	0.10 39 10.50 40 22.36 40 33.48 40 0.21 40	75.00 3926.00 08.00 3947.00 14.00 3922.00 15.00 3947.00 15.00 3947.00	8 9
TÖTAL LAP / WT CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X	120.59 3.49 -2.08	-6.52 40 -264.98 37 -134.89 36	19.00 3941.00 19.00 3935.00 44.00 3935.00 02.00 3935.00	19
SEAT LAKTY LF SEAT PAN Z AT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z ATS SEAT FORCE	-3.27 48.30 539.92 369.78 678.82 1581.65	-106.05 40 36.28 39 20.59 39 37.26 39 97.39 39 0.62 39	45.00 3932.00 35.00 3636.00 38.00 3603.00 38.00 3605.00 38.00 3605.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FUOT X TOTAL FOOT X	1633.39 10.34 21.66 30.01 -7.96 39.33	0.62 39 -67.83 38 -81.01 38 -207.42 38 -349.93 36	38.00 3605.00 38.00 3605.00 81.00 3916.00 79.00 3931.00 81.00 3931.00 81.00 3931.00	20 23 26
LF FOOT Y BT FOOT Y CT FOOT Y	121.22 12.26 52.86	-8.93 39 -159.90 38 -22.78 38	23.00 3887.00 80.00 3914.00 82.00 4070.00	21 24 27
TOTAL FOOT Y LF FOOT Z RT FCOT Z CT FOOT Z TOTAL FOOT Z RES FCOT FORCE	81.35 172.33 194.06 186.98 532.23 592.61	0.62 39 8.51 39 -91.41 39 -52.82 39	81.00 3886.00 40.00 3875.00 16.00 3970.00 21.00 3875.00 23.00 3875.00 24.00 4191.00	22 25 28

NEG 4-10 -00 013 JEST: 513	SUBJ: FH4 W	7: 149.0	S: 10 SP:	2 0611: 3	
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	10.05 1.02 0.34 11.89 -1.06 2.08	9.9740 -0.173 -0.173 -29.188	1045.00 39:00 38:00 39:04.00 39:04.00 4:55:00 38:64.00	1765. 33 3453. 33 3774. 33 3824. 33 3824. 33 3824. 30	BUST WOOD #
	12.78 11.33 3.454 15.66 16.17 29.94	-0.24 -0.14 -1.78 -2.37 -1.02 1.27	3913.00 3912.00 3930.00 3962.00 3987.00 3987.00 3987.00	373700 375700 376700 396700 39700	5 6 7
변인 : 역동의 : 6 대통령인 호텔의 인도의 : 14 대통령인 : 14	1.95 13.90 13.92 20.90 16.55	0.55 -1.09 0.72	3980.00 3926.00 3926.00 3873.00 3904.00	3927.00 3927.00 3774.00 4002.00 4002.00	2394
	35.75.55 9.056 9.056 9.056 9.056 9.056 9.057 9.057 9.057	3.60 2.40 6.11 5.08 14.74 12.80 7.38 2.88	3978.00 3978.00 3957.00 3998.00 3998.00 3998.00 3998.00 39977.00	\$\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\	146
Property of the control of the contr	29.08 37.65 6.03.83 93.89 93.86 34.98 26.98 66.50	0.58 2.84 6.95 2.46 -228.75 -110.26 -338.39 -98.81 36.07	3931.00 3984.00 3999.00 3999.00 3994.00 3874.00 3874.00 3874.00	3916.000 39124.000 39124.000 39224.000 39228.000 39228.000	99
TOTAL TERM Z TOTAL T TOTA	439.57 688.26 1588.20 10.63 1621.16 10.88 -2.50 51.97 -15.94 -12.54	15.80 42.47 108.26 0.73 110.57 -123.65 -40.55 -173.36 -333.12 -2.10	3918.00 3927.00 3925.00 3926.00 3926.00 3926.00 3913.00 3869.00 3869.00 3869.00	3611.000 3605.000 3605.000 3606.000 3606.000 3915.000 3921.000 3921.000	12 :3
	21.33 0.61 56.83 186.03 190.51 199.16 490.55 506.05	-79.83 -78.60 -48.50 -14.86 -11.52 -57.24 50.03	3960.00 4003.00 3888.00 3908.00 3913.00 3914.00 3914.00	3908.00 3909.00 3917.00 3817.00 4009.00 3886.00 3878.00 4009.00	21 24 27 22 25 28

Total	NEG SHLD HAR ANG I	EST: 463 SUBJ:	G-3 M	T: 161.0	G: 10 GP:	1 CELL:	i i
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	TO THE STATE OF TH	EST: 463 SUBJ:	X 2990181816526602776996507776928555515949895773339965691711040122222222222222222222222222222222	N 93.61446099388869954 70339 3361024302548198777746772866169 845674302548198777746772866169 84567430254819877774677286616 11 12 6 4 5 1 1 1 2 6 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	11 200000000000000000000000000000000000	2	10 10 10 10 10 10 10 10 10 10 10 10 10 1

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NEG SHLO HAR ANG TEST: 465 SUBJ: G-2 WT: 118.0 G: 10 GP: 1 CILL: J
                                                                                                                                                                    DATA ID
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NEG SHLD HAR ANG	TEST:	524	SúBJ:	H-3	WT: 186.	0 G: 10 (SP: 2 CELL:	G
<u></u> <i>Del</i> 10				MAX	MI		12	CH
10V EXT PWR CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE VEL SEAT X SEAT X				10.05 0.83 0.79 11.31 10.64 -1.22	9. -1. -0. -0. -25.	12 3869.0 85 3992.0 22 3865.0 07 3864.0 87 4159.0	3852.33 3997.00 3775.00 3775.00 3807.00	48 35 31 1
SEAT Z SEAT Z (SM) CHEST X CHEST Z CHEST RES				0.99 11.68 11.04 4.28 0.22 17.56	-0. -0. -3. -1. 0.	86 3997.0 17 3870.0 12 3871.0	0 4021.00 0 3674.00 0 3675.00 0 3949.00 0 3919.00 0 3933.00	29 32 33 34 5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD SES HEAD SI HEAD HIC SHD REFL LF				36.85 3.05 2.02 11.37 11.76 19.03 16.56	-2.5 -0.1 -1.2	3823.0 32 3880.0 11 4132.0 28 3883.0 17 3883.0 3835.0	0 3950.00 3940.00 3860.00 3660.00 3963.00 3963.00	2 3 4
SHD REEL LF LF SHOULDER SHD REEL BY				44.39 52.59 81.87	0.3 -2.5 3.9	3953.0 0 3934.0	0 3871.00 4025.00	1 ų 1 6
SHD REEL AT RT SHOULDER TOTAL SHLD REEL TOTAL SHLD REEL TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER LER BELT				45.64 66.58 103.83 87.34 113.38 178.60 0.96	4.9 -3.6 7.7 -1.3 7.8	10 3940.0 18 3930.0 18 3931.0 19 3953.0 12 3931.0 14 3935.0	0 4017.00 3872.00 3872.00 3873.00 4013.00	15 17
BT LAP BELT TOTAL LAP TOTAL LAP / NT				37.98 54.81 89.02 0.48	16.3 21.3 38.7	9 3955.00 4 3950.00 8 3954.00	3904.00 3869.00 3869.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X SEAT LNK Y				136.30 38.92 87.78 74.06 77.55	0.2 -26.7 -182.8 -87.7 -267.2	9 3814.00 8 3934.00 1 3950.00	3878.00 3879.00 3888.00 3888.00	10 18 19
LF SEAT PAN Z Z RT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z WT RES SEAT FORCE / WES SEAT FORCE / WES SEAT X	ī			363.11 557.62 899.85 1808.15 9.72 1827.47 9.83	-18.4 19.4 28.9 56.5 121.3 0.6 123.5	8 3879.00 7 3888.00 8 3884.00 6 3887.00 5 3887.00 4 3887.00	3604.00 3643.00 3643.00 3642.00 3642.00 3642.00	35 11 12 13
RT FOOT X CT FOOT X TOTAL FOOT X LF FOOT Y RT FOOT Y				0.13 14.96 -0.75 12.53 123.44	-138.2 -77.5 -230.6 -434.9 -5.8	2 3827.00 3827.00 3894.00	3685.00 3902.00 3886.00	20 23 26
CT FOOT Y TOTAL FOOT Y LF FOOT Z AT FOOT Z CT FOOT Z OTAL FOOT Z RES FOOT FORCE				33.76 22.77 49.85 180.82 206.45 207.67 539.15 672.39	-103.7 -57.3 -52.3 -52.3 -4.4 -65.5 -42.5 -69.1	3920.00 3997.00 3920.00 3868.00 3894.00 3872.00	3869.00 3882.00 3871.00 4018.00	21 27 27 22 25 28

NEG SHLD HAP ANG	TEST: 520	SUBJ:	H-5 WT:	136.0	G: 10 GP:	2 CEUL:	5
5a1a 15			MAX	M ; N	' 1	• :	
OV EXT PWR Scandade X Scandade Z CGGGT X SCAN X SECT X			10.05 1.39 0.76 11.14 10.34 2.17 0.54 12.18 11.22	9.96 -10.75 -0.29 -0.15 -25.99 -1.75 -1.18 -0.14	1509.00 3841.00 3941.00 3908.00 39077.00 3661.00 38075.00 3915	364 20 10 10 10 10 10 10 10 10 10 10 10 10 10	A 10 MUNITER DIOLE
CHEST Y CHEST Z CHEST RES			1.35 -0.18 15.73 15.78	-3.19 -2.43 -0.65 -0.68	3998.00 3982.00 3921.00 3921.00	3957.00 3959.00 3667.00 3737.00	5 6 7
04807 SI HE400 X X HE600 RES HE600 RES HE600 RES			15.78 29.33 2.53 3.30 13.12 13.48 21.82	-2.32 1.29 -1.36 1.56	3869.00 3920.00 3990.00 3932.00 3932.00 3879.00	4010.00	2 3 +
SHI REFL LF SHI REFL LF SHI SHIPL LF			17.86 41.75 59.97 88.97	10.89 13.61 24.78	3900.00 3980.00 3950.00 3952.00	3962.80 3909.00 3910.80	14 15
TO SEE THE HERE OF THE PROPERTY OF THE HERE OF THE PROPERTY OF			38.32 56.03 93.94 69.00 114.10 181.18	14.77 13.92 35.95 25.66 34.62 50.90	3939.00 3945.00 3945.00 3981.00 3951.00 3946.00	3910-000 390-000 390-000 390-000 390-000 390-000	15
1019L 580 / WT LE LAP BELT BI LAP BELT TOTAL LAP			39.72 47.45 87.04	0.45 3.09 3.38 8.63	3946.00 4010.00 4011.00 4011.00	3914.00 3920.00	9 9
CROTOH STRAP LE SEAT LNK X BI SEAT LNK X			0.64 74.30 40.36 43.71	0.06 8.10 -160.71 -48.53	4011.00 4013.00 4164.00 3965.00	3923.00 3923.00 3922.00 3922.00	10
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z WT RES SEAT FORCE AES SEAT FORCE / LF FOOT X RT FOOT X			47.30 56.34 314.34 348.76 777.13 1434.31 10.55 1447.76	-209.23 -28.75 19.94 17.32 59.37 117.62 0.86 121.63	3868.00 3983.00 3924.00 3927.00 3926.00 3926.00	3922.50 3915.20 3915.20 3615.00 3696.20 3615.00 3615.00	35 11 12 13
LI FUUL X	WТ		10.65 -12.26 21.71 -46.60	0.89 -145.17 -61.12 -212.07	3926.00 3687.00 3894.00 3868.00	3615.00 3930.00 3929.00 3930.00	20 23 26
TOTAL FOOT X LF FOOT Y RT FOOT Y CT FOOT Y			-66.52 124.49 23.45	-414.75 -3.91 -109.67	3868.00 3921.00 3959.00 3875.00	3929.00 4084.00 3913.00 3979.00	21 24 27
TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FOOCE			7.21 42.14 200.44 189.48 121.17 474.70 595.09	-54.20 -40.09 -5.65 -154.99 -91.27 108.26	3876.00 3876.00 3919.00 3915.00 3895.00 3918.00	3993.00 3993.000 3887.00 3867.00 3864.00 3869.00	22 25 8

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NEG SHLO HAR ANG	1857: 469	Sugl:	M-c WT:	175.6	6: 13 GP:	$1 \leq 2 \leq \ell \leq \ell \leq \ell$	
DATA ID			M □ X 	MIN	<u> 7 : </u>	T 3 	; <u>-</u>
10V EXT PWR CARBIAGE X CARBIAGE X CARBIAGE Z CARBIAGE Z (SM)			10.02 1.64 0.73 11.87	9.98 -1.22 -0.84 -0.39	872.00 3819.00 3826.00 3854.00	60000000000000000000000000000000000000	45 31 1
CHRRIAGE VEL SEAT X SEAT Y SEAT Z			10.19 -1.24 1.99 1.34 11.44	-0.10 -26.20 -1.29 -2.03 -0.22	3855.00 4161.00 3825.00 3822.00 3860.00	3753.00 3822.00 3847.00 3827.00 3650.00	23
SEAT Z (SM) CHEST X CHEST Z CHEST BES			10.33 1.83 0.10 18.66 18.75	-0.12 -2.51 -1.38 -0.35 0.65	3861.00 3861.00 3867.00 3891.00	3690.70 3914.00 3661.00 3794.00	34 5 6 7
CHEST SI HEAD X HEAD Y HEAD Z			32.05 1.12 1.49 11.46 11.47	-5.15 -0.14 -0.56 0.83	3821.00 3962.00 3950.00 3875.00 3875.00	377500 377500 390500 390500 3600 4200	2 3 4
HEAD RES HEAD SI HEAD SI HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER			18.64 14.06 33.04 35.97 64.49	6.38 4.30 11.60	3827.00 3843.00 3945.00 3922.00 3922.00	4086.00 3931.00 3858.00 3865.00 3857.00	14 16
SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHDD REEL TOTAL SHOULDER			29.71 41.26 70.63 57.48 69.27 121.85	12.23 4.55 17.74 18.94 9.34	3905.00 3907.00 3907.00 3944.00 3910.00	38657.E.37.E.37.E.37.E.37.E.37.E.37.E.37.E.	15 17
TOTAL SHO / WT LF LAP BELT AT LAP BELT TOTAL LAP TOTAL LAP / WT CROTCH STRAP			0.70 28.71 43.25 69.81 0.40	29.34 0.17 4.13 14.02 19.63 0.11	3910.00 4097.00 3964.00 3966.00	3857.00 3857.00 3893.00 3864.00 3893.00	8 9
CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X			105.62 24.58 1.54	-0.78 -176.49 -133.74 -308.50	3952.00 4171.00 3826.00 3945.00	3860.00 3877.00 3876.00 3875.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z HT			67.16 408.63 428.79 921.87 1749.54	~54.37 53.89 44.80 71.32 175.25	3949.00 3876.00 3878.00 3879.00 3878.00	3868.00 3622.00 3604.00 3668.00 3604.00	35 11 12 13
RES SEAT FURCE	ыт		10.00 1776.34 10.15 24.97 21.54 0.49	1.00 175.39 1.00 -79.91 -83.32	3878.00 3878.00 3878.00 3822.00 3850.00	3604.00 3604.00 3604.00 3667.00 3882.00 3868.00	20 23 26
TOTAL FOOT X LF FOOT Y RT FOOT Y CT FOOT Y	,		116.97 14.64 43.48	-228.54 -375.88 -8.80 -136.03 -46.40	3823.00 3822.00 3856.00 3914.00 3822.00 3822.00	3867.00 3801.00 3874.00 3867.00	21 24 27
TOTAL FOOT Y LF FOOT Z AT FOOT Z CT FOOT Z TOTAL FOOT Z			60.70 195.07 204.07 200.11	-82.20 -31.81 -16.40 -97.84 -108.52	3822.00 3846.00 3873.00 3877.00 3877.00	3871.00 3815.00 3835.00 3816.00 3815.00	22 25 28
RES FOOT FORCE			574.03	74.79	3875.00	4084.00	

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-- -- -- TEST: 463 SUBJ: MIO -- WT: 148.0 G: 10 GP: 1 CELL: 3
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NEG SHLD HAR ANG TEST: 492 SUB	J: P-3 NT: 202.0	G: 10 GP:	2 CELL: G
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TOTALL PRODUCT TOTAL LATINGT FOOOT TOTAL LATINGT TOTAL LATINGT TOTAL LATINGT SEATT FOOOTT TOTAL LATINGT SEATT		795745600000000000000000000000000000000000	

NEG SHLD HAR ANG	TEST: 468	SUBJ:	R+2 ₩T:	145.0	G: 10 GP:	1 CELL:	Ĵ
D978 ID			MAX	MIN	T !	T2	54
104 A A A A A A A A A A A A A A A A A A A			10.02 1.397 12.15 10.426 1.757 11.255 0.761 10.593 4.934 15.531 13.62	9.905 -0.615 -0.155 -0.157 -1.157 -1.158 -0.71 -1.258 -0.71 -1.258	430.000 385.000 389088.000 389088.000 38889.000 388556.000 388556.000 389526.000 3895266.000 389511.000 38968.339566.000	9515210000000000000000000000000000000000	- 9611 9254 567 254 2555
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LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X LF FOOT Y RT FOOT Y TOTAL FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z TOTAL FOOT Z RT FOOT Z TOTAL FOOT Z RES FOOT FORCE			36.65 123.68 20.84 8.17 67.57 159.94 225.17	-139.33 -72.08 -207.50 -417.13 -4.72 -112.28 -87.06 -71.61 -32.34 -29.92 -103.16 -126.68 13.80	3895.00 3858.00 3858.00 3890.00 3878.00 3878.00 3914.00 3914.00 3914.00	3906.00 3905.00 3906.00 3847.00 3906.00 3908.00 3849.00 3869.00 3869.00	2036 2147 227 228

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NES 3-10 HAR AND TEST: 464 SUBJ: 8-3 MT: 147.0 G: 10 GP: 1 05:10 S
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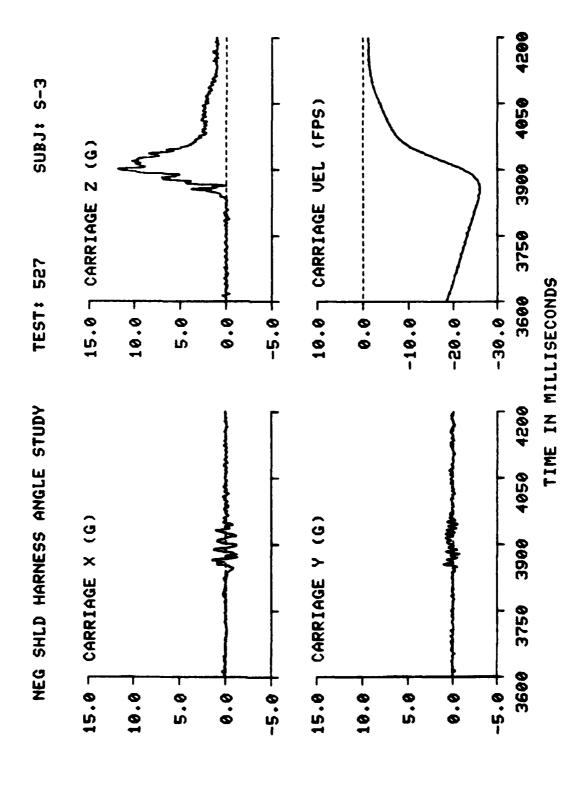
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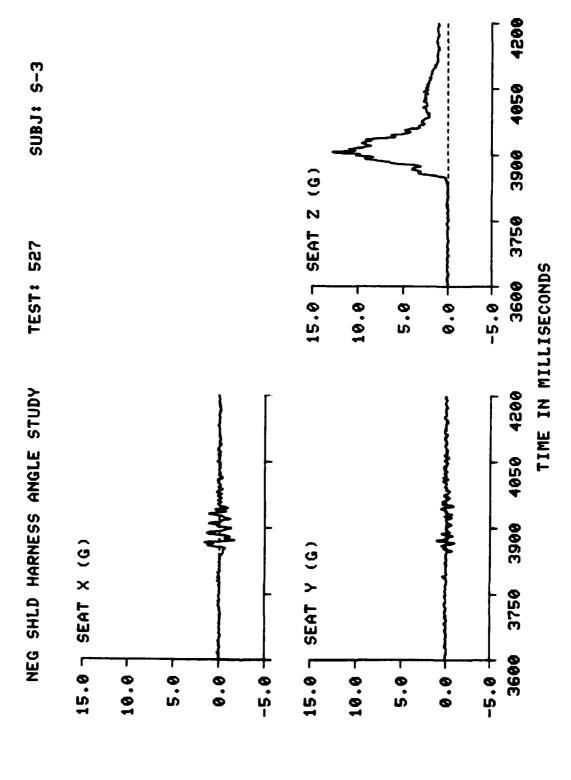
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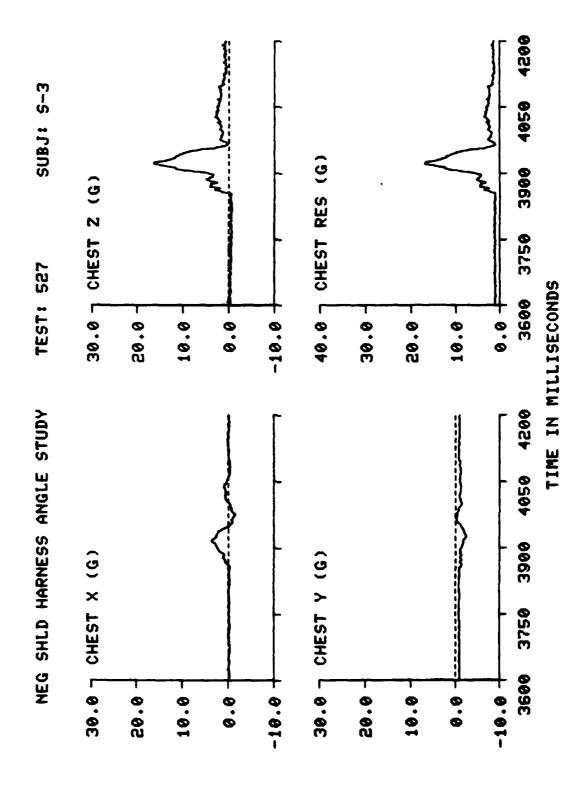
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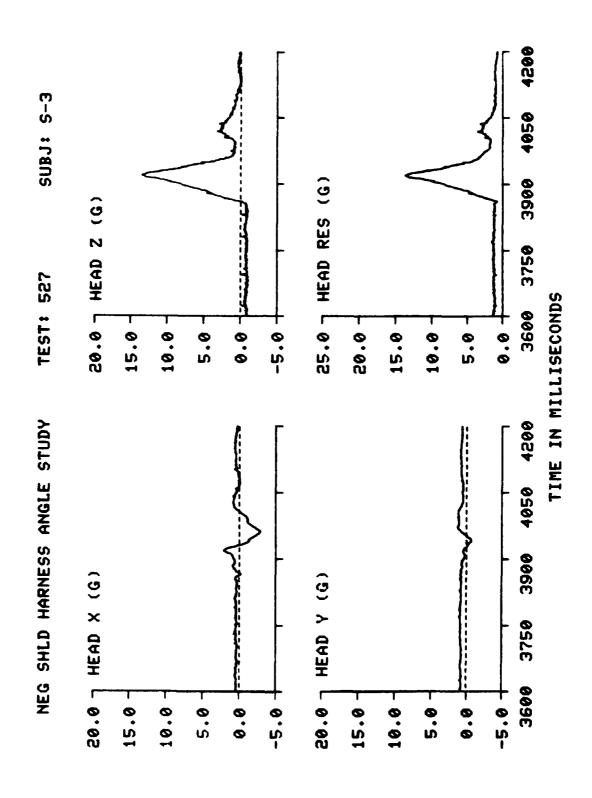
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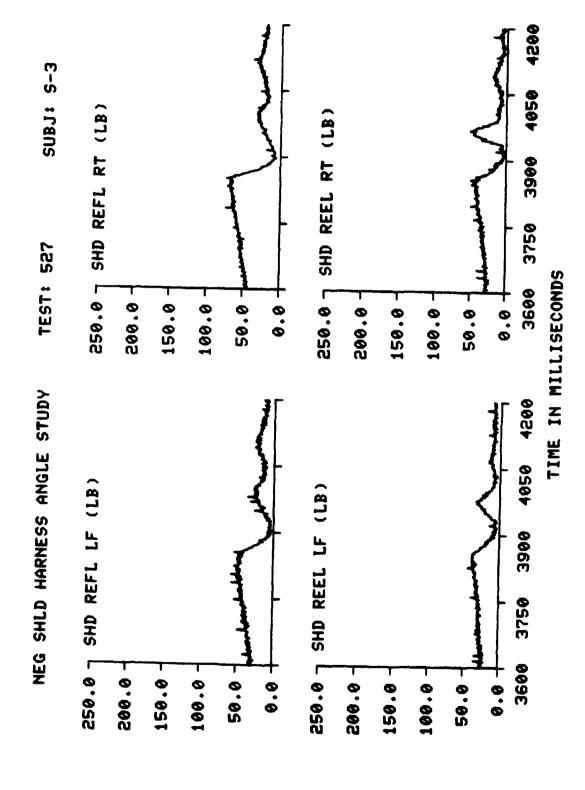
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3876.00
3875.00
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    3930.00
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  3910.00
3910.00
          TOTAL FOOT Z
RES FOOT FORCE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   -226.10
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               585.10
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    3910.00
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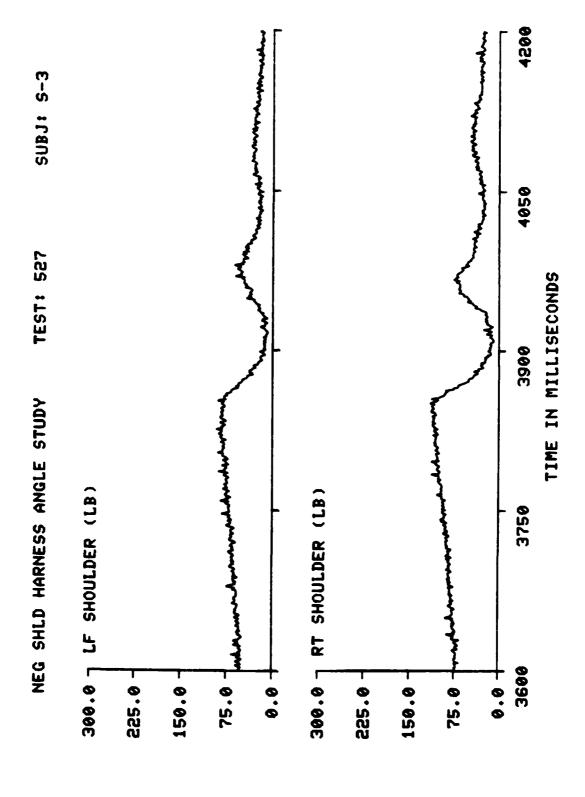


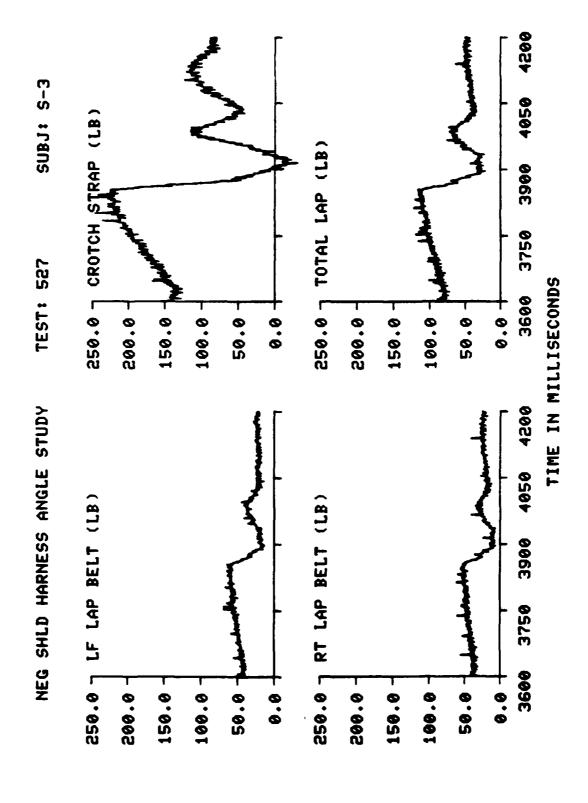


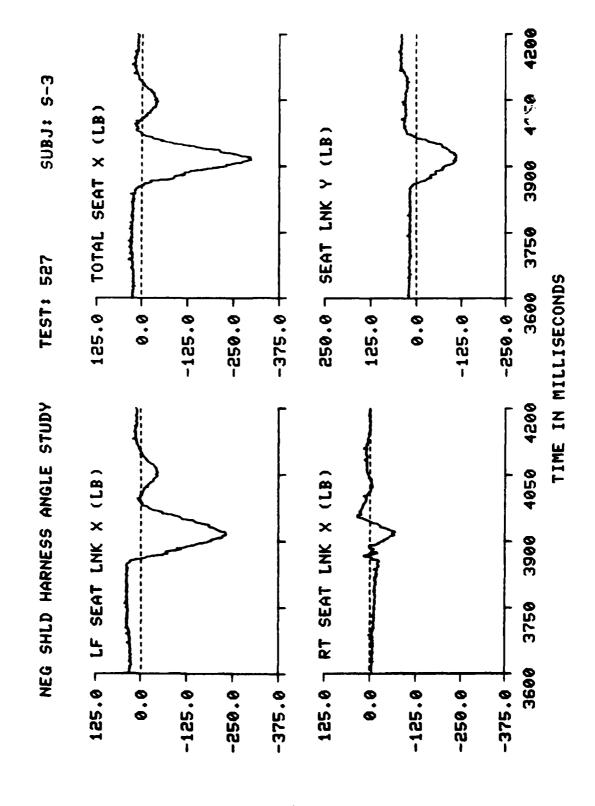


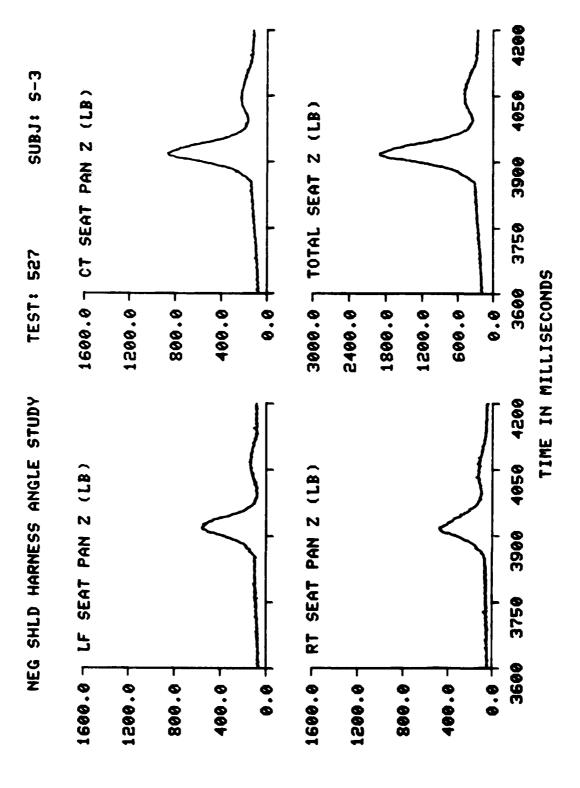


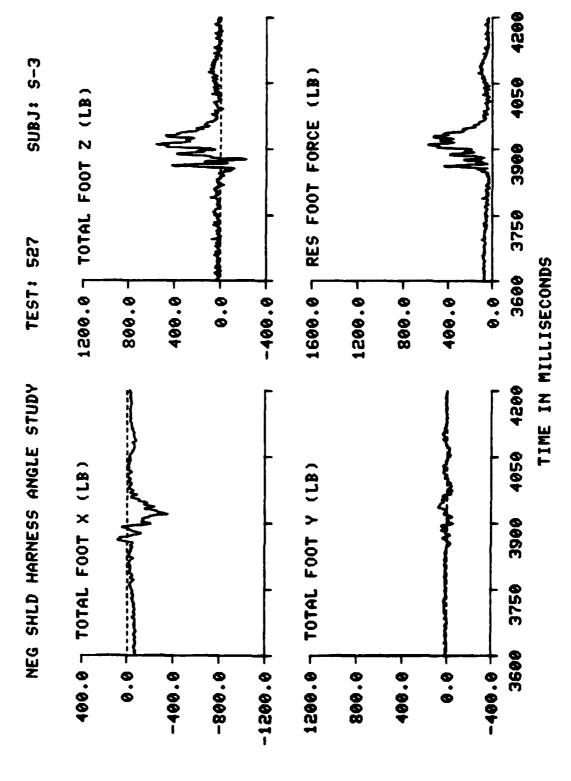












NEG 5HLD HAR ANG TEST: 528	SUBJ: F-3 WT:	158.0 G: 10 GP:	1 CELL: H
DATA ID	MAX	MIN T1	T2 CH
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z CARRIAGE Z (SM)	10.05 1.98 1.04 11.86	9.96 610.00 -1.55 3797.00 -1.18 3789.00 -0.42 3831.00	605.00 48 3830.00 36 3783.00 31 3723.00 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT T SEAT Z SEAT Z SEAT Z	10.97 -1.05 2.24 1.43 12.51	-0.13 3831.00 -26.11 4139.00 -1.64 3798.00 -1.70 3795.00 -0.22 3838.00	3723.00 3791.00 3807.00 3807.00 3800.00 33 3638.00
SEAT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES	11.46 2.04 -0.59 15.64 15.73	-0.12 3839.00 -2.12 3807.00 -2.68 3827.00 -0.96 3861.00 1.01 3861.00	3638.00 3881.00 53849.00 63748.00 73781.00
CHEST SI HEAD X HEAD Y HEAD Z HEAD RES	28.38 2.17 2.75 13.18 13.26	3795.00 -1.89 3834.00 0.25 3914.00 -1.27 3849.00 1.37 3849.00	3928.00 3883.00 3857.00 3607.00 4014.00
HEAD SI HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER	20.88 18.01 46.40 43.31 85.99	3805.00 3825.00 10.84 3866.00 3.20 3880.00 22.20 3879.00	3924.00 3887.00 4071.00 14 3839.00 16 4096.00
SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHLD REEL	47.82 60.84 102.40 92.82 102.05	22.08 3861.00 10.38 3876.00 46.51 3875.00 34.03 3865.00 13.93 3879.00	4100.00 15 3843.00 17 4094.00 4094.00 3842.00
TOTAL SHOULDER TOTAL SHO / WT LF LAP BELT BT LAP BELT TOTAL LAP TOTAL LAP / WT	186.08 1.18 54.11 68.35 119.18 0.75	69.18 3878.00 0.44 3878.00 18.72 3930.00 20.88 3925.00 42.53 3927.00 0.27 3927.00	4095.00 4095.00 3844.00 8 3840.00 9 3843.00 3843.00
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X	99.33 47.32 - 55 .68	-9.92 3933.00 -217.70 4075.00 -39.01 3799.00 -256.70 3759.00	3859.00 10 3849.00 18 3849.00 19 3849.00
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE	43.08 - 486.12 374.62 788.57 1642.26 10.39 1665.94	118.56 3785.00 29.86 3846.00 30.28 3850.00 56.37 3848.00 0.86 3848.00 147.47 3848.00	3844.00 35 3602.00 11 4198.00 12 3605.00 13 3600.00 3600.00
RES SEAT FORCE / WT LF FOOT X AT FOOT X CT FOOT X TOTAL FOOT X	26.28 8.80 -	0.93 3848.00 170.95 3798.00 -83.87 3795.00 241.64 3799.00 493.77 3799.00	3603.00 3531.00 20 3831.00 23 3832.00 26 3831.00
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y	142.55 25.63 5.59 44.47	-10.31 3838.00 -87.96 3798.00 -75.40 3801.00 -75.71 3873.00	3926.00 21 3847.00 24 3844.00 27 3845.00
LF FOOT Z AT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE	190.86 203.24 205.28 ~	-2.92 3840.00 22.39 3839.00 190.23 3798.00 122.31 3840.00 113.20 3840.00	3809.00 22 4152.00 25 3810.00 28 3809.00 4180.00

NEG SHLO HAR ANG TEST: 498	SUBJ: F-2 WT:	161.0 S: 10	GP: 1 SECU: H	
DI ATAO	MAX	MIN TI	T2 C	
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z	10.05 1.58 0.77 12.52	9.97 808. -1.48 3380. -0.55 3882. -0.47 3912.	00 3887.00 3: 00 3866.00 3 00 3632.00	6
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z SEAT Z	10.44 -1.18 1.03 0.79 11.78	-0.16 3912. -25.89 4157. -1.02 3921. -1.09 4023. -0.18 3918.	00 3870.00 29 00 3950.00 39 00 3886.00 39 00 3745.00 39	9 2 3 4
SEAT Z (SM) CHEST X CHEST Z CHEST RES	10.66 4.24 -0.42 16.98 17.63	-0.13 3919. -1.84 3935. -2.38 3897. -0.76 3935. 0.67 3935.	00 3970.00 00 3939.00 00 4026.00 00 3766.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD RES HEAD SI	37.12 3.03 2.29 13.13 13.48 21.39	3877. -2.94 3936. 0.62 4016. -2.10 3936. 1.56 3936. 3889.	00 3972.00 00 3927.00 00 4027.00 00 4139.00 00 3998.00	2 3 4
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT	17.63 35.34 45.97 77.38 50.72	3906. 8.25 3958. 0.39 3967. 19.36 3968. 11.09 3956.	00 4019.00 10 00 3925.00 10 00 4004.00	6
SHD REEL BT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER TOTAL SHO / WT	67.07 112.19 85.89 112.85 188.54 1.17	2.07 3966. 25.27 3965. 19.46 3957. 2.55 3966. 47.00 3966. 0.29 3966.	00 3926.0- 17 00 4054.33 00 4050.03 00 3925.00 00 4062.00	7
LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT	53.46 80.03 112.72 0.70	26.98 4006. 29.77 4024. 58.48 4006. 0.36 4006.	00 3914.00 4 00 3921.00 9 00 3921.00 00 3921.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X	117.39 12.43 3.98	-9.25 4022. -258.92 3730. -128.83 3644.	00 3933.00 10 00 3932.00 11	8
SEAT LAK T LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE		-385.45 3664. -134.84 4197. 30.31 3933. 31.67 3932. 24.45 3937. 101.37 3935. 0.63 3935. 101.70 3935.	00 3933.00 35 00 3628.00 1 00 3613.00 1 00 3601.00 1 00 3613.00	1 2
RES SERT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X	10.41 22.53 33.43 10.04 48.33	0.63 3935. -53.42 3879. -72.31 3878. -192.10 3881. -317.83 3880.	00 3613.00 00 3929.00 20 00 3930.00 20 00 3930.00 20 00 3930.00	3 6
LF FOOT Y AT FOOT Y CT FOOT Y TOTAL FOOT Y	110.18 15.14 42.26 57.44	-12.98 3914. -144.60 4105. -15.59 3878. -48.02 3881.	00 3913.00 25 00 4075.00 27 00 3929.00	
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE	186.77 174.92 204.02 531.05 569.08	-3.86 3915. 0.09 3922. -70.15 3921. -37.26 3922. 57.93 3915.	00 4026.00 25 00 3890.00 25	2 5 8

NEO SH.S HAR ANG TEST: Sub Su Data ID	Bere e 4 MAX	(: .ୟଞ୍ଚିତ MIN	⇒. 10 GP T1	: 2 CELL: T 2	- CH
:JV EXT PWR CRRRIAGE X CARRIAGE T	10.05 1.52 0.93	9.96	314.00 3874.00	74.00 3881.00	48 35
CARRIAGE Z CARRIAGE Z (SM) CARRIAGE VEL SEAT X	12.08 1 0. 31 -1.26	-0.58 -0.22 -0.07 25.13 -1.17	3865.00 3905.00 3906.00 4183.00	3 89 0.00 3738.00 3738.00 3867.00	31
SEAT Z SEAT Z SEAT Z CHEST X	1.42 1.90 11.22 10.41	-1.17 -2.20 -0.15 -0.07	3877.00 3872.00 3911.00	3918.00 3878.00 3748.00	29 32 33 34
CHEST X CHEST Z CHEST BES CHEST SI	2.61 0.08 15.01	-1.53 -2.11 -0.72 -0.73	3912.00 3932.00 3969.00 3922.00 3923.00	3747.00 3969.00 3936.00 3734.00 3730.00	5 6 7
HEAD X HEAD T HEAD Z HEAD RES	90.28 0.71 2.99 11.28	0.61 -1.64	3869.00 3921.00 3998.00	4105.00 3960.00 3935.00 3770.00	3 5
HEAD SI HEAD HIC SMD REFL LF SHD REEL LF	11.30 15.47 11.76 27.81 32.11	2.06 a 5i	3931.00 3891.00 3905.00	4171.00 4011.00 3966.00 4032.00	
LF SHOULDER SHD REFL RT SHD REEL RT	32.11 52.60 39.86 41.67	2, 82 16, 17 22, 50 25, 75	3973.00 3972.00 4010.00	3909.00 3910.00 3908.00	1 4 1 6 1 5
RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER	77.36 58.73 64.24 110.44	35.87 6 18	3996.00 3997.00 3944.00 3978.00	3914.00 3914.00 3907.00 3911.00	15 17
TOTAL SHO / WT LE LAP BELT BT LAP BELT TOTAL LAP TOTAL LAP / WT	0.74 41.53 68.26 109.03	42.41 0.28 16.18 23.32	3979.00 3979.00 3991.00 3989.00 3990.00	3912.00 3912.00 3938.00 3914.00	8
CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X	0.73 134.85 45.30 25.70	43.31 0.29 -2.41 -219.15 -81.90	3990.00 4001.00 3722.00	3919.00 3919.00 3926.00 3928.00	10 18
SEAT LNK Y LF SEAT PAN Z BT SEAT PAN Z	62.36 31.98 641.41	-298.67 -149.11 50.03	3841.00 3841.00 4184.00 3928.00	3934.00 3932.00 3938.00 3622.00	19 35 !!
CT SERT PAN Z TOTAL SERT Z TOTAL SERT Z / WT RES SERT FORCE	549.94 485.26 1652.22	45.79 24.41 138.84 0.93	3929.00 3937.00 3930.00 3930.00	3610.00 3603.00 3603.00 3603.00	12
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X	1684.12 11.30 25.46 53.40	149.74 1.00 -94.79 -25.03	3930.00 3930.00 3875.00 3910.00	3603.00 3603.00 3922.00 3931.00	20 20
TOTAL FOOT X LF FOOT Y AT FOOT Y	11.45 55.94 126.67 30.35	-148.65 -247.28 -6.10	3875.00 3875.00 3907.00	3923.00 3922.00 3877.00	26 21
TOTAL FOOT Y LF FOOT Z RT FOOT Z	9.63 55.06 177.71	-92.12 -74.92 -70.90 -38.77	3964.00 4091.00 3947.00 3895.00	3924.00 3925.00 3924.00 3866.00	24 27
CT FOOT Z TOTAL FOOT Z RES FOOT FORCE	176.12 137.71 366.87 423.42	-1.63 -80.29 -73.10 37.70	3895.00 3907.00 3927.00 3917.00 3917.00	3815.00 3885.00 3865.00 4147.00	22 25 28

New EMLO HAR AND TEST: 504	SuBl: S-3 Wi	1 164 0). 10 °=:	1 SELL: -	
DATA 10	MAX	MIN	T 1	12	5.4 -
OV EXT PWR CARRED X CARRELAGE X CARRELAGE Z CARRELAGE Z CARRELAGE VEL SEGO Y SEGO Y SEGO Y SEGO X CHEST X CHEST Z CHEST RES	10.05 0.16 0.54 10.52 -0.87 1.10 11.462 -0.10 21.13	9.96 -1.00 -0.536 -0.08 -25.88 -1.32 -0.24 -2.34 -3.01 -2.57	2594.00 3858.00 3851.00 3851.00 4176.00 3819.00 3819.00 3857.00 3857.00 3858.00 3872.00 3889.00	2540.00 3850.00 3831.00 3777.00 3776.00 3793.00 3821.00 3667.00 3667.00 3876.00 3876.00	98611 GARGY 567
CHEST SI HEAD X HEAD Y HEAD RES HEAD SI HEAD SIC	38.53 2.522 12.98 13.34 19.89	-5.03 0.22 -1.42 1.36	3813,00 3867.00 3971.00 3867.00 3867.00 3825.00 3847.00	4137.00 3917.00 3900.00 3696.00 4054.00 3941.00 3922.00	Med U
SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHGULDER	64.24 67.40 131.40 43.53 69.255 105.09 120.82	12.66 5.38 24.94 11.83 -1.35 11.86 27.59 6.76 39.40	3897.00 3896.00 3896.00 3904.00 3906.00 3906.00 3896.00 3905.00	4094.00 854.00 3991.00 4008.00 4008.00 3993.00 3854.00	14 16 15 17
TOTAL SHO / WT LF LAP BELT TOTAL LAP TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X AT SEAT LNK X SEATLNK Y SEATLNK Y SEATLNK Y	1.35 25.65 45.35 70.28 0.43 206.71 20.25 13.81 18.15	0.24 4.05 10.51 15.30 0.09 6.31 -220.58 -117.16 -334.74 -123.14	39C4.00 39C6.00 39C7.00 39C7.00 39C7.00 39C7.00 39C9.00 39G9.00	4001.00 3856.00 3853.00 3855.00 3874.00 3874.00 3873.00 3877.00	8 9 10 18 19
LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / HT RES SEAT FORCE RES SEAT FORCE / HT LF FOOT X RT FOOT X	525.77 584.39 689.31 1798.87 10.97 1831.66 11.17 17.52 43.14	38.94 29.94 34.35 118.05 120.13 -114.48 -69.65	3875.00 3875.00 3875.00 3875.00 3875.00 3875.00 3875.00	3609.00 3669.00 3630.00 3607.00 3607.00 3607.00 3607.00 3868.00	11 12 13 20 23 26
CT FOOT X TOTAL FOOT X LF FOOT Y BT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z CT FOOT Z	15.24 42.33 121.64 16.40 25.42 51.12 167.96 155.07 210.69	-171.69 -302.70 -8.51 -111.39 -59.13 -57.24 -87.29 -27.81	3818.00 3817.00 3853.00 3963.00 3983.00 3865.00 3862.00 3858.00	3866.00 3888.00 3819.00 3662.00 3870.00 3870.00 3809.00 3966.00	26 21 27 22 25 28
TOTAL FOOT Z RES FOOT FORCE	47 0.79 535.02	-71.46 42.50	3862.00 3862.00	3807.00 3950.00	

NEG SHLD HAR ANG TEST	497 SUBJ:	G-2 WT:	119.0	G: 10 GP:	1 CELL: H	
DATA ID		MAX	MIN	T 1	12	Cn
10V EXT PHR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z (SM)		10.05 1.33 0.86 12.97 10.52	9.98 -1.48 -0.77 -0.29 -0.04	2576.00 3862.00 3838.00 3867.00 3868.00	892.00 3867.00 3979.00 3766.00	48 36 31
CARRIAGE VEL SERT X SERT Y SERT Z SERT Z SERT Z (SM)		-1.17 1.39 1.16 12.05 10.63	-26.19 -1.20 -1.19 -0.17 -0.13	4184.00 3862.00 3835.00 3873.00 3874.00	3831.00 3866.00 3842.00 3614.00 3685.00	29 32 33 34
CHEST X CHEST Z CHEST RES CHEST SI		3.64 -0.02 18.04 18.11 30.96	-2.19 -1.96 -1.08 0.82	3892.00 3881.00 3904.00 3904.00 3835.00	3928.00 3940.00 3690.00 3828.00 3954.00	5 6 7
HEAD X HEAD Y HEAD Z HEAD RES HEAD SI		0.60 2.87 12.70 12.90 18.65	-4.62 1.83 -1.63 1.89	3832.00 3872.00 3895.00 3895.00 3843.00	3931.00 3771.00 3624.00 3837.00 3958.00	2 3 4
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT		14.95 33.68 35.63 61.91 29.35	11.83 4.07 19.45 11.26	3866.00 3912.00 3932.00 3931.00 3925.00	3937.00 3990.00 3881.00 4034.00 4011.00	14 16
SHO REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHUD REEL TOTAL SHOULDER TOTAL SHO / NT		37.33 65.92 61.61 72.28 125.76 1.06	5.97 17.82 24.47 12.36 37.84	3935.00 3935.00 3916.00 3933.00 3934.00 3934.00	4020.00 4020.00 4011.00 3881.00 4020.00	17
LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT		32.86 40.14 71.58 0.60	0.32 11.37 13.18 24.76 0.21	3971.00 3950.00 3950.00 3950.00	3876.00 3875.00 3875.00 3875.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X		41.57	-1.69 -134.45 -48.82 -182.69	3960.00 3793.00 3830.00 3809.00	3893.00 3895.00 3897.00	10 18 19
SERT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE RES SEAT FORCE / WT		47.72 479.77 362.91 1295.64 10.89 1308.52	-76.80 40.73 37.49 11.87 153.14 1.29 164.00 1.38	3802.00 3884.00 3890.00 3893.00 3891.00 3892.00 3892.00	3889.00 4143.00 3639.00 3642.00 3642.00 4197.00	35 11 12 13
LF FOOT X AT FOOT X CT FOOT X TOTAL FOOT X		34.34 23.53 9.70 49.03	-36.18 -50.49 -151.30 -237.09	3837.00 3874.00 3838.00 3837.00	3869.00 3868.00 3869.00	56 53 50
LF F00T Y AT F00T Y CT F00T Y T0TAL_F00T Y		89.93 25.50 39.85 68.30	-17.50 -109.62 -25.11 -88.30	3877.00 4176.00 3997.00 3836.00	3946.00 3868.00 3880.00 3844.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		170.04 128.91 139.68 335.93 349.97	-46.88 -24.42 -99.51 -111.81 13.66	3860.00 3834.00 3875.00 3860.00 3860.00	3830.00 3981.00 3829.00 3829.00 4148.00	22 25 28

Sugar Project HAA ANG	7837: 50 9	SuBJ:	H-3 AT:	.86.0	6: :0 6P:	2 CELL:	H
77.19.13			MAX	MIN	T 1	T 2	65 10
CAN DE LAC LEMATACE LACATORE Z CARATAGE Z (SM) LAGRETAGE			10.05 1.23 1.00 11.76 10.30	9.96 -1.48 -0.68 -0.25 -0.12	20.00 3920.00 3920.00 3913.00 3926.00	1223.00 3925.00 4040.00 3629.00 3630.00	48 36 31
THRINGT TOTAL SECTION (M) CHIST			-1.10 1.61 1.06 10.91 10.28	-25.90 -2.08 -1.43 -0.14 -0.07	4172.00 3920.00 3877.00 3917.00 3919.00	3866.00 3925.00 3884.00 3746.00 3745.00	29 32 33 34
CHIST CHIST CHIST			4.20 -0.08 19.09 19.33 40.06	-2.44 -2.60 -0.64 0.54	3929.00 3996.00 3938.00 3938.00 3877.00	3983.00 3963.00 3748.00 3639.00 4013.00	5 6 7
HI W RES HI W RES HI W RES HI W RES			3.63 1.66 11.59 11.95 20.62 17.75 39.70	-3.94 0.89 -1.25 1.26	3930.00 4046.00 3942.00 3942.00 3889.00 3910.00	3984.00 3946.00 3819.00 3882.00 4002.00 3974.00	2 3 4
SHUPEL F			49.04	10.91	3980.00 3967.00	4042.00 3927.00	1 4 1 6
LE SHOULDER SHO HEEL AT SHO HEEL AT HI SHOULDER TOTAL SHLO REEL TOTAL SHLO REEL TOTAL SHOULDER			82.30 49.41 72.10 119.14 85.48 116.98 197.09	15.00 13.32 0.92 14.97 24.65 4.76 31.52	3970.00 3984.00 3974.00 3974.00 3980.00 3973.00	4050.00 4049.00 4055.00 4054.00 4050.00 3914.00	15 17
TOTAL SHD / WT LF LAP BELT RI LAP BELT TOTAL LAP TOTAL LAP / WT			1.06 32.61 33.62 64.30 0.35	0.17 12.70 13.46 26.34 0.14	3973.00 4086.00 3999.00 3999.00	4051.00 3920.00 3919.00 3920.00 3920.00	8
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X			0.35 196.24 46.98 47.38	-16.28 -208.25 -72.54	3988.00 4174.00 3882.00	3939.00 3935.00 3935.00	10 18 19
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z TOTAL SEAT Z RES SEAT FORCE			42.72 49.57 513.77 513.77 878.69 1894.36 1915.36	-280.79 -73.78 16.97 15.96 38.74 87.97 90.02	4163.00 4004.00 3937.00 3935.00 3936.00 3937.00 3937.00	3935.00 3936.00 3658.00 3612.00 3607.00 3601.00 3601.00	35 11 12 13
LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X	WT		10.30 -6.67 14.61 -49.82 -45.45	0.48 -159.47 -95.54 -269.85 -518.61	3937.00 3880.00 3881.00 3881.00 3880.00 3931.00	3601.00 3930.00 3940.00 3932.00 3931.00 4005.00	20 23 26
ŘT FÖÖT Y CT FÖÖT Y TOTAL FÖÖT Y			142.39 34.40 9.90 56.56	-2.60 -112.92 -79.99 -68.73	3968.00 3881.00 3898.00	3915.00 3937.00 3944.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE			56.56 199.24 255.29 260.86 615.47 759.12	-14.97 24.34 -104.99 -84.96 116.15	3951.00 3940.00 3934.00 3933.00 3932.00	3871.00 4124.00 3872.00 3871.00 3788.00	22 25 28

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NEG SHLD HAR ANG TEST: 521 SUBJ: H-4 WT: 187.0 G: 10 GP: 2 CE.1; H
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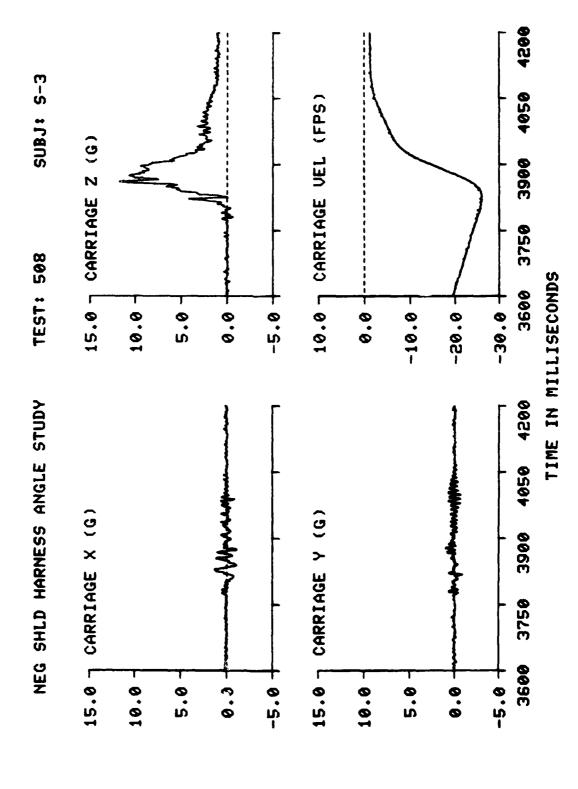
NEG SHLD HAR AND TEST: 502	SUBJ: K-1 WT:	180.0 G: 10 GP:	1 CELL: H
DATA ID	MAX	MIN T1	T2 CH
IOV EXT PHR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z CARRIAGE Z (SM)	10.05 1.18 0.68 12.01	9.96 15.00 -0.92 3912.00 -0.50 3922.00 -0.22 3905.00 -0.09 3905.00	2957.00 48 3918.00 36 3861.00 31 3821.00 1
CARRIAGE VEL SEAT X SEAT Y	10.34 -1.13 1.26 1.72 10.95	-0.09 3905.00 -26.24 4146.00 -1.31 3913.00 -1.71 3874.00 -0.32 3910.00	3834.00 3865.00 29 3918.00 32 3880.00 33 3741.00 34
SEAT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES	10.31 7.06 -0.33 16.46 17.26	-0.13 3912.00 -0.64 3927.00 -2.39 3916.00 -0.68 3936.00 0.95 3935.00	3741.00 3971.00 53936.00 63832.00 73786.00
CHEST SI HEAD X HEAD Y HEAD RES HEAD SI	34.91 1.34 1.98 13.45 13.50 19.79	3871.00 -4.93 3928.00 -0.55 3973.00 -1.35 3926.00 1.22 3926.00 3883.00	3995.00 3962.00 2 3942.00 3 3667.00 4 4199.00
HĒAD HĪC SHD REFL LF SHD REEL LF LF SHOULDER	16.24 34.53 34.48 63.36	3904.00 12.86 3952.00 2.93 3961.00 22.44 3962.00	3967.00 4037.00 14 3915.00 16 4047.00
SHO REFL RT SHO REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHLD REEL	46.91 50.00 91.03 80.19 75.83 149.82	22.51 3937.00 -1.12 3951.00 27.60 3951.00 38.24 3936.00 3.83 3956.00 57.20 3953.00 0.32 3953.00	4009.00 15 3909.00 17 3907.00 4016.00 3909.00 3908.00
TOTAL SHO / WT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT	0.83 34.70 52.33 86.52 0.48	0.32 3953.00 16.19 3994.00 18.04 3991.00 34.27 3992.00 0.19 3992.00	3908,00 4081.00 8 4082.00 9 4081.00 4081.00
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X	225.65 18.80 - -1.06 - 11.41 -	-6.18 4004.00 -238.15 3772.00 -152.32 3724.00 -369.24 3724.00	3926.00 10 3926.00 18 3927.00 19 3926.00
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z	41.64 - 723.69 821.21 627.90 2155.03	-133.98 4158.00 56.40 3929.00 39.84 3928.00 20.07 3931.00 128.65 3929.00	3929.00 35 3627.00 11 3673.00 12 3609.00 13
RES SEAT FORCE RES SEAT FORCE / WT LF FOOT X RT FOOT X	11.97 2191.97 12.18 25.82 28.34	0.71 3929.00 129.54 3929.00 0.72 3929.00 -74.54 3673.00 -25.41 3912.00	3609.00 3609.00 3609.00 3923.00 20 3924.00 23 3923.00 26
CT F00T X T0TAL F00T X LF F00T Y RT F00T Y CT F00T Y	11.84 - 57.19 - 108.68 19.52 14.44	-159.89 3873.00 -257.19 3873.00 -10.11 3907.00 -87.86 4025.00 -55.87 4109.00	3923.00 3977.00 21 3907.00 24
TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z	46.74 139.86 175.22 166.22	-58.20 3894.00 -9.02 3932.00 -22.17 3908.00 -37.98 3912.00	3937.00 3961.00 22 3961.00 25 4010.00 28
TOTAL FOOT Z RES FOOT FORCE	422.67 463.16	-20.64 3908.00 40.61 3908.00	3961.00 3846.00

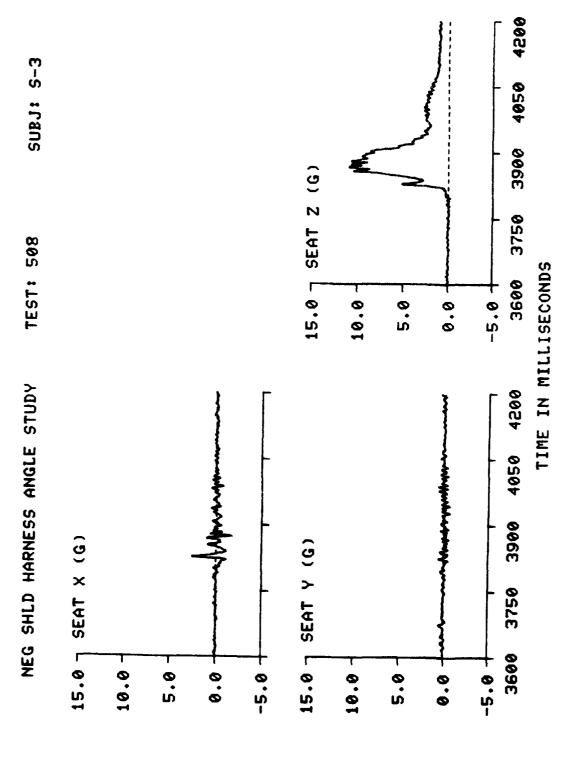
NEG SHLD HAR ANG	TEST:	512	SUBJ:	M-2	wr.	:67.2	c						
DATA ID							u :			:		;	
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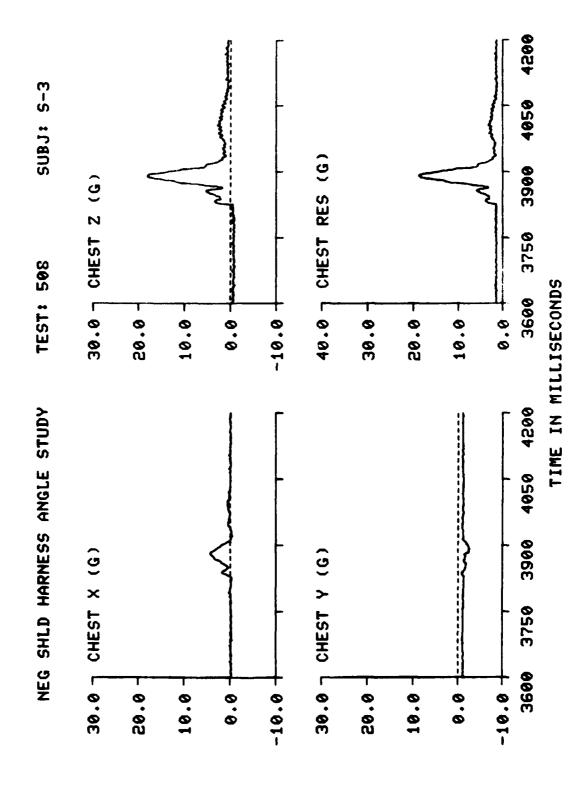
NEG SHLD HAR ANG TE	ST: 511 SUBJ:	M10 WT:	144.0	S: 10 GP:	1 CELL: H	
DATA ID		MAX	MIN	• T1	15	CH
IOV EXT PHR CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE Z SEAT X SEAT X SEAT Z		10.05 1.16 0.73 10.71 10.71 1.19 1.11 0.94 11.62	9.96 -1.25 -0.80 -0.22 -0.07 -26.15 -1.30 -0.21	379.00 3924.00 3917.00 3917.00 4146.00 4146.00 3923.00 3923.00 3920.00	1635.00 3934.00 3859.00 3606.00 3768.00 3862.00 3935.00 3881.00 3729.00	#861 92334
CHEST 7 CHEST 7 CHEST BES CHEST SI		3.63 0.19 20.24 20.38 35.77	-2.06 2.19 -0.98 1.00	3932.00 3915.00 3940.00 3940.00 3873.00	3965.00 3932.00 3864.90 3867.00 4116.00	5 6 7
HEAD X HEAD Y HEAD Z HEAD RES HEAD BIC		1.25 1.47 14.19 14.23 22.05 16.81	3 55 0.38 1.27 9.78	3924.00 4166.00 3929.00 3929.00 3883.00 3904.00	3966.00 3936.00 4122.00 3877.00 4087.00 3965.00	2 3 4
SHO REFL LF SHO REEL LF		34.94 36.40	8 25 1 64	3945.00 3962.00	4054.00 3915.00	14 16
LF SMOULDER SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHDULDER TOTAL SHOULDER TOTAL SHOULDER		62.40 26.43 27.79 47.54 60.13 57.09	11.80 14.01 0.45 15.55 22.38 2.18 29.05 0.20	3962.00 3951.00 3976.00 3977.00 3951.00 3971.00 3971.00	4055.00 4055.00 4055.00 4055.00 4055.00	15 17
LF LAP BELT BT LAP BELT TOTAL LAP		46.26 38.51 82.59	21.96 17.07 41.29	3982.00 3968.00 3982.00	3921.00 3925.00 3923.00	8 9
TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		13.02	0.29 0.59 -218.01 -77.98 -294.83	3982.00 4013.00 3840.00 3766.00 3766.00	3923.00 3934.00 3931.00 3929.00 3929.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z TOTAL SEAT Z / HT RES SEAT FORCE		23.04 556.25 452.27 702.51 1698.17 11.79 1727.98	-126.44 29.06 15.88 25.98 86.43 0.60 90.08	4159.00 3928.00 3929.00 3930.00 3929.00 3929.00	3930.00 3663.00 3744.00 3645.00 3636.00 3636.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X AT FOOT X CT FOOT X TOTAL FOOT 4 LF FOOT Y		65.13 83.47	0.63 54.19 -51.51 -160.82 -257.58 -14.36	3929.00 3875.00 3874.00 3875.00 3875.00	3636.00 3933.00 3933.00 3934.00 3933.00 3985.00	20 23 26
RT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z TOTAL FOOT Z RES FOOT FORCE		16.26 37.18 54.38 131.90 145.78 153.10 421.67 456.69	-112.42 -19.77 -41.91 -20.03 -20.24 -66.42 -74.01 36.60	3961.00 3895.00 3895.00 39923.00 3921.00 3921.00 3921.00	3919.00 4046.00 3983.00 3866.00 4027.00 3866.00 3866.00	24 27 22 25 28

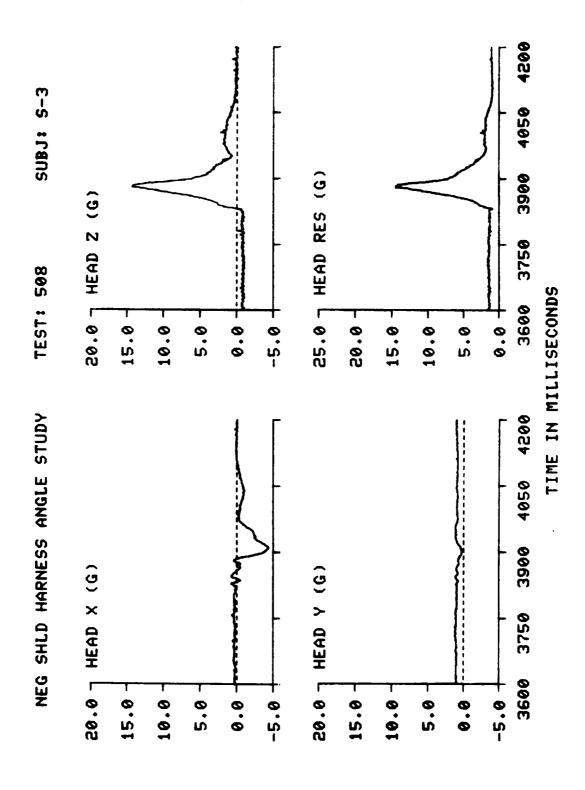
Total Phane	NEG SHLD HAR ANG T	EST: 506 SUBJ:	M11 HT:	157.0	G: 10 GP:	1 CELL: H	
CARRIAGE X CARRIAGE 7 CARRIAGE 7 CARRIAGE 7 CARRIAGE 7 CARRIAGE 7 12.81 -0.21 3915.00 3868.00 3868.00 2678.00 CARRIAGE 7 12.81 -0.21 3915.00 3868.00 3868.00 2678.00 CARRIAGE 7 10.50 -0.07 3915.00 3868.00 3868.00 2678.00 CARRIAGE 7 10.50 -0.07 3915.00 3868.00 3868.00 3869.00 27 CARRIAGE 7 10.50 -0.07 3915.00 3868.00 3869.00 3	DATA ID				T 1		5 1
TOTAL FOOT Y 51.73 -67.42 3956.00 3936.00 LF FOOT Z 197.12 -33.71 3918.00 4003.00 2 RT FOOT Z 179.81 -22.60 3943.00 4043.00 2 CT FOOT Z 174.07 -103.00 3947.00 3896.00 2	DATA ID 10 V EXT PHR CARRIAGE X CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE Z SEAT X SEAT X SEAT X SEAT Z SEAT Z SEAT Z SEAT Z SEAT Z CHEST RES CHEST X CHEST RES CHEST SI HEAD X HEAD X HEAD RES HEAD RES HEAD RES HEAD REEL RT RT SHOULDER TOTAL SHUD REEL TOTAL SHUD DER TOTAL SHUD DER TOTAL SHUD DER TOTAL SHUD N LF LAP BELT TOTAL SHUD LOER TOTAL SHUD LOER TOTAL SHUD LOER TOTAL SHUD LOER TOTAL SHUD N LF SEAT LNK X TOTAL SEAT PAN Z RT SEAT PAN Z TOTAL SEAT PAN Z TOTAL SEAT FORCE / WT RES SEAT FORCE / WT RT FOOT X TOTAL FOOT Y RT FOOT Y RT FOOT Y RT FOOT Y		X 509107777932776082577119572912277428773295232902114477.845232902114477.845232902114477.845232902114477.845232902114477.8452337761187777.845232902114477.845232902114477.845232902114477.845697777.845232902357652.129025257652.1290252576525765257652576525765257652576525	N. 96917395801887 54883 1990051631435885376663962488 91900516314883585376663962488 9190054541835885376663962488 91900545418358597000545418368359448896376663962488 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 9190054548 919000548 919000548 91900548 91900548 91900548 91900548	00000000000000000000000000000000000000	T2 135.000000000000000000000000000000000000	
HES FOOT FUHUE 492.35 15.66 3943.00 4035.00	CT FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z		16.86 51.73 197.12 179.81 174.07	-55.23 -67.42 -33.71 -22.60	3908.00 3956.00 3918.00 3943.00	3934.00 3936.00 4003.00 4043.00 3896.00	27 22 25 28

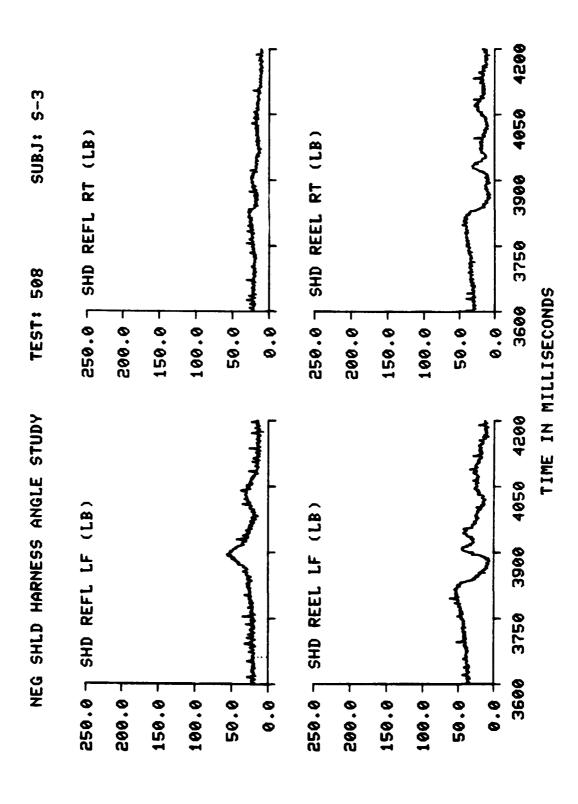
NEG SHUS HAR AND	ā\$1: 308 -30 ∂ ∪:	5-7 81	: :53.5	1 10 GP+	1	
DATE ID		MAX	MIN	T 1	<u> </u>	∪Ħ
10V EXT PHR CARRIAGE X CARRIAGE T CARRIAGE Z CARRIAGE Z (SM)		10.25 :.28 0.95 !1.77 !0.24	9.97 1.10 0.92 -0.61 0.18	829.00 3828.00 3878.00 3860.00 3861.00	1465.00 3838.JD 3817.00 3780.00 3782.00	48 36 31
CARRIAGE VEL SERT X SERT T SERT Z SERT Z (SM)		-1.03 2.54 0.64 10.88	26.23 -1.73 -0.67 -0.23	4108.00 3826.00 3989.00 3866.00	3826.00 3873.00 3929.00 3730.00	32 33 34
CHEST X CHEST Y CHEST Z CHEST RES		10.19 4.36 -0.87 16.24 18.69	0.39 -2.62 -1.11	3867.00 3882.00 3837.00 3890.00 3890.00	3679.00 3685.00 3893.00 3642.00 3736.00	5 6 7
CHEST SI HEAD X HEAD X HEAD & HEAD RES HEAD SI		34.05 0.76 1.22 14.34 14.36 19.78	-4.36 0.21 -1.09 0.83	3827.00 3845.00 3710.00 3880.00 3885.00	3936.00 3909.00 3899.00 3704.00 4119.00 3963.00	A S S S S
HEAD HIC SHD REFL LF SHD REEL LF		15.15 54.97 44.86	13.33 5.69	3857.00 3893.00 3908.00	3916.00 4092.00 3881.00	14 16
LF SHOULDER SHD REFL AT SHD REEL AT		91.44 24.55 31.01	39.26 13.22 7.75	3905.00 3903.00 3931.00	4094.00 3965.00 3860.00	15 17
RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER		48.60 78.17 63.60 125.04	25.39 28.13 16.00 66.52	3932.00 3894.00 3937.00 3905.00	3860.00 4100.00 3873.00 3862.00	
TOTAL SHO / WT LF LAP BELT RT LAP BELT TOTAL LAP		0.74 50.78 52.79 100.13	0.40 26.28 27.18 53.51	3905.00 3944.00 3921.00 3921.00	3862.00 3863.00 3861.00 3862.00	8 9
TOTAL LAP / HT CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TOTAL SEAT X		0.60 86.11 46.63 25.05	0.32 -32.07 -199.37 -91.78	3921.00 4098.00 3810.00 3828.00	3862.00 3880.00 3883.00 3883.00	10 18 19
SEAT LNK Y LF SEAT PAN Z BI SEAT PAN Z		47.70 36.18 548.54 593.04	-290.54 -96.90 47.87 44.83	3810.00 4122.00 3876.00 3877.00	3883.00 3882.00 3641.00 3640.00	35 11
CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / WT		745.25 1871.18 11.14	42.07 147.37 0.88	3884.00 3883.00 3883.00	3613.00 3600.00 3600.00	12
RES SERT FORCE / WT		1895.96 11.29 12.71	153.77 0.92 -120.19	3883.00 3883.00 3829.00	3600.00 3600.00 3873.00	20
LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		49.31 21.22 81.46	-53.79 -208.65 -379.95	3828.00 3830.00 3829.00	3873.00 3874.00 3873.00	5e 53
EF FOOT Y RT FOOT Y CT FOOT Y		119.63 14.89	-18.38 -112.90	3862.00 3691.00 3691.00	3712.00 3672.00 3678.00	21 24 27
TOTAL FOOT Y LF FOOT Z RT FOOT Z		43.15 56.79 156.69 198.27	-45.85 -36.33 -16.57 -16.93	3898.00 3888.00 3873.00	3875.00 3839.00 3983.00	22 25 28
CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		200.89 453.24 581.96	-94.28 -121.42 55.63	3868.00 3872.00 3872.00	3841.00 3839.00 4000.00	58

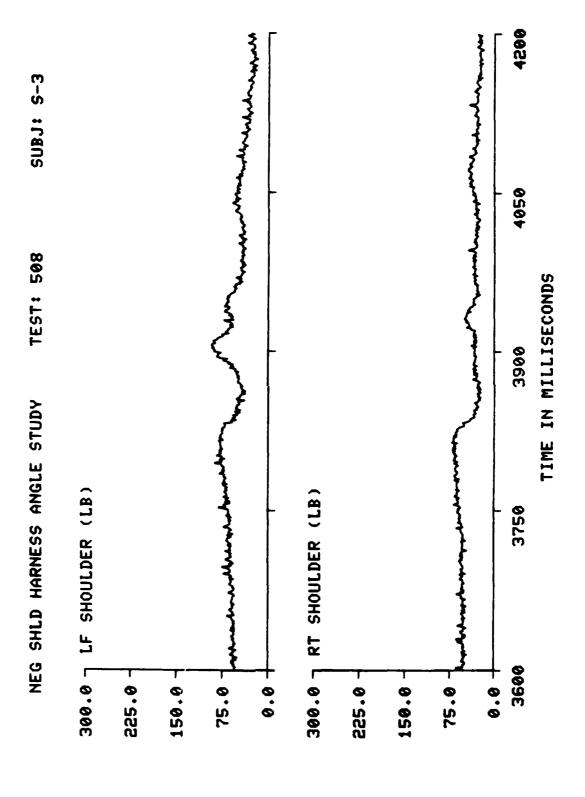


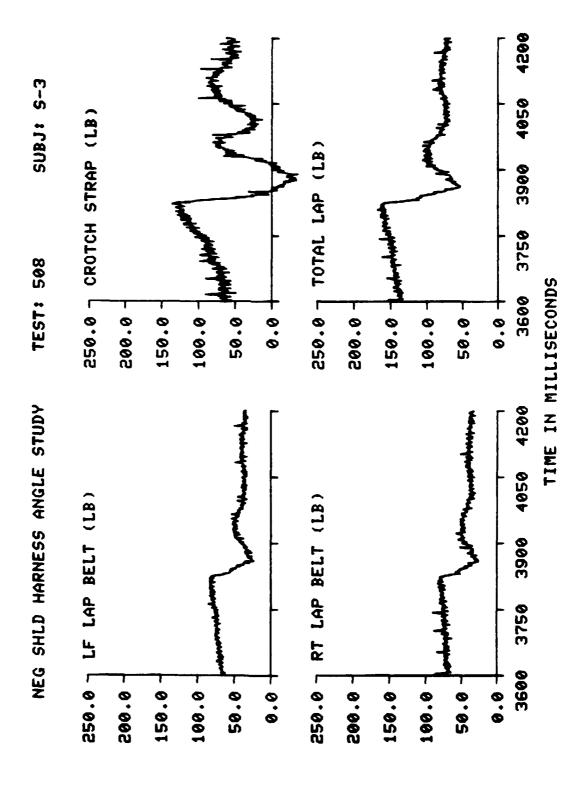


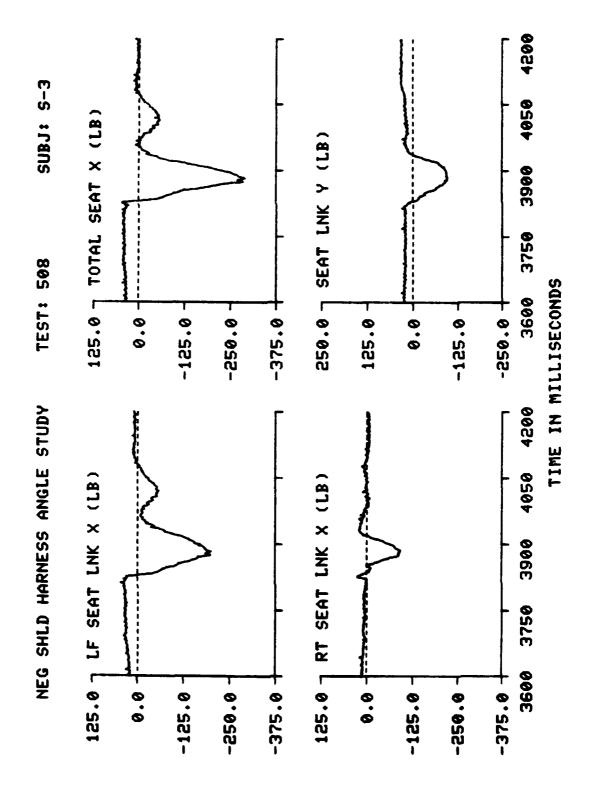


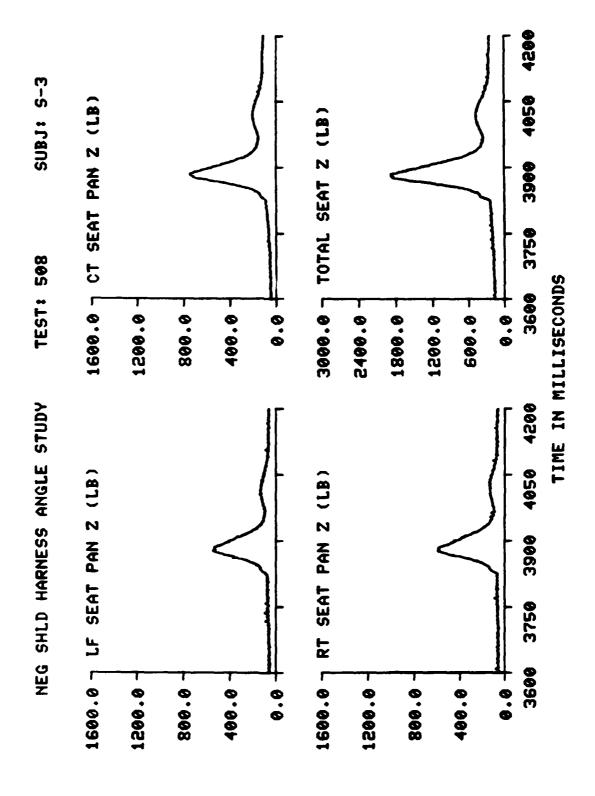


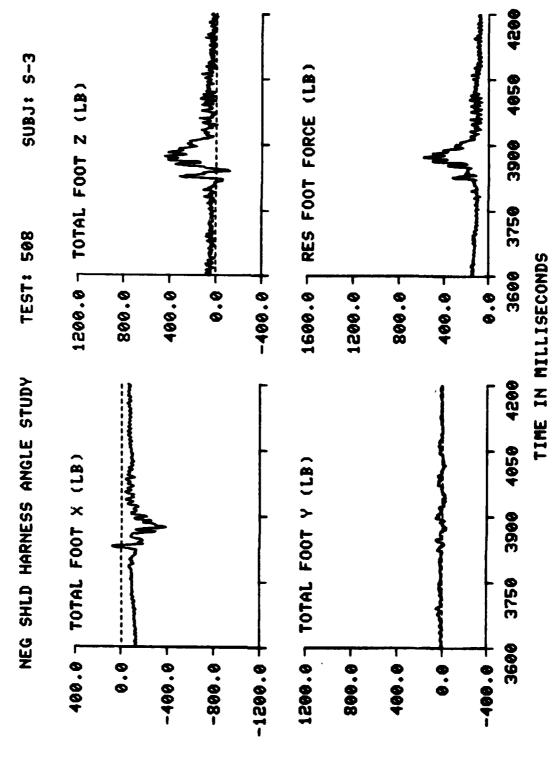












NEG SHLD HAR ANG TEST:	94 SUBJ: F-3 WT:	160.0 G: 10 GP:	1 CELL: J
DATA ID	MAX	MIN T1	T2 CH
DATA ID TOTAL I	MAX 10.18824499902704159005655456294120.0276027704519005655456294120.02990537761101.1101.1101.1101.1101.1101.1101.11	MIN 71 9.97 414.00 -0.87 3891.00 -0.87 38970.000 -0.07 3870.000 -0.07 3870.000 -0.14 3868.000 -0.24 3868.000 -0.14 3867.000 -0.14 3867.000 -0.14 3895.000 -0.18 3900.000 -0.18 3900.000 -1.65 3895.000 -1.65 3895.000 -1.61 3894.000 -1.62 3931.000 -1.63 3931.000 -1.65 3931.000 -1.65 3931.000 -1.61 3894.000 -1.62 3931.000 -1.63 3931.000 -1.65 3931.000 -1.61 3894.000 -1.62 3931.000 -1.63 3931.000 -1.65 3931.000 -1.65 3931.000 -1.61 3894.000 -1.62 3931.000 -1.63 3931.000 -1.64 3931.000 -1.65 3931.000 -	T2 26.000 356 27.000 334 28.000 334 28.000 334 29.000 334 36.000 334 36.000 336 36.000 336 36.0000 36.000 36.000 36.000 36.000 36.000 36.000 36.000 36.000 36.0000 36.000 36.000 36.0000 36.000 36.0000 36.000 36.0000 36.000 36.0000 36.000 36.00000 36.0000 36.00000 36.0000000000000000000000000000
RES SEAT FORCE / HT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X	10.99 -15.10 36.82 -13. 4 9	1.24 3894.00 -136.25 4139.00 -42.49 3878.00 -182.53 3843.00 -344.47 3842.00	3602.00 3683.00 20 3852.00 23 3884.00 26 3883.00
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X	10.99 -15.10 36.82 -13.49	1.24 3894.00 -136.25 4139.00 -42.49 3878.00 -182.53 3843.00 -344.47 3842.00	3602.00 3683.00 20 3852.00 23 3884.00 26 3883.00
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y LF FOOT Z RT FOOT Z	21.58 20.34 61.99 205.28 201.22	-11.93 3871.00 -106.21 3939.00 -59.75 3992.00 -59.93 3995.00 -24.63 3873.00 -12.15 3839.00	4001.00 21 3889.00 27 3947.00 27 3851.00 22 3851.00 22 3853.00 28
CT FOOT Z TOTAL FOOT Z RES FOOT FORCE	150.47 439.44 468.63	-181.55 3878.00 -143.33 3838.00 66.93 3872.00	3853.00 28 3852.00 3843.00

NEG SHLD HAR ANG	TEST: 481	suaJ:	F-2 NT:	160.0	G: 10 GP:	1 CELL:	J
DATA ID			MAX 	MIN	T 1	15	C H
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z			10.05 1.10 0.74 12.56	9.97 -1.07 -0.49 -0.17	272.00 3845.00 3847.00 3839.00	89.00 3853.00 3811.00 3703.00	48 36 31 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z			10.49 -0.93 1.06 0.71 11.62 10.54	-0.06 -26.11 -1.26 -1.20 -0.29	3839.00 4153.00 3848.00 3808.00 3845.00	3655.00 3797.00 3852.00 3814.00 3725.00	29 32 33 34
SEAT Z (SM) CHEST X CHEST Z CHEST Z CHEST RES			10.54 1.87 -0.30 19.04 19.11 36.00	-0.16 -3.60 -2.00 -0.58 0.83	3846.00 3860.00 3908.00 3869.00 3869.00	3724.00 3902.00 3859.00 3728.00 3723.00 3935.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD RES HEAD SI			3.01 2.11 13.50 13.85 22.56	-2.52 -0.18 -1.27 1.29	3801.00 3861.00 3862.00 3864.00 3864.00	3901.00 3973.00 3972.00 3941.00 3929.00	2 3 4
SHD REFL LF SHD REEL LF			18.58 27.33 54.76	5.28 4.66	3836.00 3907.00 3903.00	3899.00 3853.00 3851.00	1 1 1 6
LF SHOULDER SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL			80.79 31.15 50.38 81.50 58.95 104.97	10.39 4.00 3.83 10.18 10.79 10.27	3904.00 3901.00 3902.00 3902.00 3900.00 3902.00	3652.00 3848.00 3940.00 3641.00 3652.00 3939.00	15 17
TOTAL SHOULDER TOTAL SHO / WT LF LAP BELT AT LAP BELT TOTAL LAP TOTAL LAP / WT			161.82 1.01 46.99 50.71 101.53 0.63	22.17 0.14 20.93 24.65 46.53	3902.00 3902.00 3924.00 3931.00 3932.00	3852.00 3852.00 3875.00 3843.00 3875.00 3875.00	8 9
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X			126.82 10.31 -4.13	0.29 -6.51 -270.84 -108.68	3945.00 3609.00 3600.00	3869.00 3860.00 3859.00	10 18 19
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z / NT RES SEAT FORCE			1.88	-379.50 -134.90 48.37 36.05 25.10 126.27	3609.00 3962.00 3863.00 3861.00 3863.00 3863.00 3863.00	3860.00 3859.00 3602.00 3601.00 3666.00 3601.00 3601.00	35 11 12 13
	WТ		10.44 7.83 20.69 -36.69 -19.67	126.35 0.79 -61.79 -71.83 -217.36 -350.08	3863.00 3931.00 3725.00 3805.00 3804.00	3601.00 3659.00 3857.00 3857.00 3857.00	20 23 26
LF FOOT Y BY FOOT Y CT FOOT Y			120.38	-14.13 -140.20 -22.98	3841.00 4196.00 3820.00	3909.00 3850.00 3996.00	21 24 27
TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE			40.21 69.57 192.96 189.13 279.22 535.95 582.49	-71.70 3.57 16.74 -51.67 22.24 94.91	3818.00 3843.00 3843.00 3846.00 3843.00 3850.00	3812.00 3989.00 4126.00 3605.00 4126.00 4126.00	22 25 28

NEG SHLD HAR AND	TEST:	486 S	UBJ:	G-3	WT:	164.0	G: 10 GP:	1 0504:	ز
Data ib				MAX		MIN	T 1	T2	24
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z (SM)				10.0 1.1 0.7 12.6	9 5 4	9.97 -1.50 -0.67 -0.18	965.00 3868.00 3996.00 3861.00	159.00 3875.00 3991.00 3610.00	18 31 1
CARRIAGE VEL SEAT X SEAT Y SEAT Z				10.5 -0.9 1.3 1.7	3 6 7 3	-0.08 -26.30 -1.86 -1.83 -0.16	3876.00 4164.00 3870.00 3824.00 3867.00	3611.00 3823.00 3874.00 3830.00 3687.00	29 32 33 34
CHEST X CHEST Y CHEST Z CHEST RES				10.5 4.4 0.1 19.1	5 8 7 8	-0.09 -2.79 -3.46 -0.91 1.07	3868.00 3876.00 3896.00 3901.00 3901.00	3685.00 3923.00 3885.00 3681.00 3818.00	5 6 7
CHEST SI HEAD X HEAD Y HEAD Z HEAD RES HEAD SI				31.8 3.1 2.4 11.6 12.1 18.5	5 1 2	-4.08 0.39 -1.24 1.54	3825.00 3878.00 3977.00 3877.00 3877.00 3835.00	4065.00 3927.00 3908.00 3698.00 4058.00 3952.00	2 3 4
HEAD HIC SHD REFL LF SHD REFL LF LF SHOULDER SHD REFL RT				15.6 44.8 80.7 124.9 35.2	6 8 5	7.79 0.91 11.34 7.17	3855.00 3908.00 3910.00 3909.00 3916.00	3926.00 3863.00 4013.00 3863.00 3869.00	14 16
SHO REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER				68.9 103.9 75.1 143.9 218.0	2 5 8 8	-3.66 5.31 16.34 -2.63 18.22	3915.00 3915.00 3909.00 3913.00 3913.00	4015.00 3870.00 3863.00 4014.00 4009.00	15 17
TOTAL SHO / WT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT				1.3 52.3 58.2 108.6	5 9 3	0.11 24.96 25.13 50.57 0.31	3913.00 3963.00 3932.00 3939.00 3939.00	4009.00 3864.00 3870.00 3864.00 3864.00	8 9
CROTCH STRAP LF SEAT LNK X BT SEAT LNK X TOTAL SEAT X				74.6 13.4 21.4 13.8	:7 :8 - :8 -	-28.60 -191.62 -80.00 -271.00	4098.00 4110.00 3924.00 4109.00	3883.00 3875.00 3876.00 3875.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z	1.7			56.5 709.9 707.2 524.6 1911.8	3 9 6 8	-96.61 69.94 40.62 47.17 173.51	4089.00 3889.00 3888.00 3877.00 3887.00	3887.00 3604.00 3600.00 3605.00 3600.00	35 11 12 13
RES SERT FORCE	iτ ⁄			11.6 1928.4 11.7 20.7 53.0 32.6	6 0 8	1.06 174.67 1.07 -80.56 -50.90	3887.00 3887.00 3887.00 3826.00 3868.00 3826.00	3600.00 3600.00 3600.00 3889.00 3898.00	26 23 20
TOTAL FOOT X LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y				71.1 96.0 19.0 25.2 51.7	5 - 0 1	-282.17 -9.43 -77.72 -58.45 -54.54	3826.00 3863.00 3914.00 3991.00 3845.00	3899.00 4001.00 3863.00 3882.00 3882.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE				160.2 164.1 165.8 398.7 480.2	3 3 1 8	-40.52 -18.83 -60.10 -91.59 9.02	3890.00 3888.00 3870.00 3890.00 3890.00	3819.00 3985.00 3818.00 3819.00 3952.00	22 25 28

DSTS ID
CARRIAGE X
TOTAL FOOT X 27.43 -261.08 3864.00 3896.00 LF FJOT Y 101.27 -14.90 3898.00 3967.00 21 RT FOOT Y 22.35 -136.50 3863.00 3897.00 24 CT FOOT Y 42.63 -33.02 3887.00 3910.00 27 TOTAL FOOT Y 67.91 -73.45 3940.00 3908.00 LF FOOT Z 167.70 -20.74 3888.00 3893.00 22

NEG SHLO HAR AND	TEST: 491	SU B J:	h-3 WT:	198.0	S: 10 6°:	1 051	
<u> </u>			4 0 Y	·	T 1	- 2 	5 ·
10V EXT PAR CRESTOREX CRESTOREZ CRES			10.05 1.15 0.71 12.60 10.58	9.96 -1.19 -0.56 -0.23 -0.11	388.00 3909.00 3910.00 3901.00 3902.00	2677.00 390:.00 4018.00 3812.00	35 31 1
CAPATAGE VEL SEAT X SEAT X SEAT Z SEAT Z (SM)			-1.00 1.39 0.72 11.66	-25.08 -1.24 -0.92 -0.14 -0.09	4175.00 3910.00 4016.00 3907.00	3705.00 3853.00 3914.00 3918.00 3714.00	39 33 34
CHEST X CHEST X CHEST Z CHEST RES CHEST SI			1.07 17.89 18.03 32.03	-3.02 -1.54 -0.81	3918.00 3980.00 3981.00 3931.00 3931.00	3972.00 3962.00 3619.00 4163.00 4073.00	5 6 7
HEAD X HEAD Y HEAD YES HEAD ST			2.78 2.55 10.81 1.17	-2.93 0.26 -1.09 0.89	3916.00 4033.00 3921.00 3921.00 3873.00	3969.00 3938.00 3732.00 4:28.00 4059.00	2 3 4
HEAD HIC SHD REFL LF SHD REEL LF LF SHOW: DEB			19.88 16.83 44.20 54.70 85.04 48.53	3.45 7.03 10.56 13.41	3894.00 3984.00 3958.00 3959.00 3985.00	3967.00 3913.00 3919.00 3913.00 3907.00	14 16
SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLO REFL TOTAL SHLO REEL TOTAL SHOULDER			69.94 110.97 92.62 124.63 135.26	2.53 22.22 19.14 14.09 34.37	3958.00 3961.00 3985.00 3958.00 3960.00	4043.00 3907.00 3912.00 4032.00 3913.00	15 17
TOTAL SHO / WT LF LAP BELT AT LAP BELT TOTAL LAP			1.04 32.55 55.19 84.29	0.18 12.15 22.42 34.97 0.19	3960.00 4087.00 4056.00 4056.00 4056.00	3913.00 3943.00 3939.00 3942.00 3942.00	9
TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X			54.15 36.95	-11.47 -173.50 -72.52 -239.68	3980.00 4197.00 3969.00 4117.00	3935.00 3916.00 3923.00 3923.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z HT			\$9.60 356.76 560.77 842.86 1745.37	-18.79 34.66 48.04 74.91 174.61	4102.00 3918.00 3926.00 3925.00 3926.00	3913.00 3653.00 3607.00 3622.00 3602.00	35 11 12 13
RES SEAT FORCE	HT		16.30	0.93 175.00 0.93 -149.09 -95.61 +276.48	3926.00 3924.00 3924.00 3869.00 3988.00 3869.00	3602.00 3602.00 3602.00 3920.00 3936.00 3921.00	20 23 26
TOTAL FOOT X LF FOOT Y AT FOOT Y CT FOOT Y TOTAL FOOT Y			3.15 153.78 35.25 6.76	-518.52 -2.57 -127.15 -67.10	3869.00 3904.00 3959.00 3942.00 3942.00	3920.00 3993.00 3913.00 3918.00 3917.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE			82.73 214.91 234.76 197.88 573.06 748.54	-52.52 -1.52 27.67 -101.71 -35.27 100.73	3904.00 3922.00 3925.00 3922.00 3922.00	3860.00 3855.00 3881.00 3859.00 3809.00	22 25 28

NEG SHUD HAR AND	TEST:	493	SUBJ:	H=4	WT:	185.0	G: 10 GP:	2 0511;	
D1 8 20				MAX		MIN	T1	T2	Ç ri
HAYYZZV M WAYYZZV M WAYZZV M WAYYZZV M WAYZZV M WAYYZZV M W WAYYZZV M WAYYZZV M WAYYZZV M WAYYZZV M WAYYZZV M					-1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -			2- 000000000000000000000000000000000000	07 488 2889 567 284 46 57 89 089 5128 086 147 258 111 3111 200 200 200 200

NEG SHLD HAR ANG	TEST: 482	SUBJ: K-1	WT: 183.0	G: 10 GP:	1 CELL:	
DATA ID		MAX		11	T2	СН
10V EXT PWR CARRIAGE X CARRIAGE Z CARRIAGE Z CARRIAGE VEL SEAT X		10.0 1.6 0.6 12.1 10.1	6 -1.17 3 -1.00 6 -0.35 9 -0.11 9 -26.19	66.00 3816.00 3945.00 3822.00 3823.00 4193.00	666.00 3779.00 3949.00 3727.00 3622.00 3785.00	48 36 31 1
SEAT X SEAT X SEAT Z SEAT Z (SM) CHEST X CHEST X CHEST Z		2.0 1.3 10.6 10.6 5.8 0.2 16.5	0 -2.58 8 -0.19 2 -0.12	4193.00 3793.00 5788.00 3819.00 3829.00 3843.00 3835.00	3835.00 3795.00 3631.00 3632.00 3862.00 3862.00	29 33 34 56 7
CHEST RES CHEST SI HEAD X HEAD Y HEAD Z HEAD RES HEAD SI		31.34 1.51 2.17 12.75 12.75	4.35 7 0.20 -1.16	3856.00 3856.00 3789.00 3837.00 3892.00 3847.00 3847.00	4066.00 3902.00 3881.00 3850.00 3646.00	7 2 3 4
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL RT SHD REFL RT		20.12 16.42 23.12	-0.68	3797.00 3821.00 3927.00 3919.00 3927.00 3919.00	4126.00 3926.00 3885.00 3899.00 3991.00 3824.00	14 16 15
RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHCULDER TOTAL SHD / WT LF LRP BELT		39.67 39.52 31.52 51.52 42.45 48.45 48.83	11.54 0.06	3883.00 3883.00 3927.00 3883.00 3928.00 3928.00	3855.00 3832.00 3833.00 3991.00 3833.00 3833.00	iř
BT LAP BELT TOTAL LAP TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X AT SEAT LNK X		43.51 90.49 0.49 112.29 28.51	15.67 37. 7 6	3876.00 3905.00 3904.00 3904.00 3930.00 4076.00	3958.00 3971.00 3959.00 3959.00 3849.00 3844.00	10
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z		8.93 12.70 37.40 792.76 814.82 614.48	-160 77	3686.00 4075.00 3920.00 3845.00 3848.00	3845.00 3845.00 3836.00 4088.00 3618.00	18 19 35 11 12
TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE RES SEAT FORCE / WT LF FOOT X RT FOOT X		2207.06 12.06 2236.08	166.16 0.91 166.29 0.91 -77.68	3847.00 3847.00 3847.00 3847.00 3847.00 3790.00	3606.00 3618.00 3618.00 3618.00 3618.00 3635.00	13
CT FOOT X TOTAL FOOT X LF FOOT T RT FOOT T CT FOOT Y		37.14 42.23 27.17 103.80 18.76	-35.32 -150.82 -263.82 -6.50 -70.87	3791.00 3791.00 3790.00 3824.00	3835.00 3835.00 3835.00 4082.00 3823.00	23 26 21 24 27
TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		12.99 47.87 140.99 178.29 384.92	-38.62 -49.10 -48.87 -15.46 -68.88 -117.91	3894.00 3826.00 3825.00 3845.00 9790.00	3652.00 3853.00 3803.00 4090.00 9784.00 3784.00	27 22 25 28
		492.91	17.29	9845.00 9845.00	4173.00	

AIR FORCE AEROSPACE MEDICAL RESEARCH LAB WRIGHT-PATT--ETC F/6 1/3 COMPARATIVE VERTICAL IMPACT TESTING OF THE F/FB-111 CREW RESTRA--ETC(U) MAR 82 B F HEARON, J W BRINKLEY, J H RADDIN AD-A113 957 AFAMRL-TR-82-13 UNCLASSIFIED NL 3 ∞ 4 19395

NEG SHLD HAR HAG	TEST: 484	SUBJ:	M-2 WT:	162.0	G: 10 GP:	1 CELL:	j
CI RTAD			MAX 	MIN	Ť1	72	СH
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z			10.05 1.03 0.80 11.98	9.96 -1.16 -0.60 -0.22	155.00 3886.00 3866.00 3876.00	882.00 3875.00 3970.00 3686.00	48 36 31 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z SEAT Z (SM)			10.43 -0.93 1.49 0.75	-0.08 -26.11 -1.44	3893.00 4170.00 3886.00	3687.00 3835.00 3890.00	
SEAT Z SEAT Z (SM)			11.9/	-0.95 -0.21 -0.09	3857.00 3882.00 3883.00	3845.00 3699.00 3698.00	29 32 33 34 5 6
CHEST Y CHEST Z CHEST Z CHEST RES CHEST SI			10.72 -0.07 19.72 19.84 37.29 1.57 2.81	-3.72 -1.15 -0.70 0.56	3891.00 3890.00 3904.00 3904.00 3841.00	3937.00 3865.00 3791.00 3832.00 3953.00	5 6 7
HEAD X HEAD Y HEAD Z HEAD RES			10.30	-5.26 -0.07 -1.32 1.56	3884.00 3980.00 3900.00 3900.00	3943.00 3920.00 3695.00	2 3 4
HEAD SI HEAD HIC SHD REFL LF			10.95 15.78 12.63 35.59	4.48 1.66	3857.00 3870.00 3968.00 3943.00	3991.00 3960.00 3886.00	14 16
SHD REEL LF LF SHOULDER SHD REEL RT SHD RESL BT			79.78 31.14 49.34 76.67	6.16 7.03	3944.00 3950.00 3975.00	3884.00 3885.00 3879.00 4016.00	15 17
SHD RESL RT RT SHOULDER TOTAL SHED REFL TOTAL SHED REEL TOTAL SHOULDER TOTAL SHOULDER TOTAL SHOULDER			52.14 91.14 145.14	-1.52 7.88 12.22 2.88 15.12	3933.00 3949.00 3938.00 3941.00	3892.00 3887.00 3886.00 3886.00	• •
TOTAL SHO / WT LF LAP BELT BT LAP BELT TOTAL LAP TOTAL LAP / WT			0.90 41.61 63.43 102.63	0.09 13.81 21.70 35.81 0.22	3941.00 3964.00 3959.00 3965.00 3965.00	3886.00 3884.00 3886.00 3885.00	8 9
CACTCH STRAP LF SEAT LNK X AT SEAT LNK X			12.40	-29.85 -154.03 -125.95	3979.00 4103.00 3828.00	3885.00 3900.00 3892.00 3897.00 3898.00	10 18 19
TOTAL SEAT X SEAT LANK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z			58.21 515.01 550.09	-71.83 63 07	3629.00 3977.00 3900.00 3899.00 3901.00	3893.00 3603.00 3628.00 3622.00	35 11 12 13
TOTAL SEAT Z TOTAL SEAT Z / WT RES SEAT FORCE RES SEAT FORCE / I	шт		792.72 1848.46 11.41 1868.93	51.95 49.34 180.55 1.11 181.17	3901.00 3901.00 3901.00 3901.00	3628.00 3628.00 3628.00 3628.00	13
LF FOOT X BT FOOT X CT FOOT X TOTAL FOOT X	• •		14.79	-59.18 -89.19 -188.94 -333.70	3884.00 4044.00 3841.00	3897.00 3896.00 3896.00	20 23 26
LF F00T Y BT F00T Y CT F00T Y			99.26	-10.80 -106.39 -32.91 -67.96	3841.00 3878.00 3931.00 3838.00 3839.00	3896.00 3844.00 3879.00 3867.00	21 24 27
TOTAL FOOT Y LF FOOT Z RT FOOT Z CT FOOT Z				-16.27 0.38 -79.64	3888.00 3881.00 3884.00	3869.00 3958.00 3650.00 3832.00	22 25 28
TOTAL FOOT Z RES FOOT FORCE			475.95 507.65	-70.00 44.20	3881.00 3881.00	3832.00 4192.00	

NES SHLU HAR AND	TEST:	523 SUB	J: M10	WT: 144.0	G: 10 GP:	1 CELL:	j
5878 IS			MAX	MIN	T 1	72 	5 n
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z CARRIAGE Z CARRIAGE YEL			10.0 1.2 0.6 11.7	0 -0.92 9 -0.80 9 -0.25	231.00 3928.00 3930.00 3921.00 3921.00	428.00 39:3.00 40:4.00 36:80	49 31 1
CARRIAGE VEL SEAT X SEAT X SEAT Z SEAT Z (SM)			-1.0 1.1 0.6 11.2	7 -25.18 3 -0.95 9 -1.04 9 -0.23	4175.00 3929.00 3885.00 3927.00 3927.00	3837.00 3837.00 3831.00 3744.00	32 33 34
CHEST X CHEST X CHEST Z CHEST RES CHEST SI			10.9 1.8 1.2 15.9 16.0	0 -1.97 8 -1.43 7 -0.94 2 0.85	3932.00 3941.00 3948.00 3948.00	404774444404 40477444440000000000000000	5 6 7
HEAD X HEAD Z HEAD RES HEAD SI			.8 1.5 14.0 14.0	1 -3.29 7 -0.93 5 -0.87 7 0.40	3881.00 3946.00 4179.00 3937.00 3938.00 3891.00	3973.50 4129.60 3636.60 4141.60 4067.00	3 3 2
HEAD HIC SHD REFL LF SHD RESL LF LF SHOULDER			16.3 20.2 34.3	9 ~4.47	3916.00 3988.00 4009.00	00000000000000000000000000000000000000	14 16
SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REFL			53.9 27.6 43.1 65.8 45.5 63.1	0 4.49 5 12.08 7 12.53	4008.00 4013.00 3992.00 3992.00 4007.00 3991.00	3925.00	15 17
TOTAL SHOULDER TOTAL SHD / WT LF LAP BELT TOTAL LAP			106.4 0.7 41.7 41.5 81.6	7 20.17 4 0.14 1 12.07 6 12.58 3 28.66	4007.00 4007.00 4006.00 4018.00 4016.00	3936.00 3936.00 3936.00 3928.00 3923.00	8 9
TOTAL LAP / WT CROTCH STRAP LF SEAT LNK X RT SEAT LNK X			0.5 66.0 38.6 6.5	4 -43.83 1 -223.53 7 -77.06	4016.00 4014.00 3726.00 3881.00	3923.00 3954.00 3935.00 3941.00	10 18 19
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z / W RES SEAT FORCE	т		37.8 26.4 528.0 456.2 658.5 1658.5 11.5 1689.7	8 -298.85 6 -132.75 0 37.40 23.90 7 27.45 104.35 0 72	3637.00 3634.00 3936.00 3935.00 3943.00 3939.00 3939.00	3941.00 3938.00 3726.00 3754.00 3660.00 3660.00 3660.00	35 11 12 13
RES SEAT FORCE / LF FOOT X RT FOOT X CT FOOT X	WT		11.7 10.0 17.10	3 0.77 7 -75.82 8 -53.31	3939.00 3991.00 4123.00 3886.00	3660.00 3934.00 3941.00 3934.00	26 23 20
TOTAL FOOT X LF FOOT Y RT FOOT Y			0.50 9 5 .5 20.10	4 -286.73 1 -9.31 4 -117.41	3885.00 3924.00 3897.00	3934.00 3775.00 3925.00	21 24 27
CT FOOT Y TOTAL FOOT Y LF FOOT Z AT FOOT Z			31.20 65.30 178.70 179.80	8 -5.75	3882.00 3899.00 3926.00 3925.00	3930.00 3891.00 3891.00 4000.00	27 22 25 28
CT FOOT Z TOTAL FOOT Z RES FOOT FORCE			205.6 510.2 540.9	1 -62.30 6 -49.34	3929.00 3926.00 3926.00	4000.00 4000.00 4008.00	28

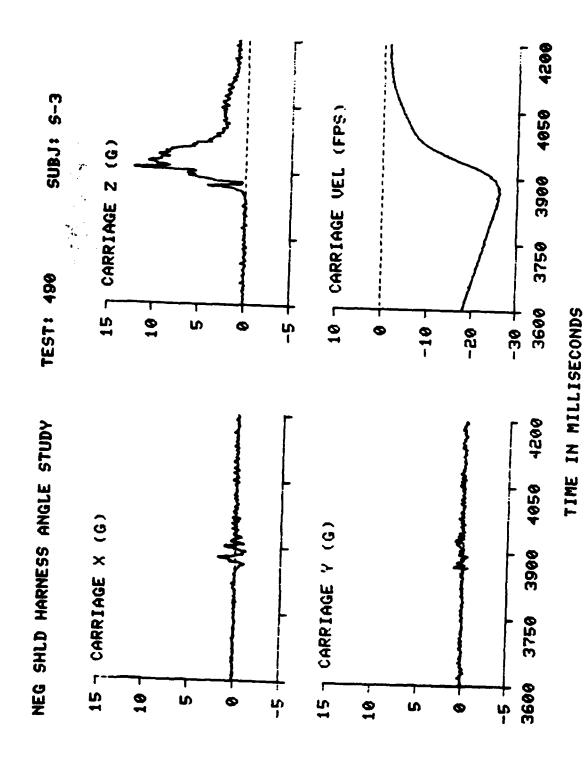
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NEG SHLD HAR ANG TEST: 487 SUBJ: MII WT: 157.0 G: 10 SP: 1 CELL: J
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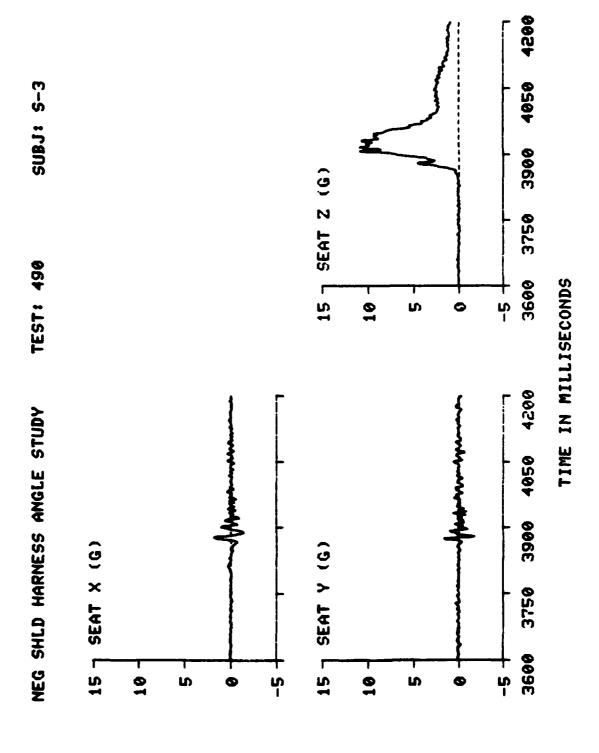
NEG SHLD HAR ANG TEST: 480	SU8J: M13 WT:	170.0 G: 10 GP:	1 CELL: J
DATA ID	MAX 	MIN T1	15 CH
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z (SM)	10.05 1.25 1.19 12.30	9.96 572.00 -1.19 3815.00 -1.16 3807.00 -0.24 3848.00	2076.00 48 3829.00 36 3803.00 31 3762.00 1
CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Y SEAT Z SEAT Z (SM)	10.51 -0.92 1.25 0.61 11.44	-0.09 3849.00 -26.16 4180.00 -1.24 3814.00 -1.02 3961.00 -0.25 3854.00	3759.00 3795.00 3826.00 3863.00 3606.00
CHEST X CHEST Z CHEST Z CHEST RES	10.51 2.79 0.13 20.97 21.10	-0.14 3855.00 -3.00 3865.00 -3.21 3879.00 -0.98 3871.00 1.00 3871.00	3605.00 3904.00 5 3863.00 6 3700.00 7 3783.00
CHEST SI HEAD X HEAD T HEAD Z HEAD RES	36.49 3.36 2.30 13.17 13.62	3813.00 -1.46 3867.00 0.79 3958.00 -1.47 3870.00 1.50 3868.00	3944.00 3922.00 3695.00 3695.00 3817.00
HEAD SI HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER SHD REFL BT	21.35 17.21 40.21 63.91 99.96	3823.00 3844.00 8.61 3774.00 10.00 3898.00 19.52 3899.00 8.70 3799.00	3935.00 3907.00 3851.00 14 3856.00 16 3858.00 3853.00 15
SHD REFL RT SHD REEL RT RT SHOULDER O.TOTAL SHLD REFL O.TOTAL SHLD REEL TOTAL SHDD REEL	25.86 30.97 55.74 64.82 92.01 152.83	-1.48 3699.00 11.39 3799.00 31.30 3899.00	3853.00 15 3863.00 17 3863.00
TOTAL SHD / NT LF LAP BELT AT LAP BELT TOTAL LAP TOTAL LAP / NT	0.90 62.26 61.07 123.33 0.73	0.18 3899.00 15.46 3719.00 19.74 3604.00 37.49 3719.00 0.22 3719.00	3658.00 3856.00 3664.00 3664.00 3864.00
CROTCH STRAP LF SEAT LNK X AT SEAT LNK X TATEL SEAT Y	110.81 20.53 - 4.59 -	18.90 3789.00 -209.92 3785.00 -148.52 3824.00 -356.71 4155.00	3843.00 10 3866.00 18 3864.00 19 3868.00
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z	31.83 408.55 478.77 862.02 1730.94 10.18	-66.61 3927.00 29.63 3874.00 33.63 3868.00 47.82 3870.00 131.39 3870.00 0.77 3870.00	3863.00 35 3602.00 11 3627.00 12 3606.00 13 3606.00
RES SEAT FORCE / WT RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X	1768.31 10.40 -4.00 - 29.80	131.61 3870.00 0.77 3870.00 -145.04 3816.00 -65.37 3816.00 -223.73 3816.00 -432.35 3815.00	3606.00 3606.00 3666.00 20 3866.00 23 3868.00 26
TOTAL FOOT X LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y	141.80 16.01 - 18.05 60.17	-10.18 3850.00 -128.64 3904.00 -44.25 3819.00 -43.73 3890.00	3866.00 4005.00 21 3867.00 24 3863.00 27 4022.00
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE	226.90 205.99	22.49 3875.00 14.21 3868.00 -121.08 3859.00 -28.90 3859.00 89.11 3868.00	4007.00 22 4101.00 25 3607.00 28 3787.00

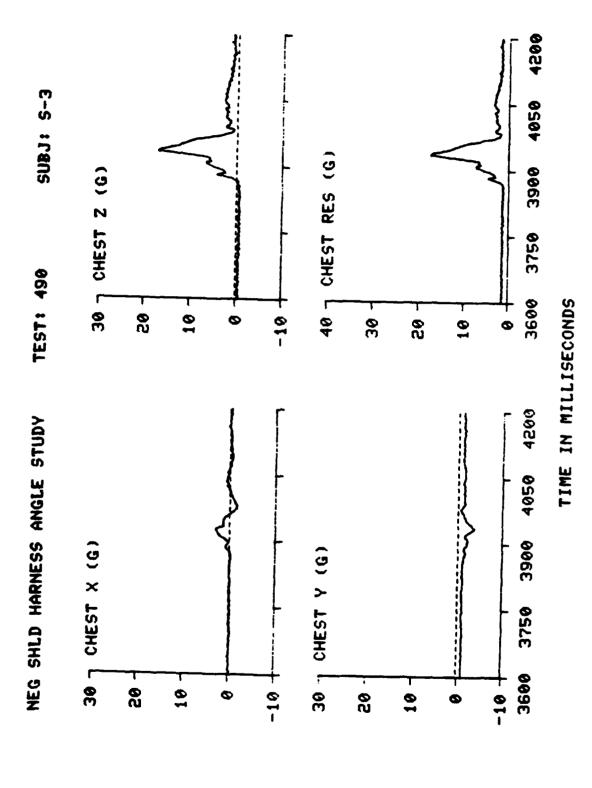
NEG SHLO HAR ANG TO	EST: 483 SUBJ:	R-2 WT:	145.0	G: 10 GP:	1 CELL:	ز
DETA ID		MAX	MIN	T 1	T2	CH
10V EXT PWR CABRIAGE X CARRIAGE Y CARRIAGE Z CARRIAGE Z (SM)		10.05 1.38 0.98 12.25 10.54	9.96 -1.20 -0.89 -0.60 -0.24	274.00 3887.00 3876.00 3918.00 3919.00	445.00 3899.00 3873.00 3838.00 3840.00	48 36 31 1
CARRIAGE VEL SEAT X SEAT Y SEAT Z SEAT Z (SM)		10.54 -1.23 1.05 11.51 10.54 3.75	-0.24 -26.19 -1.58 -1.01 -0.26 -0.13	4196.00 3884.00 4090.00 3924.00 3926.00	3881.00 3879.00 3891.00 3849.00 3743.00	29 32 33 34
CHEST X CHEST Y CHEST Z CHEST RES CHEST SI		3.79 -0.05 13.66 13.77 27.51	-1.11 -1.88 -0.92 0.55	3941.00 3977.00 3955.00 3955.00	3975.00 3887.00 3758.00 3879.00	5 6 7
HEAD X HEAD X HEAD RES HEAD HEAD HEAD		27.51 1.02 12.95 13.12 21.82 17.33 24.69	-4.33 -0.20 -0.64 0.28	3883.00 3857.00 3994.00 3943.00 3943.00	4132.00 3965.00 4121.00 3758.00 3877.00 4117.00 3975.00	2 3 4
SHD REFL LF SHO REEL LF LF SHOULDER			4.36 2.87 9.15	3912.00 3975.00 3983.00 3984.00	3919.00 3927.00 3918.00	14 16
SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL TOTAL SHOULDER		52.34 39.41 40.57 69.57 62.56 60.15	10.83 2.51 14.63 15.25 5.54 24.43	4005.00 3969.00 3969.00 3998.00 3970.00 3984.00	3918.00 3922.00 3919.00 3919.00 3927.00 3919.00	15 17
TOTAL SHO / WT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT		28.99 41.24 68.54 0.47	0.17 4.69 4.24 10.11 0.07	3984.00 3991.00 4011.00 4011.00 4011.00	3919.00 3937.00 3945.00 3932.00 3932.00	8
CROTCH STRAP LF SEAT LNK X RT SEAT LNK X		87.53 46.45	2,63 179.39 34.64	4006.00 4131.00 3886.00	3935.00 3938.00 3939.00	10 18 19
TOTAL SEAT X SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z HOTAL SEAT Z / HT RES SEAT FORCE		74.10 60.33 411.39 353.95 852.08 1590.56 10.97 1602.13	-211.58 -45.40 20.43 7.53 73.33 115.92 0.80 120.14	4140.00 4001.00 3939.00 3942.00 3943.00 3943.00 3943.00	3938.00 3932.00 3717.00 3616.00 3605.00 3616.00 3616.00	35 11 12 13
RES SEAT FORCE / WT LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X		33.18 30.93 75.50	0.63 -93.69 -49.65 -162.27 -296.78	3943.00 3867.00 3889.00 3889.00 3888.00	3616.00 3937.00 3936.00 3937.00 3937.00	20 23 26
LF FOOT Y RT FOOT Y CT FOOT Y TOTAL FOOT Y		89.30 22.20 19.92 44.50	-7.65 -88.73 -46.83 -68.56	3920.00 3975.00 3900.00 3884.00	3890.00 3937.00 3951.00 3950.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE		149.11 156.94 153.19 355.16 451.07	-39.52 -27.84 -86.76 -108.50 9.32	3944.00 3946.00 3926.00 3946.00 3946.00	3879.00 4028.00 3879.00 3879.00 4037.00	22 25 28

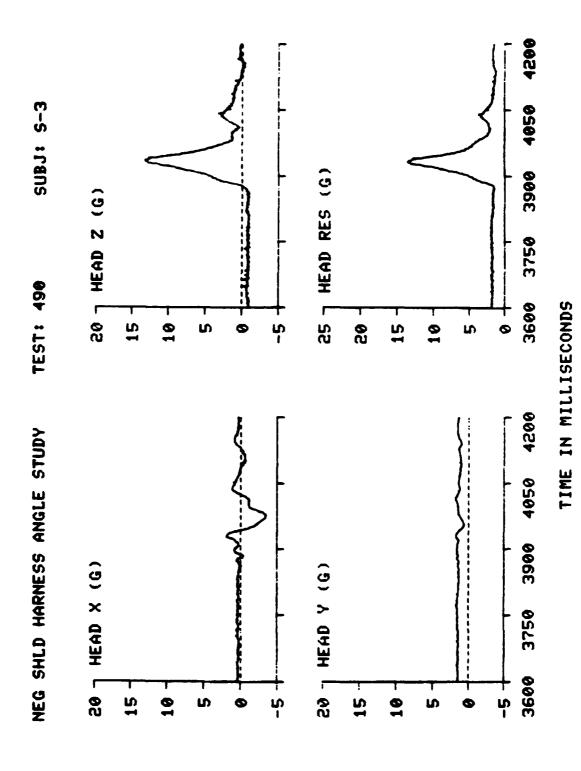
NEG SHLD HAR ANG	TEST: 518	SUBJ:	R-3 WT:	148.0	G: 10 GP:	1 CELL:	J
DATA ID			MAX	MIN	T1	T 2	СН
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z			10.05 1.36 0.57 12.18 11.11	9.97 -1.93 -0.91 -0.38	2161.00 3853.00 3896.00 3886.00	16.00 3851.00 3835.00 3782.00	48 36 31
CARRIAGE Z (SM) CARRIAGE Z (SM) CARRIAGE VEL SEAT X SEAT Z SEAT Z (SM)			-0.99 1.35 0.75 12.01	-0.13 -26.17 -1.96 -0.89 -0.23	4162.00 3855.00 3646.00 3892.00	3780.00 3840.00 3861.00 3919.00 3790.00	29 32 33 34
CHEST X CHEST Y CHEST Z CHEST RES			11.20 2.70 0.18 18.18 18.49	-0.14 -2.11 -2.98 -1.09 1.00	3894.00 3897.00 3894.00 3907.00 3908.00	3708.00 3948.00 3909.00 3690.00 4127.00	5 6 7
CHEST SI HEAD X HEAD Z HEAD Z HEAD RES HEAD SI			31.24 2.26 2.49 12.50 12.73 18.18 13.58	-3.41 0.68 -1.28 1.47	3847.00 3904.00 3981.00 3906.00 3904.00 3859.00	3963.00 3963.00 3929.00 3724.00 4197.00 3996.00	3 4
HEAD HIC SHD REFL LF SHD REEL LF LF SHOULDER			13.58 33.37 51.27 81.81 22.69	8.21 2.72 12.96	3881.00 3975.00 3952.00 3952.00 3977.00	3942.00 3893.00 3914.00 3904.00 3886.00	14 16
SHO REFL RT SHO REEL RT RT SHOULDER TOTAL SHLO REFL TOTAL SHLO REEL TOTAL SHOULDER			45.05 66.40 55.91 94.97 146.42	7.47 2.35 12.02 17.47 6.59 25.57	3955.00 3955.00 3977.00 3953.00 3953.00	3893.00 3912.00 3893.00 3913.00 3905.00	15 17
TOTAL SHD / WT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / WT			0.99 61.49 68.12 128.26 0.87	0.17 28.48 27.75 56.52 0.38	3953.00 3982.00 3979.00 3981.00 3981.00	3905.00 3886.00 3692.00 3892.00 3892.00	8
TÖTAL LAP / WT CROTEH STRAP LF SEAT LNK X RT SEAT LNK X TOTAL SEAT X			34.68 25.68 21.84	-70.34 -151.76 -69.16 -219.70	3989.00 4122.00 3963.00 3964.00	3901.00 3907.00 3904.00 3907.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z WT RES SEAT FORCE			45.10 481.73 529.30 687.02 1692.53 11.44 1708.57	-79.47 39.75 41.07 51.60 152.71 1.03 153.49	3990.00 3907.00 3908.00 3905.00 3907.00 3907.00	3907.00 3643.00 3628.00 3604.00 3628.00 3628.00	35 11 12 13
RES SEAT FORCE / LF FOOT X RT FOOT X CT FOOT X TOTAL FOOT X	МT		11.54 10.25 46.60 14.65 49.42	1.04 -128.08 -49.45 -154.39 -323.91	3907.00 3854.00 3891.00 3853.00	3628.00 3886.00 3885.00 3903.00 3866.00	20 23 26
LF FOOT Y BT FOOT Y CT FOOT Y TOTAL FOOT Y			103.49 27.81 6.76 55.47 186.91	-7.44 -77.79 -75.11 -67.46 -23.12	3693.00 3865.00 3792.00 3867.00 3875.00	3844.00 3901.00 3899.00 3901.00 3843.00	21 24 27
LF FOOT Z RT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FORCE			156.45 143.64	-24.28 -139.77 -117.49 24.46	3899.00 3852.00 3852.00 3892.00	3664.00 3686.00 3665.00 3661.00	22 25 28

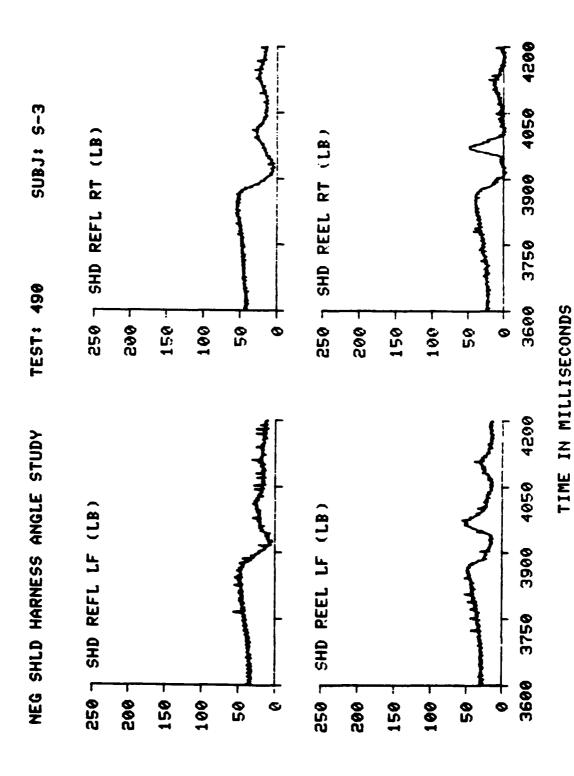
NEG SHED HAR ANG	TEST: 490	SUBJ:	S-3 WT:	168.0	G: 10 GP:	1 CELL:	j
DATA ID			HAX	HIN	<u>T 1</u>	T2	H3
10V EXT PWR CARRIAGE X CARRIAGE Y CARRIAGE Z			10.05 1.84 0.95 12.30	9.97 -1.18 -0.64 -0.24	134.00 3878.00 3869.00 3908.00	514.00 3891.00 3866.00 3760.00	48 35 31 1
CARRIAGE Z (SM) CAPRIAGE VEL SEAT X SEAT T SEAT Z SEAT Z SEAT Z SEAT Z			10.33 -1.20 1.89 1.64 10.90	-0.08 -26.22 -1.34 -1.80 -0.17	3909.00 4196.00 3878.00 3875.00 3907.00	3701.00 3857.00 3869.00 3880.00 3777.00	29 32 33 34
CHEST X CHEST Y CHEST Z CHEST RES			10.27 3.09 -0.70 17.12 17.55	-0.09 -1.55 -3.70 -0.97 1.05	3916.00 3926.00 3979.00 3935.00 3935.00	3726.00 3976.00 3936.00 3771.00 3741.00	5 6 7
CHEST SI HEAD X HEAD Z HEAD RES HEAD SI			31.95 1.90 1.81 13.21 13.43 19.97	-3.44 0.68 -1.13 1.16	3873.00 3929.00 4015.00 3932.00 3932.00 3883.00	4117.00 3977.00 3959.00 3737.00 4139.00 4001.00	2 3 4
HEAD HIC SHD REFL LF SHD REEL LF			15.76 27.09 51.22	6.37 13.46	3903.00 4009.00 3973.00	3960.00 3924.00 4060.00	14 16
LF SHOULDER SHD REFL RT SHD REEL RT RT SHOULDER TOTAL SHLD REFL TOTAL SHLD REEL			74.11 28.47 47.97 67.24 55.55	20.11 4.90 -2.77 3.11 11.64 11.95	3973.00 4009.00 3972.00 3972.00 4009.00 3972.00	3924.00 3926.00 3915.00 3916.00 3925.00	15 17
TOTAL SHOULDER TOTAL SHO / NT LF LAP BELT RT LAP BELT TOTAL LAP TOTAL LAP / NT			141.28 0.84 57.20 55.82 112.84 0.67	23.64 0.14 28.54 30.67 60.09	3973.00 3973.00 3996.00 3994.00 3995.00	3925.00 3925.00 3912.00 4058.00 4058.00	8 9
CROTCH STBAP LF SERT LNK X RT SERT LNK X TOTAL SERT X			61.09 4.53 8.90	-34.21 -187.32 -113.46	4004.00 3818.00 3732.00 3813.00	3928.00 3925.00 3931.00 3924.00	10 18 19
SEAT LNK Y LF SEAT PAN Z RT SEAT PAN Z CT SEAT PAN Z TOTAL SEAT Z TOTAL SEAT Z HT RES SEAT FORCE			47.51 642.53 607.40 627.61 1869.22	-296.63 -112.92 80.09 68.02 35.03 230.42 1.37 230.97	4171.00 3934.00 3932.00 3933.00 3933.00	3931.00 4195.00 4161.00 3609.00 3609.00 3609.00	35 11 12 13
	HT		1893.47 11.27 40.78 33.38 59.55 119.41	1.37 -72.24 -24.78 -146.17 -202.36	3933.00 3877.00 3876.00 3879.00 3877.00	3609.00 3945.00 3910.00 3925.00 3945.00	20 23 26
LF F00T Y RT F00T Y CT F00T Y T0TAL F00T Y			72.97 27.39 63.38 81.52	-14.38 -65.79 -62.10 -59.39	3910.00 3970.00 3948.00 3952.00	3639.00 3941.00 3921.00 3923.00	21 24 27
LF FOOT Z AT FOOT Z CT FOOT Z TOTAL FOOT Z RES FOOT FOACE			164.92 97.57 321.69	-68.83 -64.57 -75.14 -165.87	3944.00 3912.00 3941.00 3941.00 3942.00	3889.00 3938.00 3888.00 3889.00 4200.00	22 25 28

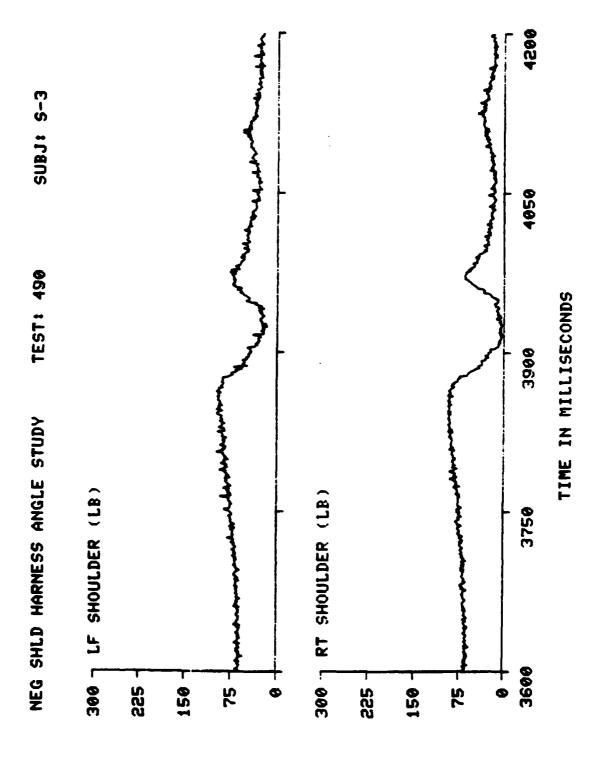


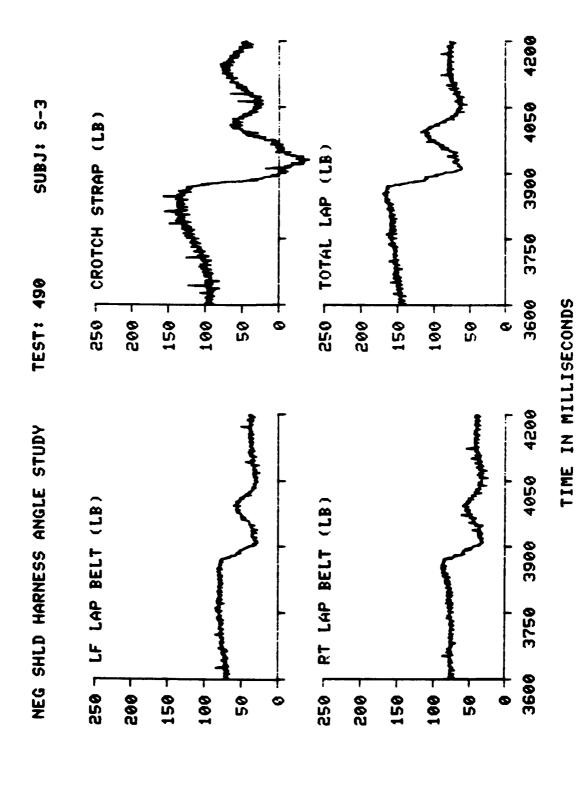


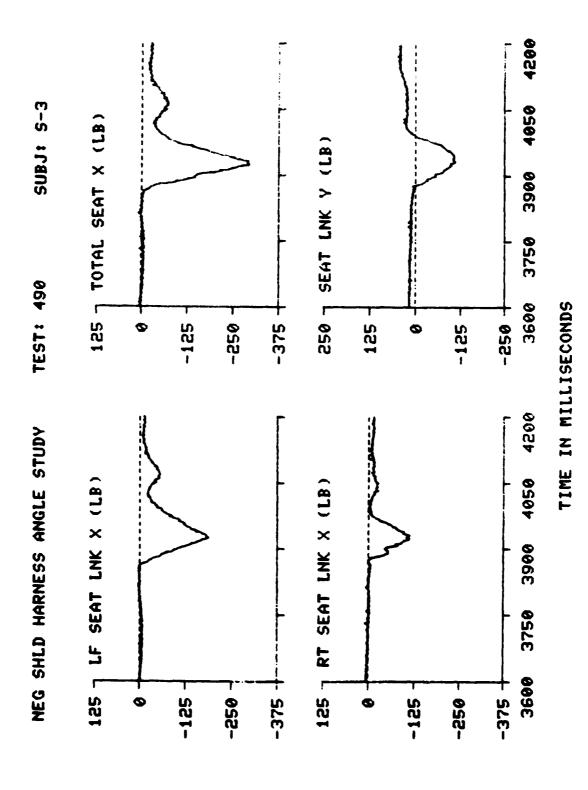


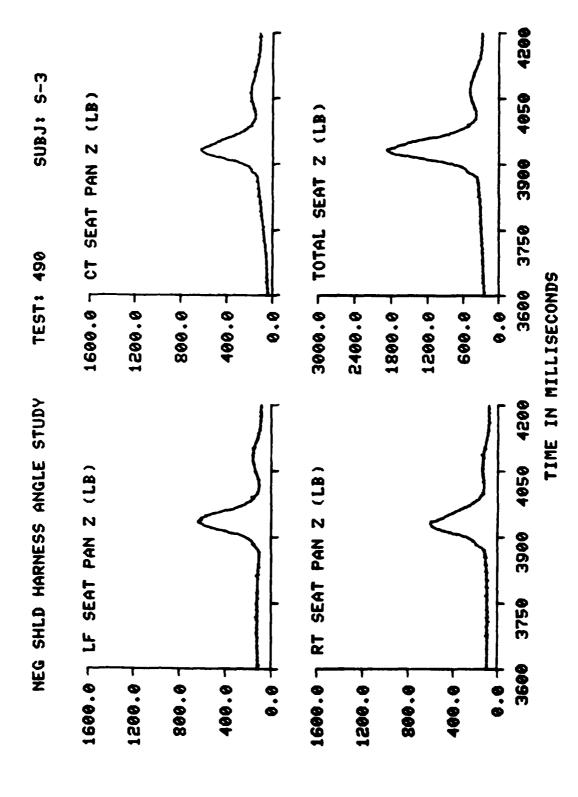


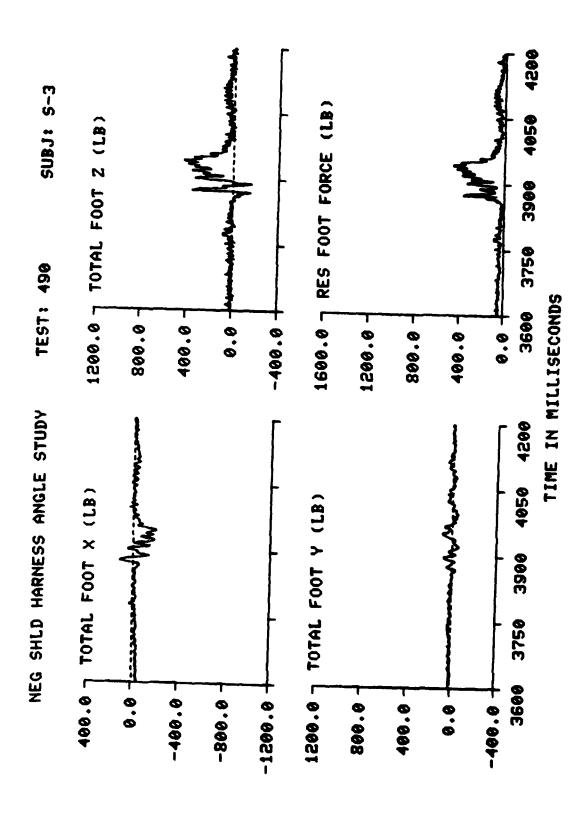












APPENDIX C

WILCOXON ANALYSES

The electronic data from this test program were analyzed by means of the Wilcoxon paired-replicate rank test (Wilcoxon & Wilcox, 1964). A total of six comparisons were made by this technique. The means and standard deviations of each parameter in each comparison are summarized in Tables C-1 through C-6. Since the number of comparable tests in one comparison may be different from the number of comparable tests in another comparison, minor variations in the means and standard deviations in a cell of the test matrix may be noted among these tables. An asterisk designates a statistically significant trend in a parameter at the 90% confidence level for a two-tailed test. The statistically significant trends in all six comparisons are summarized in the body of the report in Table 5.

The Wilcoxon analyses of parameters for which there were statistically significant differences are also presented. In these computations, the arithmetic difference between the parameter means from each condition is first computed. These differences are then rank ordered from smallest to largest, without regard to sign. An integer from 1 to n, where n is the number of pairs in the comparison, is assigned to each difference so that the smallest difference receives the rank 1 and the largest difference receives the rank n. Then, the rank integer is given the same sign as the sign of the arithmetic difference to which it corresponds. The negative integers are summed and the positive integers are summed. Finally, if either sum is greater than or equal to the critical integer sum for the specified confidence level, then the means may be considered significantly different (i.e., from two different samples).

A complete set of Wilcoxon computations for all comparisons in this test program will be maintained by the Biomechanical Protection Branch of AFAMRL until this work unit is retired. These data will eventually be recorded in a permanent data bank within the Laboratory.

TABLE C-1

COMPARISON C-G

SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA FROM WILCOXON ANALYSIS

(Peak values are tabulated for valocity, accelerations and loads.)

(n = 14)

TEST CONDITION F-111 HARNESS SEAT POSITION	í. Modified Down		G Operational Nown		Significant at 90% Confidence
CARRIAGE ACCELERATION (G) CARRIAGE VELOCITY (ft/sec) SEAT ACCELERATION (G)	Mean 10.5 -25.8 10.6	0.13 0.17 0.14	Mean 10.5 -25.9 10.6	s 0.20 0.64 0.27	
CHEST ACCELERATION (G) -X axis +X axis	-2.05 4.38	1.01 1.20	-1.96 3.57	0.51 0.93	*
+Z axis Resultant	20.3	2.85 2.79	17.3 17.6	2.44 2.41 4.22	* *
CHEST SEVERITY INDEX HEAD ACCELERATION (G) -X axis	35.4 -4.86	5.91 0.92	33.0 -3.34	1.05	*
+X axis +Z axis Resultant	1.12 13.2 13.3	0.89 0.91 0.92	1.47 12.3 12.5	0.74 0.63 0.64	*
HEAD SEVERITY INDEX STRAP LOADS (1b)	19.5	2.17	18.9	1.94	
Reflection Straps Inertia Reel Straps Tot?l Shoulder Straps	103 110 206	25 28 54	66 76 134	23 22 35	* * *
Total Lap Belt Crotch Strap	92 111	27 75	90 121	29 63	
SEAT PAN LOADS (1b) -X axis +Z axis	-290 1750	73 243	-2// 1760	67 274	
Resultant FOOTREST LOADS (1b)	1770	246	1780	276	
-X axis +Z axis Resultant	-369 456 538	89 60 95	- 387 456 556	93 74 105	

TABLE C-2

COMPARISON H-J

SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA FROM WILCOXON ANALYSIS

(Peak values are tabulated for velocity, accelerations and loads.)

(n = 14)

TEST CONDITION F-111 HARNESS SEAT POSITION	H		J		Significant
	Modified		Operational		at 90%
	Up		Up		Confidence
CARRIAGE ACCELERATION (G) CARRIAGE VELOCITY (ft/sec) SEAT ACCELERATION (G) CHEST ACCELERATION (G)	Mean 10.5 -26.1 10.7	s 0.18 0.15 0.39	Mean 10.5 -26.2 10.6	s 0.24 0.06 0.30	
-X axis	-1.79	0.88	-2.59	0.97	*
+X axis	3.73	1.39	2.68	1.34	
+Z axis	18.6	1.79	17.2	2.04	*
Resultant	18.8	1.71	17.4	2.02	*
CHEST SEVERITY INDEX	34.7	4.24	32.1	2.92	*
HEAD ACCELERATION (G) -X axis	-4.06	1.07	-3.24	1.07	*
+X axis	1.79	1.30	2.31	1.25	*
+Z axis	13.1	0.85	12.6	0.94	
Resultant HEAD SEVERITY INDEX STRAP LOADS (1b)	13.3 20.1	0.86 1.91	12.9 20.4	0.96 1.91	*
Reflection Straps	81	13	63	13	*
Inertia Reel Straps	88	22	95	34	
Total Shoulder Straps	160	33	150	45	*
Total Lap Belt	96	20	102	27	
Crotch Strap	140	52	93	32	
SEAT PAN LOADS (1b) -X axis	-295	70	-281	57	•
+Z axis	1770	209	1770	215	
Resultant	1790	213	1790	217	
FOOTREST LOADS (1b) -X axis +Z axis	-345 467	93 82	-324 443	79 71	*
Resultant	538	107	508	99	

TABLE C-3

COMPARISON C-J

SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA FROM WILCOXON ANALYSIS

(Peak values are tabulated for velocity, accelerations and loads.)

(n = 12)

TEST CONDITION F-111 HARNESS SEAT POSITION	C		J		Significant
	Modified		Operational		at 90%
	Down		Up		Confidence
CARRIAGE ACCELERATION (G) CARRIAGE VELOCITY (ft/sec) SEAT ACCELERATION (G) CHEST ACCELERATION (G)	Mean 10.5 -25.8 10.6	s 0.12 0.17 0.15	Mean 10.5 -26.2 10.5	s 0.26 0.06 0.32	*
-X axis	-2.21	0.98	-2.40	0.86	*
+X axis	4.25	1.25	2.84	1.37	
+Z axis	20.6	2.88	17.3	2.18	
Resultant	20.9	2.83	17.4	2.17	*
CHEST SEVERITY INDEX	36.1	5.93	32.2	3.10	
HEAD ACCELERATION (G) -X axis +X axis	-4.93 1.23	0.98 0.91	-3.36 2.02	1.10 0.96	*
+Z axis	13.2	0.95	12.7	0.84	*
Resultant	13.4	0.94	12.9	0.84	
HEAD SEVERITY INDEX	19.4	2.27	20.2	1.90	
STRAP LOADS (1b) Reflection Straps Inertia Reel Straps	103	27	59	10	*
	111	30	87	28	*
Total Shoulder Straps Total Lap Belt	206 88 89	59 26 51	139 99 86	37 24 29	*
Crotch Strap SEAT PAN LOADS (1b) -X axis	-286	76	-283	60	
+Z axis Resultant FOOTREST LOADS (1b)	1680 1700	175 181	1750 1770	220 222	*
-X axis	-347	70	-304	58	*
+Z axis	440	43	430	65	
Resultant	509	60	484	74	
i/ean i calle	303	00	707	7 7	

TABLE C-4
COMPARISON H-G

THE FOR ELECTRONICALLY MEASURED AND COMPUTED DATA FROM WILCOXON ANALYSIS

(PERC aties are tabulated for velocity, accelerations and loads.)

(n = 15)

PIST SUNDITION 1-111 HARNESS SUAT POSITION	Mod	H if ie d Jp	Opera	G ationa! own	Significant at 90% Confidence
CARRIAGE VELOCITY (ft/sec) CARRIAGE VELOCITY (ft/sec) SEAT ACCELERATION (G) CHEST ACCELERATION (G) -X axis +X axis +Z axis Resultant	10.7 -2.01 3.66 18.3 18.6	0.58 1.3/ 1.96 1.88	Mean 10.5 -25.9 10.7 -1.77 3.54 17.1 17.4		*
CHEST SEVERITY INDEX HEAD ACCELERATION (G) -X axis +X axis +Z axis Resultant HEAD SEVERITY INDEX	34.4 -4.07 1.72 13.0 13.2 19.8	1.03 1.29 0.94 0.97	32.6 -3.31 1.67 12.5 12.6 19.4	4.29 1.09 1.00 0.82 0.84 2.04	* * * *
STRAP LOADS (1b) Reflection Straps Inertia Reel Straps Total Shoulder Straps Total Lap Belt Crotch Strap SEAT PAN LOADS (1b)	80 86 156 97 140	13 22 35 20 50	61 74 129 84 104	12 24 32 20 29	* * * *
-X axis +2 axis Resultant FOOTREST LOADS (1b) -X axis +Z axis Resultant	-295 1760 1790 -338 461 530	68 204 207 94 83 107	-277 1700 1720 -381 461 552	66 211 213 90 76 102	* *

TABLE C-5
COMPARISON C-H

SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA FROM WILCOXON ANALYSIS

(Peak values are tabulated for velocity, accelerations and the

(n = 12)

TEST CONDITION F-111 HARNESS SEAT POSITION	Mod ·) ified own	H Modi L		÷	7+ C5 % 9(3) 0:
CARRIAGE ACCELERATION (G) CARRIAGE VELOCITY (ft/sec) SEAT ACCELERATION (G) CHEST ACCELERATION (G)	Mean 10.5 -25.8 10.6	s 0.12 0.17 0.15	Me 1/3 10.5 -26.1 10.7	0.10 0.10 0.10 0.5		
-X axis +X axis +Z axis Resultant CHEST SEVERITY INDEX HEAD ACCELERATION (G)		0.98 1.25 2.88 2.83 5.93	-1.65 3.88 18.5 18.9 34.3	0 58 1,34 1,89 1,80 4,30		•
-X axis +X axis +Z axis Resultant HEAD SEVERITY INDEX	-4.93 1.23 13.2 13.4 19.4	0.98 0.91 0.95 0.94 2.27	-4.20 1.38 13.2 13.3 19.7	1.06 9.80 9.7 0.7 0.86		*
STRAP LOADS (1b) Reflection Straps Inertia Reel Straps Total Shoulder Straps Total Lap Belt Crotch Strap	103 111 206 88 89	27 30 59 26 51	82 85 157 96 135	13 22 34 19 53		* * *
SEAT PAN LOADS (1b) -X axis +Z axis Resultant FOOTREST LOADS (1b) -X axis +Z axis	-286 1680 1700 -347 440	76 175 181 70 43	-294 1730 1760 -320 449	76 208 212 74 71		
Resultant	509	60	510	83		

TABLE C-6

COMPARISON G-J

SUMMARY OF ELECTRONICALLY MEASURED AND COMPUTED DATA FROM WILCOXON ANALYSIS

(Peak values are tabulated for velocity, accelerations and loads.)

(n = 14)

TEST CONDITION F-111 HARNESS SEAT POSITION	Opera	ational own] ational Jp	Significant at 90% Confidence
CARRIAGE ACCELERATION (G) CARRIAGE VELOCITY (ft/sec) SEAT ACCELERATION (G) CHEST ACCELERATION (G) -X axis	Mean 10.5 -25.9 10.7	s 0.20 0.64 0.31	Mean 10.5 -26.2 10.6	s 0.24 0.06 0.30	*
-x axis +X axis +Z axis Resultant CHEST SEVERITY INDEX HEAD ACCELERATION (G)	3.55 17.2 17.4 32.8	0.59 0.92 2.11 2.10 4.38	-2.59 2.68 17.2 17.4 32.1	0.97 1.34 2.04 2.02 2.92	*
-X axis +X axis +Z axis Resultant HEAD SEVERITY INDEX STRAP LOADS (1b)	-3.28 1.72 12.3 12.5 19.3	1.10 1.02 0.74 0.79 2.07	-3.24 2.31 12.6 12.9 20.4	1.07 1.25 0.94 0.96 1.91	* *
Reflection Straps Inertia Reel Straps Total Shoulder Straps Total Lap Belt Crotch Strap SEAT PAN LOADS (1b)	61 76 131 85 105	13 23 33 20 30	63 95 150 102 93	13 34 45 27 32	* *
-X axis +Z axis Resultant FOOTREST LOADS (1b) -X axis	-273 1710 1730	66 217 219	-281 1770 1790 -324	57 215 217 79	* *
+Z axis Resultant	458 555	79 105	443 508	71 99	

ANALYSIS	OF:	CHEST	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C G			MAX MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M13 R-3 R-3 S-3 M10 F-3 M11 D-1	5.09 3.68 6.35 6.36 5.12 5.14 3.97 5.16 3.76 3.72 3.73	B VAL 4.20 3.33 3.53 4.58 3.49 4.93 4.13 2.50 3.77 3.20 2.07 3.93 4.52 1.83	0.89 0.52 0.15 1.77 -0.13 0.76 0.99 2.64 0.96 -0.17 0.71	0.00 -0.13 0.00 -0.17 0.00 0.00 0.00 0.00 0.00 0.00 0.00	0.10 0.00 0.15 0.00 0.20 0.52 0.71 0.76 0.89 0.96 0.99	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00	0.00 2.00 0.00 4.00 0.00 0.00 0.00 0.00	1.00 3.00 3.00 5.00 6.00 7.00 9.00 10.00 11.00 12.00
MEAN A: MEAN B:	4.38 3.57	STD DEV	A: B:	1.20 0.93	SUM OF N	MINUS: PLUS :	6.00	99.00
					MMM SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON	ANALYSI:	5						
	a	-	_					
		CHEST		r			MAX	
		10 10					MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-13 M-3 R-3 P-3 S-10 F-3 M11 D-1	21.86 22.74 22.88 18.62 25.04 18.92 19.48 25.39 17.67 19.12 18.77 16.73 20.08	21.87 14.91 16.31 15.98 16.55 15.32 21.53 20.91 16.44 16.25 16.11 16.84 14.63 18.66	-0.01 7.83 6.57 2.64 8.49 2.961 -1.43 8.95 1.401 1.93 2.10	-0.01 0.00 0.00 -1.43 0.00 0.00 0.00 -4.61 0.00 0.00	0.00 1.42 1.42 0.00 1.93 2.10 2.54 2.93 3.01 0.00 6.57 7.83 8.95	1.00 2.50 4.00 5.00 6.00 7.00 8.00 9.00 11.00 12.00 13.00	1.00 0.00 0.00 4.00 0.00 0.00 0.00 10.00 0.00	0.00 2.50 0.00 5.00 6.00 8.00 9.00 11.00 12.00

ANALYSIS	S OF:	LHE	ST REG					
FUNCTION FUNCTION	V A = G: V B = G:	10 10	CELL:				ABS ABS	
SUBJ	A VAL	B VAL	A - B	ORD -	08U +	N	N -	N +
G-9 G-2 F-13 M12 8-3 M10 M10 M10 M1-2	22.29 22.87 23.17 19.23 25.23 18.75 17.30 20.04	22.06 14 94 16.57 16.43 16.69 15.53 21.07 16.87 16.63 16.19 17.20 14.94	1.02 1.02 1.02 1.02 1.02 1.02	0.00 1.03 0.00 0.00 0.00 0.00 0.00	0.23 0.00 1.41 1.40 1.67 2.18 2.02 3.02 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 0.00 5.00 4.00 5.00 5.00 5.00
MEAN A: MEAN B:	20.55 17.55	5.0 br 570 BF	+ fi: v R;	2 19 - 31			12.30	
	OF :	HEAN 10 10					IN	
SUBU	A VAL	8 V60			09p +		IN	
GG-2 GG-2 K-13 M1-2 B-3 P-3 M1-3 M1-3 M1-1 M-1	-5.27 -5.75 -9.47 -5.38 -9.49 -5.39 -4.00 -4.00 -3.43 -3.95 -3.67 -5.40	4.35 4.35 4.35 4.35 4.35 4.35 3.86 1.35 8.75 1.36 1.36 1.36 1.36 1.36 1.36 1.36 1.36	1.70 2.40 1.62 1.57 3.07 0.37 1.44 1.74 0.88 2.52 0.07	0.07 -0.37 -0.84 -0.88 1.14 1.32 1.44 1.57 1.52 1.70 2.29 2.40	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.08 2.00 3.90 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00 13.00	N - 1.00 2.00 3.00 4.00 5.00 7.00 8.00 9.00 10.00 11.00 12.00 13.00	N +
MEAN A; MEAN B:	-4.86 -3.34	SID DEV	f!: ! fr: 1	· , 설탕 - 교육	SUM OF N	MINUS: PLUS: -	105.00 -	0.00

HAR SIGNIFICANT DIFFERENCE ----

HNHL 1212	UF:	HEAD	t.					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C G		ĺ	MAX MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M13 R-3 P-3 S-3 M10 F-3 M11 D-1	13.02 12.39 14.62 12.31 13.15 14.21 13.32 12.43 13.69 14.37 13.56 12.89 13.74	B VAL 11.80 12.40 13.10 11.22 12.76 12.82 12.30 13.46 12.69 11.86 11.86 11.46	1.22 -0.01 1.52 1.09 0.39 1.39 1.12 0.13 0.23 1.68 1.70 0.69 1.88	-D.01 0.00 -0.15 0.00 0.00 0.00 0.00 0.00 0.00 0.00	0.00 0.13 0.00 0.23 0.39 0.63 1.09 1.12 1.22 1.39 1.52 1.68 1.70	1.00 2.00 3.00 4.00 5.00 6.00 7.00 9.00 10.00 11.00 12.00	1.00 0.00 3.00 0.00 0.00 0.00 0.00 0.00	0.00 2.00 4.00 5.00 6.00 7.00 9.00 10.00 12.00 13.00
		STD DEV						
					*** \$16	NIFICAN	DIFFERE	NCE +++
WILCOXON	ANALYSI:	5						
ANALYSIS	OF:	HEAD	RES					
FUNCTION FUNCTION	A = G:	10						
SUBJ						!	985 985	
*	a vai	B VOI	a_a	MBD _	ORD +	N.	ABS	N +
G-3 G-2 F-2 K-1 M19 R-2 R-3 P-3 S-3 M10 F-3 M11 O-1	a vai	B VAL	a_a	MBD _	0RD + 0.00 0.14 0.16 0.24 0.27 0.75 0.86 0.90 1.02 1.46 1.49 1.70 1.72	N.	ABS	N + 10.000 2.000 4.000 5.000 7.000 10.000 11.000 14.00

NAM SIGNIFICANT DIFFERENCE NAM

ANALYSIS	OF:	TOTAL	SHLD I	REFL				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C C			1AX 1AX	
SUBJ	A VAL	8 VAL	A-8	ORD -	ORD +	N	N -	N +
G-2 G-2 K-13 R-3 R-3 P-3 S-10 F-3 M13 M11 D-12	119.18 72.40 107.94 80.21 135.88 65.21 110.82 109.74 107.03 73.21 134.52 84.35 105.38 142.32	B VAL 68.61 37.22 53.74 61.26 62.45 57.47 665.57 103.17 57.13 46.58 81.51 52.19 122.12 57.48	50.57 54.20 18.95 73.43 7.74 44.25 49.90 26.63 53.01 32.16 -16.74 84.84	0.00 0.00 -16.74 0.00 0.00 0.00 0.00 0.00 0.00 0.00	6.57 7.74 0.00 18.95 26.63 32.16 35.18 44.25 49.90 50.57 53.01 54.20 73.43 84.84	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00 12.00 13.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	1.00 2.80 0.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00 12.00 13.00
		STD DEV STD DEV						
					HHH SIG			
	ØF:	TOTAL					10 V	
. FUNCTION	B = G:	10 10	CELL:	C			1AX 1AX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-1 M132 R-3 R-3 M11 F-3 M111 D-2	137.25 98.47 141.31 74.81 130.38 62.10 114.87 109.72 115.46 71.71 133.82 100.75 95.09	115.75 39.83 62.66 57.45 83.32 59.15 78.49 107.85 72.82 62.54 110.45 62.06 85.17	21.50 58.65 17.36 47.06 2.95 36.387 42.64 9.17 23.37 23.39 83.88	08D	1.87 2.95 9.17 9.98 17.36 21.50 23.37 36.69 47.06 47.06 58.65 83.46	1.00 2.00 3.00 4.00 5.00 6.00 7.00 9.00 10.00 11.00		1.00 2.00 3.00 4.00 5.00 6.00 7.00 9.00 10.00 11.00
	136.73							

		TOTAL						
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C G			1AX 1AX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M13 R-2 R-3 F-3 M10 F-3 M11 D-1	256.32 167.12 240.70 137.30 265.70 120.70 224.50 216.66 209.19 133.52 259.04 170.31 189.95 292.51	B VAL 184.29 75.90 116.39 109.17 144.84 108.69 142.36 184.03 123.68 106.31 185.90 107.22 172.02 121.85	72.03 91.22 124.31 28.13 120.86 12.01 82.14 32.63 85.51 27.21 73.14 63.09 17.93 170.66	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	12.01 17.93 27.21 28.13 32.63 63.09 72.03 73.14 85.51 91.22 120.86 124.31 170.66	1.00 2.00 3.00 4.00 5.00 7.00 8.00 9.00 10.00 11.00 13.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.00 3.00 4.00 5.00 7.00 8.00 9.00 10.00 11.00 12.00 13.00
MEAN A: MEAN B:	205.97 134.48	STD DEV	A: B:	54.32 35.12	SUM OF N	MINUS: PLUS :	0.00	105.00
					MMM SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON								
	HNHLTSIS	5						
		-	v					
ANALYSIS	0F:	- CHEST		ш			4 T NJ	
ANALYSIS FUNCTION	OF: A = G:	- CHEST	CELL:	H		ţ	A T N	
ANALYSIS FUNCTION	OF: A = G:	- CHEST	CELL:	H J ORD -	ORD +	ì	A T N	N +
ANALYSIS FUNCTION	OF: A = G:	- CHEST	CELL:	H J GRD	0RD + 0.00 0.27 0.45 0.50 0.50 0.52 0.58 0.75 0.82 1.01 1.16 1.36 1.63 1.76	ì	A T N	N + 0.00 2.00 3.00 4.50 4.50 6.00 7.00 8.00 10.00 11.00 13.00

MMM SIGNIFICANT DIFFERENCE MMM

		-						
ANALYSIS	OF:	CHEST	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: H	!		ы	A.: 10.	
SUBJ	A VAL	B VAL	A-B	ORD ~	ORD +	N	NJ	N =
F-3 H-4 M13 R-3 G-2 S-3 K-1 F-2 M11 H-3 M10 G-2	2.04 1.47 3.27 3.79 4.51 1.36 7.06 4.29 4.20 3.63 3.64	B VAL 1.57 1.15 2.79 2.70 4.45 1.32 3.09 5.89 1.87 3.79 3.19 2.29 1.80 1.56	0.47 0.32 0.48 1.09 0.06 0.39 1.27 1.17 2.37 0.30 1.03 1.91 1.83 2.08	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.06 0.30 0.32 0.39 0.47 0.48 1.03 1.09 1.17 1.27 1.83 1.91 2.08 2.37	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 13.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	10000000000000000000000000000000000000
MEAN A: MEAN B:	3.73 2.68	STD DEN	/ A: / B:	1.39 1.34	SUM OF N	MINUT.	1.7. 1.7.	105.22
WILCOXON 		S - CHESI	T Z					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: H	l I		M	AX AX	
SUBJ	A VAL	B VAL	A-B	OAD -	ORD +	N	N -	₹ +
F-3 H-4 M13 R-3 G-3 M-2 S-3 K-1 F-2 M11	15.64 17.19 20.47 20.70 21.09 20.55 18.24 16.98 17.05	B VAL 16.14 16.07 20.97 18.18 19.17 19.72 17.12 16.05 19.04 13.66 16.15 17.89 15.97 14.78	-0.50 1.12 -0.50 2.52 1.92 0.83 1.12 0.41 -2.06 3.39 2.12	0.00 -0.50 -0.50 0.00 0.00 0.00 0.00 -2.06 0.00	0.41 0.00 0.00 0.83 1.12 1.12 1.20 1.92 0.00 2.12 2.52	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00	0.00 3.00 0.00 0.00 0.00 0.00 0.00 0.00	9.00 0.00 0.00 9.00 9.00 6.00 0.00 11.00

NAM SIGNIFICANT DIFFERENCE NAM

SUM OF N MINUS: 14.00 ------SUM OF N PLUS: ----- 91.00

MEAN A: 18.57 STD DEV A: 1.79 MEAN B: 17.21 STD DEV B: 2.04

ANALYSIS	G::	CHEST	RES					
FUNCTION FUNCTION		i.i f	CETT:				ARX MAX	
SUBJ	H VAL	p A.ir	A-B	Ø60 -	080 +	N	N -	N +
54883223122:5	15.70 17.40 20.64 21.15 20.569 17.263 17.263 18.33 19.33 18.33 18.31	16.15 16.47 16.46 19.18 19.55 16.54 19.17 16.77 16.08	0.42 0.98 0.461 1.952 0.72 -1.48 3.57 1.85 1.36	-0.42 -0.46 -0.00 0.00 0.00 -1.46 0.00 0.00	8.00 0.72 0.72 0.98 1.14 0.00 1.88 2.61 2.25 3.57	00000000000000000000000000000000000000	8. 00 0.00 0.00	1.50 6.00 7.00 5.00 11.00 12.00
MEAN 8:	18.83 17.40	STD DEV	H: B:	1.71 2.02	SUM OF N	หไทบว่: PLUS :	11.00	
					яня 516	NJE , Calvi	MIFIERE	ÎNÎL SKA
FILEBXON	ANALYSIS							

ANAL/SIS	ÐF.	CHEST	51					
FUNCTION FUNCTION			CELL:	l i			MAX MAX	
5085	A VAL	8 VAL		650 ·	⊕ 0 -	ř.	N	, ·
	28,355,430 37,05,430 38,005,305 38,005,126 37,166 37,166 40,06	33.19 29.88 36.49 31.29 31.95 31.95 31.79 31.79 31.79 31.79 31.79 31.79	0.58 7.20 6.73 1.71 2.10 3.57 1.12 -1.85 -0.67 8.03	0.67 0.00 0.00 -1.85 0.00 0.00 0.00 -4.81 0.00 0.00	0.56 0.00 1.12 1.71 0.03 2.10 2.70 3.57 4.70 0.00 6.19 6.73 7.20 8.03	4.00 5.00 7.00 8.00 9.00 10.00	0.0 <u>5</u>	9.00 0.00 6.00 8.00 9.00 11.00 13.00
MEAN A: MEAN B:	34.72 32.06	GTD DEV		4.24 2.92	SUM OF N	PLUS:	17.00	38.00
					HYH SIG	NIFICANT	CHERR	Vit nex×

HINE 1313	OF:	HEAD	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: F	1 I		ì	41N 41N	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N ~	N +
F-3 H-13 M13 G-3 K-23 K-23 K-23 K-22 M13 M10	-1.89 -2.41 -5.41 -5.03 -5.57 -4.99 -4.99 -4.99 -4.99	B VAL -1.65 -2.13 -1.16 -3.41 -4.08 -5.24 -4.52 -4.52 -4.52 -4.52 -4.33 -3.40 -3.93 -3.99 -3.09	-0.24 -0.24 -1.95 -1.95 -0.95 -0.31 -0.92 -0.54 -0.10 -1.05 -1.01	-0.10 -0.24 -0.26 -0.31 -0.42 -0.42 -0.92 -0.95 -1.01 -1.53 -1.82	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00 12.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00 12.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0
MEAN A: MEAN B:	-4.06 -3.24	STD DEV STD DEV	A: B:	1.07 1.07	SUM OF N	MINUS: PLUS :	105.00	0.00
					*** SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON	DNO! YS1							
	HINT 127	o .						
ANALYSIS		- HEAD	X					
FUNCTION	OF: A = G: B = C.	- HEAD	CELL: H			ì	1AX 1AX	
FUNCTION	OF: A = G: B = C.	- HEAD	CELL: H	OAD -	ORD +	ì	YON	N +
FUNCTION	OF: A = G: B = C.	- HEAD	CELL: H	0RD0.00 -0.17 -0.00 -0.42 -0.52 -0.62 -0.63 -0.00 -0.87 -1.14 -1.68 -2.23	0R0 + 0.02 0.00 0.17 0.00 0.44 0.00 0.00 0.00 0.00 0.00 0.0	ì	YON	N + 1.00 0.00 3.00 0.00 5.00 0.00 0.00 0.00 0

ANALYSIS	OF:	HEAD :	Z					
FUNCTION FUNCTION		10 10				M	IAX IAX	
\$UB J	A VAL	B VAL	A-8	ORD -	080 +	N	N -	N +
F-3 H-4 M13 G-3 G-3 K-1 S-3 K-1 F-2 M11 H-3 M10 G-2	13.18 13.66 13.31 11.85 12.98 11.80 14.34 13.15 13.13 13.81 13.15 11.59 14.19	12.25 13.11 13.17 12.50 11.64 10.90 13.21 12.70 13.50 12.95 12.72 10.81 14.05 13.06	0.93 0.55 0.14 -0.65 1.34 0.90 1.13 0.75 -0.37 0.43 0.78 0.14 -0.36	0.00 0.00 -0.36 -0.37 0.00 -0.65 0.00 0.00 0.00	0.14 0.14 0.00 0.00 0.43 0.55 0.00 0.75 0.78 0.86 0.90 0.93 1.13 1.34	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00	0.00 0.00 3.00 4.00 0.00 7.00 0.00 0.00 0.00	1.00 2.00 0.00 5.00 6.00 8.00 9.00 11.00 12.00 14.00
		STD DEV STD DEV						
					MMM SIG			
MILCOXON ANALYSIS	ØF:	- HERD (ш		c	1 8 S	
FUNCTION	B = G:	10 10	CELL:	Ĵ		ρ	185	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
F-34 M13 R-3 G-3 M-2 S-1 F-2 M11 H-3 M10 G-2	13.26 14.48 13.46 12.07 13.34 11.85 14.36 13.50 13.48 13.97 13.17 11.95 14.23 12.90	B VAL 12.56 14.00 13.62 12.73 12.11 10.95 13.43 12.75 13.85 13.12 12.84 11.17 14.07 13.08	0.70 0.48 -0.16 -0.66 1.23 0.90 0.75 -0.37 0.85 0.78 0.78 -0.18	-0.16 0.00 -0.18 0.00 -0.37 0.00 -0.66 0.00 0.00 0.00 0.00	0.00 0.16 0.00 0.33 0.00 0.48 0.70 0.75 0.78 0.85 0.90 0.93	1.50 1.50 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	1.50 0.00 3.00 0.00 5.00 0.00 0.00 0.00 0	0.00 1.50 0.00 4.00 0.00 6.00 9.00 10.00 11.00 12.00 14.00
		STD DEV						88.50

WILEUKON ANALISIS

ANALYSIS	OF:	TOTAL	SHLD F	9EFL				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	7			1AX 1AX	
508.	A VAL	B VAL	A-8	OHD -	080 +	N	<u>t.</u> :	N +
E H M R C M S K F R M H M G C M S K F R M M H M G C M S K F R M M M M M M M M M M M M M M M M M M	92.82 67.10 87.41 82.90 78.79 78.17 80.19 85.89 95.77 75.42 85.48 60.13	70.89 77.13 64.62 55.91 752.14 55.55 51.97 56.95 64.62 92.57 43.15	21.93 -10.93 -22.79 26.99 29.91 16.62 28.62 28.62 28.62 11.10 -7.14 14.56 18.46	-7.14 -10.03 -0.00 -0.00 -0.00 -0.00 -0.00 -0.00 -0.00 -0.00 -0.00 -0.00	0.00 0.00 11.10 14.56 16.65 18.46 21.93 22.62 22.79 26.99 28.62 29.91 33.21	1.00 2.00 3.00 4.00 5.00 7.00 8.00 10.00 11.00 12.00	1.00 2.00 0.00 0.00 0.00 0.00 0.00 0.00	0.000 3.000 5.000 7.000 7.000 10.000 11.000 13.000
MEAN A: MEAN B:	31.20 62. 73	STD DEV STD DEV	A: B:	12.64 13.19	SUM OF N	MINUS: PLUS :	3,00	102.00
					*** SIG	NIF1CANT	DIFFERE	NCE ***
WILCOXON	ANALYSIS	ŝ						
BNALTSIS	UF:	CACTO					10 1	
ANALYSIS FUNCTION	OF: A = G: B = G:	CACTO	CELL:	H		j	4A X 4A X	
ANALYSIS FUNCTION	OF: A = G: B = G:	CACTO	CELL:	H	ORU -	j	4AY	N ·
ANALYSIS FUNCTION	OF: A = G: B = G:	CACTO	CELL:	H	3.35 0.00 0.00 21.74 23.55 25.02 26.39 46.00 50.85 61.43 68.25 103.57 113.36 132.04	j	4AY	N 1.00 0.000 4.000 5.000 7.000 11.000 12.000 14.00

HAM SIGNIFICANT DIFFERENCE HAM

ANALYSIS	OF:	TOTAL	FOOT	7				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	T H		ı	MAX MAX	
SUBJ	A VAL	B VAL	A-B	OAD -	ORD +	N	N -	N +
F-3 H-4 M13 R-3 G-3 M-2 S-3 K-1 F-2 R-2 M11 H-3 M10 G-2	562.33 539.27 525.48 370.07 470.79 505.76 453.24 422.67 531.05 400.11 387.92 615.47 421.67 335.93	B VAL 439.44 459.18 511.56 401.21 398.78 475.95 430.19 384.92 535.95 375.90 573.06 510.26 345.49	122.89 80.09 13.92 -31.14 72.01 29.81 23.05 37.75 37.75 12.02 44.95 12.02 42.41 -88.59	-4.90 -9.56 0.00 0.00 0.00 -31.14 0.00 0.00 0.00 0.00 -88.59	0.00 0.00 12.02 13.92 23.05 29.81 0.00 37.75 42.41 44.95 72.01 80.09 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 13.00	1.00 2.00 0.00 0.00 0.00 7.00 0.00 0.00 0	0.00 0.00 3.00 4.00 5.00 6.00 0.00 9.00 11.00 12.00 0.00
MERN A: MERN B:	467.27 442.65	STD DEV	A: B:	81.76 70.77	SUM OF N	MINUS: PLUS :	23.00	82.00
					MMM 5IG	NIFICANT	DIFFEREN	NE www
								TOL ARA
MIFCOXON	ANALYSIS	5						TOL ARA
		S - CARRI	AGE VE					10 2 XXX
ANALYSIS FUNCTION FUNCTION	OF: A = G: B = G:	CARRI 10 10	CELL:	C J			MIN Min	
ANALYSIS FUNCTION FUNCTION SUBJ	OF: A = G: B = G:	CARRI 10 10 B VAL	CELL: CELL:	L C J Ord -	O RO →	N	MIN MIN N	N +
ANALYSIS FUNCTION FUNCTION SUBJ	OF: A = G: B = G:	CARRI	CELL: CELL:	L C J Ord -	O RO →	N	MIN MIN N	N +

MEAN A: MEAN B:

20.61 17.25

STO DEV A: STO DEV B:

ANALYSIS	OF:	CHEST	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:) C			IAX IAX	
SUBJ	A VAL	B VAL	A-B	0RD -	0RD +	N	N -	N +
F-3 M19 R-3 G-3 M-2 S-3 K-1 F-2 R-2 M11 M10 G-2	3.03 3.36 5.12 5.09 1.93 3.97 6.35 3.68 5.69 3.76 5.16	1.57 2.79 2.70 4.45 1.32 3.09 5.89 1.87 3.79 3.19 1.80	1.46 0.57 2.42 0.64 0.61 0.88 0.46 1.90 0.57 3.36 2.29	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.46 0.57 0.57 0.64 0.68 1.46 1.90 2.42 2.42	1.00 2.50 2.50 4.00 5.00 6.00 7.00 8.00 9.00 10.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.50 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00
MEON O.	11 25	STO DEV	A:	1.25	SUM OF N	MINUS:	0.00	70.00
MEAN B:	2.84	STD DEV STD DEV	B:	1.37	SUM OF N	PLUS :		75.00
MEAN B:	2.84	STD DEV	B:		SUM OF N			
MILCOXON			B:					
WILCOXON	ANALYSI							
WILCOXON	ANALYSI OF: A = G: B = G:	S - CHEST 10	Z CELL: CFLL:	C	нын SIG	NIFICANT M	DIFFEREN	ICE ***
WILCOXON	ANALYSI OF: A = G: B = G:	S - CHEST	Z CELL: CFLL:	C	нын SIG	NIFICANT M	DIFFEREN	ICE ***

MAK SIGNIFICANT DIFFERENCE MAK

76.00

2.00

SUM OF N MINUS: SUM OF N PLUS:

2.88

ANALYSIS	OF:	CHEST	RES					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C C		F	IBS IBS	
SUBJ	A VAL	B VAL	A-B	ORD -	0RD +	N	N -	N +
F-3 M13 R-3 G-3 M-2 S-1 F-2 R-2 M11 M10 G-2	19.21 25.23 17.30 22.29 20.15 25.49 19.23 23.17 18.75 19.07 17.74 22.87	B VAL 16.15 21.10 18.49 19.18 19.84 17.55 16.54 19.11 13.77 16.48 16.02 14.87	3.06 4.13 -1.19 3.11 0.31 7.94 2.69 4.90 2.59 1.72 8.00	0.00 -1.19 0.00 0.00 0.00 0.00 0.00 0.00 0.00	0.31 0.00 1.72 2.59 3.69 3.11 4.06 4.13 4.98 7.94 8.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 0.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00
MEAN A: MEAN B:	20.88 17.43	STD DEV STD DEV	A: B:	2.83 2.17	SUM OF N	MINUS: PLUS :	2.00	76.00
					*** SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON	ANALYSIS	ò -						
		CHEST	SI					
ANALYSIS FUNCTION	OF: A = G: B = G:	CHEST	CELL:	C			ABS ABS	
ANALYSIS FUNCTION	OF: A = G: B = G:	CHEST	CELL:	C J ØRD -	ORD +		IRC	N +
ANALYSIS FUNCTION	OF: A = G: B = G:	CHEST	CELL:	C J ORD	ORD + 2.10 2.13 0.00 0.00 5.48 0.00 6.26 6.87 6.99 7.38 10.82 11.11		IRC	N + 1.00 0.00 0.00 5.00 7.30 8.00 9.00 10.00 11.00

MERN A: MERN B:

13.24 STD DEV A: 12.72 STD DEV B:

MICCOXON		-						
ANALYSIS	OF:	HEAD	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: C	j			IIN IIN	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
F-3 M13 R-3 G-3 M-2 S-3 K-1 F-2 R+2 M11 M10 G-2	-3.95 -5.39 -6.37 -5.99 -4.03 -4.47 -4.49 -3.47	-1.65 -1.46 -3.41 -4.08 -5.26 -3.44 -4.35 -2.52 -4.33 -3.29 -3.09	-2.30 -3.92 -1.98 -2.19 -0.73 -0.58 -1.68 -1.95 -0.16 -0.53 -0.14 -2.66	-0.14 -0.16 -0.53 -0.58 -0.73 -1.68 -1.95 -1.98 -2.19 -2.30 -2.66 -3.92	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0
		STD DEV						
					*** SIG	NIFICANT	DIFFEREN	CE ***
MILCOXON	ANALYSI	5						
ANALYSIS	OF:	HEAD	2					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: C	; j			IAX IAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N 	N -	N +
F-3 M13 R-3 G-3 M-2 S-3 K-1 F-2 R-1	13.56 13.15 13.32 13.02 11.31 13.69 12.31 14.62 14.21	12.25 13.17 12.50 11.64 10.90 13.21 12.70 13.50 12.72	1.31 -0.02 0.82 1.38 0.41 0.48 -0.39 1.12 1.26	-0.02 0.00 0.00 -0.39 0.00 0.00 -0.67 0.00 0.00	0.00 0.17 0.32 0.00 0.41 0.49 0.00 0.82 1.12	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00	1.00 0.00 0.00 4.00 0.00 0.00 7.00 0.00	0.00 2.00 3.00 5.00 6.00 8.00

SUM OF N MINUS: 12.00 ------SUM OF N PLUS: ----- 66.00 *** SIGNIFICANT DIFFERENCE ***

66.00

0.95 0.84

ANALYSIS	OF:	HEAD (RES					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C		i	985 985	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
F-3 M13 R-3 G-3 M-2 S-3 K-1 F-2 M11 M10 G-2	13.58 13.21 13.33 13.03 11.63 13.87 12.43 14.82 14.41 13.11 14.49 12.42	B VAL 12.56 13.62 12.73 12.11 10.95 13.43 12.75 13.85 13.12 12.84 14.07 13.08	1.02 -0.41 0.60 0.92 0.68 0.44 -0.32 0.97 1.27 0.42 -0.66	0.00 -0.32 -0.41 0.00 0.00 -0.66 0.00 0.00 0.00	0.27 0.00 0.00 0.42 0.44 0.60 0.00 0.68 0.92 0.97 1.02	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00	0.00 2.00 3.00 0.00 0.00 0.00 0.00 0.00	1.00 0.00 0.00 4.00 5.00 6.00 0.00 8.00 9.00 11.00
MEAN A: MEAN B:	13.36 12.93	STD DEV STD DEV	A: B:	0.94 0.84	SUM OF N	MINUS: PLUS:	12.00	66.00
					MMM SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON	ANALYSIS	-						
ANALYSIS	OF:	TOTAL	SHLD F	REFL				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C		ı	IAX IAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
F-3 M13 R-3 G-3 M-2 S-3 K-1 F-2 R-2 M11	134.52 135.88 110.82 119.18 142.32 107.03 60.21 107.94 65.21	B VAL 70.89 64.62 55.91 75.18 62.14 55.55 51.57 562.56 64.32 45.57 43.15	63.63 71.26 54.91 44.00 80.18 51.48 28.64 50.99 2.65	0.00 0.00 0.00 0.00 0.00 0.00	2.65 20.03 27.64 28.64 29.25 44.00 50.99 51.48 54.63	1.00 2.00 3.00 4.00 5.00 7.00 8.00	0.00 0.00 0.00 0.00 0.00 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00
M10 G-2	73.21 72.40	45.57 43.15	27.64 29.25	0.00 0.00	71.26 80.18	11.00 12.00	0.00	11.00

HHH SIGNIFICANT DIFFERENCE HHH

ANALYSIS OF:	TOTAL SHL) REEL		
FUNCTION A = FUNCTION B =	G: 10 CEL- G: 10 CEL-	.: C .: J		AX AX
SUBJ A VI		3 ORD - 0	JRD + N	N - N +
F-3 133.4 M13 130.3 R-3 114.6 G-3 137.6 M-2 152.5 S-3 115.4 K-1 74.8 F-2 141.3 R-2 62. M11 100. M10 71.6 G-2 98.4	32 104.54 29. 38 92.01 38. 37 94.97 19. 25 143.98 -6. 73 91.14 61. 36 99.11 16. 31 42.42 32. 31 104.97 36. 10 60.69 1. 75 92.11 8. 71 63.10 8.	28	1.41 1.00 0.00 2.00 8.61 3.00 8.64 4.00 16.35 5.00 19.90 6.00 29.28 7.00 32.39 8.00 32.39 8.00 36.34 9.00 38.37 10.00 47.58 11.00 51.59 12.00	0.00 1.00 2.00 0.00 0.00 3.00 0.00 4.00 0.00 5.00 0.00 6.00 0.00 7.00 0.00 8.00 0.00 9.00 0.00 10.00 0.00 11.00 0.00 12.00
MEAN A: 111 MEAN B: 86	STO DEV A: .66 STO DEV B:			2.00 76.00
		ж	<≍ SIGNIFICANT	DIFFERENCE ***
WILCOXON ANAL	(S 1S			
ANALYSIS OF:	TOTAL SHO	JLDER		
FUNCTION A = FUNCTION B =	G: 10 CEL G: 10 CEL	.: C .: J	M	AX AX
SUBJ A VE		3 ORD - O	38D + N	N - N +
F-3 259.0 M13 265. R-3 224.5 G-3 256. M-2 292.5 S-3 209. K-1 137.7 F-2 240. R-2 120. M11 170. M10 133.5 G-2 167.	163.51 95.70 152.83 112.70 152.83 112.70 146.42 78.32 218.04 38.51 145.14 147.39 67.30 82.01 55.70 161.82 78.37 153.60 16.37 153.60 1	0.00 0.00	08D + N 6.55 1.00 16.71 2.00 27.05 3.00 88.28 4.00 55.29 5.00 67.91 6.00 78.08 7.00 78.08 8.00 79.58 9.00 95.53 10.00 12.87 11.00	0.00 1.00 0.00 2.00 0.00 3.00 0.00 4.00 0.00 5.00 0.00 6.00 0.00 7.00 0.00 8.00 0.00 9.00 0.00 10.00 0.00 11.00 0.00 12.00
MEAN A: 206. MEAN B: 139.	41 STO DEV A: 40 STO DEV B:			0.00 78.00

HNHLIST	S OF:	TOTE	AL LAP					
FUNCTION FUNCTION	N A = G: N B = G:	10 10	CELL:	C C			IRX IRX	
SUBJ	A VAL	B VAL	A-B	080 -	ORD +	N	N -	N +
F-3 R-3 R-3 G-2 S-1 F-2 R-2 M10 G-2	131.86 59.03 125.24 103.27 91.44 101.50 102.82 62.81 73.47 80.26 69.12 54.69	130.30 123.33 128.26 108.63 102.44 112.84 90.49 101.53 68.54 83.99 81.63 53.71	1.56 -64.30 -3.02 -5.36 -11.00 -11.34 12.33 -38.72 4.93 -3.73 -12.51 0.98	0.00 0.00 -3.02 -3.73 0.00 -5.36 -11.00 -11.34 0.00 -12.51 -38.72 -64.30	0.98 1.56 0.00 0.00 4.93 0.00 0.00 0.00 12.33 0.00 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	0.00 0.00 3.00 4.00 0.00 6.00 7.00 8.00 0.00 11.00	1.00 2.00 0.00 5.00 0.00 0.00 9.00 0.00
					SUM OF N			
					*** SIG	NIFICANT	DIFFEREN	CE ×××
	N ANALYSIS	-	AL SEAT Z					
ANALYSI:		- 1016					18 X 18 X	
ANALYSI: FUNCTION FUNCTION SUBJ	5 0F: N A = G: N B = G: A VAL	- TOTA 10 10 B VAL	CELL:	C J Ord -	0RD +	N N	IAX N -	N +
ANALYSIS FUNCTION FUNCTION SUBJ F-3 M13 R-3 G-3 M-2 S-3 K-1 F-2 R-2 R-1 M10	5 0F: N A = G: N B = G: A VAL	- TOTA 10 10 B VAL	CELL:	C J Ord -	0.00 39.32 0.00 0.00 0.00 0.00 81.05 0.00 0.00	N 1.00 2.00 3.00 4.00 5.00 7.00 8.00 9.00	IAX N -	0.00 2.00 0.00 0.00 0.00 7.00 0.00

ANALYSIS OF:	RES SEAT FO	RCE				
FUNCTION A = G: FUNCTION B = G:	10 CELL:	C		M	AX AX	
SUBJ A VAL	B VAL A-B	ORD -	ORD +	N	N -	N +
F-3 1690.93 M13 1702.18 R-3 1570.47 G-3 1861.29 M-2 1686.06 S-3 1716.92 K-1 2099.55 F-2 1754.69 R-2 1587.72 M11 1789.99 M10 1634.33 G-2 1338.79	1757.83 -66.90	-14.41 0.00 -55.42 -66.13 -66.90 -67.17 -75.46 0.00 -136.53 -138.10 -176.55 -182.87	0.00 36.33 0.00 0.00 0.00 0.00 0.00 83.92 0.00 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	1.00 0.00 3.00 4.00 5.00 6.00 7.00 0.70 9.00 10.00 11.00	0.00 2.00 0.00 0.00 0.00 8.00 0.00 0.00
MEAN A: 1702.74 MEAN B: 1774.35	SID DEV A: STO DEV B:					10.00
			*** SIGN	IFICANT	DIFFEREN	CE ***
WILCOXON ANALYSI	S -					
ANALYSIS OF:	TOTAL FOOT	X				
FUNCTION A = G: FUNCTION B = G:	10 CELL:	C		M	IN IN	
SUBJ A VAL	B VAL A-B	ORD -	ORD +	N	N -	N +
F-3 -363.74 M13 -414.06 R-3 -300.25 G-3 -337.72 M-2 -312.88 S-3 -282.10 K-1 296.95 F-2 -425.15 R-2 -495.03 M11 -353.10 G-2 -246.99	B VAL A-B -344.47 -19.27 -432.35 18.29 -323.91 23.66 -282.17 -55.55 -333.70 20.82 -202.36 -79.74 -263.82 -33.13 -350.08 -75.07 -296.78 198.25 -275.05 -59.79 -286.73 -66.37 -261.08 14.09	0.00 0.00 19.27 0.00 0.00 -33.13 -55.55 -59.79 -66.37 -75.07 -79.74	14.09 18.29 0.00 20.82 23.66 0.00 0.00 0.00 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	0.00 0.00 3.00 0.00 0.00 6.00 7.00 8.00 9.00 10.00 11.00	1.00 2.00 0.00 4.00 5.00 0.00 0.00 0.00 0.00

ANALYSIS	OF:	RES F	001 F0	RCE				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C J			MAX MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
F-3 M13 R-3 G-3 M-2 S-1 F-2 R-1 M10 G-2	515.42 570.14 481.05 464.89 504.36 488.29 616.31 585.77 504.75 515.64 395.06	B VAL 468.63 639.47 427.41 480.27 507.65 457.04 402.91 592.49 451.07 440.20 540.97 404.40	46.79 -69.33 53.64 -15.38 -3.29 25.038 25.038 23.82 134.70 -25.33	-3.29 -9.34 -15.38 0.00 0.00 -25.33 0.00 0.00 0.00 -69.33 0.00	0.00 0.00 0.00 23.82 25.02 0.00 46.79 53.64 64.55 65.38 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	1.00 2.00 3.00 0.00 6.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 4.00 5.00 0.00 7.00 8.00 9.00 10.00
MEAN A: MEAN B:	508.65 484.38	STD DEV STO DEV	A: 8:	59.82 73.75	SUM OF N	MINUS: PLUS :	23.00	55.00
WILCOXON	ANALYSI:	S -						
ANALYSIS	OF:	CHEST	Z					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	H G			MAX MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	0AD +	N	N -	N +
G-3 G-2 F-2 M12 R-3 F-4 H-3 S-0 F-3 M11 M-2 H-4	21.09 18.04 16.98 16.46 20.47 17.05 20.70 15.01 19.09 18.24 20.55 17.19	B VAL 21.87 14.91 16.31 15.98 16.55 15.32 21.53 15.66 17.56 16.25 16.11 16.84 18.66 16.35	-0.78 3.13 0.648 3.973 -0.85 1.53 1.899 -0.43 1.89	-0.47 0.00 -0.65 0.00 -0.78 -0.83 0.00 0.00 0.00 0.00 0.00	0.00 0.48 0.00 0.67 0.00 0.00 0.04 1.43 1.53 1.73 1.80 1.89 3.13 3.92	1.00 2.00 3.00 4.00 5.00 7.00 8.00 9.00 11.00 12.00 14.00	1.00 0.00 3.00 0.00 5.00 6.00 0.00 0.00 0.00 0.00 0	0.00 0.00 0.00 0.00 0.00 7.00 8.00 9.00 11.00 14.00 14.00
MEAN A: MEAN B:								

HHH SIGNIFICANT DIFFERENCE HHH

HNHLTSIS	OF:	CHEST	RES					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:				MAX MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M12 R-3 F-3 M10 F-3 M12 H-2 H-4	21.13 18.11 17.63 17.26 20.64 17.34 21.10 15.35 18.35 18.69 20.38 15.73 18.36 20.56 17.40	B VAL 22.06 14.94 16.57 16.43 16.69 15.53 21.72 16.17 17.98 16.63 16.19 17.20 18.75 16.55	-0.93 3.17 1.06 0.83 3.95 1.86 -0.82 1.82 3.75 -0.46 1.85	-0.46 -0.62 -0.82 0.00 -0.93 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.83 0.85 0.00 1.06 1.16 1.35 1.81 1.81 1.82 3.17 3.75	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 10.00 11.00 12.00 14.00	1.00 2.00 3.00 0.00 6.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 4.00 5.00 7.00 8.00 11.00 12.00 14.00
MEAN A: MEAN B:	18.60 17.35	STD DEV STD DEV	A: B:	1.88 2.05	SUM OF N	MINUS: PLUS:	12.00	108.00
					*** SIG	NIFICANT	DIFFERE	NCE +++
WILCOXON	ANALYSIS	6						
		-						
ANRLYSIS	OF:	- CHEST						
ANALYSIS FUNCTION	OF: A = G:	CHEST	CELL:	H 5			MAX MAX	
ANALYSIS FUNCTION	OF: A = G:	CHEST	CELL:	H 5 Ond -	ORO +		MO V	N +
ANALYSIS FUNCTION	OF: A = G:	- CHEST	CELL:	H G ORD	0RD +		MO V	N + 1.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00

	OF:	пени	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	H G			IIN IIN	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N 	N -	N +
G-3 G-2 F-2 K-1 M12 R-3 F-4 H-3 S-3 M10 F-3 M11 M-2 H-4	-5.62 -2.93 -2.93 -3.43 -3.43 -4.23 -4.11 -3.55 -4.55 -4.55 -5.55	-4.57 -3.35 -2.85 -4.31 -4.12 -3.95 -3.65 -2.88 -2.55 -1.43 -3.85 -1.55	-0.46 -1.27 -0.09 -0.47 -1.10 -0.31 -1.28 -0.46 -1.02 -1.48 -1.00 -0.46 -0.59 -0.42	-0.09 -0.31 -0.42 -0.46 -0.46 -0.47 -0.59 -1.00 -1.00 -1.02 -1.10 -1.27 -1.28 -1.48	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.00 3.00 5.00 5.00 5.00 7.00 8.00 9.50 9.50 11.00 12.00 14.00	1.00 2.00 3.00 5.00 5.00 5.00 7.00 8.00 9.50 9.50 11.00 12.00 13.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0
		STO DEV						
					*** SIG	NIFICANT	DIFFERE	NCE ***
HILCOXON ANALYSIS		S - HEAD	7					
FUNCTION	A = G:			H			IAX	
FUNCTION SUB.	A = G: B = G:	10	CELL:	H G GBD -	GRD +		4A Y	N +
FUNCTION SUBJ G-2 F-2 K-1 M12 R-3 F-4 H-3 S-3 M10 F-3 M11 M-2 H-4	A = G: B = G: A VAL 12.98 12.70 15.13 13.31 13.81 11.85 11.28 11.59 14.34 14.19 13.18 13.15 11.80 13.66		CELL:	H G G GRD	08D + 		4A Y	N + 1.00 2.00 3.00 4.00 5.00 7.00 9.00 10.00 12.00 14.00

		HEAD						
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	Б Б			IAX IAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M12 R-3 F-4 H-3 S-10 F-3 M10 H-3 H-4	13.34 12.90 13.48 13.50 13.47 12.07 11.30 11.95 14.36 14.23 13.26 13.17 11.85 14.48	B VAL 12.17 12.43 13.36 11.41 12.94 12.92 12.43 13.92 11.76 13.63 12.79 11.86 12.36 11.47 14.01	1.17 0.47 0.12 2.09 0.52 -0.36 -2.62 0.79 0.73 1.44 1.40 0.38	0.00 -0.36 0.00 0.00 0.00 0.00 0.00 0.00 0.00	0.12 0.19 0.00 0.38 0.47 0.52 0.73 0.81 1.05 1.17 1.40 1.44 2.09 0.00	1.00 2.00 3.00 4.00 5.50 5.50 7.00 8.00 9.00 10.00 11.00 12.00 13.00	0.00 0.00 3.00 0.00 0.00 0.00 0.00 0.00	1.00 0.00 4.00 5.50 5.50 8.00 9.00 11.000 13.00 14.00
MEAN A: MEAN B:	13.15 12.63	STD DEV STD DEV	A: 8:	0.97 0.84	SUM OF N	MINUS: PLUS :	18.00	102.00
					*** SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON	ANALYSI:	5						
		-	SHIO R	IFFI				
ANALYSIS FUNCTION	OF: A = G:	TOTAL	CELL:	н		M M	inv	
ANALYSIS FUNCTION	OF: A = G:	TOTAL	CELL:	н	0 RD +		inv	N +
ANALYSIS FUNCTION	OF: A = G:	TOTAL	CELL:	н	0.00 0.00 0.00 8.56 11.31 13.55 16.33 18.93 21.04 21.31 23.23 24.39 24.39 32.15 36.48 38.30		inv	N + 0.00 0.000 3.000 4.000 5.000 7.00 8.000 10.000 11.000 13.000 14.000

ANALYSIS	OF:	TOTAL	SHLD R	REEL				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	H G			IAX IAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-2 G-2 F-2 K-1 M13 R-2 R-3 F-4 H-3 S-3 M10 F-3 M10 F-3 H-1	120.84 72.28 112.85 75.83 95.86 72.87 64.24 116.98 63.60 57.09 102.05 65.72 109.96 87.32	B VAL 115.75 39.83 62.66 57.45 83.32 59.15 76.49 43.20 113.38 72.82 62.54 110.45 62.06 69.27 72.39	5.09 32.45 50.19 18.38 12.59 18.60 21.04 3.60 21.04 3.60 40.69 14.93	0.00 0.00 -5.45 -5.62 -8.22 -0.00 0.00 0.00	3.60 3.66 5.09 0.00 0.00 0.00 12.54 13.09 14.93 18.38 21.04 32.45 40.69 50.19	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00 12.00 14.00	0.00 0.00 0.00 0.00 5.00 0.00 0.00 0.00	1.00 3.00 0.00 0.00 0.00 9.00 11.00 11.00 14.00
MEAN A: MEAN B:	85.98 73.52	STD DEV	A: B:	21.88 23.55	SUM OF N	MINUS: PLUS:	22.00	98.00
					MMM SIG			
	OF:	- Total						
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	G G			AX IAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M13 R-3 F-4 H-3 S-10 F-3 M11 M-2 H-4	221.62 125.76 188.54 149.82 174.82 174.27 110.44 197.09 125.11 186.08 132.17 184.46 153.48	B VAL 	37.33 49.86 72.15 40.65 29.98 40.61 3.89 18.49 -4.20 24.95 62.61 31.68	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.18 1.36 1.91 3.89 0.00 18.49 24.95 29.98 31.68 37.33 40.65 49.861 72.15	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00 14.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.00 3.00 4.00 6.00 7.00 8.00 9.00 11.00 12.00 14.00
MERN A: MEAN B:	156.33 128.90	STD DEV	A: B:	34.61 32.29	SUM OF N	MINUS: PLUS:	5.00	115.00

MMM SIGNIFICANT DIFFERENCE MMM

HNAL YSIS	OF:	TOTA	L LAP					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	H G		1	MAX MAX	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M13 R-2 R-3 F-4 H-3 M10 F-3 M11 M-2 H-4	70.28 71.58 112.72 86.52 101.43 107.63 114.72 109.03 64.30 100.13 82.59 119.18 72.19 114.84 120.30	120.59 51.31 99.35 73.62 84.98 83.00 107.31 63.65 89.02 68.63 67.60 93.54 68.24 69.81	-50.31 20.27 13.37 12.90 16.45 24.63 7.41 45.38 -24.72 31.50 14.99 25.64 3.95 45.03 8.09	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	3.95 7.41 8.09 12.90 13.37 14.99 16.45 20.27 24.63 0.00 25.64 31.50 45.38 0.00	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00 14.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 0.00 11.00 12.00 14.00
MEAN A: MEAN B:	96.50 83.52	STD DE STD DE	V A: V B:	19.93 19.95	SUM OF N	MINUS: PLUS :	25.00	95.00
					*** SIG	NIFICANT	DIFFERE	NCE ***
III. CAVAN								
MILLOYUM	ANALYSI	5						
		-	CH SIBOI	5				
ANALYSIS	0F:	- CROT				!	мах	
ANALYSIS FUNCTION	OF: A = G:	- CROT 10	CELL:	Ĥ		i 1	MAX MAX	
ANALYSIS FUNCTION	OF: A = G:	- CROT 10	CELL:	Ĥ	ORD +	N 	MAX MAX N -	N +
ANALYSIS FUNCTION	OF: A = G:	- CROT 10	CELL:		0RD + 0.00 4.99 0.00 23.86 40.91 0.00 49.09 529.39 72.58 75.62 80.98	N 2.00 3.00 4.00 5.00 6.00 7.00 9.00 10.00 11.00 14.00 15.00	N - 1.00 0.00 3.00 0.00 0.00 0.00 0.00 0.00	N +

ANALYSIS OF:	TOTAL SEAT Z	
FUNCTION A = G: FUNCTION B = G:	10 CELL: H 10 CELL: G	MAX Max
SUBJ A VAL	B VAL A-B ORD -	ORD + N N - N +
G-3 1798.87 G-2 1295.64 F-2 1625.27 K-1 2155.03 M13 1729.38 R-2 16678.47 F-4 1652.22 H-3 1894.36 S-3 1871.18 M10 1698.17 F-3 1642.26 M11 1773.12 M-2 1928.02 H-4 2023.30	1778.35	0RD + N N - N + 0.00 1.00 1.00 6.60 13.00 2.00 0.00 2.00 20.52 3.00 6.00 3.66 0.00 4.00 4.00 6.00 6.00 21.21 5.00 0.66 5.00 0.00 6.00 6.00 0.66 37.78 7.00 0.00 7.00 43.62 8.00 0.00 6.00 68.40 9.00 0.00 9.00 70.00 10.00 0.00 10.06 66.21 11.00 0.00 11.00 103.33 12.00 0.00 12.00 111.29 13.00 0.00 13.00 147.51 14.00 0.00 14.00 178.48 15.00 0.00 15.00
		SUM OF N MINUS: 11.00 SUM OF N PLUS : 109.00
		*** SIGNIFICANT DIFFERENCE ***
WILCOXON ANALYS	S -	
	S - RES SEAT FORCE	
ANALYSIS OF: FUNCTION A = G: FUNCTION B = G:	RES SEAT FORCE 10 CELL: H	MAX MAX
ANALYSIS OF: FUNCTION A = G: FUNCTION B = G:	RES SEAT FORCE 10 CELL: H	MAY
ANALYSIS OF: FUNCTION A = G: FUNCTION B = G:	RES SEAT FORCE 10 CELL: H	MAY
RNALYSIS OF: FUNCTION A = G: FUNCTION B = G: SUBJ A VAL G-3 1831.66 G-2 1508.52 F-2 1675.75 K-1 2191.97 M13 1765.20 R-2 1609.85 R-3 1695.69 F-4 1684.12 H-3 1915.36 S-3 1895.96 M10 1727.98 F-3 1665.94 M11 1808.70 M-2 1951.28 H-4 2052.99	RES SERT FORCE 10 CELL: H 10 CELL: G 8 VAL A-B ORD - 1797.70 33.96 0.00 1200.05 108.47 -20.64 1633.39 42.36 0.00 2182.86 9.09 0.00 1743.29 21.91 -27.05 1630.49 -20.64 0.00 1672.95 22.74 0.00 1672.95 22.74 0.00 1621.16 62.96 0.00 1827.47 87.89 0.00 1827.47 87.89 0.00 1923.01 -27.05 0.00 1609.17 118.81 0.00 1589.22 76.72 0.00 1767.23 41.47 0.00 1776.34 174.94 0.00 1896.95 156.04 0.00	

HNHL 1313	OF:	TOTAL	FOOT >	(
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	H G		,	IIN IIN	
รบвว	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-3 G-2 F-2 K-1 M13 R-3 F-4 H-3 S-3 M10 F-3 M10 F-4 H-2 H-4	-302.70 -237.09 -317.83 -257.19 -388.32 -235.18 -247.28 -518.61 -379.95 -257.58 -257.58 -293.77 -493.77 -324.22 -319.14 -468.03	B VAL -305.25 -355.25 -349.91 -367.29 -417.13 -305.55 -333.12 -434.90 -355.24 -376.39 -558.37 -346.81 -375.88 -590.72	2.55 118.16 32.10 -7.28 -21.03 92.33 70.37 85.84 -83.71 -24.71 118.81 64.69 56.74 122.69	0.00 -7.28 -21.03 0.00 -24.71 0.00 0.00 0.00 -83.71 0.00 0.00 0.00	2.55 0.00 0.00 22.59 0.00 32.10 56.74 64.60 70.37 0.00 85.84 92.33 118.16 118.81	1.00 2.00 3.00 4.00 5.00 6.00 9.00 10.00 11.00 12.00 14.00	0.000 3.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000	1.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00
		STD DEV						
					*** SIG	NIFICANT	DIFFERE	NCE ***
WILCOXON	ANALYSI:							
ANALYSI5	OF:	CARRI	AGE VEL	-				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	C H			IIN	
SUBJ	A VAI						ITN	
		B VAL	A-B	ORD -	ORD +		ITN	N +
G-2 K-1 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2 M13	-25.62 -25.63 -25.85 -25.85 -25.64 -25.64 -25.64 -25.64 -25.667 -25.67 -25.96	B VAL -26.19 -26.24 -25.88 -25.82 -25.89 -26.11 -26.13 -25.99 -26.15 -26.17	A-B 0.57 0.61 0.03 0.68 -0.15 0.49 0.51 0.27	0RD 0.00 -0.03 -0.15 0.00 0.00 0.00 0.00 0.00 0.00	0.03 0.00 0.00 0.15 0.21 0.27 0.44 0.49 0.51 0.57 0.61		ITN	N 1000000000000000000000000000000000000

ANAL YSIS	OF:	CHEST	X					
FUNCTION FUNCTION	A = 0: B = 6:	10 10	CELL:	С Н			MIN MIN	
SUBJ	A VAL	B VAL	A-8	ORD -	ORD +	N	N	N +
G-2 K-1 G-3 5-3 M-2 F-3 M11 R-3 M10 R-13	-2.13 -1.03 -1.93 -2.62 -4.10 -2.56 -1.46 -3.79 -1.98 -2.17 -0.79 -2.01	-2.54 -2.54 -2.59 -3.84 -3.84 -2.00 -1.10 -2.09 -1.09 -1.09	0.06 -0.39 0.41 -2.23 -0.88 -0.72 0.66 -1.78 -0.88 0.11 -0.50	0.00 -0.11 -0.37 -0.39 0.00 -0.50 0.00 -0.72 -0.88 -0.88 -1.78 -2.23	0.06	1.00	0.00 2.00 3.00 4.00 0.00 6.00	1.00
		STD BEY STD BEY						
					HHH SIG	NIFICANT	DIFFEREN	CE ***
MILEOXON	ANPLYSIS							
ANALYSIS	OF:	CHEST	Ž					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: CELL:	H C		1	MAX MAX	
SUBJ	P VAL	R VAL	A-B	ORD -	ORD +	N	N -	N *
G-2 K-1 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2	22.74 18.62 21.86 25.39 20.08 22.88 19.12 18.77 16.92 17.67 18.25 25.04	8 VAL 16.46 21.09 18.24 20.55 16.98 15.64 18.27 20.70 20.24 17.05 20.47	4.70 2.16 0.77 7.15 -0.47 5.90 3.48 0.50 -3.78 -2.57 1.20	-0.47 0.00 0.00 0.00 -2.57 0.00 -3.78 0.00 0.00	0.00 0.50 0.77 1.20 2.16 0.00 3.48 0.00 4.57 4.70 5.90 7.15	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 11.00	1.00 0.00 0.00 0.00 0.00 6.00 0.00 0.00	0.00 2.00 3.00 4.00 5.00 0.00 9.00 10.00 11.00
MEAN A:	20.61 18.64	STD DEV	A: B:	2.88	SUM OF N	MINUS: PLUS :	15.00	63.00

HER SIGNIFICANT DIFFERENCE HER

MEAN A: MEAN B: -4.93 -4.20 STD DEV A: STD DEV B:

		•						
ANALYSIS	Of:	CHES	T RES					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL: C			A	BS BS	
SUBJ	A VAL	B VAL	A-B	ORD -	GRD +	N	N -	N +
G-2 K-1 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2 M13	22.87 19.23 22.29 25.49 20.15 23.17 19.21 19.07 17.30 17.74 18.75 25.23	18.11 17.26 21.13 18.69 20.56 17.63 15.73 18.36 21.10 20.38 17.34 20.64	4.76 1.97 1.16 6.80 -0.41 5.54 3.48 0.71 -3.80 -2.64 1.41 4.59	-0.41 0.00 0.00 0.00 -2.64 0.00 -3.80 0.00 0.00	0.00 0.71 1.16 1.41 1.97 0.00 3.48 0.00 4.59 4.76 5.54 6.80	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	1.00 0.00 0.00 0.00 0.00 6.00 0.00 0.00	0.00 2.00 3.00 4.00 5.00 0.00 7.00 9.00 10.00 11.00
MEAN A: MEAN B:	20.88 16.91	STD DE	V A: V B:	2.83 1.80	SUM OF N	MINUS: PLUS :	15.00	63.00
					*** SIG	NIFICANT	DIFFEREN	CE ***
WILCOXON	ANALY519	6 -						
ANALYSIS	OF:	HEAD	X					
		10 10				M		
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
G-2 K-1 G-3 S-3 M-2 F-3	-5.75 -6.03 -6.27 -4.02 -5.99 -4.47	-4.62 -4.93 -5.03 -4.36 -5.57 -2.94	-1.13 -1.10 -1.24 0.34 -0.42 -1.53	-0.06 0.00 -0.16 0.00 -0.42 0.00	0.00 0.12 0.00 0.34 0.00 0.52 0.00 0.00 0.00	1.00 2.00 3.00 4.00 5.00	1.00 0.00 3.00 0.00 5.00	0.00 2.00 0.00 4.00 0.00 6.00

*** SIGNIFICANT DIFFERENCE ***

66.00

12.00

SUM OF N MINUS: SUM OF N PLUS:

0.98 1.06

ANALYSIS	OF:	TOTAL	SHLD	REFL				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	С			IAX IAX	
SUBJ	A VAL		A-B	ORD -	ORD +	N	N -	N +
G-2 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2	72.40 80.21 119.18 107.03 142.32 107.94 134.52 84.35 110.82 73.21 65.21 135.88	61.61 80.19 105.09 78.17 78.79 85.89 92.82 75.42 82.90 60.13 95.77 87.41	10.79 0.02 14.09 28.86 63.53 22.05 41.70 8.93 27.92 13.08 -30.56 48.47	0.00 0.00 0.00 0.00 0.00 0.00 0.00 -30.56 0.00 0.00	0.02 8.93 10.79 13.08 14.09 22.05 27.92 28.86 0.00 41.70 48.47 63.53	3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 9.00 0.00	1.00 2.00 3.00 4.00 5.00 7.00 8.00 0.00 10.00 11.00
MEAN A: MEAN B:	102.76 82.02	STD DEV STD DEV	A: B:	27.14 12.97	SUM OF N	MINUS: PLUS :	9.00	69.00
					HHH SIG	NIFICANT	DIFFEREN	CE ***
LITE COVON	0001 2010							
MILCOYON	ANALYSIS	-						
		TOTAL	SHLD	REEL				
ANALYSIS	OF:		SHLD 1	REEL C H		M	IAX IAX	
ANALYSIS FUNCTION FUNCTION SUBJ	OF: A = G: B = G:	TOTAL 10 10 B VAL	SHLD I	REEL C H ORD -	0RD +	M 	IAX IAX N -	N +
ANALYSIS FUNCTION FUNCTION SUBJ G-2 K-1 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2 M13	OF: A = G: B = G: A VAL 98.47 74.81 137.25 115.46 152.73 141.31 133.82 100.75 114.87 71.71 62.10 130.38	TOTAL 10 10 8 VAL 72.28	SHLD CELL: CELL: A-B -1.02 16.41 51.86 42.77 28.76 31.77 35.03 42.00 14.62 14.62	ORD - -1.02 -10.14 0.00 0.00 0.00 0.00 0.00 0.00 0.00	0RD + 0.00 0.00 14.62 16.41 26.19 28.46 31.77 34.52 35.03 42.00 42.77 51.86	N 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00	N - 1.00 2.00 0.00 0.00 0.00 0.00 0.00 0.00	N + 0.000 3.000 4.000 5.000 6.000 7.000 8.000 10.000 11.000
ANALYSIS FUNCTION FUNCTION SUBJ G-2 K-1 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2 M13	OF: A = G: B = G: A VAL 98.47 74.81 137.25 115.46 152.73 141.31 133.82 100.75 114.87 71.71 62.10 130.38	TOTAL 10 10 B VAL 72.28 75.83 120.84 63.60 109.96 112.85 102.05 65.72 72.87 57.09 72.24 95.86	SHLD CELL: A-B -1.021 151.86 42.77 28.46 31.77 342.00 14.62 -10.14 34.52 A: B:	C H	SUM OF N	PLUS:	N - 1.00 2.00 0.00 0.00 0.00 0.00 0.00 0.00	75.00

ANALYSIS	OF:	ገርተ	วะ ระบบแบ	EH.				
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	С		14 i 17 l	e,	
SUBJ	A VAL	B VAL	A-8	ORD -	OAD +	N	N -	N +
G-2 K-1 G-3 S-3 M-2 F-2 F-3 M11 R-3 M10 R-2 H13	167.12 137.30 256.32 209.19 292.51 240.70 259.04 170.31 224.50 133.52 120.70 265.70	125.76 149.82 221.62 125.04 184.46 188.54 186.08 132.17 144.27 102.11 149.30 174.82	41.36 -12.52 34.76 84.15 108.85 52.16 72.16 38.14 80.23 31.41 -28.60 90.88	0RD12.52 -28.60 0.00 0.00 0.00 0.00 0.00 0.00 0.00	0.00 0.00 31.41 34.70 36.14 41.36 52.16 72.96 80.23 84.15 90.88 108.05	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00	1.00 2.50 5.50 6.60 6.00 6.00 6.00 6.00 6.00 6	
MEAN A: MEAN B:	206.41 157.00	STD 0: STD 0:	EV A: EV B:	58.58 54.3 6	SUM OF N	MINUS. Plug	3.5.	
					FHH SIG			
MILCOXON		_						
	115:	CDG	TOU CIDOR	,				
			TCH STRAF			iu:	av	
FUNCTION	A = G: B = G:	10 10	CELL: CELL:	C		M		
FUNCTION	A = G: B = G:	10 10	CELL: CELL:	C	0RD +	M	A Y	N +
SUBJ G-2 K-1 G-3 S-3 M-2 F-2 F-2 F-2 M11 R-3 M10 R-2 M13	A = G: B = G:	10 10	CELL: CELL:		08D + 6.74 0.09 0.00 30.87 38.78 0.00 65.82 0.00 0.00 0.00 0.00	M	A Y	N 1.000000000000000000000000000000000000

HER STONE CAND DISERRICE WAS

ANALYSIS OF: CHEST X

FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:				IIN IIN	
SUBJ	A VAL	B VAL		0R0 -	0RD +	N		N ·
F-3 H-14 M1-3 G-3 M3 K2 S1 F2 M1-3 M1-3 M1-3	-2.67 -2.33 -2.04 -1.39 -1.96 -2.51 -1.52 -1.12 -2.19 -1.49 -2.30 -3.31 -1.45 -2.07	-2.87 -4.41 -3.00 -2.11 -2.79 -3.72 -1.55 -1.16 -3.60 -1.11 -2.28 -3.02 -1.97 -2.69	0.20 2.08 0.96 0.72 0.83 1.21 0.03 0.04 1.41 -0.38 -0.29 0.52	-0.02 0.00 0.00 -0.29 -0.38 0.00 0.00 0.00 0.00	0.00 0.03 0.04 0.20 0.00 0.52 0.62 0.72 0.83 0.96 1.21 1.41 2.08	1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00	1.00 0.50 0.60 0.60 0.00 0.00 0.00 0.00 0	7.000 6.000 7.000 6.000 6.000 7.000
MEAN A: MEAN B:	-2.03 -2.59	STO DEV STD DEV	A: B:	0.59 0.97	SUM OF N	MINUS: PLUS :	12.00	93.00
					MMM SIG	NIFICANT	DIFFEREN	CC ***
WILCOXON	RNALYSIS							
ANALYSIS	DF:	CHEST	X					
FUNCTION FUNCTION	A = G: B = G:	10 10	CELL:	J O		١	IAX IAX	
		B VAL			OAD +	N ~	N -	N +
F-3 H-4 M133 G-32 M-32 M-32 F-22 M13 H-30 G-21	2.07 2.42 3.49 4.13 4.20 1.83 3.77	1.57 1.15 2.79 2.70 4.45 1.32 3.09	0.50 1.27 0.70 1.43 -0.25 0.51 0.68	-0.25 0.00 0.00 0.00 0.00 0.00	0.00 0.50 0.51 0.68 0.70 0.74 1.14	1.00 2.00 3.00 4.00 5.00 6.00 7.00	1.00 0.00 0.00 0.00 0.00	0.00 2.00 3.00 4.00 5.00 6.00
F-2 R-2 M11 H-3 M10 G-2	4.58 3.53 4.93 3.93 4.28 3.20 3.33	5.89 1.87 3.79 3.19 2.29 1.80	-1.31 1.66 1.14 0.74 1.99 1.40	-1.31 0.00 0.00 0.00 0.00	1.14 1.27 0.00 1.40 1.43 1.66 1.77	8.00 9.00 10.00 11.00 12.00 13.00 14.00	9.00 0.00 0.00 0.00 0.00	7.00 8.00 0.00 10.00 11.00 12.00 13.00 14.00

WILCOXAN CHARTER

ANALYSIS 8	of:	HEAD :	x					
FUNCTION A	$\theta = G: 10$ $\theta = G: 10$		CELL: C	; !			MAX MAX	
SUBJ	A VAL	3 VAL	A-B	ORD -	ORD +	N	11 -	N +
F-3 H-43 A-33 K-23 S-12 K-22 M113 M10	0.83 3.96 1.60 0.98 2.26	2.80 5.30 3.36 2.26 3.14	-1.97 -1.34 -1.76 -1.28 -0.88	-0.03 -0.08 0.00 -0.27 0.00	0.00 0.00 0.19	1.00 2.00 3.00 4.50 4.50	1.00 2.00 0.00 4.50 0.00	0.00 0.00 3.00 0.00 4.50
MEAN A: MEAN B:	1.72 2.31	STD DEV STD DEV	A: B:	1.02 1.25	SUM OF N	MINUS: PLUS :	89.50	15.50
					MMM SIGN	IFICANT	DIFFERE	NCE ***
WILCOXON A	ANALYSIS							
ANALYSIS O	JF:	HEAD F	RES					
FUNCTION A							ABS ABS	

ANALYSIS	OF:	HEAD	RES					
FUNCTION FUNCTION			CELL:				ABS ABS	
SUBJ	A VAL	B VAL	A-B	ORD -	ORD +	N	N -	N +
	14.01 12.94 12.43 12.17 11.47 13.63 11.41 13.35 12.92 12.36 11.76	12.56 14.00 13.62 12.73 12.11 10.95 13.43 12.75 13.85 13.85 13.84 11.17	-0.70 0.01 -0.68 -0.30 0.06 0.52 0.20 -1.34	-0.30 -0.48 -0.49 0.00 -0.65 -0.68 -0.70 -1.28	0.06 0.20 0.00 0.00 0.52 0.59 0.00 0.00	1.00 2.00 3.50 3.50 5.00 6.00 7.00 8.00	3.50 5.00 6.00 7.00 0.00 10.00 11.00 12.00 13.00	0.00 0.00 5.20 9.00 0.00 0.00
	12.54 12.88	STO DEV		0.79 0.96				23.50

WILCOXON BNALTSIS

RNALY515	OF:	HEAD	51					
FUNCTION FUNCTION	A = 6: B = 6:	10 10	CELL:	e G			MAX MAX	
SUBJ	A VAL	B VAL	A-8	ORD -	0 RD →	N	N	h -
F-3 H-13 R-3 G-3 K-2 5-3 K-1 F-2 M11 H-3 M10 G-2	16.92 22.65 18.29 17.40 16.57 18.64 20.07 16.32 21.33 18.62 19.03 22.74	20.15 22.97 21.35 18.18 18.52 15.78 19.97 20.12 22.56 21.82 21.45 19.88 21.49 20.99	1.23 -0.32 -3.06 -0.78 -1.95 2.86 0.10 -3.46 -1.24 -0.49 -2.83 -0.85 1.46 -3.25	0.00 -0.32 -0.49 -0.78 -0.85 -1.23 -1.24 0.00 -1.95 -2.83 -3.06 -3.25 -3.46	0.10 0.00 0.00 0.00 0.00 0.00 1.46 0.00 0.00 2.86 0.00	1.06 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00 11.00 12.00 13.00	0.00 2.60 3.00 4.00 5.00 6.00 7.60 0.00 10.00 10.00 12.00 13.00	1.000000000000000000000000000000000000
MEAN A: MEAN B:	19.30 20.37	STD DE	V A: V B:	2.87 1.91	SUM OF N	MINUS: PLUS :	85.00	20.00
					www SIG	NIFICANT	DIFFERE	NCE ***
MILCOXON	ANALYSIS	5						
		5 - T o ta	L SHLD F	BEEL				
ANALYSIS FUNCTION	OF: A = 6: B = 6:	10	CELL:	G			MAX Max	
ANALYSIS FUNCTION	OF: A = 6: B = 6:	10	CELL:	G	ORD →		MOY	۸ +
ANALYSIS FUNCTION	OF: A = 6: B = 6:	10	CELL:		0.00 0.00 0.00 5.91 0.00 0.00 15.03 0.00 0.00 0.00 0.00		MOY	N +

MAN SIGNIFICANT DIFFERENCE MAN

WILCOXON ANALYSIS

MAX MAX		
- ORD + N N - N +		
- 08D + N N - N + 6 0.00 1.00 1.00 0.00 6 0.00 2.00 2.60 0.00 6 0.00 4.00 4.50 0.00 6 0.00 5.00 5.00 0.00 6 0.00 6.00 6.00 0.00 6 0.00 7.00 7.00 0.00 7.00 0.00 9.00 9.00 0.00 22.39 8.00 0.00 8.00 9 0.00 9.00 9.00 0.00 27.16 10.00 0.00 10.00 27.16 10.00 0.00 10.00 3 0.00 12.00 12.00 0.00 8 0.00 13.00 13.00 0.00 8 0.00 14.00 14.00 0.00		
SUM OF N MINUS: 87.00 SUM OF N PLUS: 18.00		
*** SIGNIFICANT DIFFERENCE ***		
MAX MAX		
- ORD + N N - N +		
3 0.00 1.00 1.00 0.00 0 0.00 2.00 2.00 0.00 0 4.73 3.00 0.00 3.00 0 11.96 4.00 0.00 4.00 3 0.00 5.00 5.00 0.00		
8		

HWH SIGNIFICANT DIFFERENCE HWW

WILCOXON RNALYSIS

ANALYSIS OF: TOTAL SEAT Z								
FUNCTION A = G: 10 GELL: G FUNCTION B = G: 10 CELL: J	MAX MAX							
SUBJ A VAL B VAL A-8 ORD -	ORD + N N - N +							
SUBJ A VAL B VAL A-8 ORD - F-3 1572.26 1729.73 -157.47 -14.56 H-4 1875.79 2024.12 -148.33 0.00 R-3 1657.26 1692.53 -35.27 -35.27 G-3 1778.35 1911.88 -133.53 -43.44 M-2 1749.54 1848.46 -98.92 -48.34 S-3 1897.34 1869.22 28.12 0.00 K-1 2158.72 2207.06 -48.34 -71.66 F-2 1581.65 1625.09 -43.44 -97.40 R-2 1621.90 1590.56 31.34 -98.92 M11 1735.34 1843.98 -108.64 -108.64 H-3 1808.15 1745.37 62.78 -133.53 M10 1586.88 1658.54 -71.66 -148.33 G-2 1192.31 1289.71 -97.40 -157.47	0.00 1.00 1.00 0.00 28.12 2.00 0.00 2.00 31.34 3.00 0.00 3.00 0.00 4.00 4.00 0.00 0.00 5.00 5.00 0.00 0.00 6.00 6.00 0.00 62.78 7.00 0.00 7.00 0.00 8.00 8.00 0.00 0.00 9.00 9.00 0.00 0.00 10.00 10.00 0.00 0.00 11.00 11.00 0.00 0.00 12.00 12.00 0.00 0.00 13.00 13.00 0.00 0.00 14.00 14.00 0.00							
MERN R: 1709.42 STD DEV R: 216.55 MERN B: 1769.09 STD DEV B: 215.41								
	MMM SIGNIFICANT DIFFERENCE MMM							
WILCOXON ANALYSIS								
ANALYSIS OF: RES SERT FORCE								
FUNCTION A = G: 10 CELL: G	MAX MAX							
FUNCTION A = G: 10 CELL: G	MAX MAX ORD + N N - N +							
FUNCTION A = G: 10 CELL: G	MAX MAX ORD + N N - N +							
	ORD + N N - N + 0.00 1.00 1.00 0.00 28.36 2.00 0.00 2.00 29.54 3.00 0.30 3.00 0.00 4.00 4.00 0.00 0.00 5.00 5.00 0.00 0.00 6.00 6.00 0.00 66.60 7.00 0.00 7.00 0.00 8.00 8.00 0.00 0.00 9.00 9.00 0.30 0.00 10.00 10.00 0.00 0.00 11.00 11.00 0.00 0.00 12.00 12.00 0.00 0.00 13.00 13.00 0.00 0.00 14.00 14.00 0.00							

WILCOXON ANALYSIS

ANALYSIS	S OF:	TOTAL FOOT					
			MIN MIN				
	A VAL	B VAL A-B	ORD -	ORD +	N	N -	N +
	-590.72 -367.29 -305.25 -305.25 -375.88 -355.24 -249.91 -349.93 -417.13 -348.90 -376.39	-344.47 -213.90 -368.57 -222.15 -432.35 65.06 -323.91 18.36 -282.17 -23.08 -333.70 -42.18 -202.36 -152.88 -263.82 13.91 -350.08 0.15 -296.78 -120.35 -275.05 -71.76 -518.52 83.62 -286.73 -89.66 -261.08 -94.17	0.00 0.00 -23.08 -42.18 0.00 -71.76 0.00 -89.66 -94.17 -120.88 -152.88	0.00 0.00	7.00 8.00 9.00 10.00 11.00 12.00	0.00 0.00 4.00 5.00 7.00 0.00 9.00 11.00	1.00 2.00 3.00 0.00 6.00 0.00 0.00 0.00
MEAN A: MEAN B:	-384.90 -324.26		92.72 79.13	SUM OF N		85.00 	20.00
				*** SIG	NIFICANT	DIFFERE	NCE ***

APPENDIX D

SUMMARY OF PHOTOMETRIC DATA

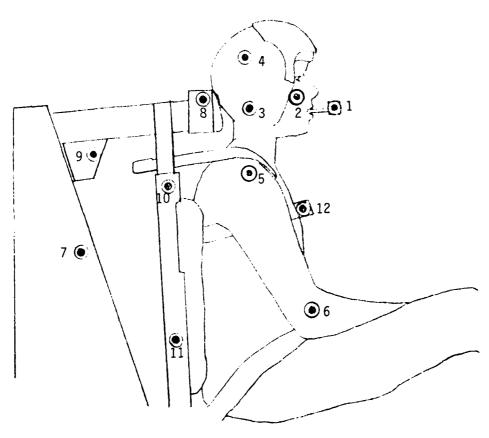
The photometric data obtained from this test program were analyzed to characterize the motions of photometric targets (fiducials) fixed to the test subject and thus describe the subject's dynamic response to impact. Reduction of the film data included digitization of target position information and computer plotting of the position-time, velocity-time, and acceleration-time history of each fiducial.

Fiducials were placed on subjects and the test fixture in accordance with the guidelines provided in "Film Analysis Guide for Dynamic Studies of Test Subjects" (SAE J138, March 1980). The positions of subject-mounted fiducials relative to reference fixture-mounted fiducials were documented for each subject prior to each test. The locations and number designations of each fiducial are shown in Figure D - 1. The distance between the "mouth pack" target (Target No. 1) and the center of the triaxial accelerometer in the mouth was four inches.

The photometric data were obtained by three 16 mm Milliken cameras, two mounted on the test carriage and one mounted off the carriage. The off-board camera and one on-board camera were positioned to provide a frontal view of the subject and the other on-board camera was positioned to provide a right lateral view of the subject. Each camera lens had a focal length of 10 mm. During the impact, the cameras were operated at 500 frames/sec.

The Photo Digitizing Systems Model 200 processor consists of an Automatic Film Reader (AFR), an electronic scanning camera, and a Data General Corporation (DGC) Nova 3/12 computer. This system was utilized for target position digitization. The semi-automatic ARF is manually initialized by selecting, with a cursor, targets of interest in the first frame of data. Targets on subsequent frames are automatically scanned, acquired, and identified. The target coordinates are then digitized by the Nova computer and the digitized data are then stored on magnetic tape. The coordinate resolution of the ARF is 0.025% of the major film dimension.

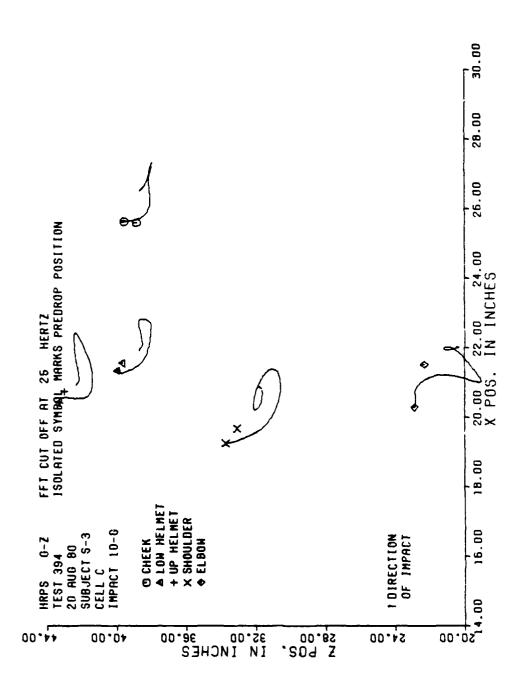
These digitized data were then processed on the Control Data Corporation (CDC) Cyber 74 computer system. The computer analysis routine used to process the film data has been described elsewhere (Graf et al., 1978; Brinkley et al., 1981). The program permitted the graphic presentation of position-time, velocity-time, and acceleration-time histories of fiducials and abscissa-ordinate position histories as well. Typical data from each cell of the experimental matrix are presented. The tests selected are the same as those selected for presentation in Apendix B.

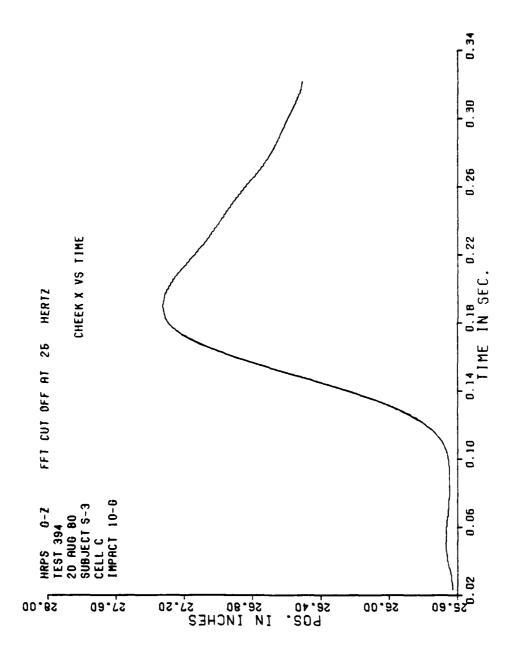


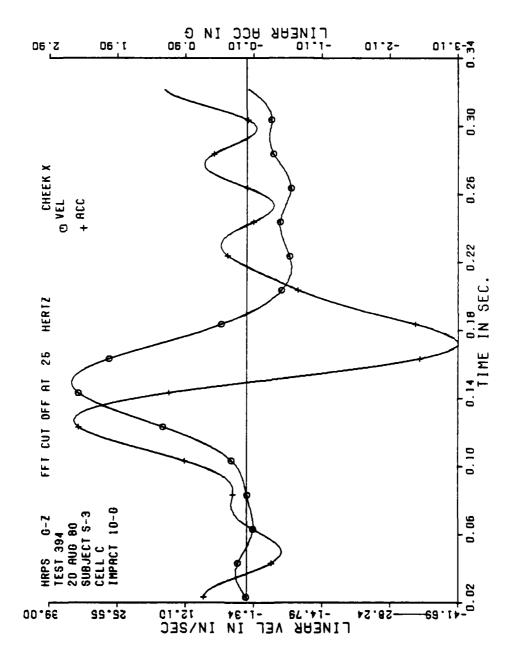
- Mouthpack
- Cheek
- Lower Helmet Upper Helmet Shoulder

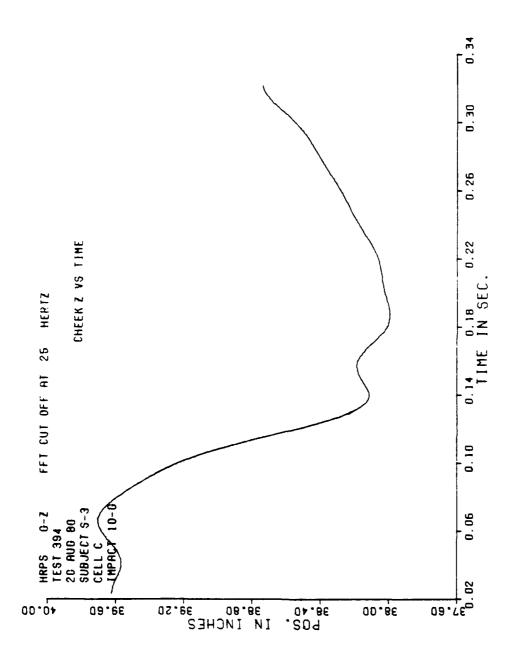
- E1bow
- Upper Frame
- Front Head Rest
- 9. Rear Head Rest
- 10. Upper Seat Back
 11. Lower Seat Back
 12. Chest Pack

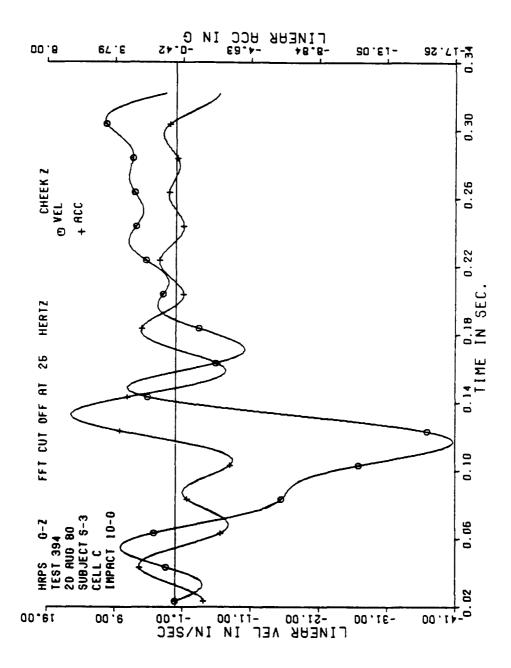
Figure D - 1. Locations of Fiducials.

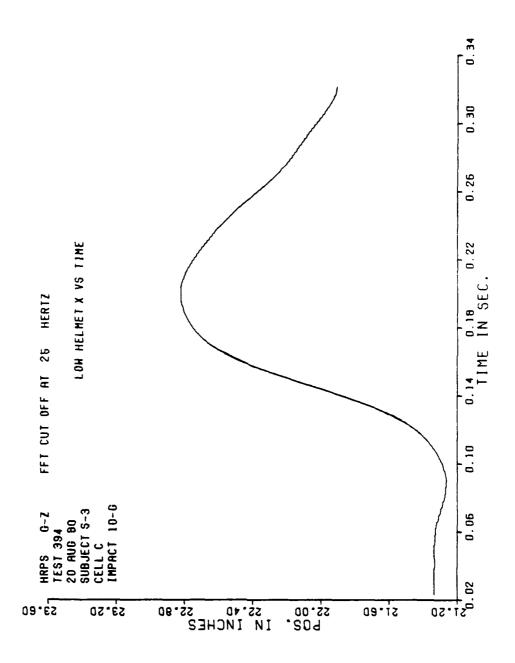


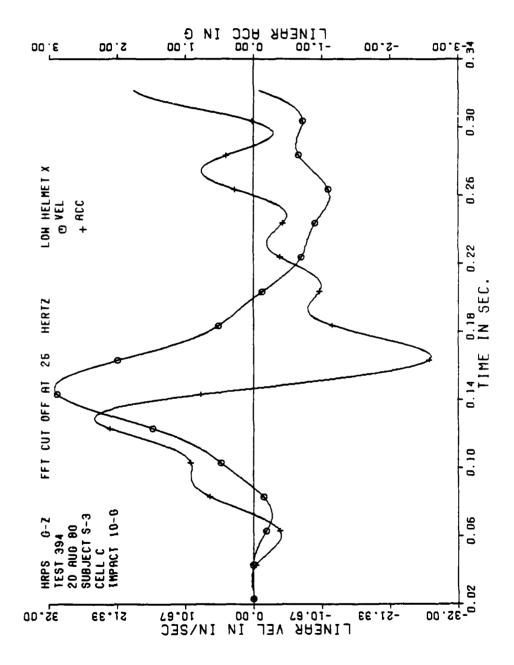


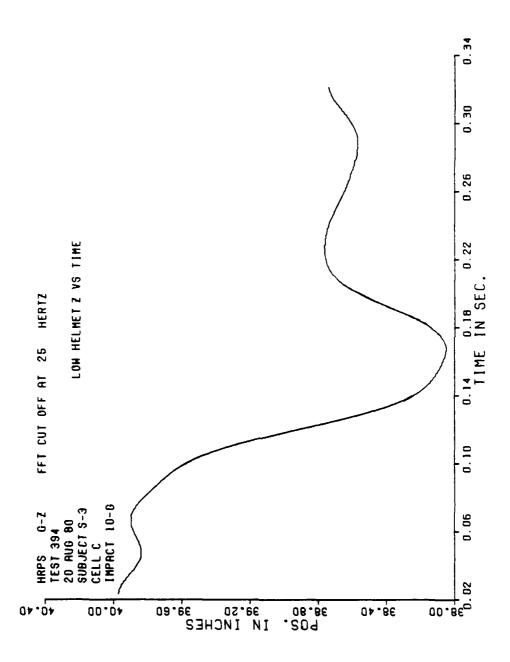


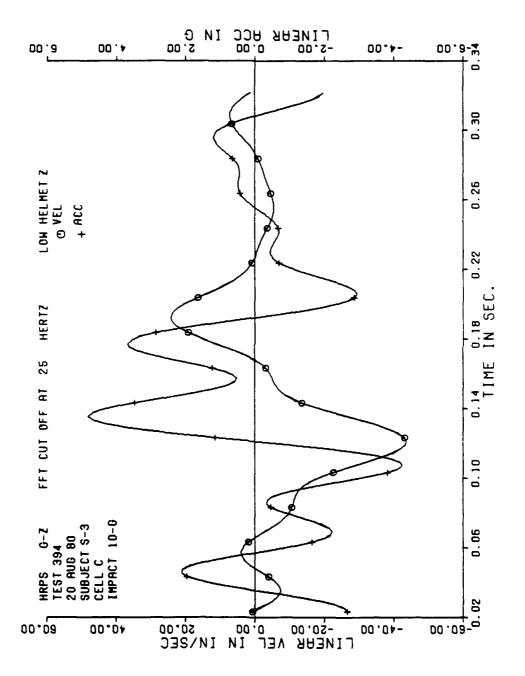


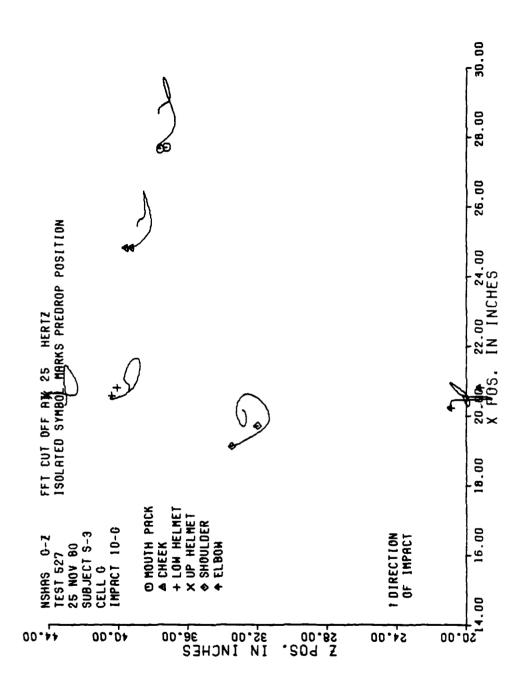


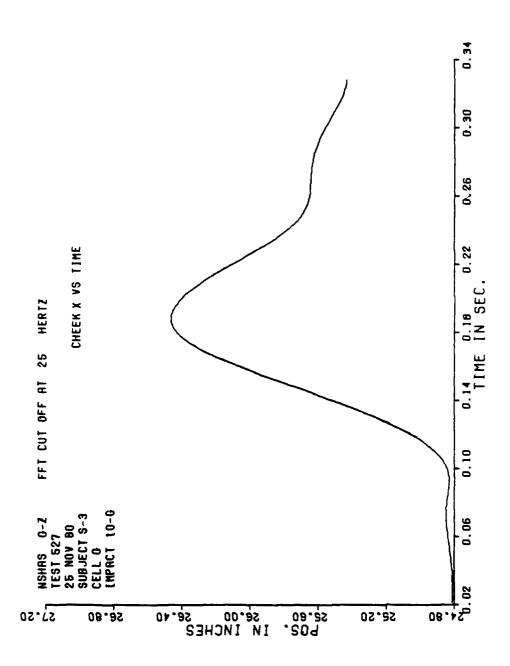


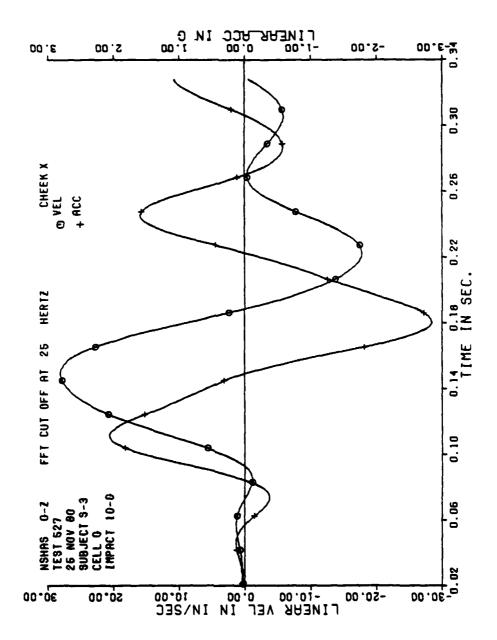


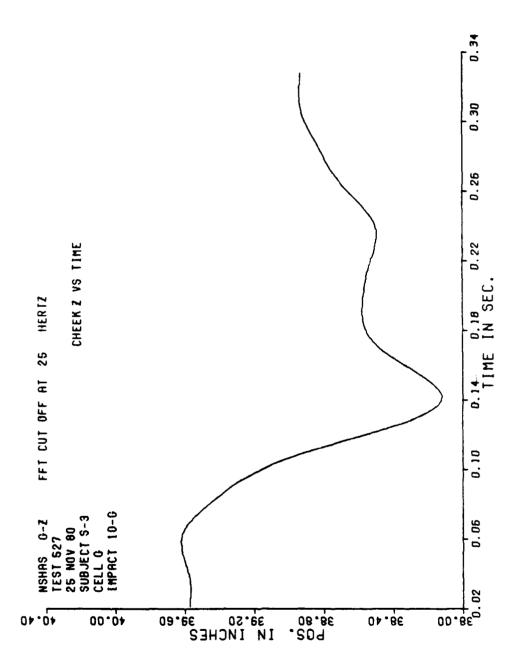


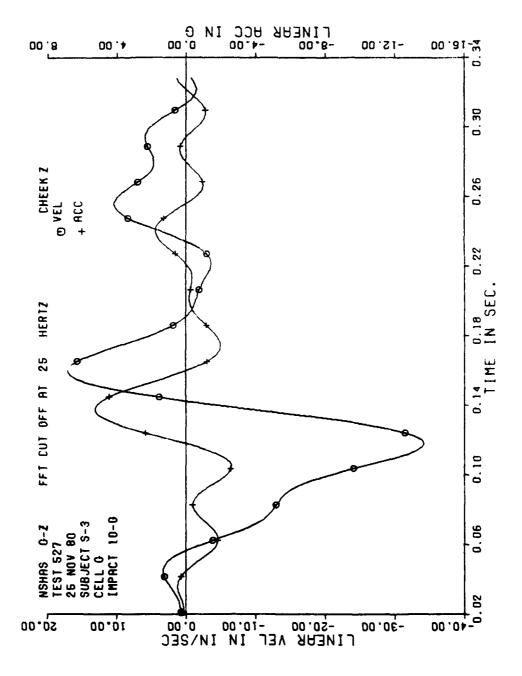


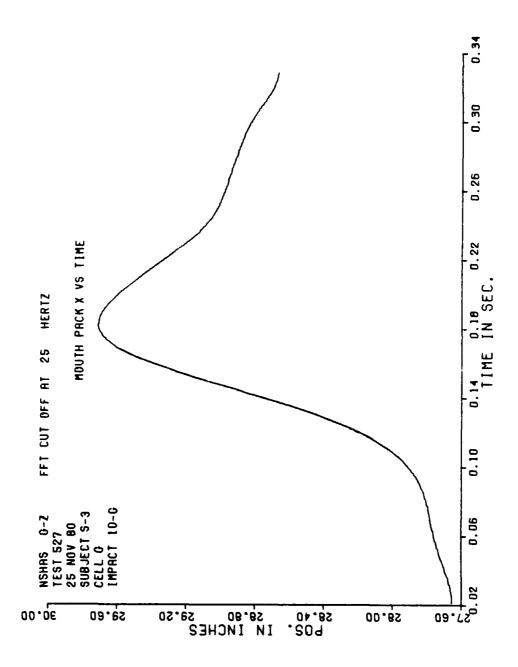


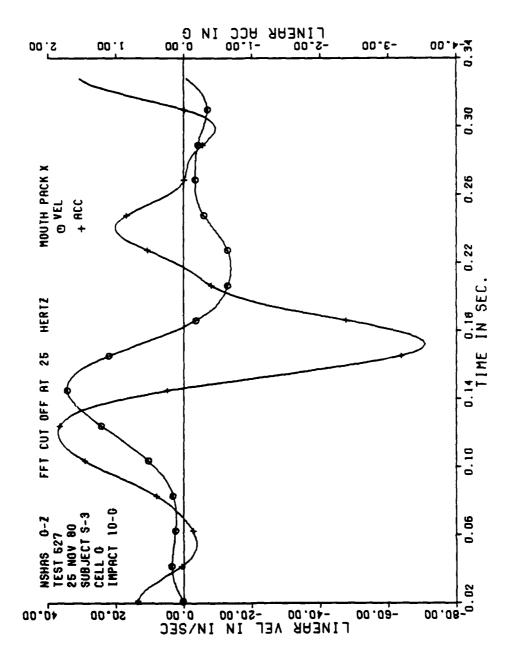


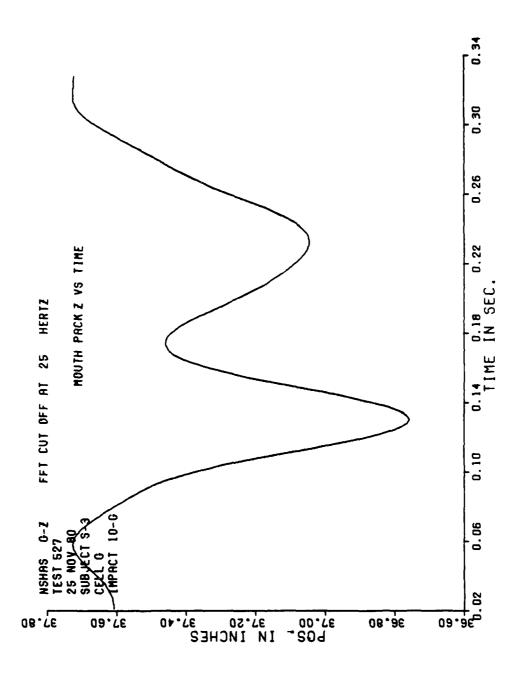


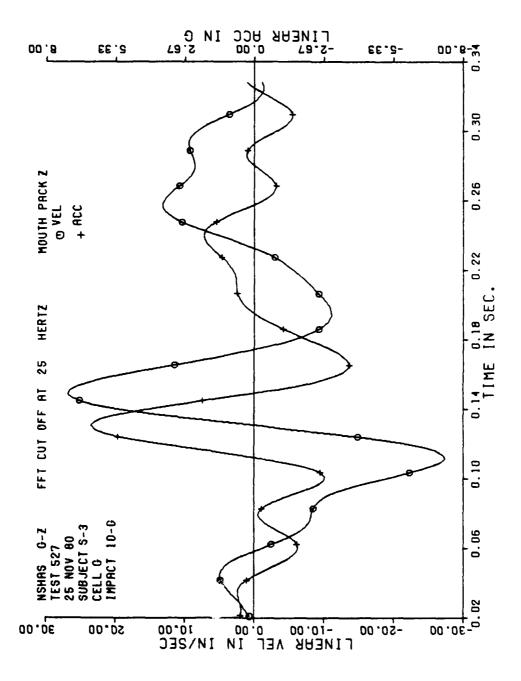


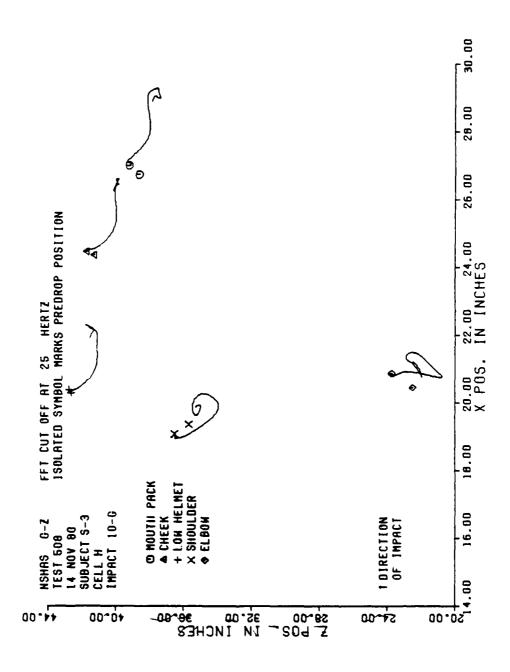


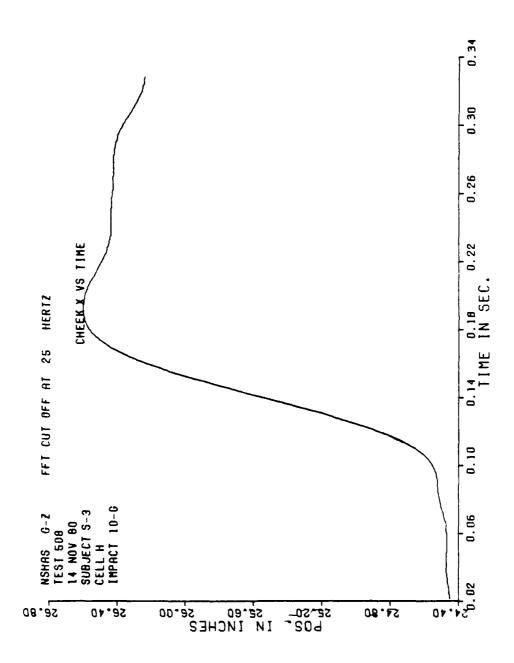


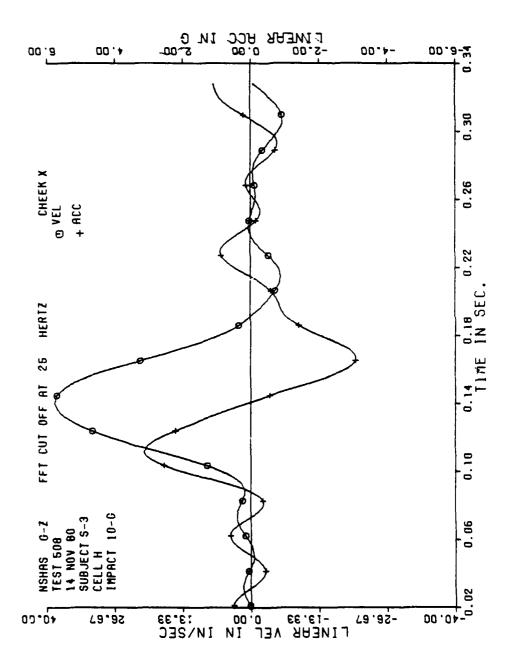


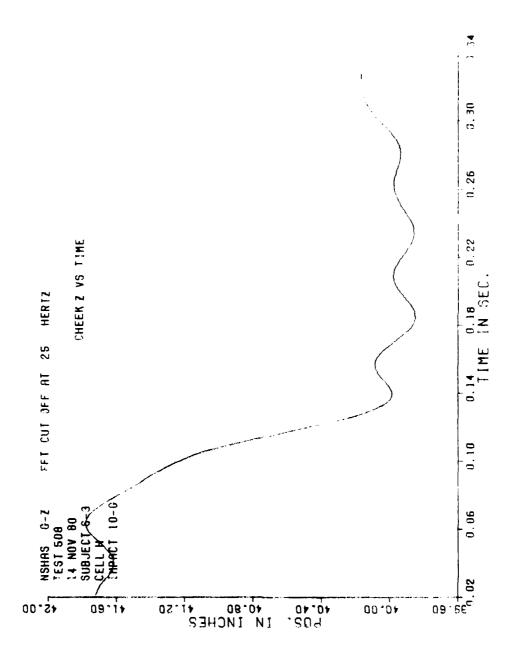


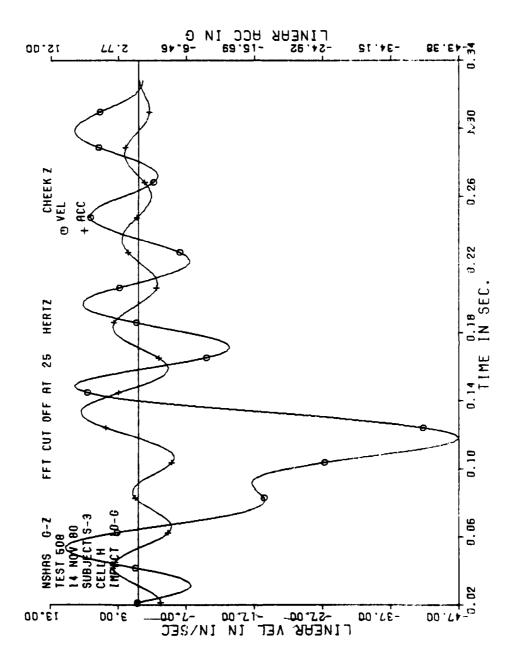


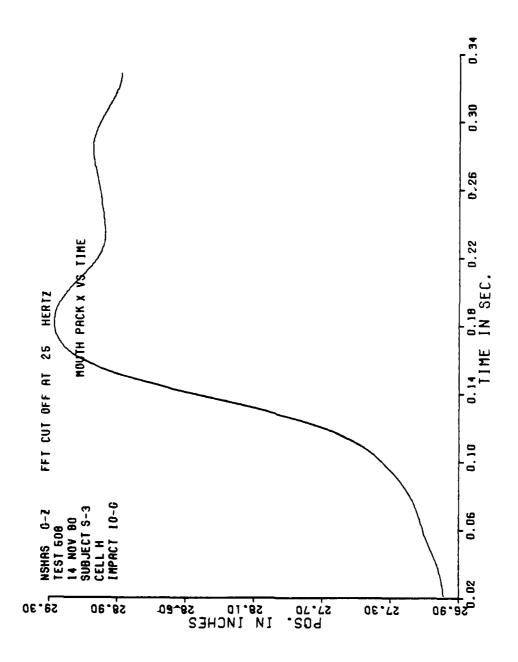


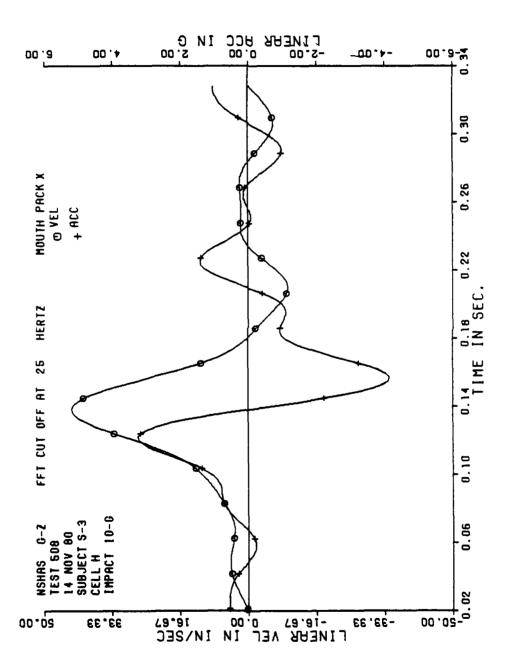


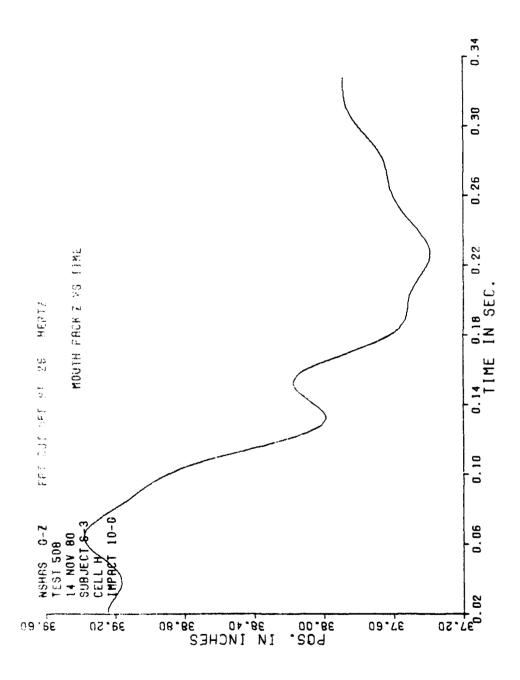


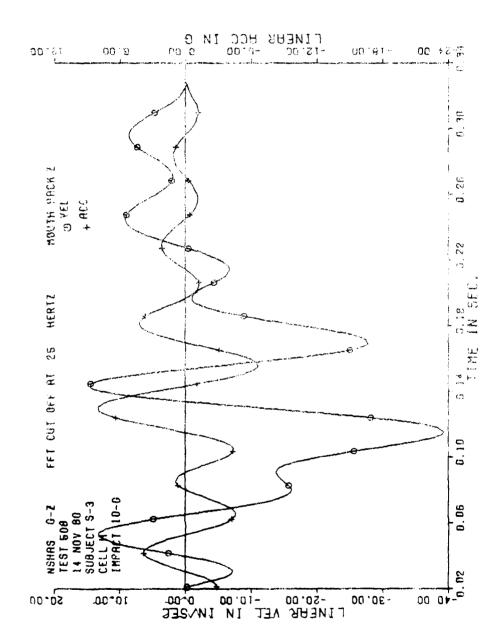


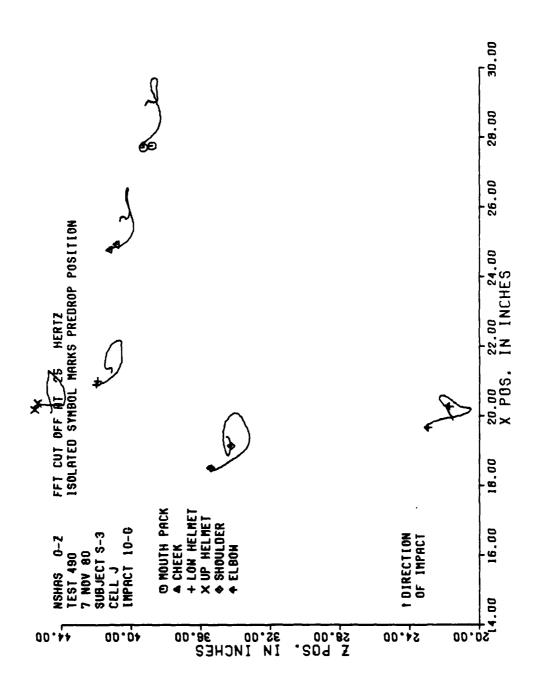


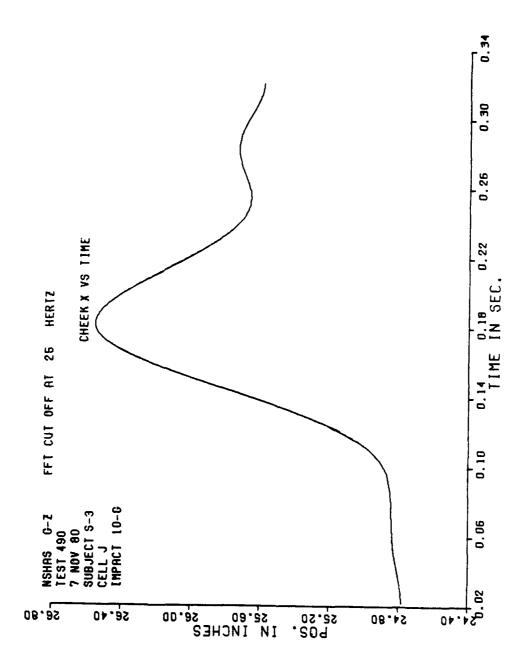


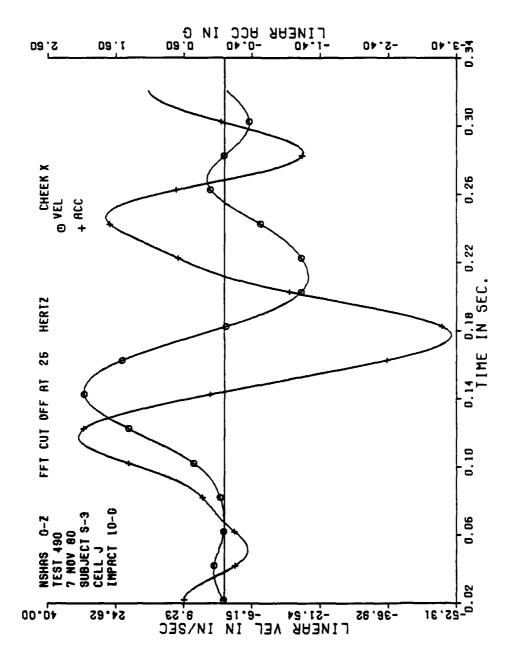


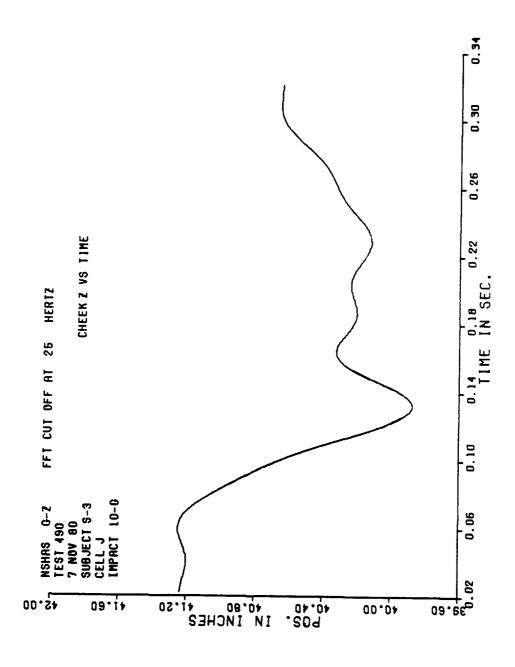


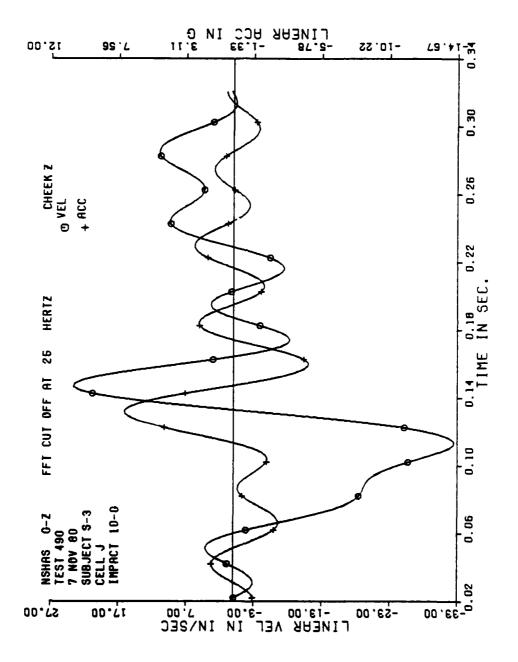


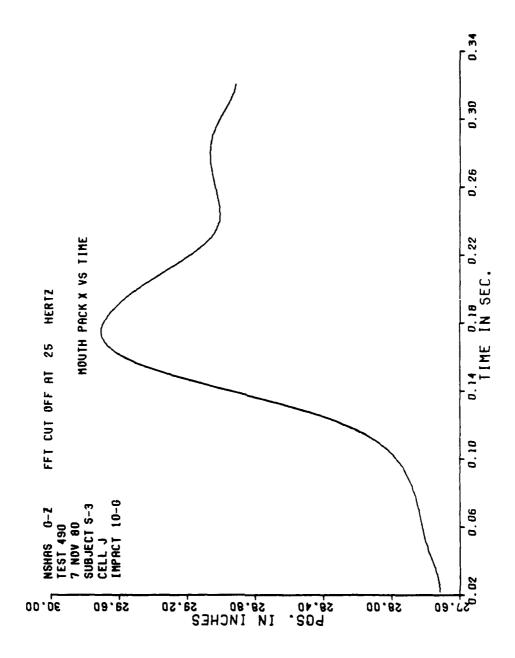


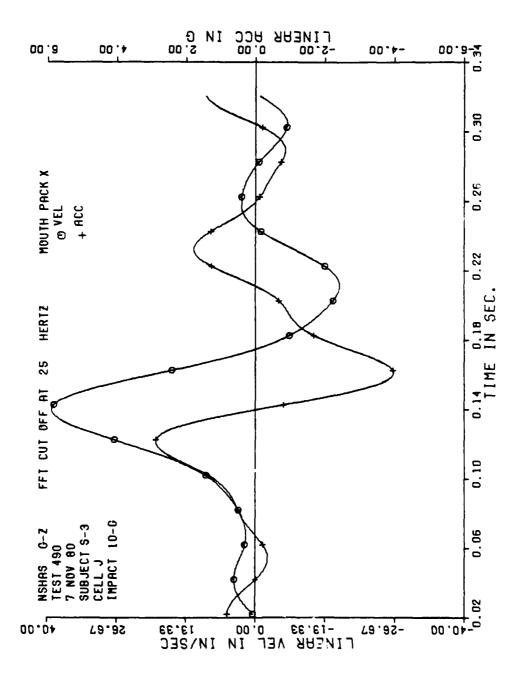


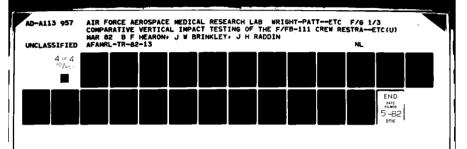


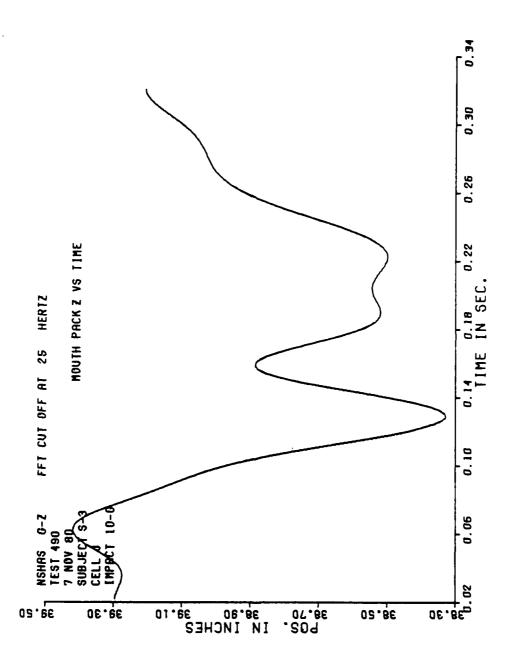


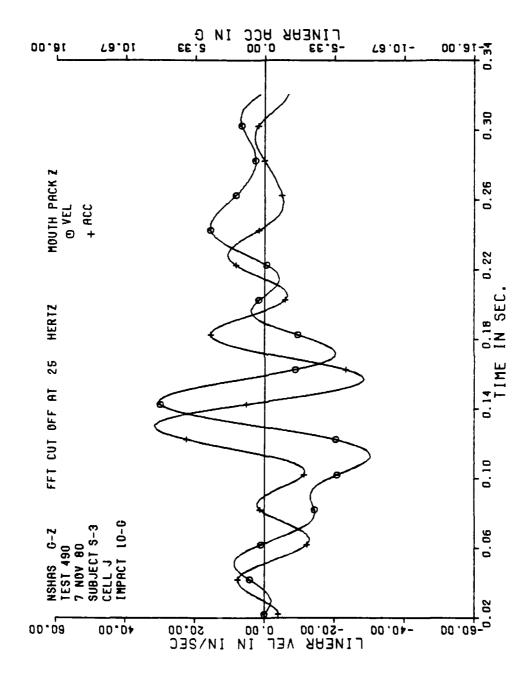












APPENDIX E

PRETEST MEASUREMENT DATA

Inertia reel strap angles were measured by inclinometer for each subject who participated in this test program. Measurements were obtained in the operational and the modified F/FB-111 harnesses. (Due to scheduling conflicts, two subjects, D1 and M2, were not measured in the modified harness.) Measurements were obtained in three different seat back angle conditions and in as many as six different seat elevations for each seat back angle condition. Measurements for both the left and right inertia reel strap are provided. The seat elevations for which there were "contact" or "impingement" of the inertia reel straps with the lower aspect of the headrest are indicated. ("Contact" and "impingement" are defined in Section 2A of this report.)

SUBJECT ID: D-1 SITTING HEIGHT: 39.7 MID-SHOULDER SITTING HEIGHT:

28.0

OPERATIONAL F-111 HARNESS

	FULL DOWN	NMC			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				FULL UP
		"(=	2	=	, m	=	4	=	5	5"
SEAT BACK ANGLE	ر ـ	R	,	8	ר	æ	-	~	ب	~		~
920	-5	-3	Impin -8	Impingement -8 -7								
1030	0.5		1.5 -3	-2	Contact	tact -7	Impingement	gement -11				
1100	5	7	0.5	0.5 2.5 -3.0 -1.5 -8 -7	-3.0	-1.5	Con -8	Contact	Impingement	Impingement		

MODIFIED F-111 HARNESS

		7.70			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJL	ISTMENT				
	בונר אב	DOWN										FULL UP
	ر	,,	1	=	2	Ξ.	(1)	=	4	2	5	5#
SEAT BACK ANGLE	٦.	R	٦	~		×		~		~		~
950												
1030												
1100												

SUBJECT ID: F-3 SITTING HEIGHT: 36.4 MID-SHOULDER SITTING HEIGHT: 25.5

OPERATIONAL F-111 HARNESS

0" 1" 5" 6 1 12.5 9.5			2			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				
ANGLE L R L S 6 1 1 12.5 9.5		בותר ה	NA.								·		FULL UP
L R L 5 6 1 13 12.5 9.5)"	1	=	2	=	Ê	=	4	=	98	
5 6 1 13 12.5 9.5	SEAT BACK ANGLE		R	-1	R	٦	~	7	~		~		~
13 12.5	920	5	9	1	1.5	1.5 -3.5	-2	Contact -6.5 -5	cact -5				
	1030	13	12.5	9.5	6	5	7	0	0	Contact -5	act -4		
1100 15 16 12 13	1100	15	91	12	13	8.5	8	7	7	0	+1	-4.5 -3	-3

MODIFIED F-111 HARNESS

	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	JSTMENT				FULL UP
		0"	1	=	2		(4)		4		5.	
SEAT BACK ANGLE	٦	R	ار	R	ı	æ		~	ر	~	رد	8
920	17	18	12	13.5	6	9.5	3	·4	-2	-1	-6.5	-6.5 -5.5
1030	24	25	20	20.5 14	14	15.5	6	10	3	7	-3	-2
1100	77	24	20	21.5 15	15	16	12	13	6.5	x	1	۲,

SUBJECT ID: F-2 SITTING HEIGHT: 37.5 MID-SHOULDER SITTING HEIGHT: 26.3

OPERATIONAL F-111 HARNESS

SEAT VERTICAL ADJUSTMENT

SEAT BACK ANGLE L R L R 920 6 5.5 1.5 1.5	1" R	ا ا 2،		۲	=				
1 9	R	٦		Э.		7	=	2	
9			~	ب	~	٦	œ	٦	~
9		•		Imping	Impingement				
	1.5 1.5	-3	-3.5	-7	-8				
						Contact	act		
1030 10 9 6.5	6.5 8.0	1	2	-3.5	-3.5 -3.5 -8.5 -8.0	-8.5	-8. 0		
1100 15 17 5 11	13	3	7	1	3.0	0 8	6	6 5 7 0 1 0 3 0 3 0 3 0 0 0 1 0 0 0 0 0 0 0 0	act
12 1112 11		0:5	٠.١	1.0	3.0	-5.0	73.0	-0.0	0./-

MODIFIED F-111 HARNESS

	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT				FULL UP
	J	0"	1		2	2"	8	₽_	4	=	нĠ	E
SEAT BACK ANGLE	٦	٣	J	æ	٦	œ	1	R	R	~	٦	œ
920	16	16.5	11	16.5 11 11.5 7		6.5	2.5	6.5 2.5 2.5 -3.5 -2.5 -7.5 -7.5	-3.5	-2.5	Cont -7.5	act -7.5
1030	23	24	19.5	24 19.5 20 15 16	15	16	10	10 10.5 3.5 4	3.5	4	-4.5 -3	-3
1100	26	26.5	23	23.5	19	19.5	15	26.5 23 23.5 19 19.5 15 15.5 10 10 5 5 5.5	0.1	10.5	5	5.5

MEASURED INERTIA REEL STRAP ANGLES

SUBJECT ID: F-4 SITTING HEIGHT: 36.4 MID-SHOULDER SITTING HEIGHT: 2

				OPERAT	IONAL F	OPERATIONAL F-111 HARNESS	RNESS					
·	FULL DOWN	NMC			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT			,	FULL UP
		0"	1	2	2	14	3	ш	4	=	5	5#
SEAT BACK ANGLE		R	ר	æ	-1	R	1	~	١	٣	7	~
920	9	5.5	5.5 1.5 3	3	-1	0	Cont 5	Contact -5				
1030	9.5	9.5	9.5 5.5 6.5	6.5	1	1	-4.5	-4.5 -4	8-	-6.5	Impin	Impingement
1100	12	12	8	8	7	3	0	1	-5	7-	Contact	Contact -9

FULL UP 5.5 œ Contact -2 ខ្ម -3 ~ 13 2 **4** 4 16.5 SEAT VERTICAL ADJUSTMENT 12 ~ ₽ 15 11 8.5 ~ 17 21 15.5 19.5 26.5 13.5 22 12 21 25 18.5 28.5 ~ 26 FULL DOWN 5 17 25 27 SEAT BACK ANGLE 920 1030 110^{0}

MEASURED INERTIA REEL STRAP ANGLES

SUBJECT ID: G-3 SITTING HEIGHT 34.8 MID-SHOULDER SITTING HEIGHT: 25.0

				OPERAT	OPERATIONAL F-111 HARNESS	-111 HA	RNESS					
•	FULL DO	DOWN			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				FULL UP
		0"	1	=	7	н	3	=	4	=	5"	_
SEAT BACK ANGLE		æ	L	æ	٦	٧	ب	~		~		~
920	10	10	9	6.5	6.5 2.5 2.0 -2.0	2.0	-2.0	Contact -2.0 -5.0 -5.5	Cont -5.0	act -5.5		
1030	16	16.5	12.5	16.5 12.5 11.5 9.0	0.6	10	7	7	0.5	0.5	0.5 0.5 -5 -4.5	act -4.5
1100	18	18	15	14	11.5	12	8.5	14 11.5 12 8.5 7.0 3.5 3.5 -0.5 -0.5	3.5	3.5	-0.5	-0.5

FULL UP 6.5 œ Contact ~ 5 \sim 5 8.5 α C1 11 4= 10.5 φ, 13.5 15.5 SEAT VERTICAL ADJUSTMENT ~ <u>-</u> 13.5 15 9 20.5 12 18 œ 5 10.5 20 19 25.5 \propto 16 22 17.5 23 24 20.5 25.5 œ 28 FULL DOWN . 21 27 27 SEAT BACK ANGLE 920 1030 1100

SUBJECT ID: G-2 SITTING HEIGHT: 33.3 MID-SHOULDER SITTING HEIGHT: 23.2

				OPERAT	OPERATIONAL F-111 HARNESS	-111 HA	RNESS					
	FULL DOWN	NM(SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT	·			FULL UP
		"(1	=	5"		ĸ	=	4	=	-2	
SEAT BACK ANGLE		æ	٦	R	٦	R		~	-1	~		~
920	13	13.5 10	10	10	5	9	2	3	-2	-1	Impingement	ement
1030	17	18	14.5 16	16	6	10	5	6.5 1	1	2.5	2.5 -3	-2
1100	20	21.5	21.5 17.5 18.5 13.5 15	18.5	13.5	15	8.5	6	4.5	4.5 6.5 0	0	0
			,									

FULL UP C 13 6 œ 2 8.5 ن Ξ 7.5 17.5 17 ~ 15.5 _ 15 SEAT VERTICAL ADJUSTMENT ~ 12 22 23 <u>.</u> 11.5 21.5 20 26.5 \simeq 17 26 16 25 25 29.5 22 30 ~ 28.5 28.5 21 32.5 32.5 ~ 27 FULL DOWN 5 33 25 31 SEAT BACK ANGLE 920 1030 1100

SUBJECT 10: $_{\rm H-3}$ SITTING HEIGHT: $_{38.0}$ MID-SHOUL DER SITTING HEIGHT: $_{26.1}$

OPERATIONAL F-111 HARNESS

	F18 1 D	777			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	INT			
	- 1	COMIN									FILL 13D
		0		•	•	=	7	_			
					1		7	+	4	2	5"
SEAT BACK ANGLE	١	R	٦,	×	ب	~		_	٥	-	,
								,	۷		¥
920	4.5	3.5	3.5 0	201	Contact	act	Impingement	t			
				Ç.,	<u></u>	+7-					
1030	8.0	7.5	3	7.5 30 30 0 5 13	(, !			Impingement		
				0.0			8 <u>-</u> 0-		-10		
1100	8.5	9.5	5	6.5	6.5 0.5 1.5	7.				Imping	Impingement
							5 5	0.80.	-6.5	-13	17 5

	FULL D	DOWN			SEAT	r VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				į
												FULL UP
		0"		=	2	=	_	=		=	i	
											C	
SEAT BACK ANGLE	1	œ	_	~	_	α	***	۵	•		•	
							,	٤	ار	¥	י	-
920	16	15.5 12	12		9	U	-		ı		lmpingement	ement
						7:7	-	Ç	-5	-3.5		
1030	21	21	21 16.5 17		11 5 11 5	711		i				
							0.0	6.5] –		9-	٦.
1100	23	23.5 19.5 20	19.5	20	15 15		10 6	•				
					`		70.7		7		-	

MEASURED INERTIA REEL STRAP ANGLES

SUBJECT ID: H-5 SITTING HEIGHT: 35.6 MID-SHOULDER SITTING HEIGHT: 24.1

OPERATIONAL F-111 HARNESS

SEAT VERTICAL ADJUSTMENT

	FULL DOWN	NMO			SEA	r VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				
	-	_										FULL UP
		<u>.</u>		=	•	=		-				
							7		4	ا ح	2	5"
SEAT BACK ANGLE		×	ا۔	~	-	٥						
,						4	_	¥	ٍا	~		Ω
920	9.5	11.5	11.5 5.5	8.0	0	С		i c				
							2.3 -2.0 -0.5	-0.5				
1030	15	17.5	17.5 11	13	7	o		0				100
							<u> </u>	0.0	С	2.5	-7	קרר
1100	19	19.5	19.5 15.5 16.5 11.0	16.5	11.0	12.5	α	3				
Ī				7			5	0	·†	· ^	C	+
					•					,		

MODIFIED F-111 HARNESS

	FULL DOWN	NMC			SEA.	T VERTI	SEAT VERTICAL ADJUSTMENT	ISTMENT				
												Fill 11P
		۳.		1								
									7		u,	5"
SEAT BACK ANGLE		ď		~		~	-	c		,		
((()							1	۲	-	×	_,	~
920	22.5	23	31	1 -	13.5	L/	Ç	(
							ΩŢ	10.5	<u>ر</u> ا	4.5	C	<u> </u>
1030	29	31	25	26.5	2.5	٠,		î				
						1	5.0	, ,	11	11.5		9
1100	32	31.5	25	25.5	28	ω α ς		r.				
							1	74.5	20	20	15	16
					•				•			

SUBJECT ID: H-4 SITTING HEIGHT: 37.0 MID-SHOULDER SITTING HEIGHT: 25.7

OPERATIONAL F-111 HARNESS

	FIII P DOWN	NA NA			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	JSTMENT			•	
												ייור טליי
	J	"(~ 1	=	. 2	=	(*)		4	=	" S	
SEAT BACK ANGLE	ļ	R	ر_	~		~		~		c		~
					Cont	Contact	Imping	Impingement				
920	7	3	0	0	-4	-5.5		-8.5				
1030	·	-	, , ,	ľ	·	Ç			Cont	Contact	Impingement	ement
201	1.4	11	10.5	10.5	γ	3	٠.٠	-2	-5.5 -6.5	-6.5	-11	-12
0011	ι.	, '	-	-	c	,	,		,		Contact	act
2011		14	1.1	7.7	Ø	٥	5	2.5 -1	- -	-2.5	-7.5	-7.51 -8.5

	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	ISTMENT				FULL
)	,,	1	=	2	=	e e		4	=	5	z
SEAT BACK ANGLE	٦	R		~		æ		~	٦	æ	ب	œ
920	11	11.5	11.5 6	7.5	2	3.5 -2	C3	1	Con	Contact	Impin	[mpingement
1030	18.5	19.5	19.5 13	14.5 10		1.1	7	5.5	ç i	-]	α.	-6.5
1100	12	22.5	22.5 16.5 18		11.5 13	13	7	رد	5	+1.5	67	# 1 # 1

SUBJECT ID: K-1 SITTING HEIGHT: 35.7 MID-SHOULDER SITTING HEIGHT:

24.8

OPERATIONAL F-111 HARNESS

•	FULL DO	NMO			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT				FULL UP
	O	,,		=	2	2"	3	=	4	=	2	. S.
SEAT BACK ANGLE	٦	8	٦	æ	٦	~	٦	~	-	œ	۰	~
920	7	5	0	1.5	+3	Contact	Impin	Impingement				
1030	10.5	12	9	7	2.5	3.5	3.5 -2.5 -1.5 -7	-1.5	Con -7	Contact		
1100	12	13	8.5 10	10	3	7	0	1	Con5	Contact	Imping	Impingement

	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT				FULL UP
		.0	1	Ε.	2	.	ω,		4		.5	
SEAT BACK ANGLE		R	-1	R	٦	æ	٦,	~	٦	~	رد	~
920	14	14	8	8.5	8.5 3.5	7	2	-1.5	1	Contact ' -6	Imping	mpingement
1030	21	21.5 17	17	17.5 11	11	12	6.5 6	9	O	C	-7.57	7
1100	23	22.5	18.5	22.5 18.5 19 15	ł	15.5 10 11 4 5	01	11	7	5	-2	-1.5

MEASURED INERTIA REEL STRAP ANGLES

SUBJECT ID: M-2 SITTING HEIGHT: 35.2 MID-SHOULDER SITTING HEIGHT:

24.0

OPERATIONAL F-111 HARNESS

	FULL DO	OWN			SEA	r VERTI	SEAT VERTICAL ADJUSTMENT	STMENT				
		=										FULL UP
				<u> </u>		Ξ.		=	_	=		
SEAT BACK ANGLE									3			=
SEN! BACK ANGLE	_	~		~	_	٥						
					,	٤	,	×		~	_	6
920	11,5	10.5	c	,	_				Contact	30.4		و
		()	7.0	8.5		6.0 4.5	1.5	-	:	,	.mpingement	ement
1030	7.	1							-	-3.5	-7.0	
		14.5 12.5 10.5	12.5	10.5	0.8	8.0	8.0 8.0 4.0 8.0	`			Contact	act
1100	,						2	7	0.1-	0	-5	-5.5
1100	1.5	16	13	13	0	(
					0.0	9.0	0.2 9.0 6.0 6.5 2.0	6.5	=			,
					•				н			4

	FULL UP		**************************************		2				
SEAT VERTICAL ADJUSTMENT		= 0		-	L 12				_
SEAT VERTICA		2							
		1.		ر ـ					
FULL DOWN		ō		٦					
				SEAL BACK ANGLE		920	1030	1100	

SUBJECT 1D: M-10 SITTING HEIGHT: 36.1 MID-SHOULDER SITTING HEIGHT: 24.8

OPERATIONAL F-111 HARNESS

•	FULL DO	NMC			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT			į	FULL UP
	J	11	Ī	8	2	Ξ	E	=	4	=	5	5"
SEAT BACK ANGLE	ر ـ	R	٦	×	1	~	ر	~		~		~
920	7	6.5	6.5 3.5	3	0	0	Cont -3.5	Contact -3.5 -4.5	Impingement -8 -8	gement -8		
1030	+15	13.5	13.5 10	8.5	5	7	3	2	-2.5	-2.5 -2.5	Impingement	ement -7.5
1100	16	16.5	16.5 12	11.5 11	11	9 5.6	9	5	1.5	5.	-2	-2.5

•	FULL DO	DOWN			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT		:	!	FULL UP
	כ	0"	1	=	"Z	2	нĈ	=	4	4"	нS	
SEAT BACK ANGLE	ļ	R	Ţ	R	ļ	æ	٦	R		×	٦,	æ
026	61	61	14	14 13.5 9	6	6	3.5 2.5 -1	2.5	-1	-1.5	-1.5 Contact.5	act7.5
1030	28.5	29	24.5 25	25	20	20	20 13.5 14	I	10	8	0	-1
1100	56	27	22	22.5 18	18	18	18 15.5 13.5 7	13.5		7.5 3 2.5	3	2.5

SUBJECT ID: M-11 SITTING HEIGHT: 35.7 MID-SHOULDER SITTING HEIGHT: 25.4

OPERATIONAL F-111 HARNESS

	FIR I DO	NA.			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMEN				FULL UP
				=	2	=		=	4	=	5	E
CFAT BACK ANGIE	_	~	لين	~	ب	œ	-1	~		٣		R
000	٥			3 5	-,	-2 5		Contact	Impingement	gement		
926	٥	٥	,	7:5	1		L				1800	100
1030	71	13	8	∞	2	5	1	.5	7 -	-5		6-
1100	17	16	13	12.5	6	8.5	9	5	1	1	7-	-5

MODIFIED F-111 HARNESS

	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU:	STMENT				FULL UP
	"0	=		=	2"	=	3"	=	 ф	=	5"	
CEAT BACK ANGLE	_	~		~	٦	~	٦	œ	-	æ	ار	œ
920	20	19	16	15	12.5	12.5 11.5	8	7	3	2	-2	-3
1030	1	28	22.5	22	18.5	18.5 17.5 13.5 12	13.5	12	&	6.5 -2	-2	l-
1100	28	1		21	18.5 18 14 13.5 9	18	14	13.5		6	5.5	5.5 5.5

SUBJECT ID: M-13 SITTING HEIGHT: 37.3 MID-SHOULDER SITTING HEIGHT: 26.3

OPERATIONAL F-111 HARNESS

	FULL DOV	NWC			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT		,		FULL UP
	J	,,	1	=	2	=	3	11	4	=	5	=
SEAT BACK ANGLE	Ļ	R	٦	æ		8	٦	~	٦	~	ب.	~
950	4.0		2.5 0	0	Con -4.0	Contact Impingement	Imping -8.0	gement -				
1030	9.0	8.0	8.0 6.0 5.5	5.5	1.0	2.0	Contact 2.0 -3.0 -3.5	_act _3.5	Impingement -8.0	gement		
1100	12.5	13.5	13.5 7.5 9.0 6.0 7.0 1.0 2.5 -4.0 -2.5 -6.5 -6.0	9.0	6.0	7.0	1.0	2.5	-4.0	-2.5	Cont -6.5	act -6.0

	FIII DO	NMOG			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				FILL
		.0	1	=	2	2"	3	_	4		5.	
SEAT BACK ANGLE	ب	R		~	7	~	ب	~		~		æ
920	12	10	6.5	6.5 6.5 +3	+3	+2	-5	-2	Impingement	ement		
1030	20	22.5	22.5 15	17	10	12	3.5	3.5 6.5 -1.5	-1.5	0	∞-	-5
1100	26	26	20	23	14	17	10 13	13	3.5	5.5	51	5.

MEASURED INERTIA REEL STRAP ANGLES

SUBJECT ID: P-3 SITTING HEIGHT: 39.1 MID-SHOULDER SITTING HEIGHT: 27.7

OPERATIONAL F-111 HARNESS

	FULL DOWN	NA.			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				FII : 10
		(-4	=	1,2	=	3	=	4	=	5.	
SEAT BACK ANGLE	ر.	~		~		~	٦	~		~		α
920	Con -3	Contact	Imping	Impingement								
1030	0.5	0	-3.5 -1	-1	Contact -9 -7	act -7	Impin	Impingement				
1100	9	8	2	3.5	3.5 -2.55	5	-7	-5	Contact -12 -10	Contact 	[อินเป⊯]	mpingement

MODIFIED F-111 HARNESS

•	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT				FULL UP
	0	.0	1	=	2	2	3	=	Þ	11	5	=
SEAT BACK ANGLE		R	-1	æ	Γ	æ	ן	R	ר	В	١	œ
920	15	9	.5	1	-3	7-	Impin	Impingement				
1030	11	12	7	5.5	0	5.	7-	-5	Contact -11 -11	taet -11	guidmī	Impingement
1100	18	17	12	12 11.5	8	7	3	3	-2.5	-2.5 -3 -10	-10	b-

SUBJECT ID: R-2 SITTING HEIGHT: 35.9 MID-SAGULDER SITTING HEIGHT: 24.3

OPERATIONAL F-111 HARNESS

	FULL DO	DOWN			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT		·		FULL UP
	0	84	1	2	2	=	S.	=	4	=	က	***
SEAT BACK ANGLE	7	æ		~	J	~		œ	-1	~	7	~
926	10	6	7	5	0	0.5	Contact 0.5 -3.5 -4	act -4	Impingement	ement		
1030	17.5	17	11	11	5.5	5.5 5.5 2.5 2.5	2.5	2.5	Contact -3	act -3	Impingement	ement
1100	18.5	18	14	13.5 10.5 11	10.5	11	7	7	2	2	Cont	Contact -4

	FULL DO	NMOC			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT				FULL UP
	0)"			2"	3	.33		4		5	2
SEAT BACK ANGLE		R	J	R	-1	æ	٦	~	-1	~	Ĺ	~
920	20	20	18.5	18.5 18.5 11	11	10	5	9	- 5	0	Contact -5	act -4.5
1030	27.5	28.5	28.5 23.5 24		19	19.5	13	14	8	7.5	0	0
1100	31	31	26.5 26		22.5 23	23	18	18.5	13	13.5	7.5	8

SUBJECT ID: R-3 SITTING HEIGHT: 35.2 MID-SHOULDER SITTING HEIGHT: 23.9

OPERATIONAL F-111 HARNESS

FULL DUMN SEAT BACK ANGLE L R			:			SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				
L R L R L R L R L R 9 8 5 4.5 1 1 -3.5 -3 Impingement 16 17 11 11 8 9 3 3 Contact 18 19 14 15 10 9 7.5 8 1.5 2	_		2							!			FULL U
L R L R L R L R 9 8 5 4.5 1 1 -3.5 -3 Impingement 16 17 11 11 8 9 3 3 Contact 18 19 14 15 10 9 7.5 8 1.5 2)"	T	=_,		=	က	=	Þ	=	3	=
9 8 5 4.5 1 1 Contact -3.5 Impingement 16 17 11 11 8 9 3 3 Contact -3.5 18 19 14 15 10 9 7.5 8 1.5 2	SEAT BACK ANGLE	٦	~	٦	~	٦	œ	ند	~	د	æ		~
16 17 11 11 8 9 3 3 Contact 18 19 14 15 10 9 7.5 8 1.5 2	920	6	8	5	4.5	1	1	Cont	act -3	Imping	ement		
18 19 14 15 10 9 7.5 8 1.5 2.5 -2	1030	16	17	11	11	8	6	3	3	Cont 2	1 ~	Imping -8	ement -9
	1100	18	19	14		10	6	7.5	8	1.5	2.5	-2	-3

MODIFIED F-111 HARNESS

	FULL DO	DOWN	,		SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				FULL UP
		=		H	2	2"	3	=	4	=	2	5"
SEAT BACK ANGLE	لب	æ		R	٦.	~	ب	~	ر	œ	ب	œ
920	19.5	20.5 15	15	16.5 10	10	11	9	7.5	1.5	2	-3	-",
1030	27	53	24	25	18.5	20	18.5 20 14.5	91	8.5	ł	51	~
1100	29	29.5	29.5 25.5 26	26	20	21.5	15.5	21.5 15.5 16.5 :0.5 11	:0.5	11	6	£.

SUBJECT ID: S-3
SITTING HEIGHT: 36.6
MID-SHOULDER SITTING HEIGHT: 25.6

OPERATIONAL F-111 HARNESS

	FULL DO	NMO		·	SEAT	VERTIC	SEAT VERTICAL ADJUSTMENT	STMENT				FULL UP
	0	u	1	=	2	=	(m)	=	4	=	2	5#
SEAT BACK ANGLE	L	R	١	2	1	~		~	ر	æ	ب.	~
950	7	5.5	0	[-3.5 -3	-3	Contact	act -7	Impin	[mpingement		
1030	6	10	5.5	9	5.	1.5	1.5 -4	-3	Con -8	Contact	Impingement	ement
1100	17	15	12	13.5	6	9.5	5	3.5 -5	-5	-1	-5.5 -7	7-

MODIFIED F-111 HARNESS

	1 1 1 1	NAIO			SEAT	SEAT VERTICAL ADJUSTMENT	AL ADJU	STMENT				
	3											דטרו טיי
	.0)"	, T	=	5	=		ε	4		5	2
SEAT BACK ANGLE	1	R	-1	æ	-	~	ب	œ		~	ب	œ
920	18	18	12.5 13	13	8	8.5 3	3	7	-2	-1	Con -7.5	Contact -7.5 -6.5
1030	23.5	23.5 23.5 21	21	21	15.5	15.5 15.5 10	10	9.5	4.5	9.5 4.5 4.5 -3	-3	-2
1100	25	24	24 21.5 20		16.5	17.5	12.5	16.5 17.5 12.5 12.5 7.5 7.5 2.5	7.5	7.5	2.5	2.5

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